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## HAI –

## A new, airborne, absolute, twin dual-channel, multi-phase TDLAS-hygrometer

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14 **Abstract**

15 The novel Hygrometer for Atmospheric Investigations (HAI) realizes a unique concept for simultaneous  
16 gas-phase and total (gas-phase + evaporated cloud particles) water measurement. It has been developed  
17 and successfully employed for the first time on the German HALO research aircraft. This new instrument  
18 combines direct Tunable Diode Laser Absorption Spectroscopy (dTDLAS) with a first-principle evaluation  
19 method to allow absolute water vapor measurements without any initial or repetitive sensor calibration  
20 using a reference gas or a reference humidity generator. HAI contains two completely independent dual-  
21 channel (closed-path, open-path) spectrometers, one at 1.4  $\mu\text{m}$  and one at 2.6  $\mu\text{m}$ , which allow together to  
22 cover the entire atmospheric  $\text{H}_2\text{O}$  range from 1 to 40 000 ppmv with a single instrument. Both spectrometers  
23 comprise each a separate, wavelength-individual extractive, closed-path cell for total water (ice and gas-  
24 phase) measurements. Additionally, both spectrometers couple light into a common, open-path-cell outside  
25 of the aircraft fuselage for a direct, sampling-free and contactless determination of the gas-phase water con-  
26 tent. This novel twin dual-channel setup allows for the first time multiple self-validation functions i.e. in  
27 particular a reliable, direct, in-flight validation of the open-path channels. During the first field campaigns,  
28 the in-flight deviations between the independent and calibration-free channels (i.e. closed-path to closed-  
29 path and open-path to closed-path) were on average in the 2% range. Further, the fully autonomous HAI  
30 hygrometer allows measurements up to 240 Hz with a minimal integration time of 1.4 ms. The best preci-  
31 sion is achieved by the 1.4  $\mu\text{m}$  closed-path cell at 3.8 Hz (0.18 ppmv) and by the 2.6  $\mu\text{m}$  closed-path cell at  
32 13 Hz (0.055 ppmv). The requirements, design, operation principle and in-flight performance of the hy-  
33 grometer are described in this work.

34 **1. Introduction**

35 Water vapor is in many ways one of the most important measurand for atmospheric investigations  
36 (Ludlam, 1980; Möller et al., 2011; Ravishankara, 2012). Water is the most important greenhouse gas (Kiehl  
37 and Trenberth, 1997), and known as a key atmospheric coupling element of almost all microscopic (e.g.  
38 droplets/ice crystals formation), macroscopic (e.g. clouds/precipitation) and global processes (e.g. hydrolog-  
39 ical cycle). Therefore, it is strongly related to the numerous highly relevant topics of atmospheric science  
40 and closely related to “climate change” (Held and Soden, 2000; Houghton, 2009; Kiehl and Trenberth, 1997;  
41 Maycock et al., 2011). Unsurprisingly, numerous water vapor studies have been carried out targeting its



42 atmospheric trends and variability (Lu and Takle, 2010; McCarthy et al., 2009; Ross and Elliott, 1996; Scherer  
43 et al., 2008; Trenberth et al., 2005; Xie et al., 2011), its influence on transport models (Kiemle et al., 2012;  
44 Schäfler et al., 2010), or its impact on radiation balance models (Lockwood, 1990; Ramanathan et al., 1989;  
45 Schneider, 1972). One reason for the high complexity of atmospheric water vapor is that it is one of the few  
46 atmospheric molecules that appears in all three phases. Water in the gas-phase is a very strong infrared  
47 absorber and significantly impacts atmospheric energy fluxes through latent heat transfers by the different  
48 phase transitions. Condensation to the liquid phase or freezing to solid particles leads to effective scattering  
49 of solar radiation, which directly raises links to the formation process of cirrus clouds (Krämer et al., 2009;  
50 Spichtinger et al., 2004). These relationships show the complexity from a theoretical as well as a modeling  
51 point of view. Today, however, the quality e.g. particularly accuracy and comparability of atmospheric  
52 water measurements frequently limit a better understanding of key atmospheric processes (Krämer et al.,  
53 2009; Peter et al., 2006; Scherer et al., 2008; Sherwood et al., 2014). Despite the outlined importance and the  
54 large effort invested in the developments of hygrometers in recent years, water vapor remains a target mol-  
55 ecule that is very difficult to measure accurately.

56 Several major issues exacerbate water vapor measurements. Atmospheric water vapor encompasses a very  
57 large concentration range: 3 – 40 000 ppmv from troposphere up to the lower stratosphere. The spatial fluc-  
58 tuations of H<sub>2</sub>O in the atmosphere are high, which leads on fast aircraft (approx. 800 km/s cruising speed) to  
59 highly dynamic H<sub>2</sub>O variations of up to several 1000 ppmv/s in the gas-phase and up to several  
60 10 000 ppmv/s for total water measurements. These issues require, especially for aircraft based hygrome-  
61 ters, very high time resolution in combination with very high precision and accuracy. Additionally, since  
62 water vapor readily changes from one phase to the other, it would be extremely helpful if hygrometer were  
63 able to differentiate between the phases or could differentiate at least between water vapor and total water  
64 in order to minimize systematic uncertainties caused by the sampling process. Last but not least, water va-  
65 por is very effectively absorbed from nearly any surface. This challenges in a highly complex manner not  
66 only the entire gas sampling system but also the calibration infrastructure which is typically required for  
67 most hygrometers. By waiving the entire calibration process, special hygrometers (Wolfrum et al., 2011)  
68 circumvent all calibration related issues efficiently which will explained in chapter 3.2.

69 This brief compilation illustrates the complex challenges associated with developing a water vapor instru-  
70 ment, especially if it should be able to measure in tropospheric and stratospheric atmospheric conditions.  
71 Numerous (mostly single-channel) hygrometers have been developed in the last decades with various ad-  
72 vantages and drawbacks (see (Wiederhold, 1997) and e.g. (Buck, 1985; Busen and Buck, 1995; Cerni, 1994;  
73 Desjardins et al., 1989; Diskin et al., 2002; Durray et al., 2008; Ebert et al., 2000b; Gurlit et al., 2005; Hansford  
74 et al., 2006; Helten et al., 1998; Hunsmann et al., 2008; Karpechko et al., 2014; Kley and Stone, 1978; May,  
75 1998; Meyer et al., 2015; Ohtaki and Matsui, 1982; Roths and Busen, 1996; Salasmaa and Kostamo, 1986;  
76 Schiff et al., 1994; Silver and Hovde, 1994b, 1994a; Thornberry et al., 2014; Webster et al., 2004; Zöger et al.,  
77 1999a, 1999b)). Consequently, the question should be raised from the opposite point of view: What are the  
78 important and required properties to be covered and combined for a near-universal “Hygrometer for At-  
79 mospheric Investigation” to serve as an innovative and cutting-edge tool to explore open and new scientific  
80 questions related to atmospheric water vapor?



81

82 **2. Requirements for HAI, a “Hygrometer for Atmospheric Investigation”**83 **2.1. Specific instrumental boundary conditions defined by the atmosphere**

84 Currently, atmospheric water vapor data are generated from two instrument types: Remote sensing ap-  
85 proaches (like satellites (Oelhaf et al., 2004) or ground-based – often FTIR based - monitoring stations  
86 (Zachariassen et al., 2003)) or in-situ hygrometers deployed directly inside of the environment to be meas-  
87 ured. The latter are roughly separated in ground-base “weather stations”, balloon-borne radiosondes  
88 (Miloshevich et al., 2006) and hygrometers on airborne carriers like aircraft (Marenco et al., 1998). While  
89 ground-based stations permanently provide local information, radiosondes deliver “only” one point like  
90 vertical profile per ascend/ descend but for heights up to 50 km above ground. Airborne vehicles like air-  
91 craft and helicopters, however, combine the possibility for arbitrary flight paths and localized measure-  
92 ments, e.g. for cloud investigations, with a beneficial long operation range. In particular the most modern  
93 research aircrafts such as the German HALO aircraft (Anon, 2014; Krautstrunk and Giez, 2012) or the US  
94 American HIAPER (Anon, n.d.) bridge the gap and combine broad spatial coverage (> 10 000 km), high  
95 altitudes (up to 15 km), and large payloads of up to 3000 kg payload, in a favorable pressurized, and air-  
96 conditioned cabin. The relatively high traveling speed of these aircraft (of up to 230 m/sec), however, also  
97 causes disadvantageous influences on the measurements themselves, which are difficult to take care of  
98 compared to the very slowly moving (low m/sec range), quasi-static balloon-borne radiosondes. Even rela-  
99 tively simple meteorological variables such as air temperature and pressure need complex retrieval algo-  
100 rithms (Giez, 2012) if measured on a high speed aircraft. In short, aircraft in the contemporary working  
101 equipment are indispensable for both the investigation of spatially confined effects too small to retrieve  
102 from remote sensing data and the validation of the remote sensing instruments (Oelhaf et al., 2004). It is  
103 highly desirable for an airborne system to be constructed in a way that it operates on the ground in the  
104 same manner as in flight and that every change of environment and “boundary conditions” of the instru-  
105 ment is logged. This allows, besides multipliable deployments e.g. in ground based stations, extensive vali-  
106 dation as well as laboratory comparisons with other instruments and avoids systematic, barely detectable  
107 deviations only occurring in flight. This notion related to water vapor leads directly into the everlasting  
108 discussion about sampling via open-path systems operation in the free flow versus closed-path systems  
109 extracting the air to be analyzed into the instrument. Open-path hygrometers offer numerous great benefits  
110 such as: the prevention of any sampling errors or uncertainties (caused by surface absorbing effects) as well  
111 as the high response time that is limited only by the transfer functions of optical or electrical components  
112 but not by the gas exchange rate. The latter circumvents the complicated and adulterant deconvolution of  
113 smoothing effects with time-response functions caused by a sampling system. On the other hand, the  
114 boundary conditions such as gas pressure and gas temperature as well as possible spatial inhomogeneities  
115 of both parameters are difficult to accurately take into account for an open-path system. Additionally, an  
116 airborne open-path sensor has to operate in harsh boundary conditions, i.e. over a large range of tempera-



117 tures (-80 to 50 °C), pressures (70 to 1000 hPa), for large ram pressures (900 km/h gas velocity) and mechan-  
118 ical stress through accretion of ice or liquid water.

119 The major problem of all present open-path systems is their highly complex calibration, or even just valida-  
120 tion, since realistic flight conditions, in particular the dynamics, are extremely difficult to realize in the lab  
121 with sufficient accuracy. A direct metrological link to test the inflight performance, e.g. a dynamic calibra-  
122 tion facility for open-path hygrometers, is therefore missing.

123 Closed-path systems, on the other side, are simply installed inside an air-conditioned cabin, in a much more  
124 protected environment. Gas is sampled with a suitable inlet and led via a tubing system to an “internal”  
125 measurement chamber, such as an optical absorption cell or a suitable “cavity”, e.g. for a dew point mirror  
126 hygrometer. On one hand, it is much easier to accurately control and maintain the physical boundary con-  
127 ditions of the sample gas, e.g. temperature, pressure, flow, etc. in the measurement volume. On the other  
128 hand, it is difficult to ensure and maintain a representative sampling process and to quantify and correct  
129 sampling related deviations. These maybe caused by adsorption and desorption effects, which occur on all  
130 surfaces of the sampling system, and have to be carefully minimized by heated (HAI ≈80 °C) stainless steel  
131 sampling pipes, along with an instrument design ensuring high gas flows (HAI ≈100 liter/min) under all  
132 flight conditions.

133 Gas-phase H<sub>2</sub>O measurements in clouds are often carried out via backward facing sampling inlets. But such  
134 inlets are readily sampling small liquid or solid water particles possibly causing systematically positive  
135 offsets. For every H<sub>2</sub>O measurement using airborne extractive (=closed-path) instruments on aircrafts, it  
136 needs to be taken into account that the instruments response reflects contributions from the sensor element  
137 itself as well as the sampling/tubing system and their dynamic properties. One indisputable, major ad-  
138 vantage of typical extractive instruments is the possibility for careful tests outside of the aircraft, i.e. in a  
139 hangar or laboratory. But, to take full advantage of this, it is desirable for a sophisticated instrument to in-  
140 tegrate supervising and monitoring functions in a way that a performance comparable to the laboratory can  
141 be ensured during any inflight situation. This generates the great benefit of transferring the performance  
142 from the laboratory to the field situation, quite similar how a metrological transfer standard is typically  
143 used. This directly reinforces the question, how to assess the accuracy of airborne hygrometers.

## 144 2.2. Accuracy and state of the art instrumentation

145 In general, the highest measurement and preparation accuracy is realized by the validated primary stand-  
146 ards of national metrology institutes such as PTB (Germany) or NIST (USA). The international, metrological  
147 water vapor scale is defined by traceable primary water vapor generators (Brewer et al., 2011). The mixing  
148 ratio range, required to cover the entire tropospheric and stratospheric coverage of about 3 - 40000 ppmv, is  
149 realized by a combination of generators based on different physical principles. Their typical uncertainty is  
150 in the order of 0.5% relative, but varies with the physical principle used and their realization (Brewer et al.,  
151 2011; Buchholz et al., 2014a; Mackrodt, 2012). In other words, it is not possible to validate any hygrometer  
152 with better accuracy due to the lack of a suitable accurate reference.



153 Comparing the available metrological accuracy to some results from field comparisons of airborne hygrom-  
154 eters demonstrates the large potential for improvement. For example, long-term (> 10 years) change studies  
155 of stratospheric H<sub>2</sub>O (Oltmans and Hofmann, 1995; Rosenlof et al., 2001; Solomon et al., 2010) suffer from  
156 significant, difficult to quantify relative deviations between different instruments in the range of 50–100%  
157 (Fahey et al., 2014; Peter et al., 2006; Vömel et al., 2007) which recent studies such as (Rollins et al., 2014)  
158 confirm ( $\pm 40$ –50% for < 3ppmv and  $\pm 20$ % for > 3ppmv). Radiosonde comparisons with polymer sensors and  
159 chilled dew point mirror hygrometer, covering the entire troposphere and lower stratosphere region such  
160 as (Miloshevich et al., 2006), show averaged overall agreement in the 10% range (but also, local deviations  
161 in the 30% range). These deviations are quite common for many airborne campaign results e.g. (Smit et al.,  
162 2014) and become even worse when focusing on the relative deviations in regions containing highly varia-  
163 ble H<sub>2</sub>O structures. Hence in 2007, an international comparison exercise, “AquaVIT” (Fahey and Gao, 2009),  
164 was organized to compare the world’s best airborne hygrometers under well-controlled, quasi-static,  
165 equivalent conditions to evaluate the accuracy under well controlled laboratory conditions, without the  
166 influence of any typical airborne sampling and dynamic effects. AquaVIT comprised 22 hygrometers (tuna-  
167 ble diode laser spectrometers, TDL, dew or frost point mirror hygrometers, D/FPH, Lyman alpha fluores-  
168 cence and absorption hygrometers, LAFH and other principles) from 17 international research groups. The  
169 instruments were categorized in well validated “core” instruments (APicT, FISH, FLASH, HWV, JLH, CFH,  
170 see (Fahey and Gao, 2009) for details) and “younger, less mature” non-core instruments. Even the core-  
171 hygrometers deviated in the important 1 to 150 ppmv WVMR range by up to  $\pm 10$  % from the mean. In oth-  
172 er words, core instruments differed by up to 20% from each other, even under quasi-static conditions. Oth-  
173 er, less representative and extensive comparisons such as (Hoff, 2009; Mangold and Wodca Team, 2003)  
174 yielded similar results.

175 Of course, the assessment of the required accuracy depends strongly on the purpose of the data. In terms of  
176 climatologies and strongly averaged or coarse validation studies such deviations can be acceptable, but as  
177 far as e.g. retrieval models for satellite data improve, such uncertainties and deviations can become critical.  
178 In many other cases such as the currently often discussed atmospheric super saturations (Peter et al., 2006),  
179 the instrument uncertainties prevent deeper investigations and therefore a better understanding. Reconsid-  
180 ering the entire situation and perceiving that after so much development effort over the past decades these  
181 deviations remain quite high, leads to the inevitable question of the concealed, common impact factors.  
182 Contemplating the typical metrological efforts needed at national metrology institutes (NMI) to generate an  
183 accurately humidified gas stream (with a sub-percent uncertainty) suggests that the uncertainties generated  
184 by typical calibration processes under field conditions could be a major contribution to these hygrometers  
185 deviations found in AquaVIT and other studies. In particular, comparing the performance and strategies of  
186 lab-based, metrological and portable field calibration facilities (Friehe et al., 1986; Helten et al., 1998;  
187 Podolske et al., 2003; Smit et al., 2000; Smorgon et al., 2014; Zöger et al., 1999b) show three significant dis-  
188 crepancies: required time for calibration, frequency of calibration, and traceability of the humidity reference  
189 itself. Calibrations in the low concentration ranges at NMIs take several hours up to days per individual  
190 humidity value. During airborne campaigns, however, calibrations often *have* to be realized (for practical  
191 reasons) in a short time, certainly less than a few hours (max) for a large number of concentration steps



192 often including several pressure levels, thereby taking the obvious risk that the instrument/reference is not  
193 fully stabilized or equilibrated. Ideally, the time between two calibrations should be shorter than the ex-  
194 pected time required for a drift/change exceeding the boundaries of the instrument uncertainties. Some  
195 airborne instruments require for the same reason calibrations before and after each flight in order to inter-  
196 polate between both calibrations (Zöger et al., 1999b). Some even work with in-flight calibrations  
197 (Kaufmann et al., 2016) sacrificing measurement time and shifting the accuracy issue to the necessary air-  
198 borne H<sub>2</sub>O-source. Undoubtedly, many of these instruments have benefits e.g. in terms of precision, space,  
199 weight, prime cost etc., which justifies the calibration effort. However, vice versa, it is condoned that the  
200 calibration process is hampered and turns out to be the major influence on the accuracy of such a sensor.  
201 Lastly, it seems necessary to implement a traceable link to the metrological humidity scales to improve the  
202 overall accuracy of airborne hygrometry (Joint Committee for Guides in Metrology (JCGM), 2009). By real-  
203 izing an unbroken chain of calibrations, it is possible to link the instrument performance and the metrologi-  
204 cal water scale respectively the SI system of units. This also guarantees an accurate measure-  
205 ment/generation value with defined uncertainties.

206 To summarize: Fulfilling all these demands in the field similar like in a NMI laboratory is a tough task.  
207 However, as discussed later, many of the covered issues can be circumvented using first principle tech-  
208 niques like dTDLAS (Ebert and Wolfrum, 1994; Schulz et al., 2007) to realize optical, absolute hygrometers  
209 which avoid over certain defined operating ranges any water vapor sensor calibration.

210 From a user's point of view, precision and response time of an airborne hygrometer appears equivalent to  
211 accuracy if he is interested in fine structure resolving data. Precision and response time are under certain  
212 circumstances reciprocally correlated to each other (Allan, 1966). Typical figures for response time of air-  
213 borne hygrometer in the literature are 0.5 – 1 Hz (Petersen et al., 2010; Szakáll et al., 2004; Zöger et al.,  
214 1999b); some instruments deliver faster data 4 Hz (Weinstock et al., 1994) or special instruments up to 25 Hz  
215 (Zondlo et al., 2010). Typical precisions at 1 Hz are in 0.1-0.2 ppmv range (Sargent et al., 2013; Zöger et al.,  
216 1999b; Zondlo et al., 2010). While in the stratosphere (<10 ppm), the precision certainly can become a limit-  
217 ing factor. This is much less the case inside clouds or within the troposphere, where frequent, very strong  
218 spatial variations (up to 1000 ppmv per 100 m flightpath in the gas-phase, or up to 20000 ppmv per 100 m  
219 during total H<sub>2</sub>O phase) pose a larger problem. An instrument with a time response of just a few Hz causes  
220 significant under-sampling which can lead to strong aliasing effects at the high velocities (approx. 700-  
221 900 km/h) of many research aircrafts. Important under such conditions is the instrument's linearity to accu-  
222 rately cover the entire H<sub>2</sub>O concentration range up to five magnitudes for total water vapor measurements.

223 Numerous additional requirements have to be fulfilled by an airborne instrument to ensure a successful  
224 operation. Due to the high operation costs for aircrafts and the high scientific demand, H<sub>2</sub>O data have to be  
225 measured continuously without any interruptions. The instrument thus has to be highly reliable, robust,  
226 and require low maintenance. The restrictions in weight and space as associated with operation on an air-  
227 craft result in the necessity for the device to be of a compact and lightweight construction. The utterly com-  
228 plex and mandatory certification process (at least in Germany) quickly enforces an instrument design freeze  
229 before a campaign; this results in very stiff constraints for improvements/repairs during a campaign.

230



### 231 **3. Design decisions and approach**

232 HAI reflects on these by providing four independent but coupled spectrometers in one single housing and  
233 by simultaneously combining open- with closed-path measurements in one single instrument. For all  
234 channels the evaluation is done with one common spectroscopic method: calibration-free direct Tunable  
235 Diode Laser Absorption Spectroscopy (cal-free dTDLAS).  
236

#### 237 **3.1. Direct Tunable diode laser absorption spectroscopy (dTDLAS)**

238 The requirements for fast measurements and high chemical selectivity in combination with a robust and  
239 small system calls to choose a contact-less, spectroscopic (hence optical) measurement technique rather than  
240 contact-mediated sensing methods such as dew point mirror hygrometer (DPH) or capacitive polymer  
241 sensors e.g. “Humicap” (Salasmaa and Kostamo, 1986; Smit et al., 2000). These are quite often used in air-  
242 borne hygrometry and especially in meteorological environments (Anon, 2010; Busen and Buck, 1995;  
243 Hansford et al., 2006; Wiederhold, 1997). Optical hygrometers can be set up to become quite immune to  
244 hydrophilic/hydrophobic substances (unavoidable in the vicinity of aircraft) as well as particles (dust, soot,  
245 ice, etc.) contained by the gas to be analyzed. These capabilities were e.g. extensively demonstrated via  
246 measurements inside of combustion processes in industrial power plants (Ebert et al., 2000a; Schlosser et al.,  
247 2002; Sun et al., 2013; Teichert et al., 2003). Even response times in the several 10 kHz range could be  
248 demonstrated recently by (Witzel et al., 2013) for measurements in combustion engines.

249 Tunable diode laser absorption spectroscopy (TDLAS) is a powerful as well as versatile diagnostic tech-  
250 nique which is frequently employed in the near infrared spectral range and led to numerous applications in  
251 atmospheric hygrometry (Diskin et al., 2002; Fahey and Gao, 2009; Gurlit et al., 2005; May, 1998; Schiff et al.,  
252 1994; Thornberry et al., 2014). Advantageous diode lasers properties are the very high spectral resolution  
253 and power density, the continuous tuneability in combination with interesting technical features such as  
254 low cost, very low size/weight/power consumption, long life time, excellent beam quality, and optical fiber  
255 coupling to name just a few.

256 The typical setup and working principle of a TDLAS instrument has been frequently described in detail  
257 (Lackner, 2011; Schiff et al., 1994; Schulz et al., 2007; Werle, 1998). Therefore, only the HAI design relevant  
258 topics are discussed here. Important for an understanding of the novel HAI instrument is the classification  
259 of TDLAS instruments by their optical detection schemes in classical single- (Ebert et al., 2000b) or multi-  
260 path (Gurlit et al., 2005; Hunsmann et al., 2008; Lübken et al., 1999; May, 1998; McManus et al., 1995) beam  
261 setups- Further major categorizing distinguishes between the wavelength-modulation schemes like single  
262 modulation frequency = direct TDLAS or dTDLAS (Ebert and Wolfrum, 1994) or double modulation  
263 schemes like wavelength modulation spectroscopy WMS (Podolske and Loewenstein, 1993; Silver, 1992;  
264 Silver and Hovde, 1994a; Silver and Zondlo, 2006; Vance et al., 2011; Webster et al., 2004). WMS which is  
265 often used for very compact airborne sensors, provides on the first glance higher sensitivities by using lock-  
266 in technologies to efficiently filter noise. This, however, sacrifices the possibility of direct physics-based  
267 quality and reliability checks, since the actual measured WMS raw signal contains less spectral information



268 than a dTDLAS raw signal. This aggravates or sometimes even prevents detailed signal analysis based on  
269 fundamental physical explanations. Using dTDLAS instead with a special, but less common, first principles  
270 evaluation procedure (Ebert and Wolfrum, 2000; Farooq et al., 2008; Mihalcea et al., 1997; Schulz et al., 2007)  
271 yields a sophisticated evaluation, characterization, and quality management and a holistic view on the  
272 physical principles behind the data. This circumstance can even be used to avoid typical calibration proce-  
273 dures with reference gas standards.

274

#### 275 *Explanation of the term “calibration-free”*

276 The term “calibration-free” is often used in different communities with dissimilar meanings. To distinguish  
277 one should consider how calibration is defined by metrology (JCGM 2008, 2008): “calibration (...) in a first  
278 step, establishes a relation between the measured values of a quantity with measurement uncertainties pro-  
279 vided by measurement standards (...), in a second step, uses this information to establish a relation for ob-  
280 taining a measurement result from an indication (of the device to be calibrated)”. In other words, an in-  
281 strument with a deterministic relation between indication and quantity can be calibrated, without knowing  
282 the physics behind it. This particularly allows for compensating of non-linearity, offsets, drifts or response  
283 changes over time as long as they are stable, predictable, or can be extrapolated. We use the term “calibra-  
284 tion-free” to emphasize that HAI does not rely on such a correction process. The hygrometer described in  
285 this paper has never been calibrated or adjusted to a water vapor primary standard. Of course, parameters  
286 like gas pressure and temperature that are used for the calculation of the water vapor content via a first  
287 principle model are measured with calibrated sensors. This is done from a practical point because a) prima-  
288 ry standards for temperature and pressure are by themselves large facilities and b) the influence in the final  
289 uncertainty budget doesn’t justify that approach. Calibration-free doesn’t mean that the whole setup only  
290 uses primary principles. The whole idea behind traceability (JCGM 2008, 2008) is to use other sizes, higher  
291 in the hierarchy of the SI units of course, to generate/analyze the target value. Metrologists name that an  
292 unbroken chain of measurements. To visualize, this means that the first water vapor value delivered by the  
293 HAI instrument is the final value. Everything has to be characterized in advance on such a level that the  
294 first measurement value is determined within its uncertainty limits. Therefore, since there was no calibra-  
295 tion, it’s termed *calibration-free*.

296 Obviously, the calibration-free approach can always be enhanced with a calibration (such as (Muecke et al.,  
297 1994)) at any time (Buchholz et al., 2013b), even after a campaign if this seems advantageous, since the re-  
298 quirements for a calibrated instrument are lower than for a calibration-free instrument.

### 299 **3.2. Non-calibrated direct TDLAS (dTDLAS)**

300 The principle of non-calibrated absolute dTDLAS is very briefly presented in the following section. More  
301 detailed information regarding TDLAS is referred to in the above-mentioned literature. The sketch in Fig-  
302 ure 1 shows the schematics of a dTDLAS spectrometer with two independent channels. For low light inten-  
303 sities  $I_0(\lambda)$  in the mW-range, the transmitted light  $I(\lambda)$  can be described by the extended Lambert-Beer equa-



304 tion (Equation 1) including possible disturbances or the absorption measurement by background radiation  
 305  $E(t)$  or broadband transmission losses  $Tr(t)$ .

306 Equation 1:  $I(\lambda) = E(t) + I_0(\lambda) \cdot Tr(t) \cdot \exp[-S(T) \cdot g(\lambda - \lambda_0) \cdot N \cdot L]$

307 By applying the ideal gas law Equation 1 can be used to retrieve the  $H_2O$  volume mixing ratio  $c$ .

308 Equation 2: 
$$c = -\frac{k_B \cdot T}{S(T) \cdot L \cdot p} \int \ln \left( \frac{I(v) - E(t)}{I_0(v) \cdot Tr(t)} \right) \frac{dv}{dt} dt$$

309 The *amount fraction*  $c$  is in metrological units officially specified as [mol/mol = mol absorber per mol gas]  
 310 which is in the environmental community better known as “volume fraction” e.g. in units of ppmv or  
 311 Vol.-%. The term  $\frac{dv}{dt}$  is called the *dynamic tuning coefficient* of the used laser. It can be determined experimen-  
 312 tally and is directly linked to the SI units (length) by using the Airy signal of the laser light passing through  
 313 a planar, air-spaced etalon (Ebert and Wolfrum, 2000; Schlosser et al., 2002). Yet, unpublished, on-going,  
 314 and long-term  $\frac{dv}{dt}$  measurements for the used 1.37  $\mu\text{m}$  DFB laser type over several years indicate a long-term  
 315 stability of its tuning characteristics better than 1%, which is within the current uncertainties of the tuning  
 316 characterisation.  $k_B$  is the *Boltzmann constant* and  $L$  is the *optical path length*.  $S(T)$  is the *line strength* of the  
 317 selected molecular transition (see chapter below) and therefore a physical property of the molecule to be  
 318 measured. The gas *pressure* ( $p$ ) and gas *temperature* ( $T$ ) can be accurately acquired in a closed-path cell; the  
 319 quality of the respective measurements in the open-path cell are discussed in the following chapter describ-  
 320 ing the construction of the open-path sensor. Equation 2 also “explains” the term calibration-free quantita-  
 321 tively as there are no other “hidden” parameters used to derive the water vapour concentration which re-  
 322 quire a calibration.

### 323 3.3. Absorption line selection

324 Suitable absorption lines have to be selected for a specific application by several criteria (Wunderle et al.,  
 325 2006) (Wagner et al., 2012). Besides a line strength maximization to ensure high sensitivity, other important  
 326 parameters have to be taken into account. For atmospheric measurements, the cross sensitivity to other  
 327 gases such as  $CO_2$  needs to be minimized. This ensures a better control of the fitting process due to the few-  
 328 er degrees of freedom. Similarly, the line should be isolated from other lines to simplify the retrieval of the  
 329 baseline function. For the open-path measurements, it is highly important to minimize temperature de-  
 330 pendence of the line strength in order to minimize the influence of gas temperature uncertainties. Lastly,  
 331 sometimes the primary constraint is the availability of suitable laser diodes and additional accessories such  
 332 as fibers and optic components. As the certification for airborne instruments nearly prevents improve-  
 333 ments/repairs during a campaign, all components need to be very reliable. For HAL, we selected two specifi-  
 334 c water lines at 2596 nm and 1370 nm in two wavelength ranges, which also had been used before  
 335 (Buchholz et al., 2012, 2013b; Ebert, 2006; Ebert et al., 2004; Fahey and Gao, 2009; Hovde et al., 2001;  
 336 Hunsmann et al., 2008; May, 1998; Seidel et al., 2012; Witzel et al., 2012; Wunderle et al., 2008) and for which  
 337 improved spectral parameters were generated (Hunsmann et al., 2006). The 2.6  $\mu\text{m}$  laser is not fiber-  
 338 coupled, doesn’t have an optical isolator, is less stable in terms of temperature fluctuations and has a lower  
 Page 9



339 beam quality, but accesses a factor of 20 stronger line and thus ultimately promises 20 x higher sensitivity.  
340 Both lines are shown (Figure 2) as a simulation based on the HITRAN (Rothman et al., 2009) database. As-  
341 suming an optical resolution of  $5.10^{-4}$  OD, the hygrometer is expected to provide for a  $1.4 \mu\text{m}$  closed-path  
342 cell (assuming a 1.5 m path length) under atmospheric conditions a water vapor concentration range from  
343 10 to 40 000 ppmv and for the  $2.6 \mu\text{m}$  closed-path channel from 0.5 to 2000 ppmv. The lower limit will final-  
344 ly be defined by the capability of minimizing and compensating the effects of parasitic  $\text{H}_2\text{O}$  offsets  
345 (Buchholz and Ebert, 2014b) and their related uncertainties. For the  $2.6 \mu\text{m}$  channel, the upper limit de-  
346 pends strongly on the retrieval quality of the baseline. This means e.g. under lower pressure conditions  
347 ( $<500$  hPa) evaluations up to 40 000 ppmv can be performed but for high pressures the absorption line at  
348 these high concentrations is saturated and so broadened that the baseline cannot be retrieved with a low  
349 uncertainty. The upper limit in general, i.e. also for the  $1.4 \mu\text{m}$  channels, is defined by the dew point related  
350 to the instrument temperature; 40 000 ppmv is equal to 100 %RH at  $28.5^\circ\text{C}$  and 1013 hPa.

#### 351 **4. HAI configuration**

352 HAI can be installed on an aircraft in several different ways and configurations. This paper focuses on the  
353 installation during the first HALO (Krautstrunk and Giez, 2012) campaigns ( TACTS (Engel et al., 2013) and  
354 ESMVal (Schlager, 2014)) which targeted multiphase  $\text{H}_2\text{O}$  measurements. Figure 3 (right) shows the fuse-  
355 lage of HALO with the two mountings belonging to HAI. In the very front is a trace gas inlet (TGI), where  
356 HAI uses the second opening from the top to sample air in flight. The sampling height is approx. 25 cm  
357 above the aircraft skin. Next to the TGI is the HAI open-path cell (OPc), which is a fiber-coupled, White-  
358 type three-mirror configuration mounted on two pylons. The mirrors fold the laser light beam shown in the  
359 middle (yellow). Both the TGI and the OPc are installed on HALO specific aperture plates. Figure 3 (left)  
360 depicts the cabin installations with the golden instrument in the middle being the rack of the HAI main  
361 unit. The black insulated  $\frac{1}{2}$ " pipes on the right side of the rack are the inlet lines providing the air sampled  
362 by the TGI to the main unit. These pipes are heated ( $70^\circ\text{C}$ ) to evaporate ice particles and droplets in the gas  
363 stream at a flow of approximately 100 volume liter/min. The open-path sensor is not visible in this photo  
364 since it is on the left side of the rack.

##### 365 **4.1. HAI main unit**

366 The HAI main unit is comprised of individual modules except for some electronics of the OPc which are  
367 directly mounted cabin sidewise on the OPc: preamplifier for detectors, five temperature measurement  
368 converters, mirror heating controller, and two pressure sensors. Many of the modules contain innovative  
369 developments, which are or will be published individually to prevent an overload of technical details in  
370 this paper. The important functions/modules to understand the HAI concept are as follows:

- 371 a. Automatic line identification and spectral locking to compensate temperature and electrical influences as  
372 well as drifts
- 373 b. Internal mechanism (Buchholz and Ebert, 2014b) to minimize and compensate effects from parasitic  
374 absorption which usually leads to variable offset problems in spectrometers like HAI



- 375 c. Gas handling system which tempers the air to be analyzed to the temperature of the instrument and  
376 therefore minimizes the risk of temperature inhomogeneity even at high gas flows as well as to provide  
377 detailed information about temperature, pressure, and gas flow in and through the system
- 378 d. Several individual sensors and electronics to collect in total more than 120 system parameters about the  
379 HAI system's health
- 380 e. Sequence control system for steering, controlling, and exception treatment of HAI with all its submod-  
381 ules

#### 382 4.2. HAI's closed-path cells (CPC)

383 Core parts of the HAI main unit are the extractive optical closed-path cells (Figure 4) installed in the indi-  
384 vidual 1.4  $\mu\text{m}$  and 2.6  $\mu\text{m}$  optics modules within the HAI main unit. Both optics modules are completely  
385 independent, and both cells use a new, highly compact miniature White type (White, 1976) design that folds  
386 the laser beam (total approx. 1.5m) with three mirrors (approx. 7.5 cm distance). The 1.4  $\mu\text{m}$  cell is an im-  
387 proved fiber-coupled version of the White cell in (Kühnreich et al., 2013). Combined with a new fiber fed  
388 through (Buchholz and Ebert, 2014a), it reaches a very low leakage rate of  $1.9 \cdot 10^{-6}$  hPaL/s and prevents par-  
389 asitic offsets due to ambient air. Three mirrors, one fiber, and one detector are the only optical parts in the  
390 cell. This reduced amount of optics minimizes fringing effects efficiently down to a long-term  $10^{-4}$  level (no  
391 readjustment needed for approx. 400 flight hours within 4 years). The temperature is measured via a 3 mm<sup>2</sup>  
392 accurate (0.3 K) PT100 sensor for slow temperature changes ( $t_{0.5}$  approx. 2.5sec) and also with a thermocou-  
393 ple (type T, 0.5 mm diameter) for faster temperature changes ( $t_{0.5}$  approx. 0.5sec). Due to the above men-  
394 tioned gas handling system and the thermal mass of the whole instrument (approx. 31 kg) the temperature  
395 fluctuations are below 0.1 K during aircraft operation. Each cell with a volume of approx. 300 ccm is typical-  
396 ly flushed during flight with a gas flow of 20 – 50 vol.liter/min per cell (depending of the installing scenario)  
397 leading to a 0.7 sec flushing time using bulk flow estimation. The 2.6  $\mu\text{m}$  closed-path cell is by its design  
398 quite similar to the 1.4  $\mu\text{m}$ . The most significant difference is the free-space beam guidance into the cell,  
399 since fibers at 2.6  $\mu\text{m}$  are quite critical to handle and suitable beam splitter etc. weren't available. Therefore,  
400 the 2.6  $\mu\text{m}$  laser, the reference cell for spectral stabilization, the beam splitters, the mirrors, and the fiber  
401 coupling section for the open-path cell are enclosed in a specially designed gas tight 2.6 $\mu\text{m}$  optics module  
402 box. This module is internally dried by a purging cycle containing a H<sub>2</sub>O purifier based on the same idea as  
403 for removing parasitic offsets in fiber coupled hygrometer (Buchholz and Ebert, 2014b). The 2.6  $\mu\text{m}$  closed-  
404 path cell is directly attached to this laser module. Before reaching the cell, the laser beam is therefore "free-  
405 space" guided in a water vapor free environment (typ. <0.5 ppmv) which efficiently avoids parasitic effects  
406 in the beam path.

#### 407 4.3. HAI's open-path cell (OPC)

408 HAI's open-path cell (Buchholz et al., 2013a) is – just like the closed-path cells – based on the White (White,  
409 1976) principle and installed in the fuselage of HALO. Contrary to the closed-path cells, both wavelengths  
410 (2.6  $\mu\text{m}$  and 1.4  $\mu\text{m}$ ) are simultaneously coupled into the same 3 mirror cell via two independent glass fi-



411 bers. The cell (made of 2" diameter mirrors) has a mirror base distance of approximately 15 cm and is set to  
412 an optical path length of 4.2 m. The optical measurement volume is located approximately 23 cm above the  
413 aircraft skin. The air temperature within the open-path cell is measured via two surface-mounted platinum  
414 PT100 sensors. These temperature sensors provide together with the HALO's temperature measurement  
415 (Giez, 2012) and our CFD simulation (not shown here) an estimated ( $\pm 5$  K) temperature in the actual meas-  
416 urement volume of the open-path cell. Improved temperature field simulations will be realized in the future  
417 as soon as a full CFD model of the aircraft and HAI is accessible. To prevent a dew or frost buildup on the  
418 mirror surface, all three mirrors are individually heated (using approx. 20W electrical power each). This  
419 raises the mirror temperature, defined by core temperature measurements of the copper mirrors, typically  
420 15 – 20 K above the ambient temperature. The gas pressure measurements are done by two commercial  
421 piezo pressure transmitters. Their individual ports are located on either side of the open-path cell. An in-  
422 flight pressure sensor validation has been realized recently via the first optical airborne pressure measure-  
423 ments. This calibration-free approach exploits the pressure dependence of the collisional broadening of the  
424 water absorption line (Buchholz et al., 2014b).

#### 425 **4.4. TDLAS based multi-phase H<sub>2</sub>O measurements**

426 By simultaneously combining closed-path and open-path water measurements, HAI is the first TDLAS  
427 hygrometer to allow calibration-free multi-phase water detection. The schematic of the HAI configuration is  
428 shown in Figure 4. Both independent closed-path cells are connected via heated inlet lines (visible in Figure  
429 3 left) to the forward facing TGI. The TGI thus samples in clouds both the *gas-phase water vapor* in conjunc-  
430 tion with the *ice particles* or *H<sub>2</sub>O droplets*. Droplets and ice particles are evaporated in the heated inlet lines,  
431 so that both closed-path cells analyze the so called *total water* content, i.e. the sum of the interstitial gas  
432 phase water plus the evaporated condensed phase water. This evaporation step is necessary since dTDLAS  
433 is typically evaluated in a way to suppress the broad band spectra of condensed phase water and only tar-  
434 gets the narrow spectral structure of water vapor. The combination of the narrow absorptions line width  
435 with a narrow tuning range of diode lasers allows this discrimination of gas-path water vapor from the  
436 broad spectra of ice or liquid water. The open-path cell, which is directly located on the fuselage of HALO,  
437 however, measures the pure interstitial gas-phase water vapor of the air which flows through the cell. By  
438 combining the total water and the gas phase water measurement, i.e. closed-path and open path signals, it  
439 is possible to derive from the difference of both HAI signals the pure condensed water phase, i.e. the  
440 ice/droplet phase. To our knowledge, HAI is the first airborne instrument that can measure all water phases  
441 with the same detection principle, the same evaluation strategy, and with two channels, each spectroscopi-  
442 cally and temporally linked by one single laser. When independent devices have been used previously to  
443 measure these water products, even with similar TDLAS instruments like in (Vance et al., 2015), this always  
444 led to the discussion on how much of the deviations are caused by atmospheric effects and how many are  
445 linked to discrepancies between the participating instruments caused by different characteristics, response  
446 times or nonlinearities of the individual instruments. Indeed, the calibration or even validation of all other  
447 open-path hygrometers cited above is based on laboratory measurement or comparisons with other hy-



448 grometers with different properties which makes it nearly impossible to justify more than statistical  
449 (=averaged) statements. HAI avoids these discussions by various self-validations.

#### 450 **4.5. Self-validation**

451 The new self-validating capabilities are a consequence of the unique 2x2 multi-channel configuration, which  
452 enables six individual validation pathways, as illustrated in Figure 5. Under clear-sky (= cloud-free) condi-  
453 tions, the independent open-path and closed-path measurements should each deliver the same value. Im-  
454 portant to note is that both the 1.4  $\mu\text{m}$  and the 2.6  $\mu\text{m}$  spectrometer channels are evaluated without com-  
455 mon parameters in calibration-free first principles mode. This means in detail: two different lasers, two  
456 different absorption paths, two different absorption lines, different line parameters, different electronics etc.  
457 Nevertheless, both spectrometer channels have to agree within their uncertainty if the first principle evalua-  
458 tion is applicable. Due to the large overlapping H<sub>2</sub>O concentration range of approximately 50 to 5000 ppmv,  
459 all disturbances can be clearly identified as long as they affect both spectrometers differently. Outside of  
460 clouds (clear-sky) – i.e. in ice particle/droplet-free air – all four channels spectrometer channels have to  
461 agree. This in particular allows to validate (or even if necessary to calibrate) the closed-path cells in a labor-  
462 atory environment and to use them as an airborne transfer standard to validate or even calibrate in turn the  
463 open-path sensor in flight. This is a unique property of HAI, since each open- and closed-path spectrometer  
464 are coupled by one single laser and all four spectrometers are employing the same evaluation procedure.  
465 Hence, HAI is called a 2x2 spectrometer.

## 466 **5. Results**

### 467 **5.1. Signal assessment**

468 Figure 6 shows - at the same instant of time - all seven raw optical signals from HAI's measurement chan-  
469 nels at approx. 15000 ppmv and 900 hPa. For visualization, the signals of each laser are scaled to the same  
470 peak maximum, since transmission changes are irrelevant according to the equation for concentration re-  
471 trieval (Equation 2). The time axis (240 Hz laser repetition rate) is already converted to the wavelength axis  
472 using the dynamic tuning rate. The triangular diode laser current modulation yields to a triangular modula-  
473 tion of the beam intensity and a similar shape in the wavelength. The tuning range of the 2.6  $\mu\text{m}$  laser is set  
474 to be broader in order to allow the evaluation of high absorbance spectra when the peak absorption is in  
475 saturation in the absorption line centre. In this case, the Lorenz wings carry the concentration information.  
476 The absorbance of the 1.4  $\mu\text{m}$  open-path signal (purple) is higher than the one from the closed-path cell, due  
477 to the threefold longer optical path length in the open-path cell. The 1.4  $\mu\text{m}$  reference cell signal (blue) is  
478 used for spectral stabilization and acquired from a low pressure (approx. 20 hPa) fibre coupled reference  
479 cell. The signal for retrieving and compensating parasitic absorption is shown in green and the associated  
480 evaluation is described in (Buchholz and Ebert, 2014b). Accordingly, the three 2.6 $\mu\text{m}$  signals are shown. The  
481 parasitic absorption compensation of the 2.6  $\mu\text{m}$  system is due to the "free-space" signal in the sealed laser



482 module (see description above) at least as important as the one for the 1.4  $\mu\text{m}$  spectrometer. This compensa-  
483 tion treatment is done using an ultra-low pressure (approx. 0.5 hPa) reference cell which provides infor-  
484 mation for both, the spectral stabilization and the parasitic absorption correction. The latter approach is  
485 described as “two pressure separation method” in (Buchholz and Ebert, 2014b).

486 Figure 7 and Figure 8 show four typical absorption signals in flight with a pre-average of 50 raw measure-  
487 ments done at 240 Hz. This leads to 4.8 measurements per second with a time resolution of 70 ms for each  
488 reading while only approx. 1/6 of the whole spectral scan is necessary for the evaluation. Figure 7 allows the  
489 simultaneous comparison of both closed-path cells. Gas pressure and temperature are slightly different due  
490 to the different piping and installation placement inside of HAI. Both  $\text{H}_2\text{O}$  concentrations agree ( $\Delta = 1.6\%$ )  
491 within the combined uncertainties (7.9%) of both channels. Each graph in Figure 7 shows in the top the  
492 measured data (green dots) with the fitted Voigt profile (black line). The measurement data are shown as  
493 absorbance ( $\text{ODe} = -\ln(I/I_0)$ ) values, i.e. the detector ( $I$ ) signal divided by the so called “base line” ( $I_0$ ). The  
494 baseline resembles the absorption spectrum in the absence of the molecular target absorption. It is retrieved  
495 from the raw spectrum ( $I$ ) by applying a synchronous polynomial fit of the baseline and a Voigt profile for  
496 the absorption line. The residuals between measured data and the model function are depicted below. The  
497 remaining structures visible in the residual are optical interference fringes, caused by light scattered from  
498 imperfect mirrors, the fiber surface, or the detector arrangement. The repetition frequency of a fringe can be  
499 linked to the optical distances between the interfering surfaces. So far, these baseline structures as well as  
500 the cell alignment remained stable over the past years. The mirrors never required cleaning during the last  
501 four years. Of course, the entire fringe structure can be phase shifted by temperature. However, compared  
502 to WMS systems, where fringe levels have to be reduced down to the  $10^{-6}$  OD level to achieve high sensitivi-  
503 ties, our calibration-free dTDLAS evaluation is designed to require a stable baseline on the  $10^{-4}$  OD level,  
504 which makes it inherently robust. In other words, dTDLAS systems like HAI are usually not as sensitive to  
505 misalignments, dirt, and optical structures as WMS systems. On the other hand, they require higher absorb-  
506 ance for the calibration-free approach.

507 Figure 8 shows a simultaneous comparison between the open-path and the closed-path channel. The out-  
508 side gas temperature in the open-path cell is approx. 52 K lower than in the closed-path cell. The gas pres-  
509 sure in the closed-path cell, on the other hand, is much higher than in the open-path cell, due to the ramp  
510 pressure guiding the air through the TGI and the piping. The measured water vapour concentrations are  
511 slightly shifted in time due to the gas transport and the sampling delay in the pipes. The  $1\sigma$  residual is a  
512 factor of six higher in the open-path than in the closed-path sensor. This mainly results from the high wind  
513 speed through the open-path cell (approx. 800 km/h). It is important to mention that for diode laser instru-  
514 ments a change of gas temperature, pressure, and concentration over a large range can commonly cause  
515 large systematic offsets, which frequently would require at a least three dimensional calibration surface  
516 (depending on pressure, temperature, concentration) or to the integration of complex assumptions (Duffin  
517 et al., 2007; Goldenstein et al., 2014; Rieker et al., 2009) to correct the data via a simulation model. WMS  
518 closed-path systems usually avoid this problem by always keeping temperature and pressure as close and  
519 constant as possible to the calibrated level which is obviously not a viable approach for an open-path sys-  
520 tem.



## 521        5.2. Precision

522        One common figure of merit to quantify the short term optical precision of a spectrometer is the  $1\sigma$  noise  
523        level in the residuum. The idea is based on the assumption that the residual is governed by random noise  
524        which limits the precision of the system. The Signal-to-Noise Ratio (SNR) is then defined by the *signal* as the  
525         $OD_{\text{peak}}$  value and the *noise* by the statistical standard deviation of the residual and thus equals to  $OD_{\text{noise}}$ .  
526        This definition yields for the 1.4  $\mu\text{m}$ /2.6 $\mu\text{m}$  closed-path cells at 4.8 Hz time resolution a SNR of 54 respec-  
527        tively to an SNR of 204. The corresponding precision is then  $100/54 \text{ ppmv} = 1.9 \text{ ppmv}$  (1.4  $\mu\text{m}$ ) respectively  
528         $100/204 \text{ ppmv} = 0.49 \text{ ppmv}$  (2.6  $\mu\text{m}$ ). This determination is quite consistent with Figure 8 where the preci-  
529        sion of the 1.4  $\mu\text{m}$  closed-path can be estimated to 1.3 ppmv in the same way. However, as the fringe struc-  
530        ture of the closed-path cell is quite stable it is better to determine the instrument precision via the Allan  
531        variance (Allan, 1966). Figure 9 shows the Allan plot for both closed-path cells. The measurement was done  
532        at 255 ppmv by measuring gas from a big vessel. Under these conditions, the precision at 4.8 Hz is  
533        0.22 ppmv (for the 1.4  $\mu\text{m}$  closed-path cell) respectively 0.065 ppmv (2.6  $\mu\text{m}$  closed-path cell). Both values  
534        are approximately a factor of 8 lower than the one derived from the single scan analysis, which confirms the  
535        stability thesis of fringes. Vice versa we can derive an actual (= unstable, random)  $1\sigma$  residual OD noise  
536        level of  $2.3\text{E-}5$  for the 1.4  $\mu\text{m}$  closed-path and  $1.7\text{E-}4$  ppmv for the 2.6  $\mu\text{m}$  closed-path cell. The best preci-  
537        sion is achieved at 3.9 Hz (0.18 ppmv) for the 1.4  $\mu\text{m}$  cell and at 13 Hz (0.055 ppmv) for the 2.6  $\mu\text{m}$  cell,  
538        normalizing this with respect to time resolution and optical path length yields a  $1\sigma$  precision of 187  
539         $\text{ppbv}\cdot\text{m}\cdot\text{Hz}^{-1/2}$  (1.4  $\mu\text{m}$ ) and 31  $\text{ppbv}\cdot\text{m}\cdot\text{Hz}^{-1/2}$  (2.6  $\mu\text{m}$ ).

540        Comparing that precision ratio of 6 between the 1.4  $\mu\text{m}$  and the 2.6  $\mu\text{m}$  to the ratio in line strength of 20 (see  
541        line selection chapter 3.3) confirms the statement mentioned above that drawbacks of the longer wave-  
542        length technology significantly constrain the practically achievable maximum gain.

543        For the open-path sensor, we have not yet been able to calculate an Allan variance so far, since we haven't  
544        acquired a data set in flight where atmospheric water was constant enough not to dominate the Allan vari-  
545        ance via natural H<sub>2</sub>O fluctuations. However, using the short term single spectrum evaluation method in  
546        Figure 8, one can calculate a SNR of 145, which yields a precision of 2.1 ppmv (=  $4.6 \text{ ppmv}\cdot\text{m}\cdot\text{Hz}^{-1/2}$ ).

## 547        5.3. Uncertainty consideration

548        One of the major benefits of the dTDLAS evaluation done with HAI is the high level of control over the  
549        fitting/evaluation process. In particular, when it is combined with the complete storage of all the raw data,  
550        this allows for a dedicated and highly flexible post-flight analysis or post-flight processing. Raw data stor-  
551        age for HAI not only covers all raw spectra but also approximately 120 measured parameters depicting the  
552        complete status of the HAI instrument. This "housekeeping data" for example, encompasses 15 tempera-  
553        ture measurements that allow for diagnostic statements e.g. about temperature inhomogeneity, electronic  
554        drift, or sensor malfunctions. Housekeeping data like these facilitate assignments of a measurement uncer-  
555        tainty closer to the one in a metrological sense (Joint Committee for Guides in Metrology (JCGM), 2008). If  
556        measurements are done in harsh environments or under rapidly changing conditions, typical concepts of  
557        metrological uncertainty assessments fall short since most of their concepts are developed for laboratory



558 applications under quasi-static conditions. But even in this case the housekeeping parameters provide in-  
559 formation for a trustful uncertainty consideration. The uncertainty calculations below are based on a physi-  
560 cal model (Equation 2) and are performed with an approach independent from the actual measurement.  
561 Depending on the individual science community uncertainties, errors, reproducibility, or misreading are  
562 often determined by comparing the readings of an instrument to reference instrument. This is contrary to a  
563 metrological uncertainty and furthermore a non-independent approach; we call that therefore simply *devia-*  
564 *tion, standard deviation* etc.

565 According to the Equation 2 for retrieving the final concentration the individual contributions to the total  
566 uncertainty budget can be assessed as followed for both closed-path cells. The optical path length was pri-  
567 marily determined with a Zemax based ray tracing simulation. In addition, it was calculated by the me-  
568 chanical mirror distance and compared to a test measurement with a known amount of CH<sub>4</sub> gas in the cell.  
569 These measurements lead to an uncertainty of the optical-path length of 15 mm (1.1%). The uncertainty for  
570 the used H<sub>2</sub>O absorption line strength is 3.5% [44]. The temperature sensors (PT100 and thermocouple type  
571 E as described above) are calibrated against metrological transfer standards (Rosemount platinum re-  
572 sistance thermometer type 162CE (PRT-25), uncertainty 1.5 mK). Due to the well-known fact that tempera-  
573 ture measurements in moving gases suffer from many issues, we use a conservative estimation of 1K (0.3%).  
574 The manufacture of the pressure sensors (Omega PAA33X) advertises a resolution of 0.02 hPa and a long-  
575 term accuracy of 0.5 hPa. We use 1 hPa as a conservative uncertainty estimation. A general statement for the  
576 laser tuning uncertainty is difficult in a metrological sense, since it depends on local effects in the spectra,  
577 pressure range, concentration level and number of absorption lines fitted. From our experience, we assume  
578 that deviations related to the fitting process in total (tuning + fit process) are in the range of below 1.5%. As  
579 a conservative approximation we use 2% for the fit and 1% for the tuning. This yields to a total uncertainty  
580 of 4.3%. for the 1.4 μm closed-path cell.

581 The largest relative influence to this total uncertainty budget results from the line strength (66%), followed  
582 by the fitting uncertainty (21%), as well as the optical path length (7%). Additionally, the offset uncertainty  
583 is defined in a calibration-free dTDLAS system by the capability of minimizing and determining parasitic  
584 effects. It is ±3 ppmv for the 1.4 μm closed-path cell of HAI, which uses the same parasitic prevention  
585 treatment as SEALDH-II (Buchholz and Ebert, 2014b). Thus, the total uncertainty of the 1.4 μm closed-path  
586 system is 4.3% ±3 ppmv. Similar considerations lead to an uncertainty of 5.9% ±0.4 ppmv for the 2.6 μm  
587 closed-path cell.

588 Related to the open-path cell, the statement is more difficult. First of all, a full CFD model of the boundary  
589 layer around HALO is currently missing, which would allow a retrieval of the gas temperature in the open-  
590 path cell. The first CFD test runs of HAI's open-path sensor and its immediate surroundings have been  
591 realized (A. Afchine (FZJ), published in (Klostermann, 2011)), and led to a first temperature uncertainty  
592 estimate of ±7 K. The full CFD model could also provide pressure field calculations. However, in this case,  
593 we developed a method on how to retrieve the pressure directly from the dTDLAS signal (Buchholz et al.,  
594 2014b) and use that for a validation of the built-in micromechanical pressure sensor. The pressure depend-  
595 ent uncertainty is in the range of 3-5%. The uncertainty of the fitting process reveals a much higher com-  
596 plexity. It depends on all impacts on the spectral signal itself such as: background radiation on the detector,



597 misalignment due to the distortion of the HALO fuselage, forming of an ice layer on the mirrors, high opti-  
598 cal dense cloud transects, eddies around the aircraft etc. From all the data gathered and compared during  
599 the HALO missions TACTS and ESMVal, we couldn't find any deviation larger than  $\pm 5\%$  in clear sky condi-  
600 tions between the open-path and the closed-path sensor which couldn't be clearly linked to explainable  
601 effects. Besides of the availability of an entire CFD model in the future, HAI will be validated by sampling  
602 the pure water vapor gas-phase (backward measurement) with its closed-path cells in future campaigns.  
603 These datasets will sustainably prove if the  $\pm 5\%$  deviation is valid under all conditions, since then all four  
604 channels of HAI (Figure 5) would permanently measure the same atmospheric value. Compared to the de-  
605 viations, revealed in the AQUAVit (Fahey et al., 2014) intercomparison of state of the art, airborne hygrome-  
606 ter in 2007, this 5% is fourfold smaller than the span (20%) of those instruments there. Additionally, one has  
607 to keep in mind that this comparison was done under quasi-static laboratory conditions. Therefore, this in-  
608 flight 5% deviation between the open-path and the closed-path sensor shows the great performance of HAI,  
609 even if a general metrological uncertainty is difficult to be given yet at this time.

#### 610 **5.4. First airborne measurements with HAI on the HALO aircraft**

611 Figure 10 shows typical HAI measurements during the first HALO science mission "TACTS/ESMVal". The  
612 chosen flight profile shows 15 min of a slow ascent from the lower to the upper troposphere. Depicted in  
613 Figure 10 are: the  $\text{H}_2\text{O}$  concentration measured with the  $1.4\ \mu\text{m}$  CPc (black),  $1.4\ \mu\text{m}$  OPc (magenta), and  
614  $2.6\ \mu\text{m}$  CPc (red) channel of HAI (left axis) as well as the ambient gas pressure (right y-axis, in blue), which  
615 is directly connected to the flight level (height). Other instruments indicated that HALO flew in the early  
616 part of the flight (left) in clear sky conditions (no clouds). All Figure 10 data were plotted with 1 Hz tem-  
617 poral resolution. With increasing height, the  $\text{H}_2\text{O}$  concentration declines about 2.5 magnitudes from approx-  
618 imately 2000 ppmv to 75 ppmv, in combination with simultaneous relative pressure variation from approx.  
619 175 to 450 hPa ( $\Delta p=275$  hPa). The  $\text{H}_2\text{O}$  concentrations, measured with HAI's two independent closed-path  
620 spectrometer channels, fit nicely to each other and show only a small average relative deviation of 1.9%  
621 (bottom graph in red), which is entirely covered by the uncertainty range of each spectrometer (4.3% resp.  
622 5.9%). This result is consistent with other measurements, where the  $2.6\ \mu\text{m}$  path was persistently on aver-  
623 age approx. 2% higher than the  $1.4\ \mu\text{m}$  channel. This could be easily explained with the  $2.6\ \mu\text{m}$  line strength  
624 from the HITRAN database being 2% too small. The 1.9% deviation also needs to be related to the fact that  
625 both hygrometer channels are evaluated completely independently, without any calibration of the instru-  
626 ment and most importantly under real-world flight conditions. Despite these demanding conditions HAI's  
627 two closed-path channels generate a relative deviation which is just 1/10 of the above mentioned 20% in  
628 AquaVIT (Fahey et al., 2014), which demonstrates the excellent accuracy of the HAI spectrometers.  
629 The purple signal also depicted in Figure 10 shows the measurement of the  $1.4\ \mu\text{m}$  open-path sensor  
630 ( $1.4\ \mu\text{m}$  OPc). Compared to the  $1.4\ \mu\text{m}$  closed-path cell ( $1.4\ \mu\text{m}$  CPc), the  $1.4\ \mu\text{m}$  OPc shows a relative devia-  
631 tion of  $\pm 5\%$  peak-to-peak or 2% on average. This performance of the  $1.4\ \mu\text{m}$  OPc is to be compared with  
632 previously described airborne hygrometer which resemble the current state of the art. However, it has to be  
633 noted that a huge number of airborne hygrometers with an extreme diversity in sensing principles and



634 specific performances have been published and described, so that a general overall statement about the  
635 “state of the art” is very difficult to make. In addition, there is a very large variation in sensor accuracy and  
636 the performance figures stated are sometime misleading and contradictory to the results of the few rare  
637 large scale intercomparisons like AquaVIT. However, a short incomprehensive review on the performance  
638 of previous open-path instruments should be given here:

639 a) A calibrated, 1 m direct absorption, open-path sensor at  $1.4\mu\text{m}$  was developed in a feasibility study for  
640 the Strato 2C research aircraft and described to cover a concentration range of 2-2000 ppmv with 100 ppb  
641 precision at 1 Hz. It was claimed to have an accuracy of 5% , but this hygrometer never flew (Roths and  
642 Busen, 1996).

643 b) A calibrated open-path sensor with an optical path length of 375 cm, folded in a Herriot cell, and a time  
644 resolution of 25 Hz gave a precision of 80 ppbv. This instrument is designed for the American HAIPER  
645 (Anon, n.d.) aircraft, a Gulfstream 550 like the HALO aircraft. To our knowledge this is the only open-  
646 path hygrometer which was frequently installed and used during the last years on an aircraft. Typical  
647 deviations to other instruments were in the range between 2% and 10% (Zondlo et al., 2010).

648 c) Another calibrated, Herriot cell based open-path hygrometer with a total path length of 11 m, a meas-  
649 urement frequency up to 10 Hz and a precision of 50 ppbv at 0.5 Hz. This instrument claims to have an  
650 uncertainty of 5% to 10% (May, 1998). However, to the best of our knowledge, except of a few test  
651 flights, no further flight was nor published or performed.

652 In summary, HAI’s 2% deviation appears to be much better than typical statements in publications.

653 To further evaluate the performance of HAI’s open-path channels we further discuss the TACTS data. The  
654 dark purple line shows the data after smoothing them with a 60 second sliding window algorithm. Quite  
655 interesting are the two dents at 13:41 and 13:44 in the relative deviation, which means the closed-path  
656 measurement was higher than the open-path measurement. Both of them happened after the water vapour  
657 concentration dropped sharply. Hence, the dents might be explained by desorption processes inside of the  
658 TGI, its piping and the measurement cell. This idea is backed by the  $1.4\mu\text{m}$  to  $2.6\mu\text{m}$  closed-path compari-  
659 son (red). The same behaviour is also visible at 13:44, but here the  $2.6\mu\text{m}$  close path cell added less desorp-  
660 tion water to the gas steam than the  $1.4\mu\text{m}$  one. Another influence leading to that deviation could be a  
661 temperature measurement on HALO affected by fast changes of humidity. The total temperature influence  
662 in Equation 2 is at this temperature and pressure approximately 0.6%/K caused by the temperature influ-  
663 ence via the ideal gas law, the line strength temperature dependence and the Gaussian width. This brief  
664 insight in a detailed data discussion demonstrates the unique opportunities offered by HAI, which allow a  
665 deep analysis to identify and eventually classify different measurement sections of a flight in order to dis-  
666 tinguish instrumental artefacts from real atmospheric situations.

667 On the second part of the flight shown in figure Figure 10 , the aircraft passes through ice clouds (Cirrus,  
668 temperature  $< -40\text{ }^\circ\text{C}$ ). The evaporation of the ice crystals captured by the gas inlet leads to a clear en-  
669 hancement of the amount of water vapor in the closed-path cells. The relative comparison shows a  $\pm 5\%$   
670 noisy structure, which is caused by the low time resolution of just 1 Hz.  $\text{H}_2\text{O}$  gradients in this particular  
671 part of the flight are in the range of several 1000 ppmv/s. Due to SSD space limitations in the first campaign,  
672 only parts of the flights where sampled with the highest time resolution of 240 Hz, in the other parts the



673 data where online pre-averaged. Like in the clear sky region, the average deviation between both closed-  
674 path cells during cirrus cloud transects remains at 2%. To further check the accuracy of the open-path chan-  
675 nel, we use two methods: The first is based on the physical argument that the gas-phase measurement al-  
676 ways has to be lower than the total water measurement, which consists of gas-phase plus evaporated  
677 ice/liquid phase. This requirement is entirely fulfilled. The second check compares the absolute gas-phase  
678 measurement with the saturation mixing ratio (SMR). In case of a fully equilibrated cloud, the gas-phase  
679 has to be saturated at temperature  $T$ , meaning that the relative humidity is  $rH = 100\%$ , from which we can  
680 also derive a SMR with the help of the water vapour partial pressure curve. This is a weaker check, since it  
681 is a well-known fact that super saturation can occur during strong air updrafts.  
682 The transition region at the fringe of the cloud is also very interesting, where the closed-path cell sees an  
683 enhancement while being still below the SMR. This means that the fringe of the cloud was not saturated but  
684 had sublimating ice particles in it.  
685 These first detailed close-ups in HAIs multi-phase measurements show the potential to investigate many  
686 interesting questions in the following campaigns focusing on Cirrus clouds which are strongly linked to a  
687 very reliable accuracy study for the open-path cell in clouds. To really investigate and fully understand all  
688 effects, HAI needs more data in Cirrus, mixed-phase and other clouds in a multi-phase configuration as  
689 well as a validation campaign in which closed- and open-path cells measure the pure gas-phase via a back-  
690 ward measurement. The latter is necessary to distinguish between sampling-effects and optical/spectral-  
691 effects in and outside of clouds. This knowledge, especially in combination with the mentioned CFD model,  
692 will then allow to reevaluate HAI's raw data which are always saved during flights and extent statements  
693 even for this short cirrus transect.

## 694 **6. Conclusion and outlook**

695 The novel *Hygrometer for Atmospheric Investigation*, HAI, realizes in a unique concept a simultaneous gas-  
696 phase and total water measurement. Based on calibration-free direct tuneable diode laser absorption spec-  
697 troscopy, HAI allows accurate, precise (0.065 ppmv at 4.8 Hz, channel depended) and very fast (up to  
698 240 Hz) measurements with a metrologically defined uncertainty (4.3%, channel dependent). HAI contains  
699 four measurement channels, grouped as two completely independent dual-channel spectrometers, one at  
700 1.4  $\mu\text{m}$  and one at 2.6  $\mu\text{m}$  to cover the entire  $\text{H}_2\text{O}$  concentration range of the atmosphere. Each spectrometer  
701 feeds light in a wavelength-individual extractive, closed-path cell with an optical absorption path length of  
702 1.5 m for total water measurements. Additionally, both spectrometers couple their light in a common open-  
703 path cell (optical path of 4.8 m) located outside of the aircraft fuselage, for a sampling-free and contactless  
704 determination of the gas-phase water content. These four spectroscopic channels plus three additional sup-  
705plementary spectroscopic channels allow multiple self-validation strategies inside and outside of clouds  
706 and therefore solve the current lack of an integrated approach to validate open-path sensors in flight. HAI's  
707 complex control software minimizes maintenance at ground and ensures almost entirely autonomous oper-  
708ation. In addition, instrument health is permanently supervised by permanent storage of a set of more than



709 120 housekeeping data. This enables a novel, holistic quality management and a sophisticated signal cross  
710 check, which guarantees a high trust level of the final H<sub>2</sub>O values.

711 HAI was operated for the first time during the TACTS/ESMVal flight campaign for more than 120 operation  
712 hours without any malfunction. The entirely independent, never calibrated first principles evaluation of the  
713 closed-path spectrometer channels reduced in-flight deviations to only 1.9% over a large concentration  
714 (75 to 2000 ppmv) and pressure range (175 to 450 hPa). The deviation between the open-path and the  
715 closed-path measurements in the same flight segment was just 2%. Despite measuring with a single evaluation  
716 concept, over a very broad range of conditions (i.e. temperatures (-70°C to 30°C), gas speeds (cm/s vs  
717 100 m/s), optical disturbances (no background light vs sunlight, clean mirrors vs. dirty scratched mirrors  
718 outside), HAI provided a high trust level of the data over extensive science missions. Laboratory evalua-  
719 tions demonstrated the lowest achievable precision of 0.18 ppmv (at 3.8 Hz) for the 1.4 μm closed-path cell  
720 and 0.055 ppmv (at 13 Hz) for the 2.6 μm closed-path cell.

721 In conclusion, HAI proved during its first deployment a novel, highly complex and demanding set of capa-  
722 bilities. This will enable in the future a much more accurate and stringent evaluation of atmospheric multi-  
723 phase water vapor data inside and outside of clouds and foster in further HALO missions the investigation  
724 of new scientific questions in the atmospheric water cycle. HAI can serve in the future as a major, powerful  
725 tool for cutting-edge atmospheric water vapor measurements.

726

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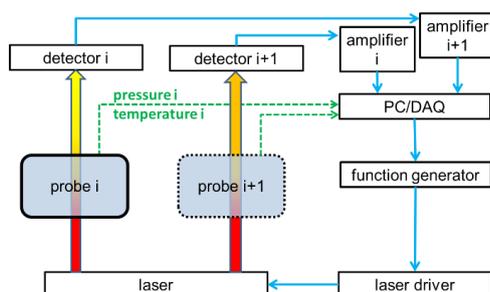
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1110 **Figures:**

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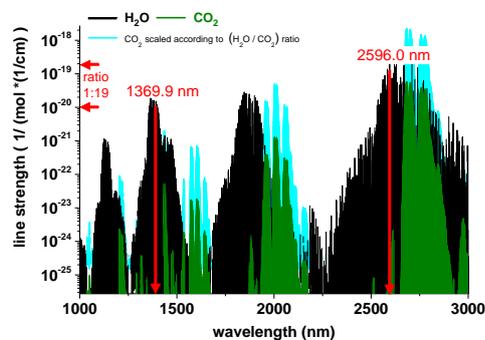
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 1114 Figure 1: Simplified schematic of a standard dTDLAS setup with several spectroscopic channels: HAI combines two  
 1115 lasers (one at 1.4  $\mu\text{m}$ , one at 2.6  $\mu\text{m}$  for high and low concentrations respectively), two with measurement cells (open-  
 1116 and closed-path), and two supplementary spectroscopic channels (parasitic water detector (Buchholz and Ebert, 2014b)  
 1117 and reference cell for spectral stabilization). Completely independent from the 1.4  $\mu\text{m}$  system is HAI's 2.6  $\mu\text{m}$  system  
 1118 with two other channels (open- and closed-path) and one additional channel for the spectral stabilization. Hence, HAI  
 1119 has in total seven spectroscopic channels.

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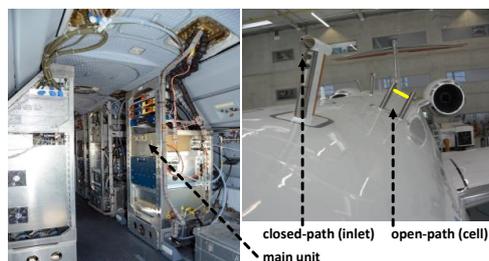
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 1123 Figure 2: HITRAN 2008 line strength plot (Rothman et al., 2009) for CO<sub>2</sub> (green) and H<sub>2</sub>O (black). The turquoise spec-  
 1124 trum in the back shows the scaled (factor 39) CO<sub>2</sub> line strength for visualization of the stratospheric situation with a low  
 1125 level of water vapor (10 ppmv) and standard CO<sub>2</sub> (390 ppmv) level.

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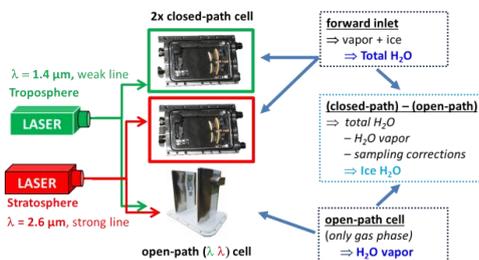


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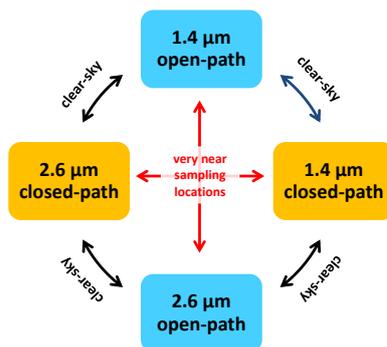
1129 Figure 3: Photos from the cabin layout of HALO with payload of the TACTS/ESMVal campaign (left). Photo of the fuse-  
 1130 lage of HALO showing the trace gas inlet (TGI) for the closed-path HAI cells (for total water detection) as well as the  
 1131 open-path HAI cell (for gas phase water detection; yellow marked are the open laser beams)

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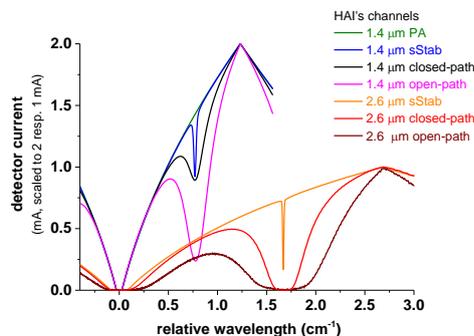
1135 Figure 4: Schematic of HAI's working principle in the multi-phase configuration. By combining selective open-path gas  
 1136 phase measurements with total water measurements in the closed-path, extractive cell (ice is evaporated before the cell  
 1137 by heated inlet lines), it is possible to derive the ice water content from the difference between closed-path and open-  
 1138 path readings.  
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1144 Figure 5: Self-validation possibilities of HAI under clear-sky conditions (black) and redundancies under all atmospheric  
 1145 conditions (red). For details see text.  
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 1149 Figure 6: dTDLAS raw signals (spectra) of all seven spectroscopic channels permanently acquired and saved by HAI.  
 1150 The x-axis is already converted to the wavelength axis via the dynamic tuning rate of the laser. Each spectrum contains  
 1151 approx. 1700 16-bit values. (PA: parasitic absorption, sStab: spectral stabilization)

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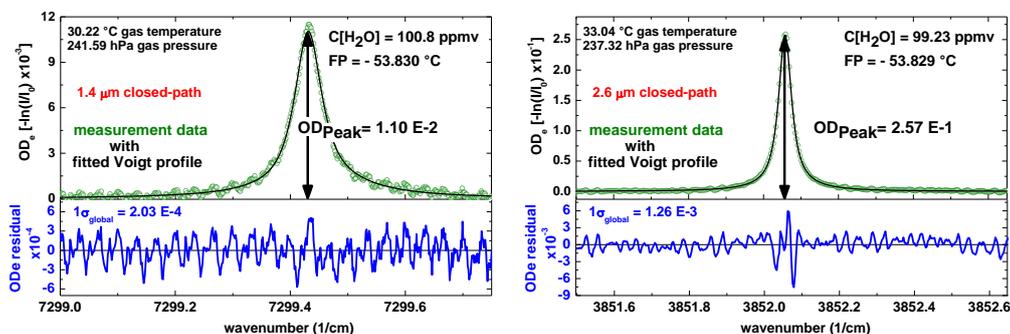


Figure 7: Typical pre-processed absorption signals (after baseline, offset and transmission correction) for both closed-path spectrometer channels (right 1.4  $\mu\text{m}$  left: 2.6  $\mu\text{m}$ ) during flight. The laser modulation frequency was 240 Hz. 50 individual raw scans are pre-averaged yielding 4.8  $\text{H}_2\text{O}$  measurements per second with 69 ms total integration time. This corresponds to a spatial resolution of 15 m at 800 km/h cursing speed. Without averaging, HAI achieves a maximum time resolution of 1.3 ms, respectively a spatial resolution of 30 cm at 800 km/h cursing speed. (It should be noted that the four vertical axes have different scales).

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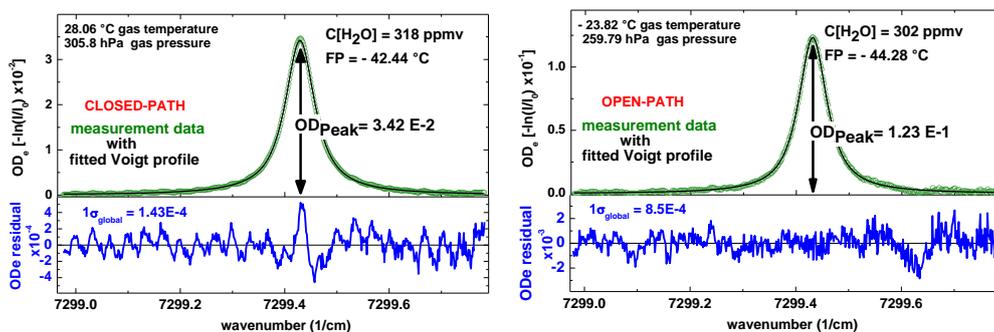


Figure 8: Comparison of pre-processed TDLAS scans (similar to Figure 7) of HAI's 1.4  $\mu\text{m}$  closed-path (left) and the 1.4  $\mu\text{m}$  open-path cell (right) during a HALO flight. Despite the entirely different measurement conditions such as wind speed (cm/s to 100 m/s) or temperature (+28 °C to -23 °C), both channels are evaluated calibration-free with the exact same methods and model.

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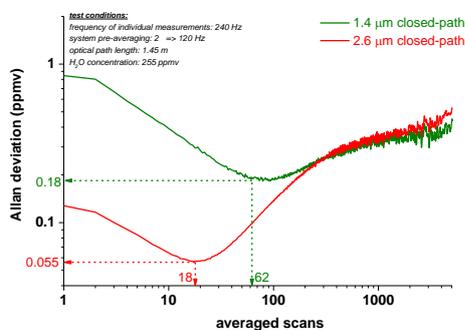
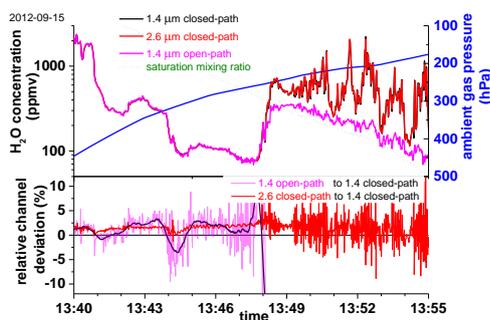


Figure 9: Allan variance plots for both closed-path spectrometer channels. The measurements were done by analysing gas from a big buffer vessel with a H<sub>2</sub>O concentration of 255 ppmv. The optimal precision from this measurement is 0.22 ppmv for the 1.4  $\mu\text{m}$  closed-path cell at 4.8 Hz (same as in Figure 7) and 0.065 ppmv for the 2.6  $\mu\text{m}$  closed-path cell. The best i.e. highest precision of 0.18 ppmv is achieved at 3.8 Hz for the 1.4  $\mu\text{m}$  cell and 0.055 ppmv at 13 Hz for the 2.6  $\mu\text{m}$  cell.

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1158 Figure 10: Typical HAI measurement on board HALO during the TACTS/ESMVal campaign. The flight  
1159 profile shows a 15 min section of a slow ascent from the lower troposphere to the upper troposphere. The  
1160 first part (under clear sky condition) can be used to perform an absolute in-flight accuracy validation of  
1161 both closed-path channels of HAI; the second part on the right side demonstrates a HAI multi-phase H<sub>2</sub>O  
1162 measurement. For further explanations of all signals and visible effects see text.  
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