Atmos. Meas. Tech. Discuss., 6, 3129–3180, 2013 www.atmos-meas-tech-discuss.net/6/3129/2013/ doi:10.5194/amtd-6-3129-2013 © Author(s) 2013. CC Attribution 3.0 License.



This discussion paper is/has been under review for the journal Atmospheric Measurement Techniques (AMT). Please refer to the corresponding final paper in AMT if available.

# Measurement of the Arctic UTLS composition in presence of clouds using millimetre-wave heterodyne spectroscopy

E. Castelli<sup>1</sup>, B. M. Dinelli<sup>1</sup>, S. Del Bianco<sup>2</sup>, D. Gerber<sup>3</sup>, B. P. Moyna<sup>3</sup>, R. Siddans<sup>3</sup>, B. J. Kerridge<sup>3</sup>, and U. Cortesi<sup>2</sup>

<sup>1</sup>Istituto di Scienza dell'Atmosfera e del Clima, ISAC-CNR, Bologna, Italy <sup>2</sup>Istituto di Fisica Applicata "Nello Carrara", IFAC-CNR, Sesto Fiorentino, Italy <sup>3</sup>RAL Space, Harwell Oxford, STFC.RAL, Chilton, UK

Received: 28 February 2013 - Accepted: 1 March 2013 - Published: 28 March 2013

Correspondence to: E. Castelli (e.castelli@isac.cnr.it)

Published by Copernicus Publications on behalf of the European Geosciences Union.



# Abstract

MARSCHALS (Millimetre-wave Airborne Receivers for Spectroscopic CHaracterisation in Atmospheric Limb Sounding) is a limb viewing instrument working in the millimetre and sub-millimetre spectral regions (from 294 to 349 GHz). The scientific rationale of
 MARSCHALS is the study of the Upper Troposphere and Lower Stratosphere region. In March 2010 MARSCHALS was deployed on-board the M-55 Geophysica stratospheric aircraft during the PREMIER-Ex field campaign. From the campaign base at Kiruna, Sweden, a research flight to investigate the Arctic atmosphere was conducted. For the first time the instrument fully exploited the three spectral bands. In this paper, we present results of the analysis of MARSCHALS measurements acquired during the PREMIER-Ex flight as processed by the Millimetre-wave Atmospheric Retrieval Code. For the second time after the SCOUT-O3 flight in 2005, MARSCHALS measured vertical distributions of temperature, water vapour, ozone and nitric acid over an altitude range of 4 to 24 km. In addition, vertical profiles of CO and N<sub>2</sub>O were ob-

- tained for the first time. The measurements were performed in clear sky conditions and in presence of low and high altitude clouds (with the potential to obscure measurements in the middle infrared spectroscopic region) and some information about thick clouds were extracted from the retrievals. The capabilities to derive information on upper tropospheric and lower stratospheric vertical profiles of temperature and minor con-
- stituents from millimetre-wave limb sounding observations in the Northern Polar region are presented and discussed for each of the individual targets. The performance of the retrieval are demonstrated from the results of data processing of MARSCHALS, deployed in the 2010 Arctic campaign with the M-55 Geophysica as an airborne simulator of the millimetre-wave limb-sounder proposed for the ESA Earth Explorer 7 candidate
- <sup>25</sup> Core Mission PREMIER (PRocess Exploration through Measurements of Infrared and millimetre-wave Emitted Radiation).





# 1 Introduction

The Upper Troposphere and Lower Stratosphere (UTLS) is a pivotal region of the Earth's atmosphere. In absolute numbers the term UTLS is generally applied to altitude ranges between 4 and 25 km at polar latitudes and between 8 and 30 km in the

- <sup>5</sup> Tropics. Chemical, radiative and dynamical processes that take place at these altitudes have a substantial and mutual influence on surface climate (ESA, 2012 and references therein). In fact, this atmospheric region – due to the very low temperatures of the layers immediately above and below the Tropopause height – is where Essential Climate Variables (Mason, 2010) such as surface air temperature and Earth radiation budget
- <sup>10</sup> are most sensitive to changes in the distribution of radiatively active gases and clouds. Of primary importance in this respect are water vapour and ozone, greenhouse gases which play a key role in the radiative balance of the planet (e.g. Solomon et al., 2010; Hurst et al., 2011; Lacis et al., 1990) and exhibit a large, global-scale, spatio-temporal variability in the UTLS. The distribution of these species, and of other atmospheric con-
- stituents in gas-phase and particulates, is ultimately determined by complex interactions between the chemical and radiative processes which represent their sources and sinks, as well as the dynamical mechanisms which are responsible for the exchange of air masses between the troposphere and the stratosphere, ranging from planetary or synoptic scales down to the mesoscale (see, for instance, Gettelman et al., 2011 and
- <sup>20</sup> references therein). In the Arctic region, an accurate representation of the processes that occur in the UTLS is an important prerequisite in understanding the perturbed chemistry in the core of the Northern polar vortex, as well as transport and mixing at its boundary. Given that extraordinary events such as the severe ozone depletion observed in the 2011 Arctic winter (January–March 2011, see Arnone et al., 2012 and
- <sup>25</sup> Manney et al., 2011) are likely to reoccur in the upcoming years, there is a clear need to acquire new observations from space with enhanced vertical and horizontal resolution, which will allow for the investigation of chemical composition, radiative forcing and dynamical transport processes occurring from the outer to innermost region of the





vortex. Moreover, the high latitudes of the Northern Hemisphere show clear evidence of the dynamic coupling between the stratosphere and the troposphere, with anomalies in the strength of the polar vortex propagating downward across the lower stratosphere, resulting in circulation anomalies and a perturbation of the Earth's surface climate.

- <sup>5</sup> However, to date the UTLS region is not adequately covered by satellite measurements, since the majority of spaceborne sensors are blinded by clouds or – as is the case with the Microwave Limb Sounder (MLS) instrument on-board the Upper Atmosphere Research Satellite (UARS) and Aura platforms (Barath et al., 1993; Waters et al., 2006) or the Sub-Millimetre Receiver (SMR) (Olberg et al., 2003) on-board the
- Odin satellite (Murtagh et al., 2002) their instruments are not fully optimised for UTLS measurements in terms of their choice of microwindows (tropospheric penetration), their low vertical resolution and their poor sensitivity (Urban et al., 2005a,b; Sandor et al., 1998; Livesey et al., 2003). To overcome this, as part of the seventh call for Earth Explorer Core missions currently under evaluation in the frame of the European
- <sup>15</sup> Space Agency (ESA)'s Living Planet Programme, one of the three mission candidates selected to enter Phase-A feasibility study is the mission "Process Exploration through Measurement of Infrared and Millimetre-wave Emitted Radiation" (PREMIER). The primary scientific objective of the PREMIER mission is to gain a better understanding of the interaction processes linking atmospheric chemistry and dynamics with climate.
- This will imply the investigation of distribution and transport of trace gases and of the radiative effects of water vapour and clouds in the mid to upper troposphere and lower stratosphere. PREMIER will achieve this through the combined observations by two limb-sounders: one of which will be operating in the thermal infrared (InfraRed Limb Sounder, IRLS), the other in the sub-millimetre-wave region (Millimetre-Wave Limb
- Sounder (MWLS) a.k.a. Stratosphere-Troposphere Exchange and Climate Monitor Radiometer, STEAM-R). To demonstrate the merits of the mission concept, an airborne demonstrator for the MWLS instrument on PREMIER – a.k.a. MARSCHALS (Oldfield et al., 2001) – was built. Respective infra-red instruments to demonstrate the IRLS capabilities are already existing, e.g. Michelson Interferometer for Passive Atmospheric





Sounding – STRatospheric aircraft (MIPAS-STR), Gimballed Limb Observer for Radiance Imaging of the Atmosphere, AirBorne version (GLORIA-AB). MARSCHALS is designed to sample the atmosphere using a limb-viewing geometry from a stratospheric balloon or aircraft. It has already been deployed from the M-55 Geophysica high altitude aircraft during the SCOUT-O3 campaign in 2005 (Dinelli et al., 2009).

As part of the preparatory studies, demonstration and testing activities carried out in parallel to PREMIER Phase-A, a series of field campaigns (PREMIER-Ex) for the deployment of airborne precursors of PREMIER instrumentation on-board the high altitude research platform M-55 Geophysica was performed in 2009–2010. The PREMIER-Ex activities consisted of a test campaign, which took place at mid-latitude (Oberpfaffenhofen, Germany, Lat. 48.1° N, Lon. 11.3° E) in November 2009 and in a scientific campaign, performed in the Arctic region (Kiruna, Sweden, Lat. 67.8° N, Lon.

- 20.4° E) in March 2010 in the frame of the RECONCILE (Reconciliation of essential process parameters for an enhanced predictability of Arctic stratospheric ozone loss and
- its climate interactions) project. The MARSCHALS instrument was present on-board the M-55 for both flights. While for the first science campaign (SCOUT-O3, 2005) only one of the three spectral bands of MARSCHALS was operational, all three bands have now been implemented for PREMIER-Ex. However, because of unfavourable flight conditions, which were inherent to the test campaign in 2009, only the data from the sciantific comparison in Kirupa are worth investigating.

<sup>20</sup> entific campaign in Kiruna are worth investigating.

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In this paper, we present the results of MARSCHALS measurements obtained during the PREMIER-Ex scientific campaign as processed by the MARC (Millimetre-wave Atmospheric Retrieval Code) code. Since all the three spectral bands were operating properly, we could retrieve CO and N<sub>2</sub>O vertical profiles in addition to the T, HNO<sub>3</sub>,

O<sub>3</sub> and H<sub>2</sub>O profiles that had already been measured using the single operating band in the SCOUT-O3 campaign. The structure of the paper is as follows: in Sect. 2 we describe the MARSCHALS instrument and its new features; in Sect. 3 we present the PREMIER-Ex campaign and the flight of 10 March 2010; results of MARSCHALS data





analysis are presented in Sect. 4; intercomparisons with other instruments and discussion are reported in Sect. 5 and in Sect. 6 conclusions are given.

# 2 The MARSCHALS spectrometer

# 2.1 Instrument description

- MARSCHALS uses a vertically scanning antenna of 235 mm diameter to measure thermal emission from the atmosphere. The limb of the atmosphere is sampled through an open aperture in the starboard side of the Geophysica aircraft at angles equivalent to 1 km tangent point steps. A single scan of 28 measurements starts just below ground level and reaches up to just above platform altitude and includes an additional spaceview at 20° above horizontal. The duration of a single scan is ~ 3 min. Integration time at each view angle is 4 times 250 ms. Incoming radiation is coupled to each of three millimetre-wave heterodyne radiometers centred at 300, 325 and 345 GHz, where it is down-converted to the range 12–24 GHz, amplified, and detected using a single channeliser spectrometer. Observation time is shared between the three spectral bands by means of alternating atmospherie agapa. The total apactral bandwidth of the above
- <sup>15</sup> by means of alternating atmospheric scans. The total spectral bandwidth of the channeliser spectrometer is 12 GHz, split up in 60 channels of 200 MHz channel bandwidth (the Intermediate Frequency (IF) bandwidth of the receivers is less than 12 GHz to a varying degree). The signal due to aliased radiances is filtered out by a high performance single sideband filter (one for each radiometer) which reduces these signals by at least 30 dB.

The antenna scan system employs a dedicated Inertial Measurement Unit (IMU) to improve the control of the pointing direction. The antenna control loop measures the instrument (aircraft) roll angle with high precision and corrects for it in the antenna control loop. This makes the instrument line-of-sight extremely stable. To help in assessing the impact of cirrus clouds on the observations, MARSCHALS incorpo-

<sup>25</sup> In assessing the impact of cirrus clouds on the observations, MARSCHALS incorporates an Optical Cloud Monitor (OCM). The spectroscopic characterisation of the three





MARSCHALS bands are summarised in Table 1 (the bands frequency range reported in the table follow the original instrument configuration). Additional information about the MARSCHALS instrument can be found in Dinelli et al. (2009), where the configuration used during the SCOUT-O3 campaign is presented together with a rendering of the instrument and it's subsystems.

# 2.2 UAMS instrument upgrades

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The first scientific deployment of MARSCHALS during the SCOUT-O3 campaign took place before the instrument was operating according to its final specifications (only the 300 GHz band was operating nominally during the remote sensing flight). In order to address this, a comprehensive instrument upgrade and characterisation project "Upgrades to Airborne Millimetre-wave Sounder" (UAMS) took place between SCOUT-O3 and PREMIER-Ex. Following is a list of points addressed by the UAMS project.

# 2.2.1 Single Sideband Filter (SSBF)

New Single Sideband Filters were developed for the three MARSCHALS bands by the Queen's University of Belfast. All of them have both good image rejection (> 30 dB) and low signal insertion loss (< 0.5 dB). The new filters for band B and D are a free-standing 3-layer linear slot-array type Frequency Selective Surface (FSS). The band C filter is a free-standing 3-layer dual polarisation annular slot array type FSS.

# 2.2.2 Band B receiver

The original band B receiver, supplied for MARSCHALS from a separate ESA technology development programme, had unsatisfactory noise performance (> 20000K in single sideband (SSB)) and a pronounced gain slope in the IF pass-band. The entire receiver was replaced by a design developed by RAL and EADS Astrium (UK). The new receiver employs a simplified single-stage down-conversion scheme which eliminates the IF gain slope. In addition, a new mixer has been specifically designed for





this channel. It has been designed to have inherently better performance in the signal sideband and includes an integral IF preamplifier, thereby improving IF match and IF noise temperature. The feedhorn used on the new receiver is a scaled version of the design developed for band C (and band D) by the University of Navarra. A new doubler

<sup>5</sup> has also been developed for this channel. The noise temperature of the new receiver, measured with a broadband power meter, is listed in Table 1. The gain slope has also been measured and was shown to have decreased from ~ 10dB to less than 1 dB. Figure 1 shows the position of the band B receiver in the new layout after the UAMS update.

### 10 2.2.3 Band C receiver

The components in the band C receiver were unchanged (only the FSS single sideband filter was replaced). However it was slightly repackaged to accommodate changes in the physical layout of the band D receiver. After the FSS upgrades the receiver performance was improved, but the improvements were less than expected. After several iterations of components, the subharmonic mixer and doubler were both replaced, producing further improvements in the noise temperature. The final value is listed in Table 1. Figure 1 shows the position of the band C receiver in the new layout after the UAMS update.

### 2.2.4 Band D receiver

Prior to the deployment of MARSCHALS in SCOUT-O3 in Darwin, the band D receiver exhibited a SSB noise temperature of approximately 19 000 K. During the deployment in Darwin the channel failed and no data was obtained; this was ascribed to a failed doubler in the local oscillator chain. The mixer originally installed in this band had a noise temperature of ~ 2000K Double Side-Band (DSB). This has now been replaced with a new mixer with a noise temperature of approximately 1500 K DSB. The failed doubler incorporating a RAL





Schottky diode. The noise temperature of the new receiver is listed in Table 1. Figure 1 shows the position of the band D receiver in the new layout after the UAMS update.

# 2.2.5 Instrument Control Unit (ICU)

The ICU stack has been fitted with external console connectors to allow field diag nostics without stack disassembly (as was necessary in the SCOUT-O3 campaign). A number of software modifications have also been performed. An error in the reading of the aircraft roll-angle data has been corrected. The aircraft roll-angle data is used to periodically calibrate the inertial navigation system of MARSCHALS. In the SCOUT-O3 campaign incorrectly read aircraft data could offset the antenna pointing; this is now not the case anymore. Roll-angle updates were hardcoded to beginning of a full 3-band measurement cycle. However, to gain spectral performance the duration of a full set of

atmospheric measurements has now increased from ~ 1 min to ~ 10 min. To make up for that, the option for a "commandable roll-angle update" has been implemented in the ICU software, which permits more regular updates to the aircraft roll-angle to mitigate pointing angle drifts.

# 2.2.6 Antenna pointing system

The antenna pointing motor and electronics are exposed systems in the open instrument bay of the aircraft, where temperature and humidity variations between the stratosphere and ground conditions can be dramatic. There were some reliability issues with the inductosyn<sup>™</sup> interface board during the SCOUT-O3 campaign, so this system has also been replaced by a completely rebuilt Antenna Pointing System module. This has so far operated with 100 % reliability both in lab tests and in flight conditions.

### 2.2.7 Thermal stability

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Outside temperatures at 20 km flight altitude can be as low as -90 °C. This puts a lot of strain on the temperature stability control system. This is addressed by increasing the



wattage of the existing heaters on the radio frequency components, as well as adding a 4th channel to the thermal stabilisation control unit to allow the calibration hot load to be stabilised. In addition to that passive thermal insulation has been added to various subsystems. The cryostat bay and the calibration hot load have been enclosed to reduce convective cooling. The receiver modules, local oscillator modules, phase lock modules and the channeliser have all been fitted with insulating styrofoam padding.

### 2.2.8 Instrument characterisation

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The significant changes to the instrument performed under UAMS required a number of instrument characterisation measurements to be repeated. Most importantly these included measurements of the antenna function (field of view), the spectral response functions (instrument line shape) and the spectral performance of the receivers (standing waves, linearity, frequency calibration).

The Azimuthally Collapsed Antenna Patterns (ACAP) and polarisation have been measured in an anechoic chamber at Queen Mary University in London. Analogous <sup>15</sup> measurements have been performed in the past, but these repeat test are of higher quality because they are performed at higher angular resolution and because of the improved spectral performance of the instrument. The measurement setup has also been better shielded from standing waves in the path between the test signal source and the detector. The ACAP and the polarisation have been measured for all three <sup>20</sup> spectral bands individually and are used in the data analysis process. We have also determined the marginal line of sight offsets between the three bands and have corrected for them in the calculation of the scan angles.

The spectral response functions have been measured in the lab at RAL. As with the antenna characterisation, a much improved setup compared to the previous measure-

ments has been applied. A line is injected into the 12–24 GHz IF port using a coupler (and amplification to compensate for the coupler loss) so that the spectral response of the instrument is measured against an input comprising a swept continuous wave



signal from the synthesiser superimposed on a broadband noise source (the receiver front end viewing an ambient blackbody load).

The use of isolators at the input and output of the line injection system means that it can be removed from the system without significantly affecting the match between the receivers and IF (and hence the spectral response of the system under test). A first

<sup>5</sup> the receivers and IF (and hence the spectral response of the system under test). A first set of spectral response measurements had been performed before the PREMIER-Ex campaign.

Initial test results showed that most spectral channels were stable, however as was observed in previous spectral response tests a small proportion of channels were unstable. These channels showed noticeable fluctuations in spectral response between

- measurement runs. After initial tests, the location of the unstable channels relative to the spectral lines of interest was investigated. The conclusion was that a re-ordering of the channeliser units was advantageous, enabling the unstable channels to be moved to spectral areas of lower importance. The re-ordering of the channeliser units was completed and additional spectral response tests were undertaken. More details about
- the channel re-ordering can be found in Spang et al. (2012).

The 200 MHz spectral resolution of the MARSHALS channeliser does not allow the detection of high-frequency standing waves in the system. To check for the presence of standing waves, the University of Bremen kindly lent RAL an Observatoire de Paris-

- Meudon (ARPEGES-LED) wideband Acousto-Optic Spectrometer (AOS). The unit was the result of the ESA funded Wideband AOS WP2411 and has a nominal bandwidth of 1.5 GHz, channel spacing of 0.89 MHz. Within the limitations of the noise performance of the receivers no standing waves have been detected. More detailed results from the instrument characterisation, as well as accounts on other minor system tests can be found in Movna et al. (2010)
- <sup>25</sup> found in Moyna et al. (2010).





# 3 PREMIER-Ex scientific campaign

### 3.1 Campaign overview

The aim of the PREMIER mission is to quantify the processes that control the composition and structure of the UTLS. The PREMIER-Ex activities culminated in a scientific

- <sup>5</sup> campaign in the northern Polar region in March 2010. The scientific objectives addressed by the PREMIER-Ex campaign were the quantification of horizontal mixing processes in the UTLS and of the cirrus formation processes in the Arctic atmosphere. The PREMIER-Ex scientific flight (hereafter denoted as PrEx 1) was performed on 10 March 2010. During the first week of March the vortex was located above the Siberian
- <sup>10</sup> Arctic and then moved backward towards the Scandinavian Peninsula. On 10 March the polar vortex was located above the Scandinavia. The PrEx 1 flight track was planned so as to sound the vortex filamentation on the vortex boundary, and thus to probe the horizontal mixing of vortex and mid-latitude air. An additional constraint was to find a match with A-Train satellites and ENVISAT overpasses for validation purposes.
- <sup>15</sup> Using European Centre for Medium-range Weather Forecasts (ECMWF) assimilated forecast model data and the Chemical Lagrangian Model of the Stratosphere (CLaMS) model, the flight path was planned going to the West, as shown in the flight path reported in Fig. 2a and then going back to Kiruna with a triangular flight pattern. The PrEx 1 flight started at ~ 07:20 UTC and finished at ~ 10:30 UTC, while the A-Train
- satellite overpasses were performed at about 10:40 and 12:20 UTC and the ENVISAT overpass was performed at about 08:45 UTC. Precise temporal coincidence with correlative satellite data is therefore only available for the first two legs of the flight. The second PREMIER-Ex flight, which was supposed to study processes related to Arctic cirrus formation and planned for the 13 March 2010, could not be performed due to a severe aircraft engine problem.

In Fig. 2a we report the values of the Potential Vorticity extracted from the ECMWF archive in Potential Vorticity Unit (PVU,  $1PVU = 10^{-6} \text{ Km}^2 \text{ s}^{-1} \text{ kg}^{-1}$ ) at 125 hPa with respect to latitude and longitude in the flight region. In the same plot we report the M-55





flight track (in black) together with the position of the MARSCHALS tangent points for each scan (in grey). The red line marks the region with PV equal to 14 PVU, that can be considered as the outer border of the vortex at this pressure level (Mariotti et al., 1997; Christensen et al., 2005). Comparing the position of MARSCHALS tangent points with

<sup>5</sup> PV values we can notice that the flight was divided into three legs: the first leg enabled MARSCHALS Lines of Sight (LOS) to look partially inside the vortex, the second to look at air masses inside and outside the vortex and the third to look at mixed air.

# 3.2 MARSCHALS measurements during PREMIER-Ex scientific flight of 10 March 2010

- <sup>10</sup> As seen in Sect. 2.2, the UAMS upgrades produced an improvement of the noise performances. Additionally the system reliability was improved, which meant that the scan sequence could be speed up by removing redundant integration cycles at each view. This has allowed to increase the number of limb-scans per flight hour and to obtain a better horizontal sampling rate with respect to the SCOUT-O3 flight. During the PrEx
- <sup>15</sup> 1 flight, MARSCHALS collected 56 limb scans (19 in band B, 18 in band C and 19 in band D) in the time interval from 07:24 UTC to 10:27 UTC. The three bands were acquired in subsequent scans, first band C then band B and finally band D. During the flight MARSCHALS collected level 1 data of good quality for all the three bands. This is therefore the first time that MARSCHALS band B and D measurements could
- <sup>20</sup> be analysed to produce Level 2 data. Table 2 reports the frequency range where the three MARSCHALS bands have good performances and that were therefore used for the analysis reported in this paper.

Figure 2b shows the altitude of the tangent points of each scan given as a function of the acquisition time. The figure shows that there are two parts of the flight (from scans 6

to 31 and from scans 35 to 53) where the aircraft altitude was nearly constant. Figure 2b also shows that there was an uneven altitude coverage of the three bands during the flight. This affects the quality of the retrieval products, because of the different vertical distribution of the information content of each scan. The few scans that had a reduced





altitude coverage or were acquired during changes of direction of the aircraft were not used in the analysis.

# 4 MARSCHALS data analysis

In this section, we briefly describe the MARC retrieval code, the new features introduced in the retrieval code with respect to the version previously used for the SCOUT-O3 data analysis and we describe the results of the analysis of MARSCHALS spectra.

# 4.1 Retrieval code

The PREMIER-Ex MARSCHALS data analysis was performed using the MARC code (Carli et al., 2007). The MARC retrieval code was developed under an ESA contract for the retrieval of the calibrated and geolocated spectra measured by MARSCHALS. It makes use of the global-fit (Carlotti, 1988), multi target (Dinelli et al., 2004) techniques and uses Optimal Estimation (OE) or maximum a posteriori approach (Rodgers, 2000) and the Levenberg-Marquardt algorithm (Levenberg, 1944; Marquardt, 1963) during the iterative retrieval procedure to reduce the stepwidth and to speed up the conver-

<sup>15</sup> gence to the minimum of the cost function. A detailed description of the main features of the code is provided in Carli et al. (2007) and a description of its use for the analysis of real data can be found in Dinelli et al. (2009).

MARC retrieves vertically distributed quantities (such as temperature and Volume Mixing Ratios (VMRs) of gases whose emission lines are present in MARSCHALS <sup>20</sup> bands) and scalar quantities, in the form of one value for each analysed scan. The Forward Model (FM) internal to the retrieval code uses a state of the art molecular continuum model developed together with a specific database (Kuhn et al., 2003) and uses an additional (retrievable) continuum profile to account for the continuum like contribution of clouds opaque in the observed frequency region. Originally the retrievable

<sup>25</sup> scalar quantities were: the pointing bias (a correction term that enables to account





for eventual instrumental mispointings) and the gain and offset (used to correct possible calibration errors). For the PREMIER-Ex data analysis the possibility to retrieve a frequency shift has been implemented into the MARC code. The frequency shift is retrieved as a frequency offset applied to the Instrument Line Shape (ILS), and is a single scalar number for each scan, because it is assumed to be due to instrumental char-

acterizations only. The frequency shift assumes positive values when the simulated spectrum has to be shifted toward higher frequencies. We tested the retrieval of the frequency shift on simulated data obtaining satisfactory results (Cortesi et al., 2012).

Along with the profiles of the target species and the instrumental parameter, the MARC algorithm provides also retrieval diagnostics that can be divided into scan diagnostic and target diagnostic. The scan diagnostic quantities are: the reduced  $\chi^2$ , the information content and the trace of the full Averaging Kernel (AK) matrix (which gives information on the total degrees of freedom of the retrieval). The target diagnostic quantities are: the biased error, that is the combination of the retrieval error due to the noise with the a priori error, the AKs of each target, and the individual information content quantifier. These quantifiers have already been discussed in Dinelli et al. (2009), Carli et al. (2007). For this analysis we have also computed the vertical resolution of the retrieval products in the form of the Full Width at Half Maximum (FWHM) of the AK

### 20 **4.2 MARSCHALS retrieval strategy**

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using the expression given by Ridolfi and Sgheri (2009).

The retrieval strategy used for the inverse processing of MARSCHALS measurements during PrEx 1 campaign was optimised through a dedicated approach. As a starting point we applied the strategy used for the data analysis performed on the SCOUT-O3 data and described in Dinelli et al. (2009). However, the different geographic location of the measurements, the upgrades of the instrument and the presence for the first time

of all three bands implied that most of the adopted choices had to be revised.

Because the noise characterisation of the new instrument setup was more reliable, in the current analysis we used the noise level as reported in the Level 1B files, that





provide a measured value for every single spectral channel. In addition the new FOV and ILS functions, measured with the strategy described in Sect. 2.2 were used.

Each MARSCHALS scan was analysed individually, retrieving four scalar instrumental parameters (gain, offset, pointing bias angle and frequency shift) along with vertical distributions of temperature, water, ozone, nitric acid, nitrous oxide, carbon monoxide and extinction coefficient. Because of the different spectral coverage of the three

bands, the vertically resolved targets of each band.

- As already mentioned in Sect. 4.1, the MARC code uses the OE method. Therefore it is important to correctly define both the a priori information of the atmospheric status and its uncertainty. Temperature, pressure,  $O_3$  and  $H_2O$  a priori data (used also as initial guess profiles) were obtained from ECMWF Meteorological Archival and Retrieval System (MARS) archive, using the operational data on 91 model levels. The a priori data for all the other gases like N<sub>2</sub>O, CO and HNO<sub>3</sub> were taken from the IG2 database
- (Remedios et al., 2007) that contains climatological averaged profiles and 1-sigma variabilities for the four seasons and for 5 latitudinal bands. The IG2 1-sigma variability was used as a priori uncertainty for all the targets. The lower limit of this uncertainty was set to a target dependent threshold, expressed as percentage of the a priori profile. This threshold was 100 % for  $O_3$  and  $H_2O$  while it was set to 50 % for  $N_2O$ ,  $HNO_3$  and CO.
- <sup>20</sup> The a priori error used for temperature was a constant value of 3 K that is the estimated error of ECMWF data.

A series of preliminary tests were performed in order to find the better retrieval strategy that could be used for the fit. These tests showed that temperature and  $H_2O$  can be retrieved from band C measurements only. However because the initial guess profiles

of H<sub>2</sub>O are very different from the real atmospheric conditions and the profiles retrieved from band C are sometimes unstable, H<sub>2</sub>O is still included into the band B and D retrievals to account for the contribution given by the far wings of its lines to the analysed spectrum. The profiles obtained in the analysis of these bands are not included into the retrieval products. Therefore the devised retrieval strategy was to start the analysis





from a band C scan (scan *N*). For the two subsequent scans, N + 1 in band B and N + 2 in band D, we used the temperature value retrieved from scan *N* and we retrieved H<sub>2</sub>O (along to the other targets) using as a priori the profile retrieved from scan *N*. The minimum altitude of the used observations, as well as the optimal target dependent retrieval grids, were chosen after some dedicated tests, using the quantifiers described in Sect. 4.1 to find the best retrieval set up. All observations with tangent altitudes above the ground level were retained in the analysis. The used retrieval altitude grids, common to all the scans of PrEx 1 flight are listed (in km) below:

- *T*: 24, 20, 18, 17, 15, 13, 11, 9, 7, 5
- <sup>10</sup> H<sub>2</sub>O: 24, 21, 18, 17, 16, 15, 14, 13, 12, 11, 10, 9, 8, 7, 6, 4
  - O<sub>3</sub>: 24, 21, 18, 17, 16, 15, 14, 13, 12, 11, 8, 5
  - HNO<sub>3</sub>: 23, 20, 17, 16, 14, 12, 9, 6
  - N<sub>2</sub>O: 20, 16, 14, 12, 10, 8
  - CO: 21, 17, 13, 7

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- external continuum: 18, 17, 16, 15, 14, 13, 12, 11, 10, 9, 8, 7, 6

As for the SCOUT-O3 analysis, one or two retrieval grid points have been added above the highest flight altitude in order to reduce the errors due to the assumed vertical distribution of the targets above the highest retrieval point.

# 4.3 MARSCHALS retrieval results

Table 3 shows the averaged Degrees Of Freedom (DOF) obtained in the analysis of the three MARSCHALS bands for all the vertically resolved targets. We see that the three MARSCHALS bands contain an evenly distributed information on  $O_3$  and on the external continuum, while for HNO<sub>3</sub> the best information is contained in band C and D.





The values reported in Table 3 confirm that the highest information on  $H_2O$  is in band C, and show that for  $N_2O$  and CO the number of retrieved DOF is close to the number of altitudes of their retrieval grid. A comparison of Table 3 with Fig. 7 of Dinelli et al. (2009) shows that the number of DOF obtained for band C in this flight for  $H_2O$ ,  $O_3$  and HNO<sub>3</sub>

- <sup>5</sup> have increased. This is due to both the improvements of the MARSCHALS instrument and the lower tropopause altitude in the geographical region where the flight took place. In fact the SCOUT-O3 campaign was performed in Darwin, at tropical latitudes, and the tropopause altitude was always higher than the flight altitude, while in this flight the tropopause was always below the aircraft.
- <sup>10</sup> The values of the reduced  $\chi^2$  obtained in the analysis is reported in Fig. 3a, where the blue, black and red symbols are used for the scans of band B, C and D respectively. We see that the  $\chi^2$  value is on average around 1.4, a part from few scans in the third leg of the flight. Band B always presents slightly higher  $\chi^2$  values than the other two bands. As an example of the quality of the retrieval and of the instrument performances,
- the bottom panels of Fig. 3 report the values of the AKs trace and of the information content for each scan using the same colour code. As expected the three bands have different information content due to both the different number of targets and the covered spectral region. Band C always has higher DOF and information content. The lower information content and AK trace value found for some scan is due to their reduced altitude coverage (see Fig. 2b).

The retrieved values of the instrumental parameters offset, gain, pointing and frequency shift stay constant along the flight indicating that the instrument did not show problems of degradation along the flight. The retrieved values of the scalar quantities, averaged over all the results of the different bands, are shown in Table 4. Offset values

are very close to zero, and gain values are close to 1, indicating good radiometric performances of the instrument. The small values assumed by the pointing bias indicates that the a-posteriori pointing knowledge is very good. Finally the values obtained for the frequency shift are of the order of few MHz, largely below the instrumental frequency step (200 MHz).





# 5 Discussion

In this section we present and discuss the results obtained for the vertically resolved quantities from the analysis of MARSCHALS data during the PREMIER-Ex campaign on 10 March 2010. All the results obtained will be presented using colour maps where

- the examined quantity is mapped as a function of altitude and of acquisition time. We report on separate panels the retrieved quantities, the retrieval error, the individual information content and the vertical resolution (FWHM). We recall here that the individual information content is not an absolute quantifier, but it represents a relative quantity that shows how much the a priori error has been reduced during the analysis. A value of the information content of 0.5 represents a reduction of the a priori error of √2 (Dinelli
- et al., 2009). Retrieved values with individual information content very low are strongly biased towards the a priori.

The results are discussed in one subsection for each retrieval target. The obtained results will also be validated through the comparison with the vertical profiles obtained by MLS/Aura or MIPAS/ENVISAT satellite sensors.

In the validation exercise, the coincidence criteria applied was 4 h maximum time delay and 200 km maximum geographical distance. The use of MLS/Aura profiles retrieved with a tomographic (2-D) approach (Livesey et al., 2006) (taking into accounts horizontal inhomogeneities), is particularly appropriate for the comparison with MARSCHALS data during the PrEx 1 flight, where horizontal mixing and vortex filamen-

tation were under investigation. For the same reason we prefer to use the MIPAS2D database (Dinelli et al., 2010), instead of the ESA Level 2 MIPAS dataset, because it has been obtained with a full 2-D analysis code as well.

### 5.1 Temperature

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<sup>25</sup> Temperature profiles retrieved from band C are presented in Fig. 4. As already said in Sect. 4.2, temperature has been retrieved from band C scans only. However, the information content on temperature is very low, confirming the findings of the previous





analysis (Dinelli et al., 2009). During the flight the temperature profiles are homogeneous in the altitude range from 24 to 12 km, while below 12 km, where the value of the individual information content becomes higher, temperature shows a minimum in the third leg of the flight.

The quality of the retrieved temperature data can be evaluated comparing MARSCHALS results with MIPAS/ENVISAT coincident data. Results of the comparison between MARSCHALS and MIPAS/ENVISAT temperature profiles show a general good agreement. In particular low temperature values at about 12 km are present also in MLS/Aura data in correspondence of the third leg of the flight (see Cortesi et al., 2012).

### 5.2 Water vapour

The results obtained for  $H_2O$  in the analysis of band C scans are presented in Fig. 5 together with the information content and vertical resolution. As can be noticed the information content distribution is not homogeneous because of the uneven altitude coverage of the MARSCHALS measurements (see Fig. 2b). In particular, we have low information below 8 km in the second half of the flight and in the altitude range from 14 to 16 km from scan 21 to 30. The retrieved VMR does not show particular features, a part from an enhancement at low altitudes in the second half of the flight, that justifies the lower information content found (higher  $H_2O$  VMRs imply more opaque spectra).

<sup>20</sup> Comparison between water vapour retrieved profiles for MARSCHALS scans with MLS/Aura and MIPAS/ENVISAT coincident profiles (not shown) show a very good agreement with MLS data especially around the hydropause.

### 5.3 Ozone

O<sub>3</sub> can be retrieved from the measurements of all the three bands (band B, C and D).
 The obtained results are reported in Fig. 6. The information content of the three bands is always high, even if the best performances are obtained for band B measurements



and the worse ones for band D. The lower values are always found in the altitude region between 15 and 10 km. In any case, the information content of  $O_3$  during PrEx 1 is high over a wider altitude range with respect to what was found during the SCOUT-O3 data analysis. This is certainly due to the different geolocation of the flight, but also to the improvements of the instrument.

In Fig. 7, we compare  $O_3$  retrieved profiles for scans from 10 to 12 with MLS/Aura and MIPAS/ENVISAT coincident profiles and we find a generally good agreement. In particular,  $O_3$  profiles retrieved from band B and D (scan 10 and 11, respectively) show good agreement with satellite data. No particular features can be seen in MARSCHALS  $O_3$  profiles during the flight.

### 5.4 Nitric acid

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In Fig. 8, we present the results obtained for the retrieval of  $HNO_3$  profile from band C and band D data. The profiles obtained from band B data are not reported since the information on  $HNO_3$  contained in this band was not sufficient to produce significative results.  $HNO_3$  values retrieved by MARSCHALS show the presence of a filament of polar air with higher  $HNO_3$  values, between 14 and 16 km. Looking at Fig. 2a, we can

- notice that MARSCHALS limb scans from 11 to 20 approximately sounded the Arctic polar vortex, while scans from 36 to 47 explored a filament of vortex air, and finally scans from 6 to 10 and from 21 to 35 and from 48 to 54 were performed outside the
- vortex. This behaviour is reflected in HNO<sub>3</sub> values retrieved by MARSCHALS at about 125 hPa, reported in Fig. 9a with respect to pressure and time (where the black line marks the 125 hPa level) with higher HNO<sub>3</sub> values find in correspondence of vortex air and filament of vortex air. HNO<sub>3</sub> data have been compared to MLS/Aura, MIPAS-STR and with MIPAS/ENVISAT data with good results.
- The coincident MLS/Aura HNO<sub>3</sub> values along the orbit track between 68° N and 76° N latitude crossing the first two flight legs (Fig. 9b) show a similar behaviour at 125 hPa (marked by the black line), thus confirming MARSCHALS results. The capability of the





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N<sub>2</sub>O profiles can be retrieved only from band B data. The obtained results are reported <sup>5</sup> in Fig. 10. As can be seen in the figure, the information content of MARSCHALS band

investigation of horizontal mixing processes in the UTLS atmospheric region.

MARSCHALS instrument to resolve the filament of polar vortex air is crucial for the

- B is not particularly high, but it is sufficient to enable the MARC code to retrieve the N<sub>2</sub>O VMR values. Due to the choice we made to use relatively large a priori errors, in order to impose a weak constrain to the measurements, the obtained N<sub>2</sub>O profiles are sometimes oscillating.
- The comparison of N<sub>2</sub>O profiles retrieved from MARSCHALS and MIPAS/ENVISAT 10 profiles and MLS/Aura show a quite good agreement, especially at low altitude (see Fig. 11 for scan 10) even if the MARSCHALS N<sub>2</sub>O profiles show some oscillations respect to the others instruments. The obtained results can be considered a proof of the robust analysis performed on MARSCHALS data of the PREMIER-Ex flight from Kiruna. 15

#### Carbon monoxide 5.6

Nitrous oxide

5.5

In Fig. 12 we present the retrieval obtained for CO from band D scans. As can be seen in Fig. 12c, the information content of band D for CO is not particularly high and the final retrieved profile is very low, with values close to zero in the altitude range from 13

- to 20 km. Only at the end of the flight the CO value increases at low altitudes, but this 20 is due to the lower information content of these scans that increases the bias toward the a priori profile. The CO retrieval was repeated using different strategies, in order to understand if the obtained results were affected by biases. None of the adopted strategies changed the final results of the retrieval.
- The very low values of CO found in this analysis are in reasonably good agree-25 ment with the CO data predicted by the CLaMS model. However, if we compare



MARSCHALS CO data to the MLS/Aura ones (an example is reported in Fig. 13a for MARSCHALS scan 23), we see that we find lower values especially at the two lowermost pressure levels. In order to verify the possibility of the MARSCHALS instrument to measure CO profiles, we report here some results obtained during the mid-latitude test flight performed on 4 November 2009 (and then named TC9 flight), where the

- CO values seem to be higher. Unfortunately the TC9 flight was performed over a very particular flight path that prevents the optimal use of measurements acquired in the limb-sounding geometry. In addition, after the TC9, the spectral response functions of the channeliser measured in the laboratory show that one of the channels with vari-
- able spectral response function was located near the CO line (Spang et al., 2012), and for this reason a channeliser re-ordering was performed before the PREMIER-Ex Campaign (Sect. 2.2). In Fig. 13b, we present the results obtained for scan 8. As can be noticed, the retrieved CO profile reaches noticeably higher values compared to the ones recorded during the PREMIER-Ex polar flight and show a better agreement with
- <sup>15</sup> MLS/Aura coincident data (version 3.3). In order to better evaluate the sensitivity of MARSCHALS data to CO, we have performed the analysis of one of band D scans (scan 23 of PrEx 1) without including CO. In this case, in the retrieval CO abundance is set to the initial guess profile. We have used two different initial guess profiles: the one contained into the IG2 database for polar atmosphere and a constant with altitude
- <sup>20</sup> profile set to a very low value  $(1.0 \times 10^{-6} \text{ ppmV})$ . The residuals (difference between the observed and simulated spectra) reported in Fig. 13c, obtained when CO is retrieved or when it is fixed to low values, are very similar (blue and red line), while the ones obtained when CO is set to the IG2 profile (yellow line) show a systematic feature in coincidence with the CO spectral position. This indicates that indeed we were in pres-
- ence of low CO values during the flight, close to the lower limit of detectability. A similar analysis was performed also for the TC9 flight, where the CO retrieved values were higher than the ones obtained in the polar flight. Repeating the analysis without retrieving CO and setting it to  $1.0 \times 10^{-6}$  ppmV, a feature in the residual in correspondence to the CO line can be seen (red line in Fig. 13d). This indicates that when the CO VMR





is higher than the ones obtained in the polar flight the MARSCHALS instrument can detect CO.

# 6 Clouds

- As already said in Sect. 4.1 to account for possible continuum-like cloud contamination of MARSCHALS spectra we retrieve external continuum profiles (Fig. 14), in the form of a cross section contribution to the spectrum (units  $10^{27} \text{ cm}^2$ ). Since its interpretation is not straightforward, in the following analysis we provide the corresponding extinction coefficient in km<sup>-1</sup> obtained from the retrieved external continuum using the ideal gas law. As can be seen in Fig. 3a, the small values obtained for the reduced  $\chi^2$  during the flight suggest that the retrieved values of the external continuum were sufficient to model the effects of clouds for almost all the flight. In one case only, for scan number 51, the retrieved extinction coefficient reaches a value ( $1.4 \times 10^{-3} \text{ km}^{-1}$ ) that suggests the presence of a cloud whose opacity is such to deserve a proper simulation in the retrieval procedure. In this case, the inclusion of the MSSF (Mie Scattering
- <sup>15</sup> Source Function) module, described in Del Bianco et al. (2007), for the calculation of the scattering contribution given by the cloud should be considered. This module computes the scattering, absorption and extinction coefficients of the cloud and the scattering source function to be included into the radiative transfer equation. This calculation is performed using as input the mean radius (*R*) and the number density of
- <sup>20</sup> the cloud particles and some information about the cloud phase (ice/water). Since the retrieved external continuum cannot provide all this information, external information about the cloud properties has to be used. In particular, we used *R* and phase values taken from the MODIS (MODerate resolution Imaging Spectroradiometer) instrument on-board the Terra satellite, whose measurements are in close temporal coincidence
- with MARSCHALS measurements (R = 25 micron, ice particles), while the cloud vertical extension (from 12 to 10 km) was inferred from the retrieved profiles of external continuum (see Fig. 15) and the number density was in the range of values observed





by in-situ instruments on-board the Geophysica during the descent. By repeating the analysis for scan 51 with the MSSF module on and these inputs, we obtain the results displayed in red in Fig. 16. In Fig. 15, we report the profile of the retrieved extinction coefficient for the standard analysis (blue) and for the analysis performed with the

MSSF module on (red). It can be noticed that, when the cloud is simulated into the forward model, the retrieved extinction profile approaches zero, confirming the correct cloud modelling into the MARC FM. The VMR profiles retrieved with the MSSF module on, presented in Fig. 16 in red, show slightly less oscillations compared to the ones retrieved with the MSSF module off (in blue). However, the fact that the external continuum retrieval is sufficient to produce good results is highlighted by the almost identical

 $\chi^2$  values obtained in the case of standard analysis and the analysis with the inclusion of the MSSF module.

The presence of high clouds in the second part of the flight is confirmed by instruments on-board the M-55 Geophysica and on-board the satellites. Information about <sup>15</sup> cloud coverage during the flight is given by the OCM module snapshots, that captured the presence of low clouds during the first leg, no cloud during the second leg and a layer of high altitude clouds on the third leg. This cloud scenario is also confirmed by the MIPAS-STR Cloud Index values (CI, Spang et al., 2004), shown in Fig. 126 of Cortesi et al. (2012). The CI map showed low clouds during the first leg of the flight

- (Cloud Top Height (CTH) below 9–8 km), no clouds up to 5 km in the second leg and high altitude clouds (CTH up to 12–11 km) in the third leg. Also MODIS and Cloud-Aerosol Lidar and Infrared Pathfinder Satellite Observations (CALIPSO) (Winker et al., 2009) coincident data confirm the presence of high altitude clouds (CTH about 11–12 km) in coincidence with MARSCHALS tangent points during the third flight leg. The
- <sup>25</sup> fact that MIPAS-STR measurements performed in the infrared spectral range were not available below cloud CTH, while the MARSCHALS measurements performed in the millimetre-wave were not affected by cloud presence apart from one sequence highlights the importance of the millimetre-wave spectral region to explore the UTLS region.





# 7 Conclusions

In this paper, we presented the results from the data analysis of limb sounding observations of the Arctic UTLS, the MARSCHALS data analysis of the PREMIER-Ex flight performed on 10 March from Kiruna, Sweden. For the first time the instrument <sup>5</sup> was operating in its full configuration with all the three spectral bands working. The availability of all the spectral bands allows to retrieve the vertical profiles of temperature, water vapour, O<sub>3</sub>, HNO<sub>3</sub>, N<sub>2</sub>O and CO, as well as to extract information about the presence of thick clouds. The PREMIER-Ex flight of 10 March 2010 was aimed at studying horizontal mixing processes in the UTLS. The MARSCHALS instrument reveals the presence of a filament of polar air with enhanced HNO<sub>3</sub> values, as predicted by the PV values from ECMWF. The filament from 16 to 14 km was well traced by the MARSCHALS instrument and the observation backed up by validation with MLS/Aura data. The presence of low and high altitude clouds during the flight did not have a significant impact on MARSCHALS spectra apart from a single scan for which a cloud

- <sup>15</sup> modelling algorithm was for the first time applied to real data with very good results. The possibility to retrieve CO profile from millimetre-wave spectra was thoroughly explored, since in the PREMIER mission the millimetre-wave sensor STEAM-R will be the only instrument able to provide CO measurements. As an air-borne demonstrator for the STEAM-R millimetre-wave limb sounder, the MARSCHALS instrument widely
- demonstrated its capability in this flight. The testing and demonstration of the inverse modelling of MARSCHALS measurements provide crucial information in the perspective of future deployment of millimetre-wave radiometers from space, like the STEAM-R instrument on-board the PREMIER mission. The quality of the retrieval products obtained from MARSCHALS limb spectra in the Arctic region is, in fact, relevant both
- for the information content of the individual measurements in the millimetre-wave region, as well as for their potential synergy with the information retrieved from independent and collocated observations in the infrared, like the ones from the InfraRed Limb Sounder (IRLS) of PREMIER. The potential of combined retrieval of millimetre-wave





and middle infrared limb measurements in the UTLS was successfully tested using MARSCHALS measurements reported in this paper and MIPAS-STR data. Full details of this activity will be reported in a paper currently in preparation and are already published in Cortesi et al. (2012).

- 5 Acknowledgements. The authors acknowledge ECMWF for access to data. The MLS dataset used in this work were produced with the Giovanni online data system, developed and maintained by the NASA Goddard Earth Sciences (GES) Data and Information Services Center (DISC). The MODIS data used in this study were acquired as part of the NASA's Earth-Sun System Division and archived and distributed by the MODIS Adaptive Processing System (MODAPS). The CALIOP data were obtained from the NASA Langley Research Center Atmo-
- spheric Science Data Center. We acknowledge J. Langen for useful comments. This activity was performed as part of the project "PREMIER Analysis of Campaign Data" under ESA contract N. 4000101374//10/NL/CT. The authors gratefully acknowledge all the PIs of the instruments on-board the M-55 Geophysica during the PREMIER-Ex campaign.

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**Table 1.** Frequency coverage and spectral performance of the three MARSCHALS spectral bands. The signal band dimensions reported in the table refer to the original instrument configuration.

Band B	Band C	Band D
317.46	337.48	360.80
294.00–305.46	316.48-325.48	342.20–348.80
329.46-341.00	349.48-358.48	372.80–379.40
11.66	9.20	6.80
0.2	0.2	0.2
44	40	32
O <sub>3</sub>	H <sub>2</sub> O (325.2 GHz),	CO (345.79 GHz),
	O <sub>3</sub>	HNO <sub>3</sub> , O <sub>3</sub>
19000	18 000	19000
6000	10 000	9000
linear slot FSS	dual polarisation	linear slot FSS
	annular slot FSS	
> 30 dB	> 30 dB	> 30 dB
< 0.5 dB	< 1.3dB	< 1.2dB
	Band B 317.46 294.00-305.46 329.46-341.00 11.66 0.2 44 O <sub>3</sub> 19000 6000 linear slot FSS > 30 dB < 0.5 dB	$\begin{array}{c c} \mbox{Band B} & \mbox{Band C} \\ 317.46 & 337.48 \\ 294.00-305.46 & 316.48-325.48 \\ 329.46-341.00 & 349.48-358.48 \\ 11.66 & 9.20 \\ 0.2 & 0.2 \\ 0.2 & 0.2 \\ 44 & 40 \\ 0_3 & H_2O(325.2GHz), \\ 0_3 & 19000 & 18000 \\ 6000 & 10000 \\ 10000 & $



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**Table 2.** Frequency range and retrieval targets of the three MARSCHALS bands. The frequency ranges of the MARSCHALS bands reported in the table, used for the PrEx1 data analysis, follow the updated instrument characterization.

Band	Frequency range	Target retrieval quantities
В	296.760-305.360	O <sub>3</sub> , N <sub>2</sub> O, HNO <sub>3</sub> , external continuum
С	317.780–325.380	T, $H_2O$ , $O_3$ , $HNO_3$ , external continuum
D	341.900–348.100	$O_3$ , HNO <sub>3</sub> , CO, external continuum

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**Table 3.** Degrees of Freedom of the three MARSCHALS bands for the vertically resolved targets.

Band	Temperature	H <sub>2</sub> O	O <sub>3</sub>	HNO <sub>3</sub>	$N_2O$	CO	Cont.
В	_	2.8	10.8	3.2	5.2	_	6.5
С	4.5	15.0	10.2	6.4	_	_	5.6
D	-	3.6	10.0	6.7	_	3.2	5.9

Band	Offset K	Gain	Pointing bias deg	Frequency shift MHz
В	$0.38 \pm 0.84$	$0.98 \pm 0.01$	$-0.036 \pm 0.071$	$10.1 \pm 0.6$
С	$-0.82 \pm 1.43$	$1.00 \pm 0.01$	$-0.014 \pm 0.041$	$12.8 \pm 2.9$
D	$-1.57 \pm 0.76$	$0.99 \pm 0.01$	$-0.005 \pm 0.025$	$11.8 \pm 1.2$

 Table 4. Average scalar quantities retrieved from the three MARSCHALS bands.





**Fig. 1.** Constellation of the MARSCHALS receivers and sideband filters after the UAMS instrument upgrades. The atmospheric signal enters the quasioptical subsystem at the bottom of the picture. The band D signal is separated by a polariser, the band B and C signals are separated by a beam splitter.



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Fig. 2. (a) PV in PVU obtained from the ECMWF archive at 125 hPa as a function of latitude and longitude in the flight region. Black line marks the flight track during the PrEx 1 flight, the position of MARSCHALS tangent points of each scan is shown in grey, while blue triangles represents the position of MLS/Aura scans. (b) Position of MARSCHALS measurements for band B, band C and band D with respect to altitude and time for the PrEx 1 flight.



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**Fig. 3. (a)** Retrieved reduced  $\chi^2$  for PrEx 1 flight for each scan as a function of time. **(b)** Trace of AK matrix for each scan as a function of time. **(c)** Information content for each scan as a function of time. Blue for band B, black for band C and red for band D.





**Fig. 4.** Maps of retrieved values and quantifiers plotted as a function of altitude and acquisition time. The black line is the aircraft altitude and the black dots mark every single scan. White dots mark the used retrieval grid points. **(a)** Temperature retrieved from band C scans, **(b)** its retrieval (biased) error, **(c)** the individual information content and **(d)** the vertical resolution (FWHM).



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Fig. 5. As in Fig. 4 but for  $H_2O$  retrieved from band C scans only.





**Fig. 6.** As in Fig. 4 but for  $O_3$  retrieved from all bands.









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Fig. 8. As in Fig. 4 but for HNO<sub>3</sub> retrieved from band C and D.



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**Fig. 9. (a)** HNO<sub>3</sub> values retrieved from MARSCHALS as a function of pressure (from 90 to 200 hPa) and time. The black line corresponds to the 125 hPa level while grey points represent the MARSCHALS retrieval grid points. **(b)** HNO<sub>3</sub> MLS/Aura values as a function of pressure and latitude for the scans marked with triangles in Fig. 2a.





Fig. 10. As in Fig. 4 but for  $N_2O$  retrieved from band B.



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Fig. 11. Comparison of N<sub>2</sub>O profile retrieved from MARSCHALS scan 10 (grey) with MI-PAS/ENVISAT (blue) and MLS/Aura (red) N<sub>2</sub>O profiles.





Fig. 12. As in Fig. 4 but for CO retrieved from band D.





**Fig. 13.** Top: CO profiles retrieved from MARSCHALS measurements (grey) and from MLS/Aura (red): (a) scan 23 of PrEx 1. (b) Scan 8 of TC9. Bottom: average residuals (observed minus simulated) spectra (blue line CO retrieved, red line CO very low, yellow line CO as in IG2 database). The green line is the average noise (missing values when spectral points are not used in the analysis): (c) scan 23 of PrEx 1. (d) Scan 8 of TC9. The grey vertical line marks the position of the CO line at 345.79 GHz.





**Fig. 14. (a)** External continuum retrieved from all bands and **(b)** its retrieval (biased) error. Units  $10^{27}$  cm<sup>2</sup>.











**Fig. 16. (a)** Temperature, **(b)**  $H_2O$ , **(c)**  $O_3$  and **(d)**  $HNO_3$  retrieved from MARSCHALS scan 51 using (red) or excluding (blue) the MSSF module.

