

The Level 2 research product algorithms for the Superconducting Submillimeter-Wave Limb-Emission Sounder (SMILES)

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1 **Abstract**

2 This paper describes the algorithms of the level-2 research (L2r) processing chain developed for
3 the Superconducting Submillimeter-Wave Limb-Emission Sounder (SMILES). The chain has
4 been developed in parallel to the operational chain for conducting research on calibration and
5 retrieval algorithms. L2r chain products are available to the scientific community. The objective
6 of version 2 of the processing, which has produced the first usable L2r data, is the retrieval of the
7 vertical distribution of trace gases in the altitude range of 18–90 km. A special care is paid on
8 the retrieval of the line-of-sight tangent altitudes (or angles) since there is no oxygen line in the
9 SMILES measurements. The line-of-sight tangent altitudes between 20 and 50 km are retrieved
10 from the strong ozone line at 625.371 GHz together with the temperature and ozone profiles.
11 The relationship between the retrieved line-of-sight angles and errors on the pressure profile is
12 examined. Another major characteristic of the processing is the use of narrow spectral windows
13 set around the spectroscopic lines of interest. The error analysis for the retrieved HOCl profile
14 demonstrates that best results for inverting weak lines can be obtained by using such approach.
15 Future versions of the L2r algorithms will improve the temperature/pressure retrievals and also
16 provide information in the upper tropospheric/lower stratospheric region (e.g., water vapor, ice
17 content, ozone) and on stratospheric and mesospheric line-of-sight winds.

1 Introduction

The Superconducting Submillimeter-Wave Limb-Emission Sounder (SMILES) is a highly sensitive radiometer used to study atmospheric chemistry with a focus on the stratosphere (Kikuchi et al., 2010). It was developed by the Japan Aerospace Exploration Agency (JAXA) and the National Institute of Information and Communications Technology (NICT, Japan). Observations have been performed from the Japanese Experiment Module (JEM) onboard the International Space Station (ISS) from October 2009 to April 2010. During that period, SMILES successfully measured the vertical distribution and the diurnal variations of various stratospheric and mesospheric species in the latitude range of 38° S–65° N.

The SMILES **operational** ground segment was developed to calibrate the raw spectrometer outputs (i.e., level-0 data) to radiance (i.e., level-1b data), and to retrieve the geophysical products (i.e., level-2 data) (Takahashi et al., 2010). Daily maps are also available (<http://smiles.nict.go.jp/index-e.html>) and level-3 products (daily and monthly averaged data regridded to fixed geographical and vertical coordinates) are under development. In addition, NICT developed tools for research on calibration and retrieval algorithms leading to the development of a level-1b to level-2 chain, hereafter named Level-2 research (L2r) chain whose products are also available to the scientific community¹.

This paper presents the retrieval algorithms used in the L2r chain. SMILES data processing benefits from the heritage of previous submillimeter missions: the Sub-Millimetre Radiometer (SMR) onboard the Odin satellite (2001–) (Murtagh et al., 2002) and the Microwave Limb Sounder (MLS) onboard the Aura satellite (2004–) (Waters et al., 2006). Their retrieval codes are described in Urban et al. (2005); Read et al. (2006); Livesey et al. (2006) (and references therein). **In the SMILES operational level-2 algorithm, the vertical profiles of geophysical parameters are simultaneously retrieved from the full spectra measured during a vertical scan of the atmospheric limb. In the L2r algorithm, the retrieval procedure has been divided into sequentially dependent processes in order to apply optimised retrieval settings to selected spectral**

¹The SMILES data distribution policy is explained at <http://smiles.tks.c.jaxa.jp/ra/indexe.shtml>

lines in a given altitude range. Such approach allows to better characterise the spectra baseline and, hence, improves the results of the weak lines inversion. Another major difference between the operational and L2r algorithms is about the line-of-sight elevation angles treatment. Most of the prior microwave limb sounding instruments used a molecular oxygen line to retrieve the tangent points pressure (or elevation angles) which is a key parameter for limb-sounding. There is no such line in the SMILES spectra. In the operational level-2 algorithm only a mean angles offset is retrieved for each scan, while in the L2r algorithm, the angles of each line-of-sight are retrieved using the strongest O₃ line in the SMILES spectra.

In Sect. 2, a brief overview of the instrument and of the level-2 research chain is presented. The strategy for the retrieval procedure is presented in Sect. 3. The forward model used to simulate the measurements is described in Sect. 4. The inversion and the retrieval error characterisation methods are presented in Sect. 5. The performance of the retrieval approach is discussed in Sect. 6. We show results for the key parameters of the retrieval procedure: line-of-sight elevation angles, and temperature and O₃ volume-mixing ratio (VMR) profiles. The retrieval error of hypochlorous acid (HOCl) is also discussed in order to illustrate the results obtained from a weak signal. Other products are not discussed in this paper. Their validation and error budgets will be described in separate papers. Finally we conclude with a summary and an outlook on the future improvements and new products.

2 Overview of the SMILES measurements and of the L2r chain

SMILES observed spectral lines from the atmospheric limb in three frequency bands, named A (624.3–625.5 GHz), B (625.1–626.3 GHz) and C (649.1–650.3 GHz). Figure 1 shows synthetic spectra for line-of-sight tangent point altitudes of 20, 30 and 40 km and for the three bands (A and B in the upper panel and C in the lower panel). The strongest spectral signature is the O₃ line at 625.371 GHz common to bands A and B. Two strong triplets can be seen, one from H³⁷Cl around 625.19 GHz (band A) and the other one from H³⁵Cl around 625.92 GHz (band B). The other lines are significantly less intense. Two bands are measured simultaneously during a continuous vertical scan (Table 1) (Ochiai et al., 2010). **The measurement of the emission**

1 from a hot and a cold load, and of a comb spectrum, are performed at the top of each scan for
2 the radiance and the frequency calibration, respectively. The cold load is the cold sky at 2.7
3 K measured with the antenna pointing at tangent altitude of ~ 200 km (above the atmosphere)
4 while the hot load is an internal load at ambient temperature measured with a switching mirror.

5 The atmospheric spectra are recorded every 0.5 s during the first 30.5 s of the scan, with
6 line-of-sight tangent altitudes ranging typically from -10 km to 90 km.

7 The instrument is a single sideband heterodyne radiometer equipped with a 400×200 mm
8 elliptical antenna. The atmospheric signal is mixed with a local oscillator signal at
9 $\nu_{lo} = 637.32$ GHz in a superconductor-insulator-superconductor (SIS) mixer cooled at 4 K which
10 gives a very low receiver temperature of 300–350 K (Kikuchi and Fujii, 2010). The bands A
11 and B lie in the lower sideband (LSB) of the local oscillator frequency and band C is in the up-
12 per sideband (USB). The signal is analysed by two acousto-optical spectrometers (AOS) with
13 similar characteristics: 1728 channels separated by 0.8 MHz with an effective spectral resolu-
14 tion of 1.4 MHz. The AOS channel frequency is calibrated for every scan and corrected from
15 the Doppler shift of ~ 8 MHz caused by the ISS velocity component parallel to the line-of-sight
16 ($\sim 4 \text{ km s}^{-1}$).

17 The data produced by the version 2 of the L2r chain are the vertical distributions of O_3 and its
18 main isotopes, two isotopomers of hydrochloric acid (H^{35}Cl and H^{37}Cl), chlorine and bromine
19 monoxides (ClO , BrO), nitric acid (HNO_3), hydroperoxyl radical (HO_2), hypochlorous acid
20 (HOCl), hydrogen peroxide (H_2O_2), and methyl cyanide (CH_3CN). The main objective for
21 version 2 algorithms is to retrieve data between 18 to 90 km. Below 18 km, the retrieved data
22 are available but they are strongly affected by the errors on the calibrated radiance and on the
23 modelling of the absorption continuum. They should be used with **caution** and the results will
24 not be discussed in the current paper.

25 The retrieval procedure is composed of sequentially dependent processes, each of them as-
26 sociated with a spectral window and a line-of-sight altitude range. The processes are denoted
27 by the name of the spectral band (A, B or C) and a window index (e.g., w1). The list of the
28 processes is given in Table 2. The same parameter can be retrieved from different processes
29 leading to different products that are defined by the parameter name (e.g., O_3 or temperature)

1 and by the process name (e.g., A-w1). The same forward and retrieval models are used for
2 each process but the setting (e.g., position and size of the window) is optimized according to
3 the characteristics of the spectral lines of interest and of the altitude range. L2r products are
4 distributed in daily files using the HDF-EOS format (<http://www.hdfgroup.org/hdfEOS.html>).

5 **3 Retrieval strategy**

6 At the SMILES observation frequencies, the atmosphere is opaque at altitudes below typically
7 10 to 12 km, depending on the upper tropospheric (UT) water vapor content. The right panel
8 of Fig. 2 shows the variation with the line-of-sight tangent altitude of the measured brightness
9 temperature at 624.547 GHz (red line). For altitudes where the atmosphere is opaque, the bright-
10 ness temperature reaches a saturated level at about 220 K: the measurement is not sensitive to
11 the atmospheric conditions below about 12 km. Scattering by ice particles can also occur in
12 the UT yielding a clear depletion of the saturated brightness temperature (Fig. 2, right panel).
13 Ice water content in the UT can be retrieved from these observations (Ekström et al., 2007; Wu
14 et al., 2008; Mendrok et al., 2008), but it requires a special algorithm that is not implemented
15 in the current processing. Inversion of ice-contaminated spectra (typically below 18 km in the
16 tropical regions) gives bad retrieval diagnostics and the retrieved data are not usable.

17 The line-of-sight tangent altitudes (or angles) are key parameters that should have been ob-
18 tained by the star tracker attached to the instrument. However the star tracker data have been
19 found to be noisy in the level-1b data (version 5). Therefore the attitudes data based on the ISS
20 Guidance, Navigation, and Control system are used instead. For each scan, the line-of-sight
21 angles must be corrected for a global offset that can be as large as 0.2° (8 km on the tangent
22 point altitude). The correction is performed before the retrieval procedure by matching the
23 measured brightness temperatures with the ones computed with the forward model (Sect. 4).
24 The precision is better than 0.001° . The atmospheric inputs for the calculation are the temper-
25 ature, pressure, H_2O and O_3 profiles from the version 5.2 of the analysis of the Goddard Earth
26 Observing System Data Assimilation System model (GEOS-5.2) (Reinecker, 2008), and other
27 species are from various climatologies. GEOS-5 data are given every 6 h on $2/3 \times 1/2$ degree

1 longitude-latitude horizontal grid. We have chosen to use the closest GEOS-5 profile to the
2 middle of the scan of interest. Figure 2 shows the forward model calculation before (dashed
3 line) and after (black solid line) line-of-sight angles correction.

4 Below typically 18 km, pressure broadening leads to overlapping wings of spectral lines.
5 Contributions from the atmospheric parameters are thus mixed and only a limited number of
6 parameters can be retrieved. In this study, the distributions of O₃, H₂O, HCl and ClO are
7 retrieved from the full spectral band (bandwidth of 1.2 GHz). Temperature and pressure profiles
8 are fixed parameters from GEOS-5, a precision of 1 K is **assumed** on the temperature.

9 Above 18 km, spectral lines of different atmospheric molecules are separated from each other
10 and from the continuum radiation background. Between 20–50 km, temperature can be retrieved
11 from the central channels of the strong O₃ line (Baron et al., 2001). Furthermore, the O₃ VMR
12 profile and line-of-sight angles have distinct signatures in the measurement: the spectral line
13 amplitude changes with VMR while the spectral line width changes with the line-of-sight angle.
14 This study shows that the low measurement noise allows **us** to distinguish and consequently to
15 retrieve both parameters.

16 The full retrieval procedure is composed of the following processes:

- 17 1. The line-of-sight elevation angles are retrieved between 20 and 50 km from the intense
18 O₃ line at 625.371 GHz (A-w0 and B-w0 in Table 2). The spectral window between
19 625.042 and 625.612 GHz is used both for bands A and B. The line-of-sight elevation
20 angles outside the retrieval vertical range are computed by linearly **extrapolating the re-**
21 **trieved angles**.
- 22 2. Temperature and O₃ are retrieved above 18 km from the same spectral range used in step 1
23 (A-w1 and B-w1). Temperature is retrieved up to ~60 km and O₃ up to 90 km. The isotope
24 OO¹⁸O is also retrieved simultaneously from two small lines at 625.901 and 625.563 GHz.
- 25 3. The strong lines of HCl are inverted between 18–90 km in the spectral ranges 624.445–
26 625.040 GHz for H³⁷Cl (A-w2) and 625.714–626.264 GHz (B-w2) for H³⁵Cl.
- 27 4. Then, a set of narrow windows (Table 2) are inverted to fit the weak lines in the middle
28 stratosphere and mesosphere. In the current version of the processing, it has been chosen

1 to not use the retrieved temperature and line-of-sight angle to invert the spectral windows
2 of band C.

- 3 5. The full spectral bands between 12 and 30 km are processed separately to retrieve the
4 VMR of O₃, H₂O, HCl and ClO in the UT/lower stratosphere (LS).

5 **4 Forward model**

6 The measured signal is simulated from the computation of the specific intensity I^{atm}
7 ($\text{W m}^{-2} \text{sr}^{-1} \text{Hz}^{-1}$) of the outgoing atmospheric radiation and by taking into account the
8 SMILES instrument characteristics, i.e., the antenna field of view, the AOS channel shape and
9 the mixing of the upper and lower sidebands. Details on the forward model computation are
10 presented in Urban et al. (2004) and Baron et al. (2008). In this section, we focus on the param-
11 eters relevant to the SMILES analysis. The signal is expressed with a brightness temperature
1 unit using the Rayleigh-Jeans linear transformation. For bands A and B that are both measured
2 in the lower sideband, the brightness temperature $T^{\text{a,b}}$ measured in the i th channel is:

$$T^{a,b}(\theta_j, \nu_i) = \frac{c^2}{2 k_b \nu_i^2} \int_{\Delta\theta, \Delta\nu} G_e^{\text{ANT}}(\theta, \theta_j, \omega) g^{\text{AOS}}(\nu^{\text{lsb}}, \nu_i) \frac{(1 - R^{\text{lsb}}(\nu^{\text{lsb}})) I^{\text{atm}}(\nu^{\text{lsb}}, \theta) + R^{\text{usb}}(\nu^{\text{usb}}) I^{\text{atm}}(\nu^{\text{usb}}, \theta)}{1 - R^{\text{lsb}}(\nu^{\text{lsb}}) + R^{\text{usb}}(\nu^{\text{usb}})} d\theta d\nu^{\text{lsb}}, \quad (1)$$

where θ_j is the mean elevation angle of the antenna during the spectrum integration, θ is the optical path elevation angle for the specific intensity calculation, G_e^{ANT} is the effective vertical antenna pattern, g^{AOS} is the AOS channel response function (see Appendix A), ν_i is the frequency of the i -th channel, ν^{lsb} and $\nu^{\text{usb}} = 2\nu_{\text{lo}} - \nu^{\text{lsb}}$ are the lower and upper bands frequencies, R^{usb} and R^{lsb} are the upper and lower sidebands rejection values (see Appendix B). The term before the integral operator is the Rayleigh-Jeans factor and the Boltzmann constant $k_b = 1.380662 \times 10^{-23} \text{ J K}^{-1}$. The formula for band C radiance is obtained by interchanging the superscript labels usb and lsb.

The contribution of the image band has been **neglected** in the current processing: $R^{\text{usb}} = R^{\text{lsb}} = 0$. It will be shown in Sect. 5.2 that the impact of such simplification is small compared to current calibration errors. The atmosphere is **assumed to be** horizontally stratified and only the integration in the vertical direction is performed. The vertical effective antenna pattern $G_e^{\text{ANT}}(\theta, \theta_j, \omega)$ includes the rotation ω of the 2-D antenna pattern with respect to the line of sight, the integration over the azimuth angle of the 2D antenna pattern and the vertical scanning motion (Shiotani et al., 2002, 40–52 pp.). The vertical integration range $\Delta\theta$ is -4.2 to 4.2° ; The contribution from outside this range is removed in the calibration procedure.

The specific intensity is computed by solving the radiative transfer equation along the optical path:

$$I^{\text{atm}}(\nu, \theta) = \int_{\text{PATH}} B(T_s, \nu) \alpha(s, \nu) \exp^{\int_{s'=s}^{\text{sensor}} -\alpha(s', \nu) ds'} ds + I_0(\nu, \theta),$$

where s and s' indicate the positions on the integration path, α is the absorption coefficient, $B(T, \nu)$ is the Planck function at temperature T and I_0 is the background radiation (Earth

3 surface, or cold space at the physical temperature of 2.7 K giving a brightness temperature
 4 of 4×10^{-4} K). The atmosphere is assumed to be in the hydrostatic equilibrium. Scattering
 5 and absorption from solid and liquid particles are not important for submillimeter wavelengths
 6 except in the presence of relatively thick clouds in the upper troposphere. The bending of the
 7 optical path by atmospheric refraction is taken into account by using a non dispersive expression
 8 of the refractive index n_0 (Rüeger, 2002):

$$9 \quad n_0 = 1 + 10^{-8} \left(77.6890 \frac{P_d}{T} + \frac{P_w}{T} \left(71.2952 + \frac{375463}{T} \right) \right),$$

10 where P_d and P_w are the dry air and water vapor partial pressures (Pa). The dry air refractive
 11 contribution assumes a CO₂ content of 375 ppm.

12 The absorption coefficient includes the molecular spectroscopic lines and the absorption con-
 13 tinua from dry air and water vapor. The continua parametrisation (Pardo et al., 2001) has been
 14 derived from atmospheric measurements at submillimeter wavelengths. It should be noted that
 15 at SMILES frequencies, the measured dry air continuum is 20–30 % below theoretical estima-
 16 tions (Boissoles et al., 2003). Such error has a significant impact on the retrieved UT/LS data
 17 and should be checked in future versions.

18 The spectroscopic line shape is a Van-Vleck and Weisskopf (VVW) profile in the lower at-
 19 mosphere where the line Doppler broadening is 40 times smaller than the collisional (pressure)
 20 broadening. At higher altitudes, a Voigt profile is used (Schreier and Kohlert, 2008). The col-
 21 lisional line broadening parameter includes the self-broadening parameter when available. The
 22 line frequency ν_q is shifted by the atmospheric pressure:

$$23 \quad \nu_q = \left(\nu_q^0 + \delta_\nu^q P \left(\frac{T_0}{T} \right)^{n_f} \right),$$

24 where ν_q^0 is the line frequency and δ_ν^q is the pressure induced frequency shift parameter at
 25 temperature T_0 and n_f is its temperature dependence.

26 A catalogue of spectroscopic line parameters between 0 and 1 THz has been created
 1 using parameters from the JPL catalogue (Pickett et al., 1998), completed with the line
 2 pressure-broadening and frequency pressure-shift parameters from the HITRAN database

(Rothman et al., 2009). The parameters have been updated with the latest laboratory measurements (Tab. 3). Less than 1200 lines per band are considered in line-by-line calculations. The line selection algorithm selects all the lines inside the bands that have an intensity larger than 10^{-4} K between 15–90 km altitudes. The most significant out-of-band lines of O_3 and H_2O are selected. For both molecules, selected out-of-band lines contribute to 99 % of the **absorption by** the out-of-band lines calculated with no selection at 5 regularly-spaced frequencies in the band. The same method is used for out-of-band lines of other molecules but the selection is performed for all species together and with a line selection threshold of 90 %. **The line selection procedure selects a large number of relatively weak spectral lines located below 100 GHz and a set of strong O_2 lines at ~ 60 GHz. These lines should not be selected since the computation of their absorption is overestimated at the SMILES frequency: *i*) the VVW lineshape overestimates the absorption with an order of magnitude at frequencies larger than 5 times the resonant frequency (Harde et al., 1995), *ii*) the actual 60 GHz- O_2 lines absorption is reduced by a line mixing effect which is not taken into account in the calculation. All lines below 100 GHz are not included in the line selection and, then, are not taken into account in retrieval processing.**

The absorption coefficient and the radiative transfer are computed on a non equidistant frequency grid composed of less than 1500 points that is defined to avoid errors larger than 0.5 K. Errors are estimated with the comparison of AOS radiances with those obtained with a fine frequency grid of 0.25 MHz resolution.

5 Methodology for the inversion and for the errors characterisation

5.1 Inversion method

Within each retrieval process, a least-squares method with regularisation is used to seek for the value of the forward model's unknown parameters x that minimizes the cost function χ^2 . The latter is given by

$$\chi^2 = \frac{(\mathbf{y} - \mathbf{F}(\mathbf{x}, \mathbf{b}))^T \mathbf{S}_y^{-1} (\mathbf{y} - \mathbf{F}(\mathbf{x}, \mathbf{b})) + (\mathbf{x}_a - \mathbf{x})^T \mathbf{S}_a^{-1} (\mathbf{x}_a - \mathbf{x})}{n_y + n_x}, \quad (2)$$

where \mathbf{y} is the measurement vector of n_y elements, \mathbf{S}_y is the measurement covariance matrix, $\mathbf{F}(\mathbf{x}, \mathbf{b})$ is the forward model (Eq. 1) depending on \mathbf{x} and on the known model parameters \mathbf{b} , n_x is the size of the vector \mathbf{x} , \mathbf{x}_a is the a priori value of \mathbf{x} and \mathbf{S}_a the a priori covariance matrix. The a priori term corresponds to n_x additional virtual measurements used to regularize the inversion process (Rodgers, 2000). The off-diagonal elements of \mathbf{S}_y are assumed to be null and the diagonal elements are set to the square of the thermal noise level. For SMILES, the latter is determined using the following form of the radiometric noise equation:

$$\epsilon_n = \frac{T_{\text{rec}}(\nu_i) + T(\nu_i, \theta)}{\sqrt{\beta d\tau}}, \quad (3)$$

with the receiver temperature (noise power generated by the receiver), T_{rec} , between 300–350 K, the atmosphere temperature received by the antenna (Eq. 1), T , between 0–250 K, the noise-equivalent spectral resolution $\beta = 2.5$ MHz, and the integration time of a single spectrum $d\tau = 0.5$ s. The actual measurement noise is thus ~ 0.3 K but a fixed value of 0.5 K is used in this analysis.

The a priori covariance matrix is built using an a priori error ϵ_a that does not correspond to the natural variability of \mathbf{x} , but to the measurement sensitivity. For instance, a large value of ϵ_a (between 100–200 % of a typical profile) is used for species such as BrO or H₂O₂ that are retrieved from weak noisy lines. The retrieved profiles are then noisy and must be averaged to increase the precision. An altitude correlation in the a priori covariance matrix is used for each product according to the vertical resolution estimated from preliminary sensitivity studies (see Sect. 5.2). The correlation values are between 2 and 6 km. The a priori value of O₃, H₂O and temperature/pressure are from the GEOS-5 analysis (see Sect. 3). As the version of GEOS-5 model used in this analysis is valid below about 70 km, the temperature and pressure profiles are extrapolated to 110 km using the MSIS temperature climatology (Hedin, 1991) with a method similar to that used for the Odin/SMR level-2 processing. Other GEOS-5 profiles are extrapolated with a constant value. Other species are described by a unique profile obtained

from different climatologies, a low confidence level is given to these data. Table 4 summarized the a priori data used in the processing.

To solve Eq. (2), the vector \mathbf{x} is scaled with respect to \mathbf{x}_a and to the a priori error ϵ_a :

$$\boldsymbol{\eta} = (\mathbf{x} - \mathbf{x}_a) / \epsilon_a \quad (4)$$

The solution is found with a Gauss-Newton iterative procedure modified by Levenberg (Rodgers, 2000, Chap. 5.7, Eq. 5.35).

$$\boldsymbol{\eta}_{i+1} = \boldsymbol{\eta}_i + (\mathbf{K}_i^T \mathbf{S}_y^{-1} \mathbf{K}_i + \mathbf{S}_\eta^{-1} + \gamma \mathbf{I})^{-1} (\mathbf{K}_i^T \mathbf{S}_y^{-1} (\mathbf{y} - \mathbf{F}_i) - \mathbf{S}_\eta^{-1} \boldsymbol{\eta}_i), \quad (5)$$

where the subscripts i and $i + 1$ indicate the iteration stage, \mathbf{F}_i is the forward model output for $\mathbf{x} = \mathbf{x}_i$, $\mathbf{K}_i = \left(\frac{\partial \mathbf{F}}{\partial \boldsymbol{\eta}} \right)_i$ is the weighting function matrix, \mathbf{S}_η is the covariance matrix of the scaled vector $\boldsymbol{\eta}$, γ is the Levenberg-Marquardt parameter and \mathbf{I} is a diagonal scaling matrix. The diagonal elements of \mathbf{I} are set to 1, except for the components of \mathbf{x} that have a small impact on the χ^2 and for which the value is zero. The a priori state of $\boldsymbol{\eta}$ is null and it is not written in Eq. (5). The covariance matrix \mathbf{S}_η is derived from the a priori covariance matrix as:

$$\mathbf{S}_\eta = \mathbf{D} (1/\epsilon_a) \mathbf{S}_a \mathbf{D} (1/\epsilon_a)^T, \quad (6)$$

with $\mathbf{D}(1/\epsilon_a)$ the diagonal matrix with diagonal elements equal to the vector $1/\epsilon_a$. The weighting function matrix \mathbf{K}_i is related to the Jacobian matrix of the unscaled parameters as below:

$$K_i[m, n] = \left(\frac{\partial F[m]}{\partial x[n]} \right)_i \epsilon_a[m], \quad (7)$$

where m and n are the measurement and parameters indexes, respectively.

If temperature is retrieved, the pressure profile can be derived from the hydrostatic equilibrium using a known pressure and temperature at a reference altitude. We use a reference altitude of 18 km and data from the GEOS-5 analysis. The pressure profile is computed above the reference altitude on a fine resolution vertical grid of 100 m. The temperature profile is linearly interpolated on this fine grid and the discrete hydrostatic equation becomes:

$$\frac{P[i + 1] - P[i]}{P[i]} = -g(\Phi) \left(1 - \frac{\bar{z}[i]}{r(\Phi)} \right) \frac{M_a}{R} \frac{z[i + 1] - z[i]}{\bar{T}[i]}, \quad (8)$$

3 where i and $i + 1$ are the indices of two consecutive vertical levels, $P[i]$ (hPa) is the pressure at
 4 the altitude z_i (m), Φ (rad) is the latitude angle, $r(\Phi) = 6\,378\,130 \left(1 - \frac{\sin(\Phi)^2}{298.25}\right)$ is the effective
 5 Earth radius (m), $g(\Phi) = 9.80616(1 - 0.0026373\cos(2\Phi) + 5.9 \times 10^{-6}\cos(2\Phi)^2)$ is the gravity
 6 at the surface (m s^{-2}), $R = 8.3145 \text{ J K}^{-1} \text{ mol}^{-1}$ is the gas constant, $M_a = 0.028964 \text{ kg mol}^{-1}$ is
 7 the molar mass of air, and $\bar{z}[i]$ and $\bar{T}[i]$ are the mean altitude and the mean temperature (K)
 8 between the levels i and $i + 1$. The pressure profile is a retrieved parameter although it is not a
 9 direct product of the inversion computation (Eq. 5). Note that it could have been retrieved along
 10 with other parameters by adding the hydrostatic equilibrium equation as an additional constraint
 11 of the inversion (Carlotti and Rodolfi, 1999).

12 The implementation of the inversion procedure is presented in the flowchart in Fig. 3. After
 13 the parameters initialisation, the forward model is run to calculate the simulated measurements
 14 and their weighting functions and to estimate χ^2 (step 1). In step 2, the scaled vector $\boldsymbol{\eta}$ is
 15 calculated and the Marquardt loop counter is set to 0. The inversion is performed in step 3
 16 according to Eq. (5), the retrieved vector is unscaled and the pressure profile is reconstructed (if
 17 needed) from the retrieved temperature. The forward model is computed for \boldsymbol{x}_{i+1} and the new
 18 χ^2 is estimated. If χ^2 increases, the process enters into the Marquardt branch: the Marquardt
 19 loop counter is incremented, γ is multiplied by 3 and the sequence of operations from step 3 is
 20 repeated. If χ^2 decreases and the final loop criteria are not fulfilled, the main loop counter is
 21 incremented, γ is divided by 3 and the procedure is repeated from step 1. These operations are
 22 performed until the change in χ^2 is less than 0.05 or the number of iterations is more than 12 in
 23 the main loop or more than 5 in the Marquardt loop. The procedure has converged if the χ^2 has
 24 decreased. A good convergence is reached for a final χ^2 between 0.6–2 and a small value of γ
 1 below 0.5. Otherwise, a warning flag is raised. Note that the lower value of the χ^2 is less than 1
 2 because of the overestimation of the error in the measurement covariance matrix \mathbf{S}_y .

3 5.2 Characterisation of the errors

4 5.2.1 Method for the errors calculation

5 The smoothing and measurement covariance matrices as defined in Marks and Rodgers (1993)
6 are naturally computed for the retrieved scaled parameters:

$$7 \mathbf{S}_n = (\mathbf{I} - \mathbf{A}) \mathbf{S}_\eta (\mathbf{I} - \mathbf{A})^T \text{ and } \mathbf{S}_m = \mathbf{G} \mathbf{S}_y \mathbf{G}^T \quad (9)$$

8 where $G = (\mathbf{K}^T \mathbf{S}_y^{-1} \mathbf{K} + \mathbf{S}_\eta^{-1})^{-1} \mathbf{K}^T \mathbf{S}_y^{-1}$ and $\mathbf{A} = \mathbf{G} \mathbf{K}$ is the averaging kernel matrix. The
9 smoothing error can be interpreted as the error caused by the limited vertical resolution of $\hat{\boldsymbol{\eta}}$
10 and as the error due to the contaminations from the simultaneously retrieved parameters (Baron
11 et al., 2002). The vertical resolution is estimated from the width of the averaging kernels. Nev-
12 ertheless interpreting the smoothing error as a **measurement** of the vertical resolution makes
13 sense only where the contribution from the a priori is small, otherwise the smoothing error
14 should be interpreted as a bias toward the a priori profiles. In order to evaluate the a priori
15 contribution, we use the measurement response: $W = \sum_j (|\mathbf{A}[i, j]|)$. In this study we consider
16 only that the vertical range of a good retrieval is for altitudes where $|W - 1| < 0.2$.

17 The measurement response, the vertical resolution and the smoothing and measurement er-
18 rors on the unscaled retrieved parameters $\hat{\boldsymbol{x}}$ are provided along with the L2r data. The errors
19 are derived from the square root of the diagonal elements of \mathbf{S}_n and \mathbf{S}_m . They will be known
20 hereafter as linear mapping errors. However the linear mapping provides a poor representation
21 of the actual errors. In our definition, the covariance matrix \mathbf{S}_a is not representative of the real
22 uncertainties on the a priori profiles (see Sect. 5.1) and the retrieval sub-grid size atmospheric
23 variability is not taken into account in the smoothing error. Furthermore, the measurement co-
24 variance matrix \mathbf{S}_y takes into account the measurement noise only, and errors on the forward
25 model parameters are not included. Parameters retrieved from one process can be used in the
26 following ones, causing the errors to propagate throughout the processes. Also, processes can
27 share common measurements components, causing errors correlation. For example, the line-
1 of-sight elevation angles and the O_3 profile are retrieved from two different processes which
2 use the same spectral line. Below, we describe the method we used to take into account the

inter-process propagation and correlation of the errors as well as the retrieval sub-grid size atmospheric variability.

The inversion method, implemented for each process, is expressed as the inverse function I of the difference between the measurement and the forward model:

$$\hat{\mathbf{x}} = I(\mathbf{y} - \mathbf{F}(\mathbf{x}_i, \hat{\mathbf{p}}, \mathbf{b}), \mathbf{x}_a), \quad (10)$$

where $\hat{\mathbf{x}}$ is the estimate of the parameter \mathbf{x} and \mathbf{x}_a is its a priori value, \mathbf{y} is the measurement and \mathbf{F} is the forward model, $\hat{\mathbf{p}}$ is the vector of parameters retrieved in a previous process and \mathbf{b} is the vector of known forward model parameters. The vector \mathbf{x}_i is initialised with \mathbf{x}_a and after convergence should be equal to $\hat{\mathbf{x}}$. In the following \mathbf{y} will not be the real measurement but a simulation depending on \mathbf{x} , \mathbf{b} and \mathbf{p} . Note that \mathbf{x}_a and $\hat{\mathbf{x}}$ are defined on a retrieval grid with steps of 2–6 km and \mathbf{x} is defined on a fine vertical grid with steps of 250 m.

The retrieved error $\hat{\epsilon}_x$ on $\hat{\mathbf{x}}$ is derived by linearising the inverse function with respect to the "uncorrect" parameters (Rodgers, 2000). It gives for Eq. (10):

$$\hat{\epsilon}_x = \left(\frac{\partial I}{\partial \mathbf{y}} + \frac{\partial I}{\partial \mathbf{F}} \frac{\partial \mathbf{F}}{\partial \hat{\mathbf{p}}} \frac{\partial \hat{\mathbf{p}}}{\partial \mathbf{y}} \right) \epsilon_y + \left(\frac{\partial I}{\partial \mathbf{F}} \frac{\partial \mathbf{F}}{\partial \mathbf{b}} + \frac{\partial I}{\partial \mathbf{F}} \frac{\partial \mathbf{F}}{\partial \hat{\mathbf{p}}} \frac{\partial \hat{\mathbf{p}}}{\partial \mathbf{b}} \right) \epsilon_b + \left(\frac{\partial I}{\partial \mathbf{y}} \frac{\partial \mathbf{y}}{\partial \mathbf{x}} + \frac{\partial I}{\partial \mathbf{F}} \frac{\partial \mathbf{F}}{\partial \hat{\mathbf{p}}} \frac{\partial \hat{\mathbf{p}}}{\partial \mathbf{x}} \right) \epsilon_x + \frac{\partial I}{\partial \mathbf{F}} \frac{\partial \mathbf{F}}{\partial \hat{\mathbf{p}}} \hat{\epsilon}_p, \quad (11)$$

where ϵ_y is the error on the measurement, ϵ_b is the error on \mathbf{b} , ϵ_x is the error on the first guess of \mathbf{x} , $\hat{\epsilon}_p$ is the error on $\hat{\mathbf{p}}$ from errors that do not explicitly appear in the error budget. The latter can include, for example, the errors on the measurement that are not used in the current process. The terms depending on $\frac{\partial \mathbf{F}}{\partial \hat{\mathbf{p}}}$ represent the inter-process error propagation and correlation.

The different terms can be estimated by perturbing the parameters of either the simulated measurement (\mathbf{y}) or the forward model (\mathbf{F}). In this study, we have chosen to perturb the measurement parameters. We used two methods, one based on finite difference calculation and the second based on statistical evaluation (e.g., standard deviation) of a set of retrieved profiles (100 simulations are performed for each parameter vector of interest). The choice of the method depends on the size of parameter vector to be perturbed. Perturbing a vector β which can be a part of either \mathbf{x} , \mathbf{p} or \mathbf{b} , with uncorrelated components gives a variance of the retrieved state vector calculated as follow:

$$\begin{cases} \hat{\sigma}_\beta = \sum_i [I(\mathbf{y}(\beta[i] + \bar{\epsilon}_\beta[i])) - I(\mathbf{y}))^2 \text{ for a finite difference analysis} \\ \hat{\sigma}_\beta = \text{STD} \{I(\mathbf{y}(\beta + \epsilon_\beta))\}^2 \text{ for a statistical analysis} \end{cases} \quad (12)$$

3 Here \mathbf{y} is the perturbed simulated measurement, $\hat{\sigma}_\beta$ is the retrieval error variance due to the
 4 vector β , $\bar{\epsilon}_\beta[i]$ is the error on the i -th element of β , ϵ_β is a random vector of standard deviations
 5 $\bar{\epsilon}_\beta$ and STD denotes the standard deviation of a set of retrieval outputs. To take into account cor-
 6 relations between the β components, independent perturbations are applied to the eigenvectors
 7 of the β covariance matrix. For example, the finite difference case becomes:

$$8 \quad \hat{\sigma}_\beta = \sum_i (I(\mathbf{y}(\beta + \mathbf{U}\bar{\epsilon}_\Lambda^i)) - I(\mathbf{y}))^2, \quad (13)$$

9 where \mathbf{U} and Λ are the eigenvectors and eigenvalues matrices of the β covariance matrix \mathbf{S}_β
 10 ($\mathbf{S}_\beta = \mathbf{U}\Lambda\mathbf{U}^T$), and $\bar{\epsilon}_\Lambda^i$ is a vector describing the error on the i -th eigenvector in the eigenvector
 11 space:

$$12 \quad \epsilon_\Lambda^i[j] = \begin{cases} \sqrt{\Lambda[i,i]} & \text{if } j = i \\ 0 & \text{if } j \neq i. \end{cases}$$

13 **5.2.2 Error assumptions**

14 The SMILES calibrated radiance (version 5) has a large error (between -1 and 10 K) likely due
 15 to the non-correction of the radiometric gain non-linearity (Ochiai et al., 2011). To study this
 16 effect, the simulated measurement \mathbf{y} is computed from a modified form of the forward model
 17 (Eq. 5):

$$18 \quad \mathbf{y} = C_{\text{cal},0} + C_{\text{cal},1} \times F(\mathbf{x}, \mathbf{p}, \mathbf{b}) + N_{\text{cal}}(F(\mathbf{x}, \mathbf{p}, \mathbf{b})) \quad (14)$$

19 where N_{cal} is a non-linear transformation to include the effect of the radiometric gain com-
 20 pression, and the coefficients $C_{\text{cal},0} = 1$ K and $C_{\text{cal},1} = 1.01$ are the antenna spill-over and the
 21 radiometric gain error. The details about the calibration procedure and the gain compression
 22 errors are given in Appendix C. The value of $C_{\text{cal},0}$, $C_{\text{cal},1}$ are upper limits deduced from the
 23 data, and the non-linear transformation is based on the latest calibration model provided by the
 24 instrument team.

1 Other errors are summarised in Table 5. They can be classified as random or systematic
2 errors with respect to a single profile measurement. Systematic errors arise from the calibra-
3 tion parameters, the AOS channel width, neglecting the image band contribution, the dry air
4 continuum and the spectroscopic parameters. Only the pressure-broadening parameters of the
5 strongest lines in bands A and B are included in this study.

6 The random errors are the measurement thermal noise and the errors on the line-of-sight
7 angles and on the atmospheric profiles. Uncorrelated errors of $0.005\text{--}0.01^\circ$ (errors of 200–
8 350 m on the tangent point altitude) are assumed on the line-of-sight angles (Shiotani et al.,
9 2002, 60p.). We have chosen to use an error of 0.007° . In reality, the value might be lower and
10 more investigations have to be conducted to assess the actual line-of-sight angle errors.

11 **6 Performance of the retrieval procedure**

12 The objectives of this section are multiple. We first demonstrate and evaluate the capability
13 of retrieving the line-of-sight elevation angles along with the O_3 and temperature profiles from
14 the strong O_3 line at 625.371 GHz. We show that temperature is retrieved with low precision
15 and accuracy and recommend to use pressure as vertical coordinates for data retrieved in the
16 altitude range where the line-of-sight tangent altitudes are retrieved. We also show the **value**
17 of reducing the width of the spectral windows containing weak lines in order to decrease the
18 contamination from neighbouring lines. This issue is discussed through the error analysis for
19 the HOCl retrieved profile.

20 The assumed uncertainties on the measurements and on the forward model parameters are
21 given in Table 2. Retrieval errors have been assessed for all the parameters listed in this table
22 but only the results with an impact larger than 1 % are shown. The O_3 , HCl and HOCl profiles
23 used for the error characterisation are shown in Fig. 4.

24 6.1 Retrieval of the line-of-sight elevation angles

25 Line-of-sight angles are retrieved from the processes A-w0 and B-w0 (Table 2). The spectral
1 window is the same for both bands. It is centred at 625.371 GHz on the strong O₃ line, and
2 extends over at least 200 MHz on both sides of the line center. Figure 4 shows the O₃ line mea-
3 sured at a tangent height of 20 km. Given a **pressure-induced line half-width** of 2.198 MHz/hPa,
4 information can theoretically be retrieved from the line shape down to ~ 90 hPa, i.e., down
5 to an altitude of ~ 17 km. The spectral window setting also reduces the contamination from
6 the neighbouring HCl lines. The profiles of O₃, temperature, HCl isotopomers, H₂O, HOCl
7 and OO¹⁸O are simultaneously retrieved on a sparse vertical grid of 3 km step below 30 km and
8 5 km above. They are considered as interfering parameters. **Three O₃ profiles are retrieved from**
9 **the fits of the main line at 625.371 GHz and of two minor lines of the molecule, respectively.**
10 **The minor lines correspond to transitions in the first level ($\nu=1$) of the O₃ vibrational states $\nu=2$**
11 **and $\nu=(1,3)$, respectively.**

12 A frequency offset for the full scan is also retrieved to account for the AOS channel frequency
13 calibration and the O₃ line frequency errors. In addition, a linear baseline (offset and slope) is
14 retrieved for each spectrum to remove radiance offsets due to reflections in the instrument optics
15 and to the antenna spill-over. The inversion iterative cycle usually converges in 5–7 iterations.

16 The averaging kernels and the measurement response (Fig. 6, left panel) show that the line-of-
17 sight angles are retrieved between 18 and 60 km. The high value of the measurement response
18 and the well-peaked averaging kernels at 18 km show that the lowest limit of the retrieval alti-
19 tude range could theoretically have been pushed further down. The lowest altitude of the
20 retrieval grid has been chosen at 18 km in order to avoid errors from the continuum absorption
21 modelling and contamination from ice particles scattering. The random errors (precision) of
22 the angles retrieved from a single scan are between 0.004–0.005° (~ 200 m on the tangent point
23 altitude) below 50 km (Fig. 7, left panel). Pressure is the dominant source of random errors.
24 Errors from the measurement noise and other atmospheric parameters are lower than 0.002°.
25 Above 50 km, the random error due to the thermal noise becomes predominant and the total
26 error is about 0.007°.

27 The right panel of Fig. 7 shows the systematic errors (accuracy) on the retrieved line-of-sight
28 angles. Neglecting the non-linearity of the calibration gain induced a systematic error of 0.004–
29 0.007° (error of 160–280 m). Between 30 and 50 km, the error due to the O₃ line pressure-
1 broadening parameter (0.003 to 0.004°) is the second source of systematic error. Above 50 km,
2 the systematic error increases strongly up to 0.008°, mainly due to the error on the AOS channel
3 width. Below 20 km, errors is estimated to 0.008° due to non-linearity calibration gain and to
4 the absorption continuum from dry air.

5 According to this results, the line-of-sight elevation angles is retrieved from the strong O₃
6 line between 20–50 km.

7 **6.2 O₃ and temperature retrievals**

8 Temperature and O₃ are simultaneously retrieved from the processes A-w1 and B-w1. The
9 spectral characteristics of the widows are the same as for the w0 window used for the line-of-
10 sight angles retrieval (Fig. 4, left panel). Spectra with line-of-sight tangent heights between
11 16–100 km are selected. The retrieval vertical grid for atmospheric parameters has a spac-
12 ing of 2 km between 18–30 km, 3 km between 30–60 km, 4 km between 60–80 km and 6 km
13 above. The retrieval grid corresponds to the vertical resolution estimated from a preliminary
14 error analysis. A mean offset on the line-of-sight angles is also retrieved. However we found
15 that in general this offset is small (less than 0.001°), indicating that the line-of-sight angles are
16 correctly updated from process w0. Other retrieval settings such as other contaminating species
17 and a priori and covariance matrices are the same as for process w0. The averaging kernels
18 analysis shows that the O₃ profile can be retrieved from 18 to 90 km with a vertical resolution
19 close to the retrieval grid sampling (Fig. 8).

20 The random errors (precision) on the retrieved O₃ profile are shown in the left panels of
21 Fig. 9. The upper and the lower panels show the absolute and relative errors, respectively.
22 Between 28–50 km, the random error is 5–7 % (0.4–0.7 ppmv). The main error source is from
23 the pressure profile. The errors for the line-of-sight angles are efficiently corrected and are
24 responsible of less than 2 % error on the O₃ profile. Below 28 km, the error on the line-of-sight

25 angles becomes the main source of random error (50 % at 20 km). However, it is smaller than
26 the value obtained if the angles were not retrieved (Fig. 10). Between 50–75 km, where it cannot
27 be retrieved, the line-of-sight angles is the main error source (20 %). Above 75 km, the O₃ line
1 intensity decreases strongly and the measurement noise becomes the main source of random
2 error on the retrieved O₃ profile.

3 The systematic errors (accuracy) on the retrieved O₃ profile are presented on the right pan-
4 els of Fig. 9. Between 20 km and 50 km, a systematic error of 5–15 % is found that is mainly
5 induced by the non-linearity of the calibration gain. Below 30 km, the error from the dry air
6 continuum is significant (5 % at 25 km), and becomes the main systematic error on the retrieved
7 O₃ profile below 21 km. The error due to the O₃ line pressure-broadening parameter is signifi-
8 cant above 30 km (4–5 %). Above 50 km, the systematic error is 7–9 %, due to the errors on the
9 AOS channel width, on the O₃ line pressure-broadening parameter and on the non-linearity of
10 the calibration gain.

11 Temperature is retrieved between 18–55 km with a vertical resolution of 6–14 km (Fig. 11).
12 A random error of 2–10 K (Fig. 12) is induced by errors on the O₃ and pressure profiles. The
13 error is significantly larger than the smoothing error shown in the central panel of Fig. 11 (less
14 than 2 K). The difference is due to oscillations on the temperature profiles retrieved from the
15 simulated measurements. It is thus an error inherent to the inversion regularisation. A different
16 regularisation should be implemented to improve the results.

17 As for O₃, the non-linearity on the calibration gain (Fig. 12, left panel) is the main systematic
18 error. It is responsible of a large error of 8–10 K. At 20 km, the dry air continuum induces a
19 significant systematic error of 2 K.

20 The results for temperature are not satisfactory and must be improved. Consequently, atmo-
21 spheric pressure is not derived from the retrieved temperature in version 2 of the processing. It
22 becomes the main source of random errors on the retrieved O₃ profile and on the line-of-sight
23 angles.

24 6.3 Use of pressure as vertical coordinates

25 If the line-of-sight elevation angle is retrieved, the height of the tangent point is adjusted to
26 the pressure level corresponding to the O_3 line width. Since the pressure at the line-of-sight
27 tangent point is correct, the error on the pressure profile should be considered as an error on the
28 altitude and not on the pressure itself. As the molecular abundance is retrieved at the tangent
1 point, it is then better to represent the retrieved molecular abundance on pressure levels instead
2 of altitude levels. Retrieved data are distributed along with the altitude and the pressure of the
3 vertical levels. Figure 13 shows the O_3 profile retrieved from a simulated measurement vector.
4 The measurement vector was computed using atmospheric profiles (true profiles) from a winter
5 mid-latitude climatology while the a priori profiles are from a tropical climatology. In addition,
6 the true pressure profile was multiplied by 0.95 in order to highlight errors induced by the non-
7 correction of the pressure profile using the retrieved temperature. A random noise of 0.4 K
8 (1-sigma) was added to the simulated radiances. As expected, the difference between the true
9 and the retrieved O_3 profiles is smaller if it is estimated on pressure levels (right panel) than on
10 altitude levels (left panel). Between 30-50 km, a difference of 4-6% remains due to the large
11 error on the a priori pressure profile (20-30%).

12 The validation works on O_3 and HCl products on both vertical representations are currently
13 under way. These studies should provide insight about the relative importance of the errors
14 induced by the GEOS-5 pressure and by the line-of-sight elevation angles.

15 In the case of a small error on the line-of-sight angle derived from the attitude data, the
16 pressure error at the tangent point can be derived from the retrieved tangent altitude using the
17 hydrostatic equation (Eq. 8). Hence, the temperature profile, retrieved from the O_3 line cen-
18 ter, and the line-of-sight angles, retrieved from the wings, provide complementary information
19 about pressure. The combination of both sources of information to retrieve the pressure profile
20 is currently under investigation.

21 6.4 HOCl retrieval

22 Two HOCl lines of similar intensity are measured at 624.378 GHz and 625.075 GHz (Fig. 1).
23 The first line is measured in band A and the second is measured both in bands A and B. Although
24 the line at 624.378 GHz is less contaminated by other spectral lines, the retrieval quality has
25 been found to be lower than for the line at 625.075 GHz. Hence only the second line is used in
26 the current processing.

27 The line is measured in windows A-w1, B-w1, A-w4 and B-w3 (Fig. 4). The same frequency
28 range of ~ 200 MHz width is used for A-w4 and B-w3.

1 The a priori error on the HOCl profile is set to 100 % or 0.1 ppbv (whichever is greater) over
2 the whole altitude range. The retrieval setting is the same as the one for A/B-w1 described in
3 Sect. 6.2. For the inversion of A-w4 and B-w3, an altitude correlation of 6 km is used in the
4 a priori covariance matrix. HOCl is retrieved simultaneously with H₂O₂, OO¹⁸O and O₃ ν 1,3.
5 A quadratic baseline is retrieved to account for radiance offset induced by the instrument and
6 the emission from the out-of-window lines. A mean frequency offset over the full scan is also
7 retrieved using the relatively strong O₃ ν 1,3 line. As the characteristics of band A windows are
8 similar to those of band B windows, only the results for band A (A-w1, A-w4) will be discussed.

9 From the analysis of the averaging kernel matrix, HOCl profile is retrieved from window
10 A-w1 between 25 and 60 km (Fig. 14, upper left panel) with a vertical resolution between
11 6 and 14 km (Fig. 14, upper right panel). The total random error on the HOCl retrieved profile
12 (Fig. 15, upper left panel) is 0.05–0.09 ppbv between 25 and 40 km. The main error source is
13 the measurement noise. Atmospheric parameters retrieved in previous processes also induce a
14 significant random error of 0.03–0.05 ppbv. The systematic error (Fig. 15, upper right panel)
15 found for window w1 is 0.05 ppbv at 25 km and decreases to below 0.005 ppbv at 50 km. The
16 line pressure-broadening parameters of nearby spectroscopic lines and the non-linear calibration
17 gain are the main errors, with amplitudes between 0.01–0.04 ppbv.

18 From window A-w4, the vertical profile of HOCl is retrieved from 30 to 65 km (Fig. 14, lower
19 panels). Hence, reducing the spectral range induces a truncation of the 25–30 km layer of the
20 retrieval vertical range. Otherwise, the vertical resolution, the total random and total systematic

21 errors have similar values to those found for the inversion of A-w1. It is important to note that
22 for A-w4, the random error is mainly due to measurement noise. Contamination from other
23 atmospheric profiles (temperature, pressure, O₃) is not significant (less than 0.02 ppbv). The
24 systematic error is from the O₃ ν 1,3 line pressure-broadening parameter. Errors from other lines
25 and calibration are not significant (less than 0.01 ppbv)

26 Using a narrower frequency bandwidth clearly decreases the contamination from other spec-
27 tral lines, but slightly degrades the sensitivity at lowest altitudes.

1 7 Conclusions

2 We have presented version 2 of the retrieval algorithms for the SMILES level-2 research pro-
3 cessing chain. The retrieval procedure is based on sequentially dependent processes. In each
4 process, a reduced spectral window is used with an optimised retrieval setting. The forward
5 model and the inversion method have been presented. The algorithms have been designed for
6 the altitude range between 18–100 km. It is not recommended to use the data retrieved below
7 this range. The line-of-sight angles are retrieved between 20 and 50 km from the strong O₃ lines
8 along with the temperature and the O₃ VMR profile. An error analysis has been conducted to
9 show that the correlations between these parameters are low and that these parameters can ef-
10 fectively be retrieved. The pressure profile is found to be the limiting parameter of the precision
11 for the O₃ profile and for the line-of-sight angles. However, the error induced by the pressure
12 can significantly be decreased if the molecular VMR profiles are represented on pressure levels
13 in the altitude range where the line-of-sight tangent altitudes are retrieved. Neglecting the non-
14 linearity of the gain in the version 5 of the calibration procedure is the main systematic error.
15 The next L2r version will use calibrated radiances including this effect.

16 An error analysis for the HOCl retrieved profile has been presented to illustrate the inversion
17 of weak lines. It is shown that using narrow spectral windows around the line of interest allows
18 to reduce the contamination from other atmospheric parameters, neighbouring spectroscopic
19 lines and calibration errors.

20 Temperature is retrieved with low accuracy and precision and must be improved in future

versions. Better calibration procedure and AOS channel response should increase the accuracy. Current works investigate ways to increase the temperature precision using a new regularisation scheme in the inversion calculation, and by joining the strong HCl and O₃ lines in the same retrieval process.

Work is also being conducted to derive reliable information in the UT/LS about molecular abundances (H₂O, O₃ and HCl) and on ice content. The retrieval of horizontal wind in the stratosphere and mesosphere along the line-of-sight direction is also under investigation.

Appendix A AOS filter shape

The AOS channel response function parametrisation in version 2 of the L2r chain has been provided before launch by the instrument teams. For both AOS units, the response function $g^{\text{AOS}}(\nu)$ is defined by three Gaussian functions:

$$g^{\text{AOS}}(\nu) = \sum_{i=1,3} \frac{A_i}{w_i \sqrt{(2\pi)}} \exp \left[-2 \left(\frac{\nu - x_{0,i}}{w_i} \right)^2 \right], \quad (\text{A1})$$

where ν is the frequency in MHz.

The amplitude (A_i), the width in MHz (w_i) and the center in MHz (x_i) are computed for each channel $j = 1 \dots 1728$ using polynomial coefficients C_p

$$X[j] = \sum_{p=0,4} C_p \cdot (j - 1)^p, \quad (\text{A2})$$

where X can be one of the Gaussian parameters A_i or w_i or x_i . The values of the coefficients C_p are given in Table 7.

Large errors on the AOS filter shape have been identified from the analysis of retrieved temperature or line-of-sight angles from spectra measured in the upper stratosphere. Ozeki et al. (2011) have estimated that the channel width (w_i) of AOS units 1 and 2 should be multiplied by 0.7 and 1.1, respectively. These corrections are applied in the version 2 of the L2r chain. The

20 multiplicative factors have been derived from the temperature retrieval and, thus, are represen-
21 tative of errors on channels close to the strong O₃ line center. Outside of this spectral domain
22 the response function remains uncertain.

23 The response function is calculated for each channel on a frequency grid of 13 MHz width,
24 sampled in 0.1 MHz steps within 2.5 MHz range from the channel center and in 0.4 MHz steps
1 outside. Note that the response function is not symmetric and that the channel frequency calcu-
2 lated from the comb spectrum is assumed to correspond to the mean frequency over the channel
3 13 MHz bandwidth weighted by the response function.

4 **Appendix B**

5 **Side band filter**

6 The upper and lower sideband rejection ratios (Eq. 1) are calculated using the equation (Ochiai
7 et al., 2008):

$$8 \quad R^X(\nu) = m^X (\nu - \nu_0^X)^2 + a^X \quad (\text{B1})$$

9 where $X = \text{LSB}$ or USB . The value for each parameter is given in Table 7.

10 The sideband rejection function depends on the temperature of the optics that varies between
11 16–22 °C. The image band contamination is less than 1 % and the estimation of its intensity is
12 done with less accuracy than in the main band, in order to minimize the CPU consumption.

13 **Appendix C**

14 **Radiometric gain compression error**

15 The raw radiometer outputs C (ADU) depend on the input signal T (K) as follows (Ochiai et al.,
16 2011):

$$17 \quad C = G (1 - \alpha \langle C \rangle) (T_{\text{rec}} + T), \quad (\text{C1})$$

18 where G is the radiometric gain, $\alpha = 1.515 \times 10^{-6}$ is the gain compression, $\langle C \rangle$ is the mean
19 value of the AOS output and $T_{\text{rec}} = 300$ K is the receiver temperature. The value of the gain
20 G is set in order to give a cold load AOS output of $C_c = 14500$ ADU, a typical value for
21 SMILES measurements. The cold load signal is the brightness temperature of the cold sky,
22 $T_c = 4 \times 10^{-4}$ K.

23 In order to estimate the error caused by neglecting the gain compression factor, simulated at-
24 mospheric brightness temperatures are converted to AOS output using Eq. (C1). The brightness
1 temperature T_{1b} as given in version 5 of the level-1b data is calculated by converting the AOS
2 counts while neglecting the gain compression:

$$3 \quad T_{1b} = \frac{T_h - T_c}{C_h - C_c} \times (C - C_c) + T_c \quad (\text{C2})$$

4 where $T_h = 275$ K is the hot load brightness temperature and C_h is the hot load AOS output
5 calculated using Eq. (C1).

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12 **References**

- 13 Baron, P., Merino, F., and Murtagh, D.: Simultaneous Retrievals of Temperature and Volume Mixing
14 Ratio Constituents from Nonoxygen Odin Submillimeter Radiometer Bands, *Appl. Optics*, 40, 6102–
15 6110, 2001.
- 16 Baron, P., Ricaud, P., de La Noë, J., Eriksson, P., Merino, F., Ridal, M., and Murtagh, D. P.: Studies for
17 the Odin Sub-Millimetre Radiometer: II. Retrieval Methodology, *Can. J. Phys.*, 80, 341–356, 2002.
- 18 Baron, P., Mendrok, J., Kasai, Y., Ochiai, S., Seta, T., Sagi, K., Suzuki, K., Sagawa, H., and Urban,
19 J.: AMATERASU: Advanced Model for Atmospheric TERahertz Radiation Analysis and Simulation,
20 *Journal of the National Institute of Information and Communications Technology*, 55(1), 109–121,
21 2008.
- 22 Boissoles, J., Boulet, C., Tipping, R., Brown, A., and Ma, Q.: Theoretical calculation of the translation-
23 rotation collision-induced absorption in N_2-N_2 , O_2-O_2 , and N_2-O_2 pairs, *J. Quant. Spectrosc. Ra.*, 82,
24 505–516, doi:10.1016/S0022-4073(03)00174-2, 2003.
- 25 Carlotti, M. and Rodolfi, M.: Derivation of temperature and pressure from submillimetric limb observa-
26 tions, *Appl. Optics*, 38, 2398–2409, 1999.
- 27 Ekström, M., Eriksson, P., Rydberg, B., and Murtagh, D. P.: First Odin sub-mm retrievals in the
28 tropical upper troposphere: humidity and cloud ice signals, *Atmos. Chem. Phys.*, 7, 459–469,
1 doi:10.5194/acp-7-459-2007, 2007.
- 2 Harde, H., Katzenellenbogen, N., Grischkowsky, D.: Line-Shape Transition of Collision Broadened
3 Lines, *Phys. Rev. Let.*, 74 (8), 1307–1310, 1995.
- 4 Hedin, A.: Extension of the MSIS Thermosphere Model into the Middle and Lower Atmosphere, *J.*
5 *Geophys. Res.*, 96, 1159–1172, 1991.

- 6 Kikuchi, K. and Fujii, Y.: Flight Model Performance of 640-GHz Superconductor-Insulator-
7 Superconductor Mixers for JEM/SMILES Mission, *Journal Infrared Millim. W.*, 31, 1205–1211, 2010.
- 8 Kikuchi, K., Nishibori, T., Ochiai, S., Ozeki, H., Irimajiri, Y., Kasai, Y., Koike, M., Manabe, T.,
9 Mizukoshi, K., Murayama, Y., Nagahama, T., Sano, T., Sato, R., Seta, M., Takahashi, C., Takayanagi,
10 M., Masuko, H., Inatani, J., Suzuki, M., and Shiotani, M.: Overview and Early Results of the Super-
11 conducting Submillimeter-Wave Limb-Emission Sounder (SMILES), *J. Geophys. Res.*, 115, D23306,
12 doi:10.1029/2010JD014379, 2010.
- 13 Livesey, N. J., Snyder, W., Read, W., and Wagner, P.: Retrieval algorithms for the EOS Microwave Limb
14 Sounder (MLS) instrument, *IEEE T. Geosci. Remote*, 44, 1144–1155, 2006.
- 15 Marks, C. J. and Rodgers, C. D.: A Retrieval Method for Atmospheric Composition from Limb Emission
16 Measurements, *J. Geophys. Res.*, 98(D8), 14939–14953, doi:10.1029/93JD01195, 1993.
- 17 Mendrok, J., Baron, P., and Kasai, Y.: Studying the potential of terahertz radiation for deriving ice
18 cloud microphysical information, vol. 7107, , *Remote Sensing of Clouds and the Atmosphere XIII*,
19 *Proceedings of SPIE*, doi:10.1117/12.800262, 710704p., 2008.
- 20 Murtagh, D. P., Frisk, U., Merino, F., Ridal, M., Jonsson, A., Stegman, J., Witt, G., Eriksson, P., Jiménez,
21 C., Mégie, G., de La Noë, J., Ricaud, P., Baron, P., Pardo, J. R., Hauchecorne, A., Llewellyn, E. J.,
22 Degenstein, D. A., Gattinger, R. L., Lloyd, N. D., Evans, W. F. J., McDade, I. C., Haley, C. S., Sioris,
23 C., von Savigny, C., Solheim, B. H., McConnell, J. C., Strong, K., Richardson, E. H., Leppelmeier,
24 G. W., Kyrölä, E., Auvinen, H., and Oikarinen, L.: An Overview of the Odin Atmospheric Mission,
25 *Can. J. Phys.*, 80, 309–318, 2002.
- 26 Ochiai, S., Nishibori, T., Ozeki, H., Kikuchi, K., and Manabe, T.: Superconducting Submillimeter-Wave
27 Limb-Emission Sounder on the International Space Station I: Radiometric and Spectral Calibration
28 and Data Processing., *Journal of the National Institute of Information and Communications Technol-*
29 *ogy*, 55(1), 83–94, 2008.
- 30 Ochiai, S., Kikuchi, K., Nishibori, T., Manabe, T., Ozeki, H., Mizukoshi, K., Ohtsubo, F., Tsubosaka,
31 K., Irimajiri, Y., and R. Sato, M. S.: Performance of JEM/SMILES in orbit, *Proceedings on 21st In-*
32 *ternational Symposium on Space Terahertz Technology*, 179p., 2010.
- 33 Ochiai, S., Kikuchi, K., Nishibori, T., Mizobuchi, S., Manabe, T., Mitsuda, C., and Baron, P.: Gain
1 nonlinearity calibration of the SMILES receiver, *IEEE International Geoscience and Remote Sensing*
2 *Symposium*, Sendai, <http://igarss11.org/>, last access: 1 June, 2011.
- 3 Ozeki, H., Mizobuchi, S., Mitsuda, C., Sano, T., Suzuki, M., and Shiotani, M.: Response characteristics
4 of radio spectrometers of the Superconducting Submillimeter-wave Limb-Emission Sounder (JEM-
5 SMILES), *IEEE International Geoscience and Remote Sensing Symposium*, Sendai, <http://igarss11.org/>.

6 org/, last access: 1 June, 2011.

7 Pardo, J. R., Serabyn, E., and Cernicharo, J.: Submillimeter atmospheric transmission measurements on
8 Mauna Kea during extremely dry El Niño conditions: implications for broadband opacity contribu-
9 tions, *J. Quant. Spectrosc. Ra.*, 68, 419–433, doi:10.1016/S0022-4073(00)00034-0, 2001.

10 Pickett, H. M., Poynter, R. L., Cohen, E. A., Delitsky, M. L., Pearson, J. C., and Müller, H. S. P.:
11 Submillimeter, millimeter, and microwave spectral line catalog, *J. Quant. Spectrosc. Ra.*, 60, 883–
12 890, doi:10.1016/S0022-4073(98)00091-0, 1998.

13 Read, W. G., Shippony, Z., Schwartz, M. J., Livesey, N. J., and Snyder, W. V.: The clear-sky unpolarized
14 forward model for the EOS aura microwave limb sounder (MLS), *IEEE T. Geosci. Remote*, 44, 1367–
15 1379, doi:10.1109/TGRS.2006.873233, 2006.

16 Reinecker, M.: The GEOS-5 data assimilation system: A documentation of GEOS-5.0, Tech Report
17 104606 V27 (NASA), 2008.

18 Rodgers, C. D.: *Inverse Methods for Atmospheric Sounding: Theory and Practise*, vol. 2 of Series on
19 Atmospheric, Oceanic and Planetary Physics, World Scientific, 2000.

20 Rothman, L. S., Gordon, I. E., Barbe, A., Benner, D. C., Bernath, P. F., Birk, M., Boudon, V., Brown,
21 L. R., Campargue, A., and Champion, J. P.: The HITRAN 2008 molecular spectroscopic database, *J.*
22 *Quant. Spectrosc. Ra.*, 110, 533–572, doi:10.1016/j.jqsrt.2009.02.013, 2009.

23 Rüeiger, J. M.: *Refractive Index Formulae for Radio Waves*, FIG XXII International Congress Washing-
24 ton DC, http://www.fig.net/pub/fig_2002/Js28/JS28_rueiger.pdf (last access: 1 June 2011), 2002.

25 Schreier, F. and Kohlert, D.: Optimized implementations of rational approximations, a case
1 study on the Voigt and complex error function, *Comput. Phys. Commun.*, 179, 457–465,
2 doi:10.1016/j.cpc.2008.04.012, 2008.

3 Shiotani, M., Masuko, H., and the SMILES Science Team: JEM/SMILES Mission Plan, Version 2.1,
4 Tech. rep., National Space Development Agency (NASDA) and Communications Research Labora-
5 tory (CRL), <http://smiles.tksk.jaxa.jp/document/indexe.shtml> (last access: 1 June 2011), 2002.

- 6 Takahashi, C., Ochiai, S., and Suzuki, M.: Operational retrieval algorithms for JEM/SMILES level 2 data
7 processing system, *J. Quant. Spectrosc. Ra.*, 111, 160–173, doi:10.1016/j.jqsrt.2009.06.005, 2010.
- 8 Urban, J., Baron, P., Lautié, N., Schneider, N., Dassas, K., Ricaud, P., and de La Noë, J.: Moliere (v5):
9 a versatile forward- and inversion model for the millimeter and sub-millimeter wavelength range, *J.*
10 *Quant. Spectrosc. Ra.*, 83, 529–554, 2004.
- 11 Urban, J., Lautié, N., Le Flochmoën, E., Jiménez, C., Eriksson, P., de La Noë, J., Dupuy, E., Ekström,
12 M., Amraoui, L. E., Frisk, U., Murtagh, D., Olberg, M., and Ricaud, P.: Odin/SMR limb observations
13 of stratospheric trace gases: Level 2 processing of ClO, N₂O, HNO₃, and O₃, *J. Geophys. Res.*, 110,
14 D14307, doi:10.1029/2004JD005741, 2005.
- 15 Waters, J. W., Froidevaux, L., Harwood, R. S., Jarnot, R. F., Pickett, H. M., Read, W. G., Siegel, P. H.,
16 Cofield, R. E., Filipiak, M. J., Flower, D. A., Holden, J. R., Lau, G. K., Livesey, N. J., Manney,
17 G. L., Pumphrey, H. C., Santee, M. L., Wu, D. L., Cuddy, D. T., Lay, R. R., Loo, M. S., Perun, V. S.,
18 Schwartz, M. J., Stek, P. C., Thurstans, R. P., Boyles, M. A., Chandra, K. M., Chavez, M. C., Chen,
19 G.-S., Chudasama, B. V., Dodge, R., Fuller, R. A., Girard, M. A., Jiang, J. H., Jiang, Y., Knosp, B. W.,
20 LaBelle, R. C., Lam, J. C., Lee, K. A., Miller, D., Oswald, J. E., Patel, N. C., Pukala, D. M., Quintero,
732 O., Scaff, D. M., Snyder, W. V., Tope, M. C., Wagner, P. A., and Walch, M. J.: The Earth Observing
733 System Microwave Limb Sounder (EOS MLS) on the Aura satellite, *IEEE T. Geosci. Remote*, 44,
734 1075–1092, doi:10.1109/TGRS.2006.873771, 2006.
- 735 Wu, D. L., Pickett, H. M., and Livesey, N. J.: Aura MLS THz Observations of Global Cirrus Near the
736 Tropopause, *Geophys. Res. Lett.*, 35, L15803, doi:10.1029/2008GL034233, 2008.

Table 1. Characteristics of the SMILES antenna and the observation sequence.

Antenna size	400 × 200 mm
Antenna pattern (vertical full width half maximum)	0.09° (~4 km at the limb)
Total scan duration	53 s
Atmospheric scan duration	30.5 s
Vertical velocity	0.1125° s ⁻¹ (upward direction)
Spectrum integration time	0.5 s
Typical altitude coverage	–10 to 90 km (60 spectra)
Scans per day	~1630

Table 2. SMILES L2 research data produced by the version-2 algorithms. The products list for each process or window is given along with the frequency range and the line-of-sight altitude range. The actual vertical range for a good retrieval depends on the signal strength.

Window/Process name	Frequency	Products	Line-of-sight tangent height (km)
A-w0	625.042–625.612	Line of sight elevation angle	18–70
A-w1	625.042–625.612	Temperature, O ₃ , OO ¹⁸ O	18–100
A-w2	624.445–625.040	H ³⁷ Cl, HNO ₃ , BrO	18–100
A-w3	624.35–624.900	BrO, HNO ₃ , CH ₃ CN, O ¹⁸ OO	25–80
A-w4	625.015–625.185	HOCl, OO ¹⁸ O, H ₂ O ₂	25–80
A-w5	625.500–625.612	OO ¹⁸ O*	25–80
A-w6	624.265–624.642	O ¹⁷ OO, HNO ₃	20–70
A-w7	625.132–625.232	HOBr*	20–70
A-w8*	Full band	O ₃ , H ³⁷ Cl, BrO, HNO ₃ , CH ₃ CN, H ₂ O	11–35
B-w0	625.042–625.612	Line of sight elevation angle	18–70
B-w1	625.042–625.612	Temperature, O ₃ , OO ¹⁸ O	18–100
B-w2	625.714–626.264	H ³⁵ Cl, O ¹⁷ OO	16–100
B-w3	625.015–625.185	HOCl, OO ¹⁸ O, H ₂ O ₂	25–90
B-w4	625.500–625.830	HO ₂ , OO ¹⁸ O, O ¹⁷ OO	20–90
B-w5	625.130–625.230	HOBr*	25–80
B-w6*	Full band	O ₃ , HNO ₃ , H ³⁵ Cl, H ₂ O	11–35
C-w0*	Full band	O ₃ , ClO, HO ₂ , HNO ₃ , OO ¹⁸ O, OO ¹⁷ O, BrO, H ₂ O	11–40
C-w1	649.882–650.382	BrO, HNO ₃ , OO ¹⁷ O	18–90
C-w2	649.500–649.900	HO ₂ , O ¹⁷ OO	25–90
C-w3	649.232–649.632	ClO, OO ¹⁷ O	15–80

The * indicates data that are not recommended for use.

Table 3. Values of the line frequency and of the air-broadening and pressure induced frequency shift parameters measured in laboratory. Only the main lines used in the processing are shown. Values not indicated are taken from the HITRAN (2008) database.

Species	line frequency (MHz)	air-broadening (MHz/Torr) / temperature dependence	frequency shift (MHz/Torr)
O ₃	625371.2420	3.08 / 0.73	
O ₃	650732.7260	3.01 / 0.60	
H ³⁵ Cl	625901.6627	3.39 / 0.72	0.145
H ³⁵ Cl	625918.7020	3.39 / 0.72	0.145
H ³⁵ Cl	625931.9977	3.39 / 0.72	0.145
H ³⁷ Cl	624964.3718	3.39 / 0.72	0.145
H ³⁷ Cl	624977.8059	3.39 / 0.72	0.145
H ³⁷ Cl	624988.2727	3.39 / 0.72	0.145
ClO	649445.0400	2.86 / 0.77	
ClO	649451.1700	2.86 / 0.77	
HO ₂	625660.3542	3.77 /	
HO ₂	649701.4773	2.74 /	
H ₂ O ₂	625044.1800	3.71 /	
HOCl	625075.1346	3.88 / 0.65	-0.076
BrO	624764.8570	3.05 / 0.80	
BrO	650175.8041	3.03 / 0.81	
CH ₃ CN	624819.3573	4.79 /	
CH ₃ CN	624926.4662	4.81 /	
H ₂ O	556936.0020	4.01 /	0.300
H ₂ O	620700.8070	4.38 /	-0.031
H ₂ O	752033.2270	3.96 /	0.197

Table 4. Summary of the a priori information used in the a priori covariance matrix (Eq. 6). Errors and correlations are not representative of the real data, but are inversion regularisation parameters. Values for O₃ are given for the inversion of the 625.371 GHz line and those for H₂O are for the full spectral band inversion.

	origin	error	correlation	Confidence level
Line-of-sight elevation angles	ISS attitudes	0.02°	No	medium
Scan frequency offset	Comb spectra	0.02 MHz	No	high
Temperature below 60 km	GEOS-5	5 K	6 km	high below 30 km, medium above
Temperature above 60 km	MSIS	5 K	6 km	low
H ₂ O	GEOS-5	5 ppmv	6 km	high below 15 km, low above
O ₃	GEOS-5	5 ppmv	3 km	medium
HCl	US standard	>50 %	4 km	low
other species	climatology	signal strength dependent	3–6 km	low

Table 5. Uncertainties assumed for single profile error estimation. The name of the parameters and the labels used for their identification in the figures are given in the parenthesis in the first column. The parameter γ is for the line pressure-broadening parameter.

Random errors	
O ₃ VMR (O ₃)	10 % of the a priori profile
HCl VMR (HCl)	10 % of the a priori profile
Temperature (Tp)	5 K
Pressure (Pr)	2 % of the a priori profile
Line of sight angles (LOS)	0.007° (300 m)
Measurement noise (y-noise)	0.5 K
Systematic errors	
Calibration offset (CAL0)	1 K
Calibration gain (CAL1)	1 %
Calibration gain compression α (CALN)	1.515×10^{-6}
O ₃ $-\gamma$ at 625.371 GHz (γ -O ₃)	2 %
H ³⁵ Cl- γ at 625.901 GHz (γ -35)	5 %
H ³⁷ Cl- γ at 624.964 GHz (γ -37)	5 %
O ₃ v1,3- γ at 625.051 GHz (γ -v1,3)	5 %
OO ¹⁸ O- γ at 625.091 GHz (γ -18 (1))	5 %
OO ¹⁸ O- γ at 625.563 GHz (γ -18 (2))	5 %
dry air continuum (N ₂)	20 %
Image band (IMB)	ON-OFF
AOS filter width (AOS)	10 %

Table 6. Polynomial coefficients used to calculate the channel response function for AOS units 1 and 2 (Eqs. A1 and A2).

	C _{0,...,4} for unit 1					C _{0,...,4} for unit 2				
w_1	0.710	-8.02e-5	2.40e-7	-1.11e-10	0.	0.699	3.84e-4	-7.31e-7	5.67e-10	-1.55e-13
w_2	1.438	-4.88e-4	1.74e-6	-2.02e-9	7.48e-13	2.021	-2.77e-3	4.64e-6	-3.29e-9	8.64e-13
w_3	7.510	-2.20e-2	5.02e-5	-4.38e-8	1.27e-11	5.532	-8.69e-3	1.75e-5	-1.48e-8	4.33e-12
A_1	0.256	1.93e-3	-6.34e-6	6.37e-9	-1.90e-12	0.911	-1.01e-3	1.03e-6	-2.83e-10	0.
A_2	1.238	-3.63e-3	1.21e-5	-1.21e-8	3.61e-12	0.137	1.67e-3	-1.67e-6	4.52e-10	0.
A_3	0.169	5.94e-6	-4.26e-8	0.	0.	0.044	2.06e-4	-1.97e-7	4.97e-11	0.
x_1	-0.115	-6.39e-5	1.88e-7	-6.29e-11	0.	0.021	-1.27e-4	1.17e-7	-2.82e-11	0.
x_2	0.138	-7.04e-5	-1.73e-8	0.	0.	-0.165	1.17e-3	-2.1e-6	1.59e-9	-4.34e-13
x_3	-0.872	1.35e-3	-1.32e-6	0.	0.	-2.416	3.36e-3	-2.78e-6	5.47e-10	0.

Table 7. Side band response parameters for the lower side band (LSB) and upper side band (USB) (Eq. B1). The parameters a and ν_0 vary with the temperature of the Ambient Temperature Optics (AOP).

	Lower side band (LSB)	Upper side band (USB)
m	0.00398	0.00470
ν_0 (GHz)	$-2.888012 \times 10^{-4} T_{\text{AOP}}^2 + 0.0253396 T_{\text{AOP}} + 624.75733$	$0.0085865 T_{\text{AOP}} + 648.75572$
a	$-2.90103 \times 10^{-7} T_{\text{AOP}} + 0.00003150$	$-3.30717 \times 10^{-7} T_{\text{AOP}} + 0.00006756$

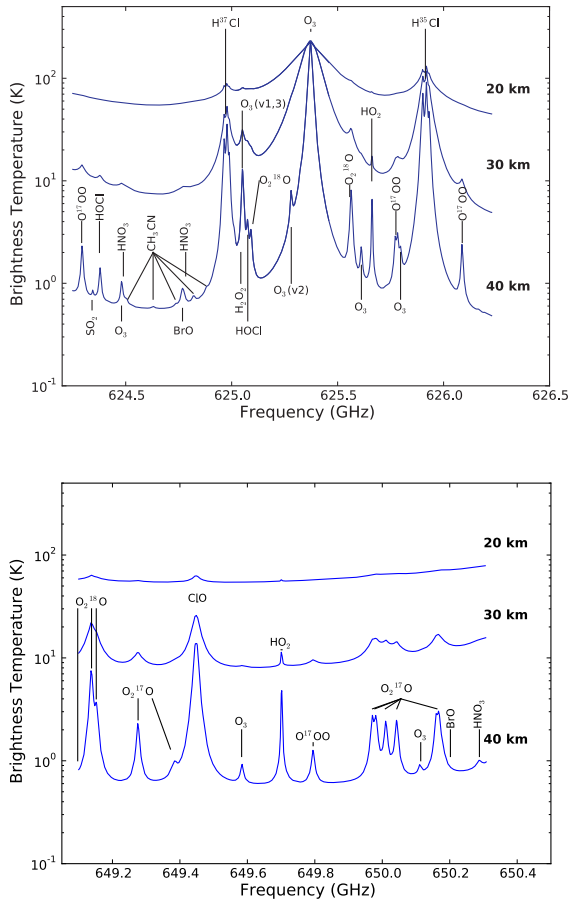


Fig. 1. Spectroscopic lines from simulated limb radiances in bands A and B (upper panel) and in band C (lower panel). The line-of-sight tangent heights are 20, 30 and 40 km.

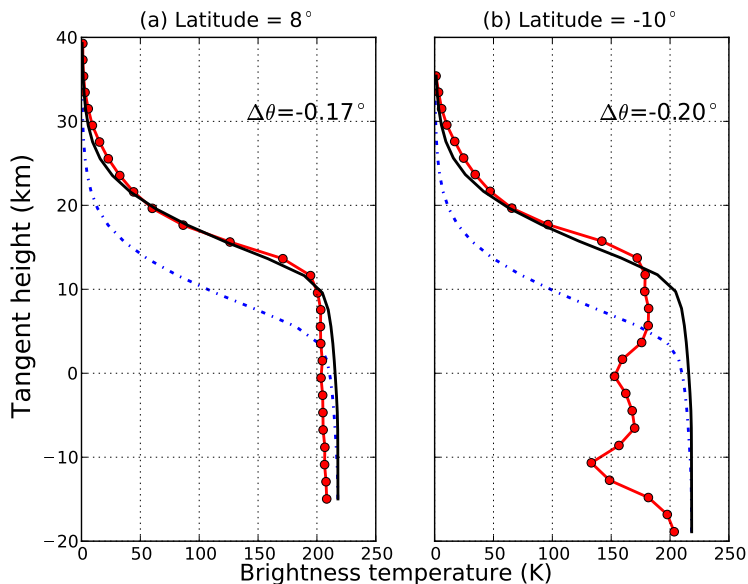


Fig. 2. Measured brightness temperature (circles) with respect to corrected tangent heights at 624.547 GHz: **(a)** clear sky case for a latitude of 8° N, and **(b)** for a latitude of 10° S with scattering ice particles. The black line is a calculation using the line-of-sight tangent altitude corrected before the retrieval calculation and the dashed line is the calculation on the original line-of-sight tangent altitude. The correction is -0.17° (about 6 km) for case **(a)** and -0.19° (about 7 km) for case **(b)**. Data are two single scans measured on 14 April 2010.

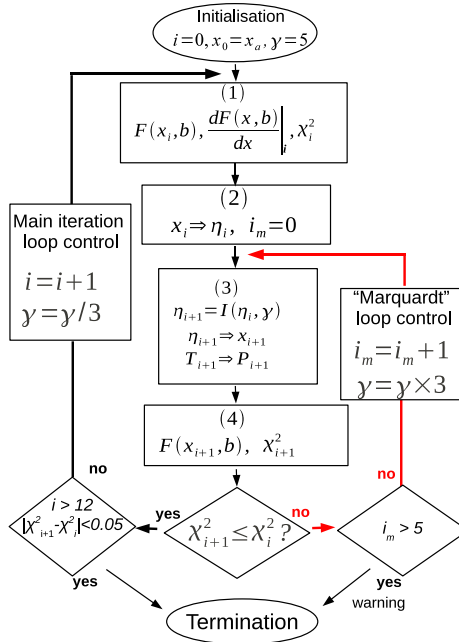


Fig. 3. Flowchart of the iterative inversion calculation. Step 1: the forward model and its Jacobian matrix, and the cost function (Eq. 2) are computed. Step 2: the retrieved parameters are scaled (Eq. 4) and the Marquardt loop counter is set to 0. Step 3: the inversion is performed (Eq. 5), the retrieved parameters are unscaled, and the pressure is derived from the retrieved temperature (Eq. 8). Step 4: the forward model and the cost function are estimated with the last estimate of the parameters. If the cost function decreases, the path indicated by the black branch indicates the Gauss-Newton iteration loop including the decrease of the Levenberg/Marquardt parameter value. Otherwise, the Marquardt loop branch is used (red branch) where the Marquardt parameter γ is increased.

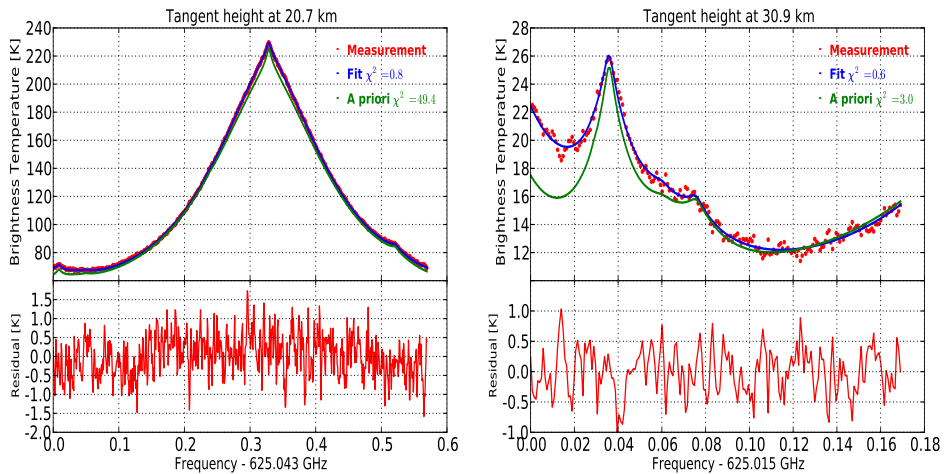


Fig. 4. Left panel: spectral line of O_3 at 625.371 GHz for a tangent height of 21 km measured in the window A-w1 along with the a priori and fitted spectra. The lower panel shows the residual of the fit. Right panel: Spectral window A-w4 for a tangent height of 31 km (upper panel) and fit residual (lower panel). The weak HOC1 line at 625.076 GHz is located between two stronger lines of $O_3 v1,3$ and $OO^{18}O$ at 625.051 GHz and 625.091 GHz, respectively. Data are from a scan measured at 24:00 LT on 14 April 2010 at a latitude of 8° N.

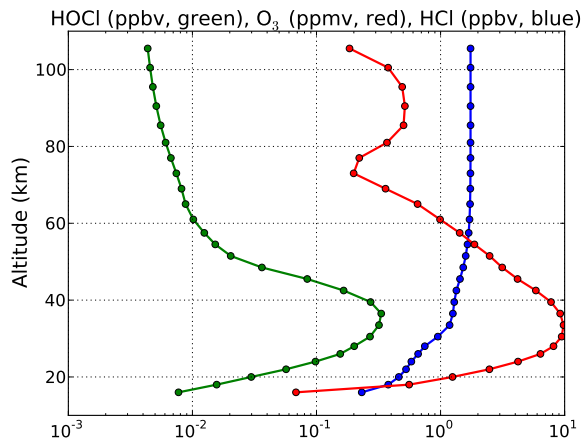


Fig. 5. VMR profiles of HOCl (green, ppbv), O₃ (red, ppmv) and HCl (blue, ppbv) used for the sensitivity analysis.

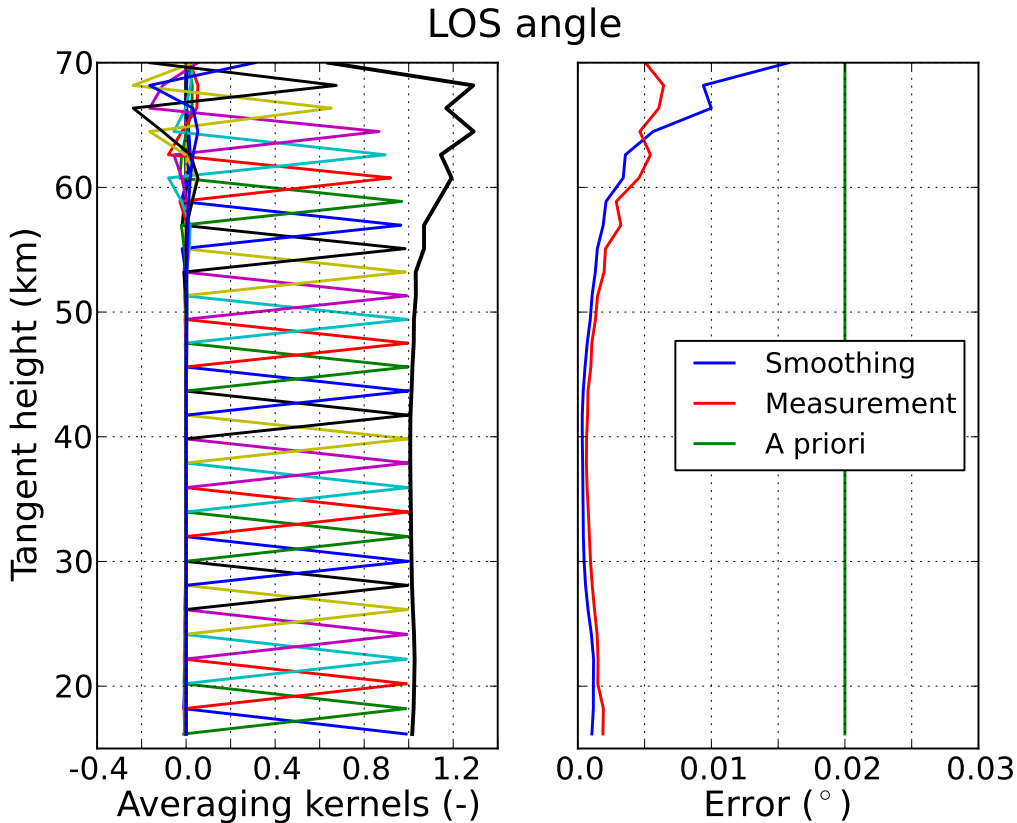


Fig. 6. Left panel: averaging kernels and measurement response (black line) for the retrieved line-of-sight elevation angle as a function of the tangent height. Right panel: line-of-sight angle measurement error (red line) and smoothing error (blue line) estimated by linear mapping (Eq. 9). The green line indicates the a priori error.

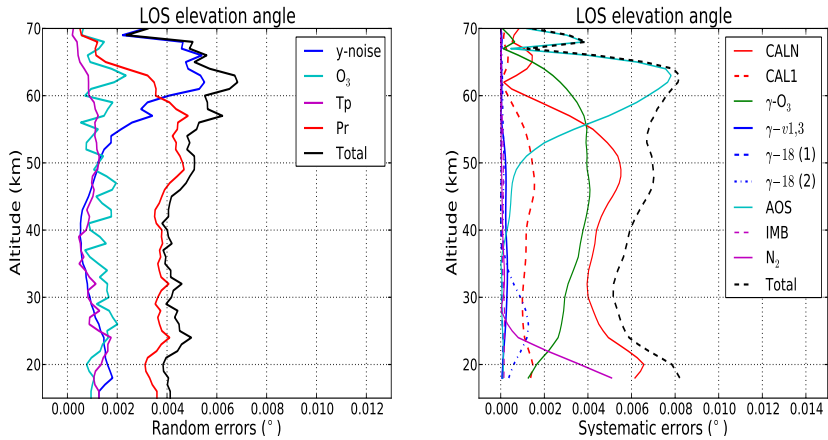


Fig. 7. Assessment of errors on the retrieved line-of-sight elevation angles from window A/B-w0. The random and systematic errors are shown in the left and right panels, respectively. The meaning of the labels and the assumed errors on the parameters are given in Table 5. The black line is the total error calculated as the root-sum-square of all sources of error.

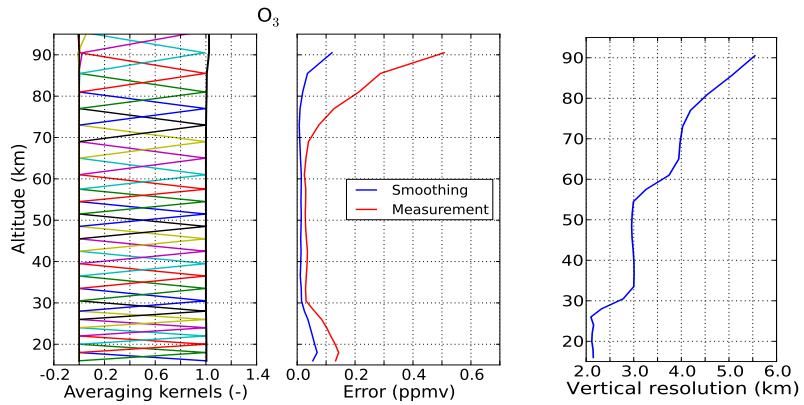


Fig. 8. Results for the O_3 profile retrieved from window A/B-w1. Left and central panels: same representation as left and right panels of Fig. 6. Right panel: vertical resolution.

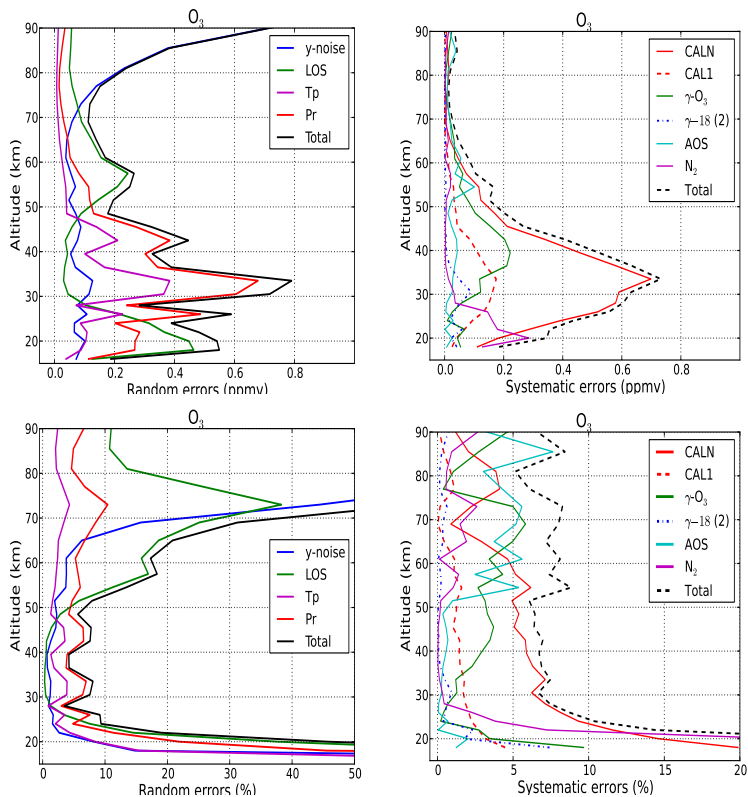


Fig. 9. Random (left panels) and systematic (right panels) errors on the O_3 profile retrieved from window A/B-w1. The absolute and relative errors are shown in the upper and lower panels, respectively. Random errors are from the measurement noise (blue line), temperature profile (magenta line), pressure profile (red line), line-of-sight angle (green) and the total random error (black line). The meaning of the labels of the systematic errors and the assumed errors on the parameters are given in Table 5. The black line is the total error calculated as the root-sum-square of all sources of error.

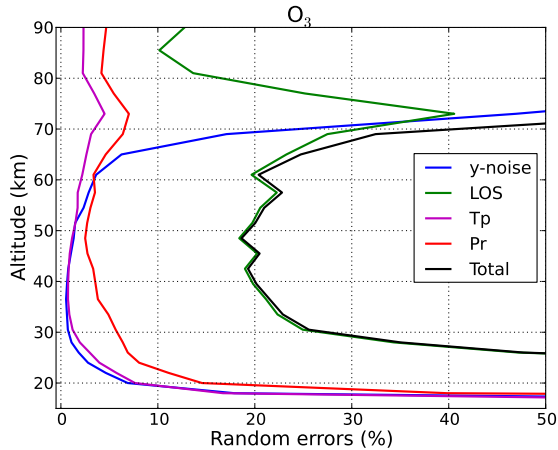


Fig. 10. Most significant random errors on a single O₃ profile retrieved from A/B-w1 if the line-of-sight angles are not retrieved. The errors from the instrument noise (blue line), temperature (magenta), pressure (red) and line-of-sight angles (green line) are shown. The assumed uncertainties are given in Table 5.

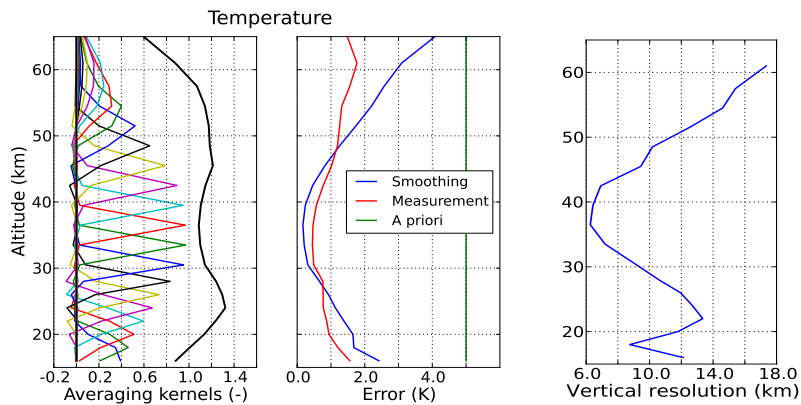


Fig. 11. Same as Fig. 8 but for the retrieved temperature profile.

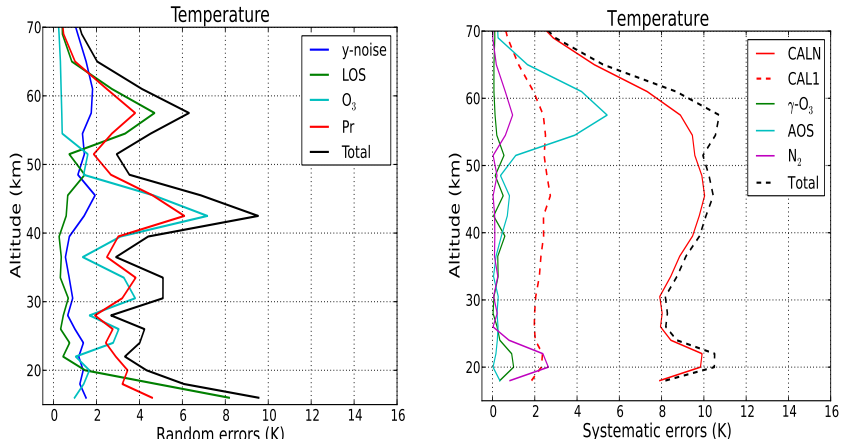


Fig. 12. Random (left panel) and systematic (right panel) errors on the temperature profile retrieved from window A/B w1. Random errors are from the measurement noise (blue line), O₃ (cyan line), line-of-sight angle (green), the pressure profile (red line). The black line is the total error calculated as the root-sum-square of all sources of error. The meaning of the labels of the systematic errors and the assumed errors on the parameters are given in Table 5.

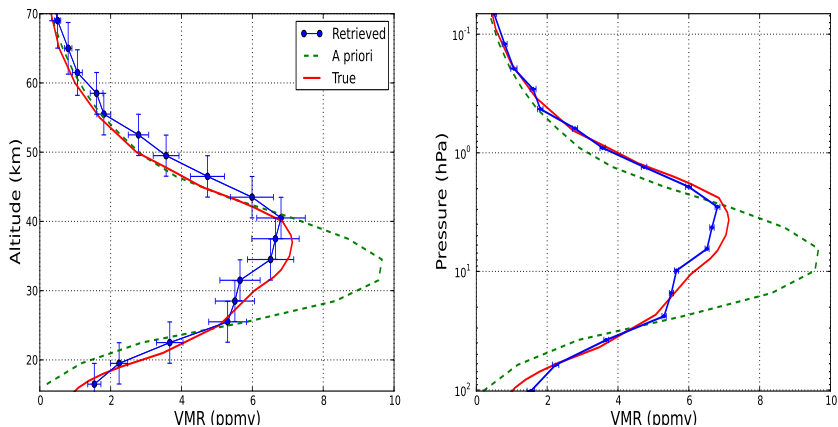


Fig. 13. Simulation of the A/B-w1 inversion process. The retrieved O_3 profile is plotted as a function of altitude levels (left panel) and of pressure levels (right panel). Atmospheric profiles for simulating the measurement vector are from a winter mid-latitude climatology and from a tropical climatology for the reference/a priori profiles. In addition, the true pressure profile was multiplied by 0.95 and a noise of 0.4 K (1-sigma) was added to the measurement vector. In the left panel the vertical bars indicate the full-width-at-half-maximum of the averaging kernels and the horizontal bars are the 1-sigma measurement noise and the pressure-induced error (Fig 9). In the right panel the horizontal bars are for the 1-sigma measurement noise.

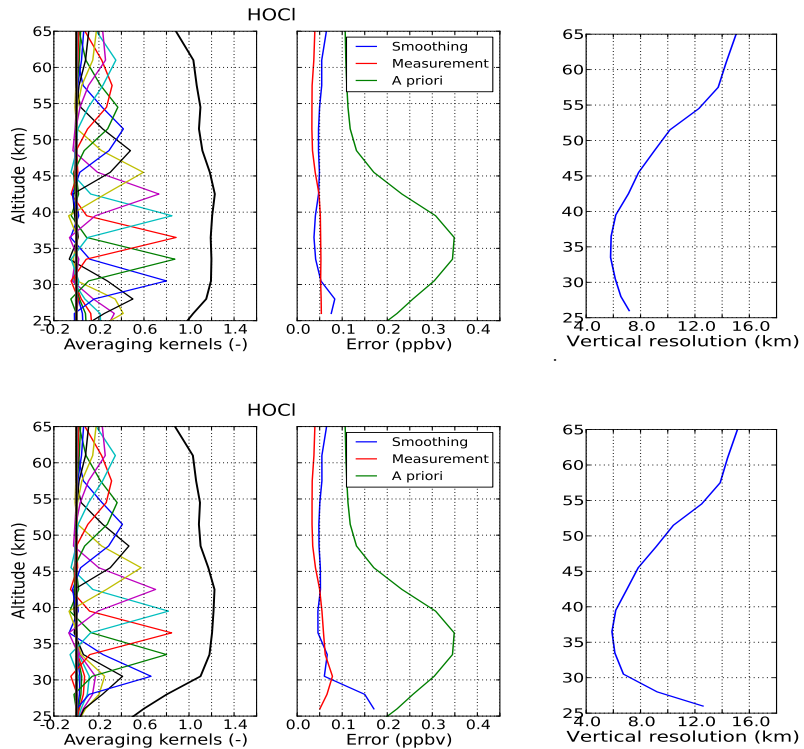


Fig. 14. Same as Fig. 8 but for the HOCl profile retrieved from windows A/B-w1 (upper row) and A-w3 or B-w4 (lower row).

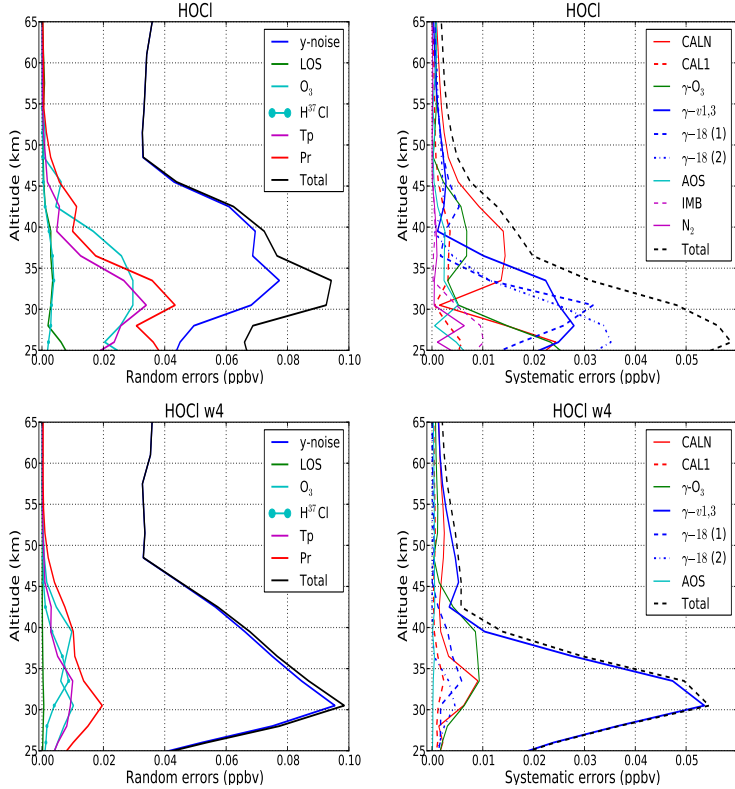


Fig. 15. Upper row: random (left panel) and systematic (right panel) errors on the HOCl profile retrieved from window A/B-w1. Lower row: Same for the HOCl profile retrieved from A-w3 or B-w4. The meaning of the labels on the systematic errors and the assumed errors on the parameters are given in Table 5.