Application of Gauss's Theorem to quantify localized surface emissions from airborne measurements of wind and trace gases

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1 Abstract

2 Airborne estimates of greenhouse gas emissions are becoming more prevalent with the advent of 3 rapid commercial development of trace gas instrumentation featuring increased measurement 4 accuracy, precision, and frequency, and the swelling interest in the verification of current emission 5 inventories. Multiple airborne studies have indicated that emission inventories may underestimate some hydrocarbon emission sources in U.S. oil and gas producing basins. Consequently, a proper 6 7 assessment of the accuracy of these airborne methods is crucial to interpreting the meaning of such discrepancies. We present a new method of sampling surface sources of any trace gas for which fast 8 9 and precise measurements can be made and apply it to methane, ethane, and carbon dioxide on 10 spatial scales of ~1000 m, where consecutive loops are flown around a targeted source region at 11 multiple altitudes. Using Reynolds decomposition for the scalar concentrations, along with Gauss's Theorem, we show that the method accurately accounts for the smaller scale turbulent dispersion of 12 13 the local plume, which is often ignored in other average "mass balance" methods. With the help of 14 large eddy simulations (LES) we further show how the circling radius can be optimized for the micrometeorological conditions encountered during any flight. Furthermore, by sampling controlled 15 16 releases of methane and ethane on the ground we can ascertain that the accuracy of the method, in 17 appropriate meteorological conditions, is often better than 10%, with limits of detection below 5 kg hr⁻¹ for both methane and ethane. Because of the FAA mandated minimum flight safe altitude of 150 18 19 m, placement of the aircraft is critical to preventing a large portion of the emission plume from 20 flowing underneath the lowest aircraft sampling altitude, which is generally the leading source of 21 uncertainty in these measurements. Finally, we show how the accuracy of the method is strongly 22 dependent on the number of sampling loops, or time spent sampling the source plume.

23

24 1 Introduction

Accurate national inventories of greenhouse gas emissions (primarily carbon dioxide (CO₂), methane 25 26 (CH₄), and nitrous oxide (N₂O)) is of paramount importance in developing strategies to understand global emissions. The multitude of sources, however, are so often highly variable in area, emission 27 magnitude, height above ground, and duration that rigorous verification is exceedingly difficult. 28 29 Nevertheless, measurement techniques have improved markedly in the past decade, and these are 30 being employed to an unprecedented extent in an effort to evaluate and refine emission inventories 31 (Nisbet and Weiss, 2010). Most so called "bottom-up" inventories are developed by aggregating 32 statistical correlates of individual process emissions to such mapping variables as population density, 33 energy consumption, head of cattle, etc., extrapolating to total emissions using a relatively small 34 number of direct measurements. On the other hand, atmospheric scientists have long striven to use 35 measurements from global surface networks, aircraft campaigns, and satellites to try to determine 36 emissions based on the amounts and build-up rates of observed trace gases. Aircraft and satellites, 37 the "top-down" approach, conveniently integrates the multitude of sources, but is heavily reliant on a 38 detailed knowledge of atmospheric transport. Top-down methods also suffer from difficulties 39 attributing sources and generalizing measurements made over a relatively short time period. 40 Attempts to reconcile these two distinct methods on global (Muhle et al., 2010) and continental scales (Gerbig et al., 2003; Miller et al., 2013) have often indicated an apparent underestimation by 41 42 the "bottom-up" methods of a factor 1.5 or more.

43 In principle, the aircraft top-down measurements can be conducted at all the atmospheric scales to

better understand and identify the emissions at comparable scales. For long-lived greenhouse gases,

45 which readily disperse throughout the atmosphere, the global scale is very instructive. The seminal

experiment began with Keeling's acclaimed CO₂ curve [1960], and has continued through more
 contemporary techniques by Hirsch et al.(2006) and Neef et al. (2010) for CH₄ and N₂O, respectively.

47 Contemporary techniques by Hirsch et al. (2006) and Neel et al. (2010) for CH₄ and N₂O, respectively. 48 At progressively smaller scales more details of the source strengths and apportionment can be made:

49 from synoptic or continental scales which can help constrain national inventories (Bergamaschi et al.,

50 2005) or specific biogeographic regions (Gallagher et al., 1994), to mesoscale investigations that

51 estimate emissions from urban areas (Mays et al., 2009; Turnbull et al., 2011; Wecht et al., 2014) or

52 specific oil and gas producing fields (Karion et al., 2013; Petron et al., 2014) and even down to

53 individual point/area sources on the order of 10-100 m size (Denmead et al., 1998; Lavoie et al., 2015;

54 Roscioli et al., 2015).

55 Aircraft in-situ measurements are particularly useful for "top-down" methods at the sub-mesoscale

56 because they can be used to measure the air both upwind and downwind of a source region.

57 However, deployments tend to be costly and thus sporadic. As far as we know, the aircraft methods

used so far can be categorized into three types. First, there is the eddy covariance technique that is

carried out at low altitudes wherein the vertical fluxes of gases carried by the turbulent wind are

60 measured by tracking rapid fluctuations of both concentrations and vertical wind (Hiller et al., 2014;

Ritter et al., 1994; Yuan et al., 2015). This method is generally thought to be the most direct, but it is
 limited to small footprint regions which must be repeatedly sampled for sufficient statistical

63 confidence, requires a sophisticated vertical wind measurement, and can be subject to errors due to

64 flux divergence between the surface and the lowest flight altitude and acceleration sensitivity of the

65 gas sensor. The second, and by far the most common approach is what chemists usually refer to as

66 "mass balance" and what is known in the turbulence community as a "scalar budget" technique.67 Many different sets of assumptions and sampling strategies are employed, but the overall goal is to

sample the main dispersion routes of the surface emissions as they make their way into the overlying
 atmosphere after first accumulating near the surface. The scales that can be addressed by this

70 method are from a few kilometers (Alfieri and Blanken, 2012; Hacker et al., 2016; Hiller et al., 2014;

71 Tratt et al., 2014) to tens of kilometers (Caulton et al., 2014; Karion et al., 2013; Wratt et al., 2001) to

even potentially hundreds of kilometers (Beswick et al., 1998; Chang et al., 2014), and this approach

has been the focus of recent measurements in natural gas production basins. These basins present a

source apportionment challenge in that emissions from multiple sources (agriculture, oil & gas wells,

75 geologic seepage, etc.) commingle as the air mass travels across the basin. The third method of

source quantification is to reference measurements of the unknown trace gas to a reference trace
 gas with a metered release (tracer) or otherwise known emission rate and assume that the tracer and

gas with a metered release (tracer) or otherwise known emission rate and assume that the tracer and
the scalar of interest have the same diffusion characteristics. Typically this tracer release technique is

79 applied to small scales of tens to hundreds of meters (Czepiel et al., 1996; Lamb et al., 1995; Roscioli

et al., 2015), but the principle has been attempted at the basin (Peischl et al., 2013) and continental

(Miller et al., 2012) scales using a reference trace gas with a suitable known emission rate such as CO₂
 or CO.

83 The airborne mass balance flight strategies can be grouped into three basic patterns: a single height

84 transect around a source assuming a vertically uniformly mixed boundary layer (Karion et al., 2013);

85 single height upwind/downwind (Wratt et al., 2001) or sometimes just downwind flight legs (Conley

86 et al., 2016; Hacker et al., 2016; Ryerson et al., 1998); and multiple flight legs at different altitudes,

87 (Alfieri et al., 2012; Gordon et al., 2015; Kalthoff et al., 2002); or just a 'screen' on the downwind face

88 of the box (Karion et al., 2015; Lavoie et al., 2015; Mays et al., 2009).

89 Here we describe a new airborne method borne out of a necessity to identify and quantify source

- 90 emissions to within 20% accuracy in a large heterogeneous field of potential sources. The novel
- 91 technique applies an aircraft flight pattern that circumscribes a virtual cylinder around an emission
- 92 source and, using only observed horizontal wind and trace gas concentrations, applies Gauss's
- 93 Theorem to estimate the flux divergence through that cylinder. By integrating the outward horizontal
- 94 fluxes at each point along the circular flight path, the flux contributions from enclosed sources can be 95 accounted for. Making an accurate estimate, however, requires the selection of an appropriate
- accounted for. Making an accurate estimate, however, requires the selection of an appropriate
 circling radius based on the micrometeorological conditions inferred in flight from measurements
- 97 onboard the aircraft. The pattern must be far enough downstream for the plume to mix sufficiently in
- 98 the vertical, yet not so far that the trace gas plume enhancements do not stand out sufficiently from
- 99 the background concentration.

100 In this study we first present the general analytical method used to derive emission estimates using

101 airborne measurements. Next, we investigate the structure of a generalized dispersing plume using

large-eddy simulation (LES) to better understand the optimal sampling strategies for quantifying
 near-surface gas sources. Because the wind fields of turbulent flows cannot be predicted in detail, we

104 do not attempt to compare specific features of our observations with specific LES results, but rather

105 we use the numerical experiments to guide the development of the observational methodology. For

106 example, by investigating the LES flux divergence profiles in the layer below the lowest flight altitude,

- 107 we are able to estimate the contribution of this unmeasured component to the overall source
- strength. We then evaluate the accuracy of the approach using coordinated planned release
- 109 experiments and by applying the method to CO_2 emitted from several power plant plumes to
- 110 compare with reported emissions.
- 111

112 2 Data Collection

113 2.1 Airborne Instrumentation

114 The airborne detection system is flown on a fixed wing single-engine Mooney aircraft, extensively modified for research as described in Conley et al. (2014). Ambient air is collected through ~5 m of 115 116 tubing (Kynar, Teflon and stainless steel) that protrudes out of backward-facing aluminum inlets 117 mounted below the right wing. In-situ CH₄, CO₂, and water vapor are measured with a Picarro 2301f 118 cavity ring down spectrometer as described by Crosson (2008), which is operated in its precision 119 mode at 1 Hz. In-situ ethane (C₂H₆) is measured with an Aerodyne Methane/Ethane tunable diode 120 infrared laser direct absorption spectrometer (Yacovitch et al., 2014). There is a 5-10 second time lag 121 in both analyzers that depends on the flow rate and tubing diameter. We use a 1/8" OD (3.175 mm) stainless line for the Picarro (~0.2 slpm flow rate), and a 6.3 mm ($\frac{1}{4}$ in) Teflon line for the CH₄/C₂H₆ 122 123 spectrometer (~4 slpm flow rate). This results in lag times of ~5 s for the Aerodyne and ~10 s for the Picarro. The lag time for the Picarro is calculated using a "breath test", whereby we exhale into the air 124 125 inlet and measure the time required for the CO₂ measurement to peak. The ethane lag time is

adjusted to maximize the correlation between the ethane and Picarro methane time series in plumes 126 127 where both gases are emitted. Both lag times are slightly dependent on pressure, i.e., with a typical 128 altitude change of ~1 km, the change in lag time is less than 10%, and is inconsequential when 129 applying this method within a few hundred meters from the surface. The horizontal wind speed and 130 direction, sampled at 1 Hz, is based on a standard aircraft pitot-static pressure airspeed measurement and a dual GPS compass that determines aircraft heading and ground speed. The accuracy of the 131 horizontal wind measurement is about 0.2 m s⁻¹ (Conley et al., 2014). The horizontal wind is 132 calibrated periodically by flying ~5 km L-shaped patterns in the free troposphere; a heading rotation 133 134 and airspeed adjustment is made to the wind calculation to minimize the dependence of the wind on aircraft heading. These adjustments typically amount to less than 2° rotation and 3% adjustment of 135 136 the airspeed. In flying the tight circle patterns described below, the pilot does not adjust the rudder 137 trim to use the same calibration coefficients in the wind measurement calculation throughout the 138 flight.

139 2.2 Large Eddy Simulations

In order to study the plume behaviour of surface emissions as it relates to sampling in the stacked 140 141 circles, we use the LES module of WRF V3.6.1. WRF-LES explicitly resolves the largest turbulent eddies 142 by filtering the Navier-Stokes scalar conservation equations at some scale in the inertial subrange, 143 and allowing the smaller motions beyond the cut-off to be modeled using a sub-grid (also called a 144 subfilter) scale turbulence parameterization that is based on properties of the larger-scale, resolved flow. Because the aircraft data is typically sampled at 1 Hz and the true airspeed is around 70 m s⁻¹, 145 we use an LES horizontal grid size roughly half (40 and 50 m) the distance between aircraft data 146 samples. Because periodic lateral boundary conditions are imposed on the WRF-LES variables, care 147 148 must be taken to ensure that the effluent does not reach the lateral boundaries of the simulation 149 domain. On the other hand, WRF-LES does not allow for parallelized computation, making the 150 simulations quite expensive in terms of computation time. We therefore struck a balance between a 151 large enough domain in horizontal extent (6 and 8 km) such that the effluent would not reach the 152 downwind boundary before the end of our simulation, while maintaining a grid size small enough to 153 resolve scales of the aircraft observations. The vertical domain needs to be large enough to 154 encompass a developing convective boundary layer (CBL), while at the same time containing 155 substantial free tropospheric flow above to serve as a reservoir that can feed momentum and freetropospheric scalars to the CBL. Moreover, the stable region (potential temperature lapse rate $d\theta/dz$ 156 157 = 5 C km⁻¹) between the CBL inversion base and the top of the domain had to be large enough to 158 damp any wave activity before it could reflect off the upper boundary and create spurious motions 159 throughout the domain.

- 160 The standard *WRF-LES* module is not set up to allow for effluent release, so we implemented a 161 modified version of the *WRF* source code (*S.-H. Chen, personal communication*) that includes a 162 surface effluent release with a specified position and release rate. Three different convective 163 simulations were run with varying resultant mean wind speeds in the boundary layer, and each was 164 allowed 4-5 hours to 'spin-up' dynamically before the effluent was released at a rate of between 2.9-165 3.5 kg hr⁻¹. The exact release time was selected to give reasonably stationary CBL depths and
- 166 turbulent kinetic energy. The conditions for the three simulations are listed in Table 1, and based on

- 167 the different wind speeds they span moderate to strongly convective boundary layers (- z_i/L_{MO} from
- 168 ~50 to ~200, where L_{MO} is the Monin-Obukhov length and z_i is the CBL depth.)
- 169 3 Methods

170 3.1 Theory of Measurement using Gauss's Theorem

171 We use an integrated form of the scalar budget equation for a passive, conservative scalar in a 172 turbulent fluid to estimate the emission of a gas of interest within a cylindrical volume V. The volume is circumscribed by a series of closed aircraft flight paths (typically circular) flown around the emission 173 source over a range of altitudes. The altitudes encompass the lowest safe flight level (usually 75-150 174 175 mAGL) up to an altitude where no discernable change in the trace gas mixing ratio, χ , is observed around the flight loop, z_{max} . The scalar in our case is the mass concentration (i.e., density of a 176 177 chemically unreactive species in a turbulent flow field, u = (-ui + vj + wk); its Reynolds decomposition is c = C + c', where C is the mean concentration around each loop and c' is the departure from the loop 178 179 mean. Figure 1 shows an actual example of the effluent sampled by the aircraft in a sequence of 180 stacked paths / that circumscribe an area, A, enclosing the source in a volume, V. The effluent is carried downwind as it mixes upward in the CBL. A virtual surface circumscribed by the circular flight 181 182 tracks is assumed enclosing the source and extending above the vertical extent of the plume so that 183 there is no vertical transport above that level. To estimate the source strength, we start with the

184 integral form of the continuity equation:

$$Q_c = \langle \frac{\partial m}{\partial t} \rangle + \iiint \nabla \cdot c \boldsymbol{u} \, dV \tag{1}$$

185 where $\langle \rangle$ denotes an average over the volume *V*, Q_c is the sum of the internal sources and sinks of *c* 186 within *V*, and *m* is the total mass of c within the volume V. At this point, we recognize that the flux 187 divergence is composed of two terms

$$\nabla \cdot \mathbf{c}\mathbf{u} = \boldsymbol{u} \cdot \nabla \mathbf{c} + \mathbf{c}\nabla \cdot \mathbf{u} \tag{2}$$

188

189 In section 3.2 we perform a scale analysis of the terms on the right-hand side (rhs) of equation 2 and 190 show that the second term, which is proportional to the horizontal wind divergence, may be 191 neglected under our normal flight protocol. This is fortunate because of the difficulty in accurately 192 estimating the horizontal wind divergence from aircraft measurements (Lenschow et al., 2007). The 193 vertical flux across the top of the flight cylinder is assumed to be zero and the flux from the bottom 194 (ground) is the surface source we are measuring. This leaves us with only the horizontal flux, i.e. cu_h 195 where u_h (= ui+vj). In order to minimize the contribution from the horizontal wind divergence term, 196 we remove the loop mean concentration, C, which does not alter the first term on the rhs because 197 ∇C =0, so that equation 2 becomes

$$\boldsymbol{u}_h \cdot \nabla \boldsymbol{c} + \mathbf{c} \nabla \cdot \boldsymbol{u}_h = \boldsymbol{u}_h \cdot \nabla (\boldsymbol{c}'). \tag{3}$$

Next, we use Gauss's Theorem to relate the volume integral to a surface integral around the volumethat is sampled by the aircraft flight loops:

$$Q_c = \langle \frac{\partial m}{\partial t} \rangle + \iiint \nabla \cdot (c' \boldsymbol{u}) \, dV = \left\langle \frac{\partial m}{\partial t} \right\rangle + \oiint c' \boldsymbol{u} \cdot \hat{\boldsymbol{n}} \, dS \tag{4}$$

where S is the surface enclosing V and \hat{n} is an outward pointing unit vector normal to the surface.

The surface integral can be broken into three elements: a cylinder extending from the ground up to a level above significant modification by the emission, the ground surface circumscribed by a low-level (virtual) circular flight path (z = 0), and a nominally horizontal surface circumscribed by a flight path above the level modified by the source ($z = z_{max}$). We assume there is no significant flux (other than the source of interest) into or out of the ground. Next, the surface integral is estimated solely from a sequence of closed path integrals measured by the aircraft at multiple flight levels to estimate the right side of Eq. 5 (blue dashed lines in Fig. 1),

$$\oint c' \boldsymbol{u} \cdot \hat{\boldsymbol{n}} \, dS = \int_0^{z_{max}} \oint c' \boldsymbol{u_h} \cdot \hat{\boldsymbol{n}} \, dl \, dz, \tag{5}$$

208 where *l* is the flight path.

209

210 Combining Eqs. 4 and 5 leads to the result that is the basis for this measurement technique where a 211 series of horizontal loops at different altitudes are flown around a source region:

$$Q_{c} = \langle \frac{\partial m}{\partial t} \rangle + \int_{0}^{z_{max}} \oint c' \boldsymbol{u}_{\boldsymbol{h}} \cdot \hat{\boldsymbol{n}} dl \, dz \tag{6}$$

Along each path the instantaneous outward flux is computed and summed over the loop to yield the mean flux divergence via Gauss's Theorem. A temporal trend of the total mass within the volume $\begin{pmatrix} \frac{\partial m}{\partial t} \end{pmatrix}$ can be estimated from the flight data and added to the flux divergence integral to obtain the emission rate.

216

217 3.2 Divergence Uncertainty

In order to estimate the relative error in the horizontal divergence term that we are eliminating, we perform a scale analysis of the relative size of the two terms that make up the path integral in Eq. 2, using some typical values of the CBL parameters (convective velocity scale $w_* = 1 \text{ m s}^{-1}$, boundary layer depth, $z_i = 1,000 \text{ m}$), and sampling geometry (flying at a radius 1 km around a point source.)

Taylor's (1922) statistical theory of dispersion in a homogenous and stationary turbulent fluid

predicts that the root mean square lateral (σ_v) and vertical (σ_z) dispersion parameters increase

linearly with time, or equivalently advection distance, downwind in the near-field. Weil (1988) shows

several examples of the growth of both of these parameters downwind to be ~ $0.5w_*$, which we use

- here for a rough estimate of a conical plume spreading to quantify the dilution of the source's
- 227 emission as it travels downwind to be intercepted by the aircraft. We use a large background mixing
- ratio characteristic of global CH₄ (~1.9 ppmv), estimate the mean gradient by the plume
- concentration divided by the distance downwind, and assume a conservatively large horizontal wind
- divergence of 10^{-5} s⁻¹, which may in fact be typical for our small sampling region (Stull, 1988). The
- results are shown in Figure 2 and, for all but the smallest sources of a few kg hr⁻¹ and wind speeds
- below 1 m s⁻¹, the divergence term is at least an order of magnitude smaller than the gradient term.
- 233

234 3.3 Applying the Theory to the LES Results

235 We calculated a comparable estimate of Q_c in the LES domain from the air density, concentration,

and wind along circular flight paths as a virtual aircraft would fly. Willis and Deardorff (1976)

237 generalized results of their convection tank experiments to downwind dispersion in the convective

boundary layer (CBL) in terms of a dimensionless length scale X, the ratio of the horizontal advection

time to the large eddy turnover time:

$$X = \frac{xw_*}{Uz_i} \tag{7}$$

where *x* is the downwind distance and *U* is the vertically averaged mean wind speed.

241 Figure 3 shows the crosswind-integrated concentration profile for the plume release in the UCD50B

242 WRF-LES run as function of X, and normalized height, $Z = z/z_i$. Because of the time limitation due to

the periodic boundary conditions, the plume is averaged for only ~15 minutes of simulation time

which is just under a large eddy turnover time for the conditions of the run. The results displayed in

Figure 3 are in good qualitative agreement with the results of Willis and Deardorff (1976) and Weil et al., (2012) save for the release being at the surface in our LES study, and at *Z* = 0.067 for the above

studies (see Fig. 1 and 2 of Weil et al., (2012)). Figure 3 shows the maximum concentration being

lofted near X~0.2 and leveling off near Z ~ 0.8 around X ~ 0.6; beyond X > 1.5 the plume is fairly well-

249 mixed throughout the extent of the boundary layer.

250 3.4 The Upwind Directed Turbulent Flux

251 Horizontal turbulent fluxes are generally ignored in boundary layer budget studies due to the fact 252 that while they are often sizeable in magnitude they do not change significantly over horizontal 253 length scales under consideration (the horizontal homogeneity assumption). In the vicinity of a point 254 source, however, this is not likely. The method outlined here estimates source emissions using a 255 measured horizontal flux that incorporates wind and scalar measurements at 1 Hz sample rate, 256 resolving scales of ~70 m (Conley et al., 2014), which should include nearly all of the turbulent 257 contributions to the horizontal flux. Here we consider the nature of this turbulent flux and the error 258 in emission estimates if only the mean transport were considered. We start with the budget equation 259 for a horizontal scalar flux in a horizontally homogeneous turbulent flow where the molecular 260 diffusive/viscous term has been neglected (Wyngaard, 2010),

261
$$\frac{d\overline{c'u'}}{dt} = -\overline{u'^2}\frac{\partial c}{\partial x} - \overline{u'w'}\frac{\partial c}{\partial z} - \overline{c'w'}\frac{\partial U}{\partial z} - \frac{\partial\overline{c'u'^2}}{\partial x} - \frac{\partial\overline{c'u'w'}}{\partial z} - \frac{1}{\rho}\overline{c'\frac{\partial p'}{\partial x}}$$
(8)

262 where ρ is density and p' is the pressure fluctuation. We then assume stationarity and integrate

across the source from a point just upwind to a point within the plume and obtain

$$264 \qquad \int_{0-}^{x} \frac{\partial \overline{c'u'}}{\partial x'} dx' = -\frac{1}{U} \int_{0-}^{x} \left[\overline{u'^2} \frac{\partial C}{\partial x'} + \overline{u'w'} \frac{\partial C}{\partial z} + \overline{c'w'} \frac{\partial U}{\partial z} + \frac{\partial \overline{c'u'^2}}{\partial x'} + \frac{\partial \overline{c'u'w'}}{\partial z} + \frac{1}{\rho} \overline{c' \frac{\partial p'}{\partial x'}} \right] dx' \tag{9}$$

We further assume that although the scalar field is not homogeneous the flow field is, and the background horizontal c-flux upwind is much smaller than the flux induced by the point source. This results in an equation for the in-plume flux

$$268 \quad \overline{c'u'} = -\frac{1}{U} \Big[\overline{u'^2} \big(C_x - C_{bckg} \big) + \overline{u'w'} \int_{0-\overline{\partial z}}^x \frac{\partial C}{\partial z} dx' + \int_{0-\overline{\partial z}}^x \overline{c'w'} \frac{\partial U}{\partial z} + \int_{0-\overline{\partial z'}}^x \frac{\partial \overline{c'u''^2}}{\partial x'} + \frac{\partial \overline{c'u'w'}}{\partial z} + \frac{1}{\rho} \overline{c'\frac{\partial p'}{\partial x'}} \Big) dx' \Big] (10)$$

269 The first three terms on the rhs of equation 10 are negative within the plume with their largest 270 magnitudes on the upwind side and diminishing downwind. On the largest, boundary layer filling 271 eddy scales the mean concentration of C downwind of a source is greater than in the upwind region, $(C_x-C_{bcka})>0$, and therefore the first term is negative, but decreases in magnitude with distance 272 downwind. However, this term is also positive on smaller scales within the plume where the mean 273 gradient is directed upwind towards the source, and is most likely responsible for the specious 274 275 intuitive impression that the horizontal turbulent flux should transport the plume downwind from the 276 source along with the mean wind advection. Moreoever, the second and third terms on the rhs are negative because the momentum flux, $\overline{u'w'}$, and mean vertical gradient, $\partial \mathcal{L}/\partial z$, are negative while the 277 vertical turbulent flux, $\overline{c'w'}$, and wind shear, $\partial U/\partial z$, are positive. Based on the vertical concentration 278 profiles shown in Weil et al. (2012) (their Figures 3 & 4) it can be inferred that the vertical 279 280 concentration gradient, $\partial C/\partial z$, changes from negative to positive near X~1 and becomes negligible for X > 2-3. Similarly, in the third term, the vertical flux, $\overline{c'w'}$, decreases with fetch. Thus the counter-281 directed flux ($\overline{c'u'} < 0$) will fade with distance downwind. Wyngaard et al. (1971) have shown that 282 283 the third-moment, turbulent transport terms (4 and 5 on rhs of equation 10) in the horizontal heat flux equation are small in the surface layer compared to the source terms, so we assume the same 284 285 holds for this scalar flux. Finally, the remaining pressure covariance term is believed to be the main 286 sink in the budget equation working to decorrelate the wind and the scalar as was shown in the 287 surface layer measurements of Wilczak and Bedard (2004). Therefore, the dominant production terms for negative $\overline{c'u'}$ (terms 1-3 on the rhs of 10) must be balanced by the pressure-correlation 288 289 term leading to an upwind-directed horizontal turbulent flux within the plume that decreases in 290 magnitude in the downwind direction.

This conclusion is supported by several previous studies. For example, in a wind-tunnel study of flux-291 292 gradient relationships Raupach & Legg (1984) reported that the mean streamwise horizontal heat flux 293 calculated by multiplying the mean wind by the mean temperature overestimates the total heat flux 294 by approximately 10%, which suggests that the turbulent component of the horizontal heat flux is 295 negative; that is, the turbulent flux is upwind, directed counter to the mean flow. Other researchers 296 have reported an even larger disparity. Field experiments by Leuning et al. (1985) indicate that the 297 horizontal turbulent flux of a trace gas is ~15% the mean flux, while Wilson and Shum (1992) suggest 298 it may be 20%. A recent LES study of particle dispersion over a plant canopy by Pan et al. (2014)

indicates magnitudes of 20% or more for the negative turbulent component of scalar fluxes in the
 vicinity of the source and decreasing with downwind fetch. We therefore conclude that when
 sampling a near-surface point source at X of order unity or less, if only the mean concentration

302 difference is measured, a significant overestimate of the scalar source is likely to occur.

303 Further evidence of this is shown in the average cospectrum of the outward wind and concentration 304 fluctuation observed in the flight loops in Figure 4. Because the integral of the cospectrum yields the 305 total flux (scalar and wind covariance), this function is useful for examining the contributions to the 306 overall flux from each of the scales of motion (represented by aircraft speed divided by frequency). 307 The results shown in Figure 4 are from a CH_4 point source with an estimated emission rate of 46 \pm 7 kg 308 hr⁻¹ which was circled 70 times at a dimensionless radius X of approximately 0.35. All cospectra of 309 sampled sources have the same structure seen in Figure 4: there is an obvious peak at the mean flight 310 loop frequency (usually ~100 s period) followed by a smaller negative dip at higher frequencies within 311 the meandering effluent plume. We believe this to be good evidence that our method captures this 312 important component of the overall flux away from the source, which cannot be obtained with a 313 traditional mean wind and an integrated concentration enhancement measurement that is so often

employed in airborne source estimates (Ryerson et al., 2001; White et al., 1976).

315 **3.5 Choosing the Downwind Sampling Distance**

Determining the optimal sampling distance from the targeted point source is a balance of several 316 317 factors. First, not surprisingly, the largest plume signal occurs closest to the source (Figure 3). Second, a high degree of confidence in the results is contingent upon sampling the majority of the plume at 318 319 and above the lowest flight altitude, which only occurs downwind after a sufficient time has elapsed 320 to loft the initially near-surface plume. And third, an attempt is made to sample the plume before it 321 reaches the top of the boundary layer so that the vertical turbulent entrainment flux does not 322 become appreciable violating the assumption of negligible flux through the top of the volume V as 323 discussed in Equation 2. Finally, close to the source, the fluctuations in concentration will be very 324 large, intermittent, at small scales, and highly variable.

- 325 To gain further insight into the second feature of the dispersing plume, Figure 5 shows the average 326 horizontal flux divergence profiles derived from the three WRF-LES runs. Here we discuss a 327 dimensionless R, which is identical to X, to emphasize that this scaled downwind distance from the 328 source is a radius of a flight loop. The flux divergence values are made dimensionless by the boundary 329 layer height, z_i , and the source emission rate, Q. Very close to the source, before the plume has had a 330 chance to loft, the flux divergence profile exhibits a strong gradient below the minimum safe flight 331 altitude, making that term difficult to measure directly, as shown in Figure 5. Farther from the source, the signal becomes weaker with increasing altitude and eventually becomes increasingly influenced 332 333 by entrainment fluxes. We therefore seek a sampling distance that is far enough to allow sufficient 334 vertical lofting yet close enough so that plume crossings are easily observable against the background
- variability and instrument noise, and are not yet influenced by entrainment mixing.
- Based on the simulation results presented in Figure 5, we see the gradient below the lowest flight
- 337 safe altitude typically becomes very small for R > 0.4, and therefore we attempt to target that
- distance to minimize the extrapolation error from the flight data to the surface. We do not currently

- measure all the necessary parameters to estimate R in-flight (primarily the surface heat flux $(\overline{w'\theta'_{\nu}})_0$
- 340 which is required to estimate w_*). Instead, we estimate w_* based on the observed boundary layer
- height, standard deviation of wind speed, and a parameterization for $w_* = \sigma_u / 0.6$ (Caughey and
- 342 Palmer, 1979).

343 3.6 Minimum number of passes

344 The atmospheric boundary layer is a turbulent medium, meaning that two passes across a plume at the same altitude and distance downwind will likely make very different measurements of the trace 345 gases of interest. A natural question arises as to how many passes are required to develop a 346 347 statistically sound estimate of the emission rate. We investigate the number of passes required to obtain a statistically robust estimate using the WRF-LES results and a controlled release experiment. 348 349 By calculating the horizontal flux divergences with a virtual airplane flying through the simulated 350 tracer field, and then randomly sampling the flux divergences from each of the legs and plotting the resultant estimated emission rate as a function of the number of samples used we obtain the results 351 352 presented in Figure 6. The gray region around the red line mean represents the standard deviation of 353 estimates based on a random set of loops. Figure 7 shows results from an analysis of actual flight data 354 from the ethane controlled release test near Denver, Colorado on November 19, 2014. It is evident from both the simulation data and the field data that a statistically stable estimate seems to be 355 356 achieved somewhere between 20-25 loops around the source.

357 3.7 Discretization and Altitude Binning the Flux Divergence Data

Measurements of the relevant scalars (e.g. CH₄) and meteorological variables are sampled at discrete time intervals. For our analyses, we interpolate all measurements including GPS (3 Hz), methane (1 Hz), temperature (1 Hz) and computed variables including horizontal wind (1 Hz), and air density (1 Hz) onto a synchronous 1 Hz time series. Next, we estimate the path integral for each individual loop of the flux normal to the flight path by summing up the flux contributions times the sample length around each loop and then summing over the height intervals,

$$Q_c = \left\langle \frac{\partial m}{\partial t} \right\rangle + \oiint \boldsymbol{F}_c \cdot \hat{\boldsymbol{n}} \, dS = \frac{\Delta m}{\Delta t} + \sum_{z=0}^{z=Z_t} (\sum_{0}^{L} (\rho \cdot u_n) \cdot \Delta s) \cdot \Delta z, \tag{9}$$

364 where ρ is the scalar air density, u_n is the wind speed normal to the flight path, Δs is the distance covered during the 1 s time interval of each measurement and L is the distance covered in one 365 complete circuit. The outer summation sums each of the discrete vertical laps from the bottom (z =366 367 0) to the highest lap $(z = z_t)$. If all laps were sampled at equidistant altitudes, the total divergence 368 could be calculated as the average divergence of all laps multiplied by the top altitude. However, 369 because there is greater horizontal transport and variability at lower altitudes, as demonstrated by 370 the widening standard deviations approaching the surface in the theoretical flux divergence profiles 371 shown in Figure 4, more sampling laps at lower altitudes increase the statistical validity of the largest 372 horizontal transport values. To ensure that all altitudes are nearly equally weighted, we divide the 373 vertical range into six equally spaced bins, save for the lowest bin which is extended to the surface, 374 and then average the measurements from the laps within each bin. The total emission is the sum of 375 the flux times the path length in each bin multiplied by the bin width. We also performed 6 flights

376 where we sampled equally at all altitudes to allow a comparison of the direct average versus the

binned results, and in all of these flights the values derived by the two methods agreed to within 5%.

378 3.8 Error Analysis

379 Our method assumes a stationary emission source. The leg-to-leg variability is primarily driven by the 380 stochastic nature of turbulence (e.g. we may sample the plume on one lap and miss it on another). By 381 aggregating the laps into vertical bins, we can use the standard deviation of the horizontal fluxes 382 within each bin as an estimate of the uncertainty within that bin. Then the total uncertainty in the 383 estimate of the flux divergence is simply estimated by adding up the individual bin uncertainties in 384 quadrature. The first term on the rhs of Equation 6 is the time rate of change of the scalar mass 385 within the cylindrical flight volume. This storage term is estimated by performing a least squares fit of 386 the methane density with time and altitude. The resulting uncertainty in the time rate of change is 387 then combined (summed in quadrature) with the uncertainty from the altitude bins to achieve a total 388 uncertainty in the measurement.

389

390 4 Results and Discussion

391

We use measurements from three sets of flights to characterize the accuracy of this estimation
 method. We flew 2 days measuring an ethane controlled release provided by Aerodyne Research,
 Inc., 4 days measuring a natural gas controlled release provided by the Pacific Gas & Electric Company
 (PG&E), and 6 power-plant flights where our estimates are compared with reported hourly power
 plant CO₂ emissions.

- 397
- 398

399 4.1 Ethane Controlled Releases

400 Two experiments with known/controlled ethane releases were performed in collaboration with the 401 Aerodyne Mobile Laboratory team. Pure ethane was released and measured with a flowmeter by the 402 Aerodyne ground crew. The Colorado site (November, 2014) was in a remote area approximately 105 403 miles NE of Denver. This site was chosen because of the flat terrain and lack of other nearby ethane 404 sources that could pollute the controlled release plume. The flux profiles for both releases are shown 405 in Figure 8 and indicate that, in both cases, the aircraft successfully flew above the ethane plume 406 (measurements tend toward zero with increasing altitude). An example of an individual lap is shown 407 in Figure 9 and indicates a clear plume signal downwind of the release. As the aircraft climbs, 408 eventually the signal disappears, as shown in the figure. Agreement was excellent, with the 409 estimated emission just 2% over the actual controlled release rate. The second Aerodyne controlled release in Arkansas on October 3, 2015 was performed at a site surrounded by nearby emission 410 411 sources and an elevation change (~70m) within the aircraft flight path. The aircraft-derived ethane 412 emission estimate was 25% higher than the controlled release rate and the calculated uncertainty 413 was significantly higher than on other sites (Table 2).

A significant upwind ethane source was observed during the Arkansas experiment. This source was 414 415 evident on roughly half of the upwind passes, suggesting that techniques which rely on a limited 416 number of upwind passes to characterize the background could have a large random error and thus 417 erroneously estimate the upwind source strength. A similar problem would affect those techniques 418 that employ a downwind transect, using the edges of that transect lying outside the plume to 419 estimate the background concentration. These observations demonstrate the complication (and 420 bias) that can arise from nearby sources. Since this method integrates all the emission sources in the 421 area within the flight circle and a small distance upwind of the circle depending on the vertical 422 mixing, estimates from Gauss's method may be biased high if there are sources within that area. The 423 average error of the two ethane releases is 13%. 424

425 4.2 Natural Gas Controlled Releases

426 In conjunction with PG&E, we performed two sets of two-day ground-level controlled release 427 experiments from existing PG&E facilities, exactly one year apart. The first set was performed southeast of Sacramento near the town of Rio Vista, CA at the Rio Vista "Y" station and the second 428 429 set near Bakersfield, CA. For the Rio Vista test, the release rate was not calibrated with a flow meter 430 but, based on the size of the orifice and the upstream pressure, the release rate was estimated at 15.2 ± 1.5 kg hr⁻¹. This release rate is an estimate of the total gas being released which is a 431 combination of primarily CH₄ and C₂H₆. We use the regression fit of ethane to methane (averaging 432 433 0.085 by mass) to estimate the actual release rate of each scalar.

434 In comparison with the C₂H₆ controlled release, CH₄ controlled releases suffer from the effect of small 435 enhancements relative to the background concentration. During the Rio Vista release, the largest 436 enhancement that we measured was 100 ppb, with 30-40 ppb being typical. Using a typical 437 background level of 1.9 ppm, a 40 ppb enhancement represents 2% of the background. In contrast, 438 for ethane the enhancements are as large or larger than the background. The results of the methane 439 controlled release tests are shown in Table 3 and indicate aircraft-derived estimates within 17% of 440 the controlled release rate. This large background results in increased uncertainty in the emission 441 calculation. The average difference between the estimated emission and the calculated flow rate is 442 13%.

443

444 4.3 Power Plant Flights

445 Power plants in the U.S. are required to report CO_2 emissions to EPA (https://ampd.epa.gov/ampd) 446 on an hourly basis. The accuracy of the reported CO_2 emissions has been determined to be ± 10.8 -447 11.0% based on reported U.S. average differences between Energy Information Administration (EIA) fuel-based estimates and EPA continuous emission monitoring-based estimates (Ackerman and 448 Sundquist, 2008; Peischl et al., 2010; Quick, 2014). Also, Peischl et al. (2010) determined an accuracy 449 450 of power plants reporting CO₂ emissions in Texas of ±14.0% based on differences between observed downwind SO₂/CO₂ and NOx/CO₂ emission ratios and those reported via EPA continuous emission 451 452 monitoring (Peischl et al., 2010). Here, we use the slightly larger uncertainty from Peischl et al.

(2010). Power plant emissions are "hot" gases and very buoyant, in contrast to a surface emission 453 454 source that is typically not buoyant. An additional uncertainty arises from temporal emission 455 variability (hourly averaged reported CO₂ emissions vs. <1 hr power plant flights that may cover parts of two reported consecutive hourly values). We estimate the total reported uncertainty by summing 456 457 in guadrature the Peischl estimate and the relative difference between two reported consecutive 458 hourly CO₂ emission values closest to the time of the power plant sampling. The aircraft frequently 459 encountered power plants during oil & gas monitoring campaigns, but usually did not have the flight time to perform a full emissions characterization of the power plant. Here we limit our comparison 460 to days when the aircraft performed a minimum of 10 laps around the plant, thus excluding the quick 461 462 fly-bys where uncertainties would be unacceptably large. The results are presented in Table 4 and 463 indicate very good agreement between Gauss's method and the reported CO₂ emissions with the averaged difference being 10.6%. A comparison plot of the reported versus measured CO₂ emissions 464 465 is shown in Figure 10. The average difference between the reported and measured emissions for the 466 5 power plants is 11%.

467

468 **5 Conclusion**

This technique was developed out of the necessity to identify and quantify individual well pads in an 469 extensive oil and gas production field. Consequently the frequent tracking of the upwind and 470 471 downwind side of the source provides a very accurate determination of the location and magnitude 472 of a given emission site. The main uncertainty arises from the effluent below the lowest flight altitude, but this is minimized by targeting a downwind distance determined by LES studies to provide 473 474 very little change in the plume flux divergence from the lowest loop to the ground. In addition to the 475 controlled release experiments, hundreds of sites have been measured using this technique with 476 varying levels of success. Ideal conditions include flat terrain, ample sunlight to promote vertical mixing, consistent winds, and no nearby competing sources. Under optimal conditions we have 477 478 demonstrated that measurement uncertainties are guite low, often better than 10%. As the 479 conditions deteriorate from the ideal to situations involving complex terrain, variable winds or nearby 480 upwind sources, measured uncertainties can increase to be as large or larger than the emission estimates themselves. In the worst case of stably stratified conditions (winter or night time), for 481 482 instance, the lack of vertical mixing may preclude the trace gases emitted at the surface from 483 reaching the minimum safe flight altitude. Complex terrain provides a challenge to the method 484 because the aircraft is unable to maintain a constant altitude above the ground. A possible future 485 refinement of this technique to be applied in complex terrain would be to fit the measurements of 486 both wind and mixing ratio to a uniform 3-dimensional surface surrounding the source, where the grid passes through the terrain and then integrate the flux normal to this irregular virtual flight path. 487 488 This would not assume level loop flight legs and would, in principle, account for individual loops being 489 flown at differing altitudes and thus more closely track mass continuity near the terrain elevation.

490

Simulation	Δx, Δy (m)	Lx, Ly (km)	∆Z (km)	∆t (s)	∆z (m)	CBL Depth (m)	CBL mean wind (ms ⁻¹)	w _* (ms ⁻¹)	$-z_i/L_{MO}$	X _{max}
UCD50A	50	8	2.5	0.30	8	750	2	0.92	210	4.5
UCD50B	50	8	2.5	0.30	8	600	3.8	0.86	73	3.6
UCD40	40	6	2.5	0.24	10	850	4.5	0.96	53	2.4

Table 1 - Domain and micrometeorological parameters for the three WRF-LES experiments in this study. *L* represents the Monin-Obukhov length.

Experiment	Date	Laps	Released	Estimated	Released	Estimated	Ethane
Location			CH_4	CH ₄	C_2H_6	C_2H_6	Difference
			kg hr⁻¹	kg hr⁻¹	kg hr⁻¹	kg hr⁻¹	
Colorado	2014-Nov-19	50	0.0	-0.1±0.3	5.5±0.5	5.6±2.9	+2%
Arkansas	2015-Oct-03	19	0.0	-3.4±12.3	8.1±0.8	10.0±6.1	+24%

Table 2 - Ethane controlled releases.

Experiment	Date	Laps	Released	Estimated	Released	Estimated	Methane
Location			CH_4	CH_4	C_2H_6	C_2H_6	Difference
			kg hr⁻¹	kg hr⁻¹	kg hr⁻¹	kg hr⁻¹	
Rio Vista	2014-Nov-03	37	13.9±2.8	12.8±8.5	1.2±0.5	0.6±0.4	-8%
Rio Vista	2014-Nov-04	27	13.9±2.8	11.5±3.2	1.2±0.5	0.5±0.3	-17%

Table 3 - Natural Gas controlled release

Power Plant	Date	Hour UTC	Laps	Reported CO_2 T hr ⁻¹	Estimated CO_2 T hr ⁻¹	Difference
Rocky Mountain	2014-Oct-06	20	19	99±14	111±24	13%
Energy						
Saint Vrain	2014-Oct-04	19	21	124±17	122±41	-1%
Pawnee	2014-Nov-19	20	14	575±81	555±160	-3%
Saint Vrain	2015-Sep-17	20	14	361±54	280±115	-23%
Four Corners	2015-Apr-11	18	12	1289±387	1119±343	-13%
Power Plant						

Table 4 - Power Plant estimates. The mid-point of the measurements (hours UTC) is indicated in the third column(Hour). The reported emissions from the hour before to the hour after that time were averaged to derive the"Reported" emissions in column 5. Emissions are reported in units of metric tons (T) per hour.

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Figure 1 - Map of the airplane flight pattern sampling a methane plume emanating from an underground storage facility. Wind direction is indicated by the white arrow and the methane mixing ratio is given by the color bar to the right. This flight was conducted on June 28, 2016 and took place between 12:46PM and 1:52PM LT at altitudes ranging from 91 m to 560 m with a loop diameter of approximately 3 km. The measured methane emission rate was 763±127 kg hr⁻¹.



Figure 2 - Graphical representation of the relative magnitude (%) of the contribution of the horizontal wind divergence to the horizontal advective terms in Equation 3, as a function of wind speed and source magnitude for methane, using a typical global background of 1.9 ppm and divergence of 10⁻⁵ s⁻¹.



Figure 3 Relative cross wind integrated concentrations of an effluent plume released at the surface in the UCD50B simulation. The data are averaged over 15 minutes of simulation time.



Figure 4 Average cospectrum of the outward directed component of the observed wind and the methane concentration from 70 laps around a point source near San Antonio, Texas. The peak at 10⁻² Hz corresponds to the period of the circle.



Figure 5 Dimensionless flux divergence profiles generated from averaging over 3 different WRF-LES runs using 30 time steps for each one. The horizontal flux per unit altitude ($d = F/\Delta z$) is normalized by the boundary layer height, z_i , and source strength, Q. The colored profiles are averages at various dimensionless distances, R=0.1, 0.2, 0.3, and 0.4 and the gray areas represent one standard deviation about the mean. The horizontal dashed lines are the approximate lowest safe flight altitude.



Figure 6 Rate of convergence toward the final leak rate estimation as a function of the number of loops for LES CASE UCD508. By 15 laps, the emissions estimate (blue line) has stabilized to 2.5 kg hr⁻¹ compared to the actual leak rate (red line) of 2.9 kg hr⁻¹. Dimensionless distance R = 0.25, 50 realizations. Grey area represents 1 standard deviation.



Figure 7 Averaged LES estimates for the Aerodyne case. This leak shows a slightly higher number of laps before convergence (~25 laps). This simulation was performed using the conditions for the Aerodyne controlled release near Denver, Colorado on November 19, 2014.



Figure 8 - Ethane horizontal transport profiles for the Aerodyne controlled releases near Denver, Colorado on November 19, 2014 (left) and in Bee Branch, Arkansas on October 3, 2015 (right). Blue dots represent individual flight loop measurements and the red circles represent the bin average values for altitude intervals represented by the red bars.



Figure 9 (Left plot) Time series of methane (blue) and ethane (red) along with (right plot) the geographic distribution of ethane (colorbar) and instantaneous winds (arrows) from a single flight loop during the second ethane controlled release.



Figure 10 Comparison of aircraft versus reported power plant emissions.

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