

# **Application of Gauss's Theorem to quantify localized surface emissions from airborne measurements of wind and trace gases**

Stephen Conley\*<sup>1,6</sup>, Ian Faloona<sup>1</sup>, Shobhit Mehrotra<sup>1</sup>, Maxime Suard<sup>1</sup>, Donald H. Lenschow<sup>2</sup>, Colm Sweeney<sup>4</sup>, Scott Herndon<sup>3</sup>, Stefan Schwietzke<sup>4,5</sup>, Gabrielle Pétron<sup>4,5</sup>, Justin Pifer<sup>6</sup>, Eric A. Kort<sup>7</sup> and Russell Schnell<sup>5</sup>

<sup>1</sup>Department of Land, Air, & Water Resources, University of California, Davis, 95616, USA

<sup>2</sup>Mesoscale and Microscale Meteorology Laboratory, National Center for Atmospheric Research, Boulder, Colorado, 80307, USA

<sup>3</sup>Aerodyne Research, Inc, Billerica, Massachusetts, 01821, USA

<sup>4</sup>Cooperative Institute for Research in Environmental Sciences, University of Colorado, Boulder, Colorado, 80305, USA

<sup>5</sup>NOAA Earth System Research Laboratory, Boulder, Colorado, USA

<sup>6</sup>Scientific Aviation, Inc., Boulder, CO, USA

<sup>7</sup>Climate and Space Sciences and Engineering, University of Michigan

Correspondence to: S. A. Conley (sacnley@ucdavis.edu)

1 **Abstract**

2 Airborne estimates of greenhouse gas emissions are becoming more prevalent with the advent of  
3 rapid commercial development of trace gas instrumentation featuring increased measurement  
4 accuracy, precision, and frequency, and the swelling interest in the verification of current emission  
5 inventories. Multiple airborne studies have indicated that emission inventories may underestimate  
6 some hydrocarbon emission sources in U.S. oil and gas producing basins. Consequently, a proper  
7 assessment of the accuracy of these airborne methods is crucial to interpreting the meaning of such  
8 discrepancies. We present a new method of sampling surface sources of any trace gas for which fast  
9 and precise measurements can be made and apply it to methane, ethane, and carbon dioxide on  
10 spatial scales of ~1000 m, where consecutive loops are flown around a targeted source region at  
11 multiple altitudes. Using Reynolds decomposition for the scalar concentrations, along with Gauss's  
12 Theorem, we show that the method accurately accounts for the smaller scale turbulent dispersion of  
13 the local plume, which is often ignored in other average "mass balance" methods. With the help of  
14 large eddy simulations (LES) we further show how the circling radius can be optimized for the  
15 micrometeorological conditions encountered during any flight. Furthermore, by sampling controlled  
16 releases of methane and ethane on the ground we can ascertain that the accuracy of the method, in  
17 appropriate meteorological conditions, is often better than 10%, with limits of detection below 5 kg  
18 hr<sup>-1</sup> for both methane and ethane. Because of the FAA mandated minimum flight safe altitude of 150  
19 m, placement of the aircraft is critical to preventing a large portion of the emission plume from  
20 flowing underneath the lowest aircraft sampling altitude, which is generally the leading source of  
21 uncertainty in these measurements. Finally, we show how the accuracy of the method is strongly  
22 dependent on the number of sampling loops, or time spent sampling the source plume.  
23

24 **1 Introduction**

25 Accurate national inventories of greenhouse gas emissions (primarily carbon dioxide (CO<sub>2</sub>), methane  
26 (CH<sub>4</sub>), and nitrous oxide (N<sub>2</sub>O)) is of paramount importance in developing strategies to understand  
27 global emissions. The multitude of sources, however, are so often highly variable in area, emission  
28 magnitude, height above ground, and duration that rigorous verification is exceedingly difficult.  
29 Nevertheless, measurement techniques have improved markedly in the past decade, and these are  
30 being employed to an unprecedented extent in an effort to evaluate and refine emission inventories  
31 (Nisbet and Weiss, 2010). Most so called "bottom-up" inventories are developed by aggregating  
32 statistical correlates of individual process emissions to such mapping variables as population density,  
33 energy consumption, head of cattle, etc., extrapolating to total emissions using a relatively small  
34 number of direct measurements. On the other hand, atmospheric scientists have long striven to use  
35 measurements from global surface networks, aircraft campaigns, and satellites to try to determine  
36 emissions based on the amounts and build-up rates of observed trace gases. Aircraft and satellites,  
37 the "top-down" approach, conveniently integrates the multitude of sources, but is heavily reliant on a  
38 detailed knowledge of atmospheric transport. Top-down methods also suffer from difficulties  
39 attributing sources and generalizing measurements made over a relatively short time period.  
40 Attempts to reconcile these two distinct methods on global (Muhle et al., 2010) and continental  
41 scales (Gerbig et al., 2003; Miller et al., 2013) have often indicated an apparent underestimation by  
42 the "bottom-up" methods of a factor 1.5 or more.

43 In principle, the aircraft top-down measurements can be conducted at all the atmospheric scales to  
44 better understand and identify the emissions at comparable scales. For long-lived greenhouse gases,  
45 which readily disperse throughout the atmosphere, the global scale is very instructive. The seminal  
46 experiment began with Keeling's acclaimed CO<sub>2</sub> curve [1960], and has continued through more  
47 contemporary techniques by Hirsch et al.(2006) and Neef et al. (2010) for CH<sub>4</sub> and N<sub>2</sub>O, respectively.  
48 At progressively smaller scales more details of the source strengths and apportionment can be made:  
49 from synoptic or continental scales which can help constrain national inventories (Bergamaschi et al.,  
50 2005) or specific biogeographic regions (Gallagher et al., 1994), to mesoscale investigations that  
51 estimate emissions from urban areas (Mays et al., 2009; Turnbull et al., 2011; Wecht et al., 2014) or  
52 specific oil and gas producing fields (Karion et al., 2013; Petron et al., 2014) and even down to  
53 individual point/area sources on the order of 10-100 m size (Denmead et al., 1998; Lavoie et al., 2015;  
54 Roscioli et al., 2015).

55 Aircraft in-situ measurements are particularly useful for "top-down" methods at the sub-mesoscale  
56 because they can be used to measure the air both upwind and downwind of a source region.  
57 However, deployments tend to be costly and thus sporadic. As far as we know, the aircraft methods  
58 used so far can be categorized into three types. First, there is the eddy covariance technique that is  
59 carried out at low altitudes wherein the vertical fluxes of gases carried by the turbulent wind are  
60 measured by tracking rapid fluctuations of both concentrations and vertical wind (Hiller et al., 2014;  
61 Ritter et al., 1994; Yuan et al., 2015). This method is generally thought to be the most direct, but it is  
62 limited to small footprint regions which must be repeatedly sampled for sufficient statistical  
63 confidence, requires a sophisticated vertical wind measurement, and can be subject to errors due to  
64 flux divergence between the surface and the lowest flight altitude and acceleration sensitivity of the  
65 gas sensor. The second, and by far the most common approach is what chemists usually refer to as  
66 "mass balance" and what is known in the turbulence community as a "scalar budget" technique.  
67 Many different sets of assumptions and sampling strategies are employed, but the overall goal is to  
68 sample the main dispersion routes of the surface emissions as they make their way into the overlying  
69 atmosphere after first accumulating near the surface. The scales that can be addressed by this  
70 method are from a few kilometers (Alfieri and Blanken, 2012; Hacker et al., 2016; Hiller et al., 2014;  
71 Tratt et al., 2014) to tens of kilometers (Caulton et al., 2014; Karion et al., 2013; Wratt et al., 2001) to  
72 even potentially hundreds of kilometers (Beswick et al., 1998; Chang et al., 2014), and this approach  
73 has been the focus of recent measurements in natural gas production basins. These basins present a  
74 source apportionment challenge in that emissions from multiple sources (agriculture, oil & gas wells,  
75 geologic seepage, etc.) commingle as the air mass travels across the basin. The third method of  
76 source quantification is to reference measurements of the unknown trace gas to a reference trace  
77 gas with a metered release (tracer) or otherwise known emission rate and assume that the tracer and  
78 the scalar of interest have the same diffusion characteristics. Typically this tracer release technique is  
79 applied to small scales of tens to hundreds of meters (Czepiel et al., 1996; Lamb et al., 1995; Roscioli  
80 et al., 2015), but the principle has been attempted at the basin (Peischl et al., 2013) and continental  
81 (Miller et al., 2012) scales using a reference trace gas with a suitable known emission rate such as CO<sub>2</sub>  
82 or CO.

83 The airborne mass balance flight strategies can be grouped into three basic patterns: a single height  
84 transect around a source assuming a vertically uniformly mixed boundary layer (Karion et al., 2013);  
85 single height upwind/downwind (Wratt et al., 2001) or sometimes just downwind flight legs (Conley

86 et al., 2016; Hacker et al., 2016; Ryerson et al., 1998); and multiple flight legs at different altitudes,  
87 (Alfieri et al., 2012; Gordon et al., 2015; Kalthoff et al., 2002); or just a 'screen' on the downwind face  
88 of the box (Karion et al., 2015; Lavoie et al., 2015; Mays et al., 2009).

89 Here we describe a new airborne method borne out of a necessity to identify and quantify source  
90 emissions to within 20% accuracy in a large heterogeneous field of potential sources. The novel  
91 technique applies an aircraft flight pattern that circumscribes a virtual cylinder around an emission  
92 source and, using only observed horizontal wind and trace gas concentrations, applies Gauss's  
93 Theorem to estimate the flux divergence through that cylinder. By integrating the outward horizontal  
94 fluxes at each point along the circular flight path, the flux contributions from enclosed sources can be  
95 accounted for. Making an accurate estimate, however, requires the selection of an appropriate  
96 circling radius based on the micrometeorological conditions inferred in flight from measurements  
97 onboard the aircraft. The pattern must be far enough downstream for the plume to mix sufficiently in  
98 the vertical, yet not so far that the trace gas plume enhancements do not stand out sufficiently from  
99 the background concentration.

100 In this study we first present the general analytical method used to derive emission estimates using  
101 airborne measurements. Next, we investigate the structure of a generalized dispersing plume using  
102 large-eddy simulation (LES) to better understand the optimal sampling strategies for quantifying  
103 near-surface gas sources. Because the wind fields of turbulent flows cannot be predicted in detail, we  
104 do not attempt to compare specific features of our observations with specific LES results, but rather  
105 we use the numerical experiments to guide the development of the observational methodology. For  
106 example, by investigating the LES flux divergence profiles in the layer below the lowest flight altitude,  
107 we are able to estimate the contribution of this unmeasured component to the overall source  
108 strength. We then evaluate the accuracy of the approach using coordinated planned release  
109 experiments and by applying the method to CO<sub>2</sub> emitted from several power plant plumes to  
110 compare with reported emissions.

111

## 112 **2 Data Collection**

### 113 **2.1 Airborne Instrumentation**

114 The airborne detection system is flown on a fixed wing single-engine Mooney aircraft, extensively  
115 modified for research as described in Conley et al. (2014). Ambient air is collected through ~5 m of  
116 tubing (Kynar, Teflon and stainless steel) that protrudes out of backward-facing aluminum inlets  
117 mounted below the right wing. In-situ CH<sub>4</sub>, CO<sub>2</sub>, and water vapor are measured with a Picarro 2301f  
118 cavity ring down spectrometer as described by Crosson (2008), which is operated in its precision  
119 mode at 1 Hz. In-situ ethane (C<sub>2</sub>H<sub>6</sub>) is measured with an Aerodyne Methane/Ethane tunable diode  
120 infrared laser direct absorption spectrometer (Yacovitch et al., 2014). There is a 5-10 second time lag  
121 in both analyzers that depends on the flow rate and tubing diameter. We use a 1/8" OD (3.175 mm)  
122 stainless line for the Picarro (~0.2 slpm flow rate), and a 6.3 mm (¼ in) Teflon line for the CH<sub>4</sub>/ C<sub>2</sub>H<sub>6</sub>  
123 spectrometer (~4 slpm flow rate). This results in lag times of ~5 s for the Aerodyne and ~10 s for the  
124 Picarro. The lag time for the Picarro is calculated using a "breath test", whereby we exhale into the air  
125 inlet and measure the time required for the CO<sub>2</sub> measurement to peak. The ethane lag time is

126 adjusted to maximize the correlation between the ethane and Picarro methane time series in plumes  
127 where both gases are emitted. Both lag times are slightly dependent on pressure, i.e., with a typical  
128 altitude change of  $\sim 1$  km, the change in lag time is less than 10%, and is inconsequential when  
129 applying this method within a few hundred meters from the surface. The horizontal wind speed and  
130 direction, sampled at 1 Hz, is based on a standard aircraft pitot-static pressure airspeed measurement  
131 and a dual GPS compass that determines aircraft heading and ground speed. The accuracy of the  
132 horizontal wind measurement is about  $0.2 \text{ m s}^{-1}$  (Conley et al., 2014). The horizontal wind is  
133 calibrated periodically by flying  $\sim 5$  km L-shaped patterns in the free troposphere; a heading rotation  
134 and airspeed adjustment is made to the wind calculation to minimize the dependence of the wind on  
135 aircraft heading. These adjustments typically amount to less than  $2^\circ$  rotation and 3% adjustment of  
136 the airspeed. In flying the tight circle patterns described below, the pilot does not adjust the rudder  
137 trim to use the same calibration coefficients in the wind measurement calculation throughout the  
138 flight.

## 139 **2.2 Large Eddy Simulations**

140 In order to study the plume behaviour of surface emissions as it relates to sampling in the stacked  
141 circles, we use the LES module of *WRF V3.6.1*. *WRF-LES* explicitly resolves the largest turbulent eddies  
142 by filtering the Navier-Stokes scalar conservation equations at some scale in the inertial subrange,  
143 and allowing the smaller motions beyond the cut-off to be modeled using a sub-grid (also called a  
144 subfilter) scale turbulence parameterization that is based on properties of the larger-scale, resolved  
145 flow. Because the aircraft data is typically sampled at 1 Hz and the true airspeed is around  $70 \text{ m s}^{-1}$ ,  
146 we use an LES horizontal grid size roughly half (40 and 50 m) the distance between aircraft data  
147 samples. Because periodic lateral boundary conditions are imposed on the *WRF-LES* variables, care  
148 must be taken to ensure that the effluent does not reach the lateral boundaries of the simulation  
149 domain. On the other hand, *WRF-LES* does not allow for parallelized computation, making the  
150 simulations quite expensive in terms of computation time. We therefore struck a balance between a  
151 large enough domain in horizontal extent (6 and 8 km) such that the effluent would not reach the  
152 downwind boundary before the end of our simulation, while maintaining a grid size small enough to  
153 resolve scales of the aircraft observations. The vertical domain needs to be large enough to  
154 encompass a developing convective boundary layer (CBL), while at the same time containing  
155 substantial free tropospheric flow above to serve as a reservoir that can feed momentum and free-  
156 tropospheric scalars to the CBL. Moreover, the stable region (potential temperature lapse rate  $d\theta/dz$   
157 =  $5 \text{ C km}^{-1}$ ) between the CBL inversion base and the top of the domain had to be large enough to  
158 damp any wave activity before it could reflect off the upper boundary and create spurious motions  
159 throughout the domain.

160 The standard *WRF-LES* module is not set up to allow for effluent release, so we implemented a  
161 modified version of the *WRF* source code (*S.-H. Chen, personal communication*) that includes a  
162 surface effluent release with a specified position and release rate. Three different convective  
163 simulations were run with varying resultant mean wind speeds in the boundary layer, and each was  
164 allowed 4-5 hours to 'spin-up' dynamically before the effluent was released at a rate of between  $2.9$ -  
165  $3.5 \text{ kg hr}^{-1}$ . The exact release time was selected to give reasonably stationary CBL depths and  
166 turbulent kinetic energy. The conditions for the three simulations are listed in Table 1, and based on

167 the different wind speeds they span moderate to strongly convective boundary layers ( $-z_i/L_{MO}$  from  
168  $\sim 50$  to  $\sim 200$ , where  $L_{MO}$  is the Monin-Obukhov length and  $z_i$  is the CBL depth.)

### 169 3 Methods

#### 170 3.1 Theory of Measurement using Gauss's Theorem

171 We use an integrated form of the scalar budget equation for a passive, conservative scalar in a  
172 turbulent fluid to estimate the emission of a gas of interest within a cylindrical volume  $V$ . The volume  
173 is circumscribed by a series of closed aircraft flight paths (typically circular) flown around the emission  
174 source over a range of altitudes. The altitudes encompass the lowest safe flight level (usually 75-150  
175 mAGL) up to an altitude where no discernable change in the trace gas mixing ratio,  $\chi$ , is observed  
176 around the flight loop,  $z_{max}$ . The scalar in our case is the mass concentration (i.e., density of a  
177 chemically unreactive species in a turbulent flow field,  $\mathbf{u} = u\mathbf{i} + v\mathbf{j} + w\mathbf{k}$ ; its Reynolds decomposition is  $c =$   
178  $C + c'$ , where  $C$  is the mean concentration around each loop and  $c'$  is the departure from the loop  
179 mean. Figure 1 shows an actual example of the effluent sampled by the aircraft in a sequence of  
180 stacked paths  $l$  that circumscribe an area,  $A$ , enclosing the source in a volume,  $V$ . The effluent is  
181 carried downwind as it mixes upward in the CBL. A virtual surface circumscribed by the circular flight  
182 tracks is assumed enclosing the source and extending above the vertical extent of the plume so that  
183 there is no vertical transport above that level. To estimate the source strength, we start with the  
184 integral form of the continuity equation:

$$Q_c = \left\langle \frac{\partial m}{\partial t} \right\rangle + \iiint \nabla \cdot c\mathbf{u} dV \quad (1)$$

185 where  $\langle \rangle$  denotes an average over the volume  $V$ ,  $Q_c$  is the sum of the internal sources and sinks of  $c$   
186 within  $V$ , and  $m$  is the total mass of  $c$  within the volume  $V$ . At this point, we recognize that the flux  
187 divergence is composed of two terms

$$\nabla \cdot c\mathbf{u} = \mathbf{u} \cdot \nabla c + c\nabla \cdot \mathbf{u} \quad (2)$$

188

189 In section 3.2 we perform a scale analysis of the terms on the right-hand side (rhs) of equation 2 and  
190 show that the second term, which is proportional to the horizontal wind divergence, may be  
191 neglected under our normal flight protocol. This is fortunate because of the difficulty in accurately  
192 estimating the horizontal wind divergence from aircraft measurements (Lenschow et al., 2007). The  
193 vertical flux across the top of the flight cylinder is assumed to be zero and the flux from the bottom  
194 (ground) is the surface source we are measuring. This leaves us with only the horizontal flux, i.e.  $c\mathbf{u}_h$   
195 where  $\mathbf{u}_h (= u\mathbf{i} + v\mathbf{j})$ . In order to minimize the contribution from the horizontal wind divergence term,  
196 we remove the loop mean concentration,  $C$ , which does not alter the first term on the rhs because  
197  $\nabla C = 0$ , so that equation 2 becomes

$$\mathbf{u}_h \cdot \nabla c + c \nabla \cdot \mathbf{u}_h = \mathbf{u}_h \cdot \nabla(c'). \quad (3)$$

198 Next, we use Gauss's Theorem to relate the volume integral to a surface integral around the volume  
 199 that is sampled by the aircraft flight loops:

$$Q_c = \left\langle \frac{\partial m}{\partial t} \right\rangle + \iiint \nabla \cdot (c' \mathbf{u}) dV = \left\langle \frac{\partial m}{\partial t} \right\rangle + \oiint c' \mathbf{u} \cdot \hat{\mathbf{n}} dS \quad (4)$$

200 where  $S$  is the surface enclosing  $V$  and  $\hat{\mathbf{n}}$  is an outward pointing unit vector normal to the surface.

201 The surface integral can be broken into three elements: a cylinder extending from the ground up to a  
 202 level above significant modification by the emission, the ground surface circumscribed by a low-level  
 203 (virtual) circular flight path ( $z = 0$ ), and a nominally horizontal surface circumscribed by a flight path  
 204 above the level modified by the source ( $z = z_{max}$ ). We assume there is no significant flux (other than  
 205 the source of interest) into or out of the ground. Next, the surface integral is estimated solely from a  
 206 sequence of closed path integrals measured by the aircraft at multiple flight levels to estimate the  
 207 right side of Eq. 5 (blue dashed lines in Fig. 1),

$$\oiint c' \mathbf{u} \cdot \hat{\mathbf{n}} dS = \int_0^{z_{max}} \oint c' \mathbf{u}_h \cdot \hat{\mathbf{n}} dl dz, \quad (5)$$

208 where  $l$  is the flight path.

209

210 Combining Eqs. 4 and 5 leads to the result that is the basis for this measurement technique where a  
 211 series of horizontal loops at different altitudes are flown around a source region:

$$Q_c = \left\langle \frac{\partial m}{\partial t} \right\rangle + \int_0^{z_{max}} \oint c' \mathbf{u}_h \cdot \hat{\mathbf{n}} dl dz \quad (6)$$

212 Along each path the instantaneous outward flux is computed and summed over the loop to yield the  
 213 mean flux divergence via Gauss's Theorem. A temporal trend of the total mass within the volume  
 214  $\left(\frac{\partial m}{\partial t}\right)$  can be estimated from the flight data and added to the flux divergence integral to obtain the  
 215 emission rate.

216

### 217 3.2 Divergence Uncertainty

218 In order to estimate the relative error in the horizontal divergence term that we are eliminating, we  
 219 perform a scale analysis of the relative size of the two terms in Eq. 2, using some typical values of the  
 220 CBL parameters (convective velocity scale  $w_* = 1 \text{ m s}^{-1}$ , boundary layer depth,  $z_i = 1,000 \text{ m}$ ), and  
 221 sampling geometry (flying at a radius 1 km around a point source.) Taylor's (1922) statistical theory of  
 222 dispersion in a homogenous and stationary turbulent fluid predicts that the root mean square lateral  
 223 ( $\sigma_y$ ) and vertical ( $\sigma_z$ ) dispersion parameters increase linearly with time, or equivalently advection  
 224 distance, downwind in the near-field. Weil (1988) shows several examples of the growth of both of  
 225 these parameters downwind to be  $\sim 0.5w_*$ , which we use here for a rough estimate of a conical

226 plume spreading to quantify the dilution of the source's emission as it travels downwind to be  
227 intercepted by the aircraft. We use a large background mixing ratio characteristic of global CH<sub>4</sub> (~1.9  
228 ppmv), estimate the mean gradient by the plume concentration divided by the distance downwind,  
229 and assume a conservatively large horizontal wind divergence of 10<sup>-5</sup> s<sup>-1</sup>, which may in fact be typical  
230 for our small sampling region (Stull, 1988). The results are shown in Figure 2 and, for all but the  
231 smallest sources of a few kg hr<sup>-1</sup> and wind speeds below 1 m s<sup>-1</sup>, the divergence term is at least an  
232 order of magnitude smaller than the gradient term.

233

### 234 3.3 Applying the Theory to the LES Results

235 We calculated a comparable estimate of  $Q_c$  in the LES domain from the air density, concentration,  
236 and wind along circular flight paths as a virtual aircraft would fly. Willis and Deardorff (1976)  
237 generalized results of their convection tank experiments to downwind dispersion in the convective  
238 boundary layer (CBL) in terms of a dimensionless length scale  $X$ , the ratio of the horizontal advection  
239 time to the large eddy turnover time:

$$X = \frac{xw_*}{Uz_i} \quad (7)$$

240 where  $x$  is the downwind distance and  $U$  is the vertically averaged mean wind speed.

241 Figure 3 shows the crosswind-integrated concentration profile for the plume release in the UCD50B  
242 WRF-LES run as function of  $X$ , and normalized height,  $Z = z/z_i$ . Because of the time limitation due to  
243 the periodic boundary conditions, the plume is averaged for only ~15 minutes of simulation time  
244 which is just under a large eddy turnover time for the conditions of the run. The results displayed in  
245 Figure 3 are in good qualitative agreement with the results of Willis and Deardorff (1976) and Weil et  
246 al., (2012) save for the release being at the surface in our LES study, and at  $Z = 0.067$  for the above  
247 studies (see Fig. 1 and 2 of Weil et al., (2012)). Figure 3 shows the maximum concentration being  
248 lofted near  $X \sim 0.2$  and leveling off near  $Z \sim 0.8$  around  $X \sim 0.6$ ; beyond  $X > 1.5$  the plume is fairly well-  
249 mixed throughout the extent of the boundary layer.

### 250 3.4 The Upwind Directed Turbulent Flux

251 Horizontal turbulent fluxes are generally ignored in boundary layer budget studies due to the fact  
252 that while they are often sizeable in magnitude they do not change significantly over horizontal  
253 length scales under consideration (the horizontal homogeneity assumption). In the vicinity of a point  
254 source, however, this is not likely. The method outlined here estimates source emissions using a  
255 measured horizontal flux that incorporates wind and scalar measurements at 1 Hz sample rate,  
256 resolving scales of ~70 m (Conley et al., 2014), which should include nearly all of the turbulent  
257 contributions to the horizontal flux. Here we consider the nature of this turbulent flux and the error  
258 in emission estimates if only the mean transport were considered. We start with the budget equation  
259 for a horizontal scalar flux in a horizontally homogeneous turbulent flow where the molecular  
260 diffusive/viscous term has been neglected (Wyngaard, 2010),



261 
$$\frac{d\overline{cru'}}{dt} = -\overline{u'^2} \frac{\partial C}{\partial x} - \overline{u'w'} \frac{\partial C}{\partial z} - \overline{c'w'} \frac{\partial U}{\partial z} - \frac{\partial \overline{c'u'^2}}{\partial x} - \frac{\partial \overline{cru'w'}}{\partial z} - \frac{1}{\rho} \overline{c' \frac{\partial p'}{\partial x}} \quad (8)$$

262 where  $\rho$  is density and  $p'$  is the pressure fluctuation. We then assume stationarity and integrate  
 263 across the source from a point just upwind to a point within the plume and obtain

264 
$$\int_{0-}^x \frac{\partial \overline{cru'}}{\partial x'} dx' = -\frac{1}{U} \int_{0-}^x \left[ \overline{u'^2} \frac{\partial C}{\partial x'} + \overline{u'w'} \frac{\partial C}{\partial z} + \overline{c'w'} \frac{\partial U}{\partial z} + \frac{\partial \overline{c'u'^2}}{\partial x'} + \frac{\partial \overline{cru'w'}}{\partial z} + \frac{1}{\rho} \overline{c' \frac{\partial p'}{\partial x'}} \right] dx' \quad (9)$$

265 We further assume that although the scalar field is not homogeneous the flow field is, and the  
 266 background horizontal c-flux upwind is much smaller than the flux induced by the point source. This  
 267 results in an equation for the in-plume flux

268 
$$\overline{c'u'} = -\frac{1}{U} \left[ \overline{u'^2} (C_x - C_{bckg}) + \overline{u'w'} \int_{0-}^x \frac{\partial C}{\partial z} dx' + \int_{0-}^x \overline{c'w'} \frac{\partial U}{\partial z} + \int_{0-}^x \left( \frac{\partial \overline{c'u'^2}}{\partial x'} + \frac{\partial \overline{cru'w'}}{\partial z} + \frac{1}{\rho} \overline{c' \frac{\partial p'}{\partial x'}} \right) dx' \right] \quad (10)$$

269 The first three terms on the rhs of equation 10 are negative within the plume with their largest  
 270 magnitudes on the upwind side and diminishing downwind. On the largest, boundary layer filling  
 271 eddy scales the mean concentration of  $C$  downwind of a source is greater than in the upwind region,  
 272  $(C_x - C_{bckg}) > 0$ , and therefore the first term is negative, but decreases in magnitude with distance  
 273 downwind. However, this term is also positive on smaller scales within the plume where the mean  
 274 gradient is directed upwind towards the source, and is most likely responsible for the specious  
 275 intuitive impression that the horizontal turbulent flux should transport the plume downwind from the  
 276 source along with the mean wind advection. Moreover, the second and third terms on the rhs are  
 277 negative because the momentum flux,  $\overline{u'w'}$ , and mean vertical gradient,  $\partial C / \partial z$ , are negative while the  
 278 vertical turbulent flux,  $\overline{c'w'}$ , and wind shear,  $\partial U / \partial z$ , are positive. Based on the vertical concentration  
 279 profiles shown in Weil et al. (2012) (their Figures 3 & 4) it can be inferred that the vertical  
 280 concentration gradient,  $\partial C / \partial z$ , changes from negative to positive near  $X \sim 1$  and becomes negligible for  
 281  $X > 2-3$ . Similarly, in the third term, the vertical flux,  $\overline{c'w'}$ , decreases with fetch. Thus the counter-  
 282 directed flux ( $\overline{c'u'} < 0$ ) will fade with distance downwind. Wyngaard et al. (1971) have shown that  
 283 the third-moment, turbulent transport terms (4 and 5 on rhs of equation 10) in the horizontal heat  
 284 flux equation are small in the surface layer compared to the source terms, so we assume the same  
 285 holds for this scalar flux. Finally, the remaining pressure covariance term is believed to be the main  
 286 sink in the budget equation working to decorrelate the wind and the scalar as was shown in the  
 287 surface layer measurements of Wilczak and Bedard (2004). Therefore, the dominant production  
 288 terms for negative  $\overline{c'u'}$  (terms 1-3 on the rhs of 10) must be balanced by the pressure-correlation  
 289 term leading to an upwind-directed horizontal turbulent flux within the plume that decreases in  
 290 magnitude in the downwind direction.

291 This conclusion is supported by several previous studies. For example, in a wind-tunnel study of flux-  
 292 gradient relationships Raupach & Legg (1984) reported that the mean streamwise horizontal heat flux  
 293 calculated by multiplying the mean wind by the mean temperature overestimates the total heat flux  
 294 by approximately 10%, which suggests that the turbulent component of the horizontal heat flux is  
 295 negative; that is, the turbulent flux is upwind, directed counter to the mean flow. Other researchers  
 296 have reported an even larger disparity. Field experiments by Leuning et al. (1985) indicate that the  
 297 horizontal turbulent flux of a trace gas is  $\sim 15\%$  the mean flux, while Wilson and Shum (1992) suggest  
 298 it may be 20%. A recent LES study of particle dispersion over a plant canopy by Pan et al. (2014)

299 indicates magnitudes of 20% or more for the negative turbulent component of scalar fluxes in the  
300 vicinity of the source and decreasing with downwind fetch. We therefore conclude that when  
301 sampling a near-surface point source at  $X$  of order unity or less, if only the mean concentration  
302 difference is measured, a significant overestimate of the scalar source is likely to occur.

303 Further evidence of this is shown in the average cospectrum of the outward wind and concentration  
304 fluctuation observed in the flight loops in Figure 4. Because the integral of the cospectrum yields the  
305 total flux (scalar and wind covariance), this function is useful for examining the contributions to the  
306 overall flux from each of the scales of motion (represented by aircraft speed divided by frequency).  
307 The results shown in Figure 4 are from a  $\text{CH}_4$  point source with an estimated emission rate of  $46 \pm 7 \text{ kg}$   
308  $\text{hr}^{-1}$  which was circled 70 times at a dimensionless radius  $X$  of approximately 0.35. All cospectra of  
309 sampled sources have the same structure seen in Figure 4: there is an obvious peak at the mean flight  
310 loop frequency (usually  $\sim 100 \text{ s}$  period) followed by a smaller negative dip at higher frequencies within  
311 the meandering effluent plume. We believe this to be good evidence that our method captures this  
312 important component of the overall flux away from the source, which cannot be obtained with a  
313 traditional mean wind and an integrated concentration enhancement measurement that is so often  
314 employed in airborne source estimates (Ryerson et al., 2001; White et al., 1976).

### 315 **3.5 Choosing the Downwind Sampling Distance**

316 Determining the optimal sampling distance from the targeted point source is a balance of several  
317 factors. First, not surprisingly, the largest plume signal occurs closest to the source (Figure 3). Second,  
318 a high degree of confidence in the results is contingent upon sampling the majority of the plume at  
319 and above the lowest flight altitude, which only occurs downwind after a sufficient time has elapsed  
320 to loft the initially near-surface plume. And third, an attempt is made to sample the plume before it  
321 reaches the top of the boundary layer so that the vertical turbulent entrainment flux does not  
322 become appreciable violating the assumption of negligible flux through the top of the volume  $V$  as  
323 discussed in Equation 2. Finally, close to the source, the fluctuations in concentration will be very  
324 large, intermittent, at small scales, and highly variable.

325 To gain further insight into the second feature of the dispersing plume, Figure 5 shows the average  
326 horizontal flux divergence profiles derived from the three WRF-LES runs. Here we discuss a  
327 dimensionless  $R$ , which is identical to  $X$ , to emphasize that this scaled downwind distance from the  
328 source is a radius of a flight loop. The flux divergence values are made dimensionless by the boundary  
329 layer height,  $z_i$ , and the source emission rate,  $Q$ . Very close to the source, before the plume has had a  
330 chance to loft, the flux divergence profile exhibits a strong gradient below the minimum safe flight  
331 altitude, making that term difficult to measure directly, as shown in Figure 5. Farther from the source,  
332 the signal becomes weaker with increasing altitude and eventually becomes increasingly influenced  
333 by entrainment fluxes. We therefore seek a sampling distance that is far enough to allow sufficient  
334 vertical lofting yet close enough so that plume crossings are easily observable against the background  
335 variability and instrument noise, and are not yet influenced by entrainment mixing.

336 Based on the simulation results presented in Figure 5, we see the gradient below the lowest flight  
337 safe altitude typically becomes very small for  $R > 0.4$ , and therefore we attempt to target that  
338 distance to minimize the extrapolation error from the flight data to the surface. We do not currently

339 measure all the necessary parameters to estimate R in-flight (primarily the surface heat flux  $\overline{(w'\theta'_v)_0}$   
 340 which is required to estimate  $w^*$ ). Instead, we estimate  $w^*$  based on the observed boundary layer  
 341 height, standard deviation of wind speed, and a parameterization for  $w^* = \sigma_u / 0.6$  (Caughey and  
 342 Palmer, 1979).

### 343 **3.6 Minimum number of passes**

344 The atmospheric boundary layer is a turbulent medium, meaning that two passes across a plume at  
 345 the same altitude and distance downwind will likely make very different measurements of the trace  
 346 gases of interest. A natural question arises as to how many passes are required to develop a  
 347 statistically sound estimate of the emission rate. We investigate the number of passes required to  
 348 obtain a statistically robust estimate using the WRF-LES results and a controlled release experiment.  
 349 By calculating the horizontal flux divergences with a virtual airplane flying through the simulated  
 350 tracer field, and then randomly sampling the flux divergences from each of the legs and plotting the  
 351 resultant estimated emission rate as a function of the number of samples used we obtain the results  
 352 presented in Figure 6. The gray region around the red line mean represents the standard deviation of  
 353 estimates based on a random set of loops. Figure 7 shows results from an analysis of actual flight data  
 354 from the ethane controlled release test near Denver, Colorado on November 19, 2014. It is evident  
 355 from both the simulation data and the field data that a statistically stable estimate seems to be  
 356 achieved somewhere between 20-25 loops around the source.

### 357 **3.7 Discretization and Altitude Binning the Flux Divergence Data**

358 Measurements of the relevant scalars (e.g. CH<sub>4</sub>) and meteorological variables are sampled at discrete  
 359 time intervals. For our analyses, we interpolate all measurements including GPS (3 Hz), methane (1  
 360 Hz), temperature (1 Hz) and computed variables including horizontal wind (1 Hz), and air density (1  
 361 Hz) onto a synchronous 1 Hz time series. Next, we estimate the path integral for each individual loop  
 362 of the flux normal to the flight path by summing up the flux contributions times the sample length  
 363 around each loop and then summing over the height intervals,

$$Q_c = \left\langle \frac{\partial m}{\partial t} \right\rangle + \oint \mathbf{F}_c \cdot \hat{\mathbf{n}} dS = \frac{\Delta m}{\Delta t} + \sum_{z=0}^{z=z_t} (\sum_0^L (\rho \cdot u_n) \cdot \Delta s) \cdot \Delta z, \quad (9)$$

364 where  $\rho$  is the scalar air density,  $u_n$  is the wind speed normal to the flight path,  $\Delta s$  is the distance  
 365 covered during the 1 s time interval of each measurement and  $L$  is the distance covered in one  
 366 complete circuit. The outer summation sums each of the discrete vertical laps from the bottom ( $z =$   
 367 0) to the highest lap ( $z = z_t$ ). If all laps were sampled at equidistant altitudes, the total divergence  
 368 could be calculated as the average divergence of all laps multiplied by the top altitude. However,  
 369 because there is greater horizontal transport and variability at lower altitudes, as demonstrated by  
 370 the widening standard deviations approaching the surface in the theoretical flux divergence profiles  
 371 shown in Figure 4, more sampling laps at lower altitudes increase the statistical validity of the largest  
 372 horizontal transport values. To ensure that all altitudes are nearly equally weighted, we divide the  
 373 vertical range into six equally spaced bins, save for the lowest bin which is extended to the surface,  
 374 and then average the measurements from the laps within each bin. The total emission is the sum of  
 375 the flux times the path length in each bin multiplied by the bin width. We also performed 6 flights

376 where we sampled equally at all altitudes to allow a comparison of the direct average versus the  
377 binned results, and in all of these flights the values derived by the two methods agreed to within 5%.

### 378 **3.8 Error Analysis**

379 Our method assumes a stationary emission source. The leg-to-leg variability is primarily driven by the  
380 stochastic nature of turbulence (e.g. we may sample the plume on one lap and miss it on another). By  
381 aggregating the laps into vertical bins, we can use the standard deviation of the horizontal fluxes  
382 within each bin as an estimate of the uncertainty within that bin. Then the total uncertainty in the  
383 estimate of the flux divergence is simply estimated by adding up the individual bin uncertainties in  
384 quadrature. The first term on the rhs of Equation 6 is the time rate of change of the scalar mass  
385 within the cylindrical flight volume. This storage term is estimated by performing a least squares fit of  
386 the methane density with time and altitude. The resulting uncertainty in the time rate of change is  
387 then combined (summed in quadrature) with the uncertainty from the altitude bins to achieve a total  
388 uncertainty in the measurement.  
389

## 390 **4 Results and Discussion**

391

392 We use measurements from three sets of flights to characterize the accuracy of this estimation  
393 method. We flew 2 days measuring an ethane controlled release provided by Aerodyne Research,  
394 Inc., 4 days measuring a natural gas controlled release provided by the Pacific Gas & Electric Company  
395 (PG&E), and 6 power-plant flights where our estimates are compared with reported hourly power  
396 plant CO<sub>2</sub> emissions.  
397  
398

### 399 **4.1 Ethane Controlled Releases**

400 Two experiments with known/controlled ethane releases were performed in collaboration with the  
401 Aerodyne Mobile Laboratory team. Pure ethane was released and measured with a flowmeter by the  
402 Aerodyne ground crew. The Colorado site (November, 2014) was in a remote area approximately 105  
403 miles NE of Denver. This site was chosen because of the flat terrain and lack of other nearby ethane  
404 sources that could pollute the controlled release plume. The flux profiles for both releases are shown  
405 in Figure 8 and indicate that, in both cases, the aircraft successfully flew above the ethane plume  
406 (measurements tend toward zero with increasing altitude). An example of an individual lap is shown  
407 in Figure 9 and indicates a clear plume signal downwind of the release. As the aircraft climbs,  
408 eventually the signal disappears, as shown in the figure. Agreement was excellent, with the  
409 estimated emission just 2% over the actual controlled release rate. The second Aerodyne controlled  
410 release in Arkansas on October 3, 2015 was performed at a site surrounded by nearby emission  
411 sources and an elevation change (~70m) within the aircraft flight path. The aircraft-derived ethane  
412 emission estimate was 25% higher than the controlled release rate and the calculated uncertainty  
413 was significantly higher than on other sites (Table 2).

414 A significant upwind ethane source was observed during the Arkansas experiment. This source was  
415 evident on roughly half of the upwind passes, suggesting that techniques which rely on a limited  
416 number of upwind passes to characterize the background could have a large random error and thus  
417 erroneously estimate the upwind source strength. A similar problem would affect those techniques  
418 that employ a downwind transect, using the edges of that transect lying outside the plume to  
419 estimate the background concentration. These observations demonstrate the complication (and  
420 bias) that can arise from nearby sources. Since this method integrates all the emission sources in the  
421 area within the flight circle and a small distance upwind of the circle depending on the vertical  
422 mixing, estimates from Gauss's method may be biased high if there are sources within that area. The  
423 average error of the two ethane releases is 13%.  
424

## 425 **4.2 Natural Gas Controlled Releases**

426 In conjunction with PG&E, we performed two sets of two-day ground-level controlled release  
427 experiments from existing PG&E facilities, exactly one year apart. The first set was performed  
428 southeast of Sacramento near the town of Rio Vista, CA at the Rio Vista "Y" station and the second  
429 set near Bakersfield, CA. For the Rio Vista test, the release rate was not calibrated with a flow meter  
430 but, based on the size of the orifice and the upstream pressure, the release rate was estimated at  
431  $15.2 \pm 1.5 \text{ kg hr}^{-1}$ . This release rate is an estimate of the total gas being released which is a  
432 combination of primarily  $\text{CH}_4$  and  $\text{C}_2\text{H}_6$ . We use the regression fit of ethane to methane (averaging  
433 0.085 by mass) to estimate the actual release rate of each scalar.

434 In comparison with the  $\text{C}_2\text{H}_6$  controlled release,  $\text{CH}_4$  controlled releases suffer from the effect of small  
435 enhancements relative to the background concentration. During the Rio Vista release, the largest  
436 enhancement that we measured was 100 ppb, with 30-40 ppb being typical. Using a typical  
437 background level of 1.9 ppm, a 40 ppb enhancement represents 2% of the background. In contrast,  
438 for ethane the enhancements are as large or larger than the background. The results of the methane  
439 controlled release tests are shown in Table 3 and indicate aircraft-derived estimates within 17% of  
440 the controlled release rate. This large background results in increased uncertainty in the emission  
441 calculation. The average difference between the estimated emission and the calculated flow rate is  
442 13%.  
443

## 444 **4.3 Power Plant Flights**

445 Power plants in the U.S. are required to report  $\text{CO}_2$  emissions to EPA (<https://ampd.epa.gov/ampd>)  
446 on an hourly basis. The accuracy of the reported  $\text{CO}_2$  emissions has been determined to be  $\pm 10.8$ -  
447 11.0% based on reported U.S. average differences between Energy Information Administration (EIA)  
448 fuel-based estimates and EPA continuous emission monitoring-based estimates (Ackerman and  
449 Sundquist, 2008; Peischl et al., 2010; Quick, 2014). Also, Peischl et al. (2010) determined an accuracy  
450 of power plants reporting  $\text{CO}_2$  emissions in Texas of  $\pm 14.0\%$  based on differences between observed  
451 downwind  $\text{SO}_2/\text{CO}_2$  and  $\text{NO}_x/\text{CO}_2$  emission ratios and those reported via EPA continuous emission  
452 monitoring (Peischl et al., 2010). Here, we use the slightly larger uncertainty from Peischl et al.

453 (2010). Power plant emissions are “hot” gases and very buoyant, in contrast to a surface emission  
454 source that is typically not buoyant. An additional uncertainty arises from temporal emission  
455 variability (hourly averaged reported CO<sub>2</sub> emissions vs. <1 hr power plant flights that may cover parts  
456 of two reported consecutive hourly values). We estimate the total reported uncertainty by summing  
457 in quadrature the Peischl estimate and the relative difference between two reported consecutive  
458 hourly CO<sub>2</sub> emission values closest to the time of the power plant sampling. The aircraft frequently  
459 encountered power plants during oil & gas monitoring campaigns, but usually did not have the flight  
460 time to perform a full emissions characterization of the power plant. Here we limit our comparison  
461 to days when the aircraft performed a minimum of 10 laps around the plant, thus excluding the quick  
462 fly-bys where uncertainties would be unacceptably large. The results are presented in Table 4 and  
463 indicate very good agreement between Gauss’s method and the reported CO<sub>2</sub> emissions with the  
464 averaged difference being 10.6%. A comparison plot of the reported versus measured CO<sub>2</sub> emissions  
465 is shown in Figure 10. The average difference between the reported and measured emissions for the  
466 5 power plants is 11%.

467

## 468 **5 Conclusion**

469 This technique was developed out of the necessity to identify and quantify individual well pads in an  
470 extensive oil and gas production field. Consequently the frequent tracking of the upwind and  
471 downwind side of the source provides a very accurate determination of the location and magnitude  
472 of a given emission site. The main uncertainty arises from the effluent below the lowest flight  
473 altitude, but this is minimized by targeting a downwind distance determined by LES studies to provide  
474 very little change in the plume flux divergence from the lowest loop to the ground. In addition to the  
475 controlled release experiments, hundreds of sites have been measured using this technique with  
476 varying levels of success. Ideal conditions include flat terrain, ample sunlight to promote vertical  
477 mixing, consistent winds, and no nearby competing sources. Under optimal conditions we have  
478 demonstrated that measurement uncertainties are quite low, often better than 10%. As the  
479 conditions deteriorate from the ideal to situations involving complex terrain, variable winds or nearby  
480 upwind sources, measured uncertainties can increase to be as large or larger than the emission  
481 estimates themselves. In the worst case of stably stratified conditions (winter or night time), for  
482 instance, the lack of vertical mixing may preclude the trace gases emitted at the surface from  
483 reaching the minimum safe flight altitude. Complex terrain provides a challenge to the method  
484 because the aircraft is unable to maintain a constant altitude above the ground. A possible future  
485 refinement of this technique to be applied in complex terrain would be to fit the measurements of  
486 both wind and mixing ratio to a uniform 3-dimensional surface surrounding the source, where the  
487 grid passes through the terrain and then integrate the flux normal to this irregular virtual flight path.  
488 This would not assume level loop flight legs and would, in principle, account for individual loops being  
489 flown at differing altitudes and thus more closely track mass continuity near the terrain elevation.

490

Simulation	$\Delta x,$ $\Delta y$ (m)	$L_x, L_y$ (km)	$\Delta Z$ (km)	$\Delta t$ (s)	$\Delta z$ (m)	CBL Depth (m)	CBL mean wind ( $ms^{-1}$ )	$w_*$ ( $ms^{-1}$ )	$-z_i/L_{MO}$	$X_{max}$
UCD50A	50	8	2.5	0.30	8	750	2	0.92	210	4.5
UCD50B	50	8	2.5	0.30	8	600	3.8	0.86	73	3.6
UCD40	40	6	2.5	0.24	10	850	4.5	0.96	53	2.4

Table 1 - Domain and micrometeorological parameters for the three WRF-LES experiments in this study.  $L$  represents the Monin-Obukhov length.

Experiment Location	Date	Laps	Released $CH_4$ $kg\ hr^{-1}$	Estimated $CH_4$ $kg\ hr^{-1}$	Released $C_2H_6$ $kg\ hr^{-1}$	Estimated $C_2H_6$ $kg\ hr^{-1}$	Ethane Difference
Colorado	2014-Nov-19	50	0.0	$-0.1 \pm 0.3$	$5.5 \pm 0.5$	$5.6 \pm 2.9$	+2%
Arkansas	2015-Oct-03	19	0.0	$-3.4 \pm 12.3$	$8.1 \pm 0.8$	$10.0 \pm 6.1$	+24%

Table 2 - Ethane controlled releases.

Experiment Location	Date	Laps	Released $CH_4$ $kg\ hr^{-1}$	Estimated $CH_4$ $kg\ hr^{-1}$	Released $C_2H_6$ $kg\ hr^{-1}$	Estimated $C_2H_6$ $kg\ hr^{-1}$	Methane Difference
Rio Vista	2014-Nov-03	37	$13.9 \pm 2.8$	$12.8 \pm 8.5$	$1.2 \pm 0.5$	$0.6 \pm 0.4$	-8%
Rio Vista	2014-Nov-04	27	$13.9 \pm 2.8$	$11.5 \pm 3.2$	$1.2 \pm 0.5$	$0.5 \pm 0.3$	-17%

Table 3 - Natural Gas controlled release

Power Plant	Date	Hour UTC	Laps	Reported $CO_2$ T $hr^{-1}$	Estimated $CO_2$ T $hr^{-1}$	Difference
Rocky Mountain Energy	2014-Oct-06	20	19	$99 \pm 14$	$111 \pm 24$	13%
Saint Vrain	2014-Oct-04	19	21	$124 \pm 17$	$122 \pm 41$	-1%
Pawnee	2014-Nov-19	20	14	$575 \pm 81$	$555 \pm 160$	-3%
Saint Vrain	2015-Sep-17	20	14	$361 \pm 54$	$280 \pm 115$	-23%
Four Corners Power Plant	2015-Apr-11	18	12	$1289 \pm 387$	$1119 \pm 343$	-13%

Table 4 - Power Plant estimates. The mid-point of the measurements (hours UTC) is indicated in the third column (Hour). The reported emissions from the hour before to the hour after that time were averaged to derive the "Reported" emissions in column 5. Emissions are reported in units of metric tons (T) per hour.

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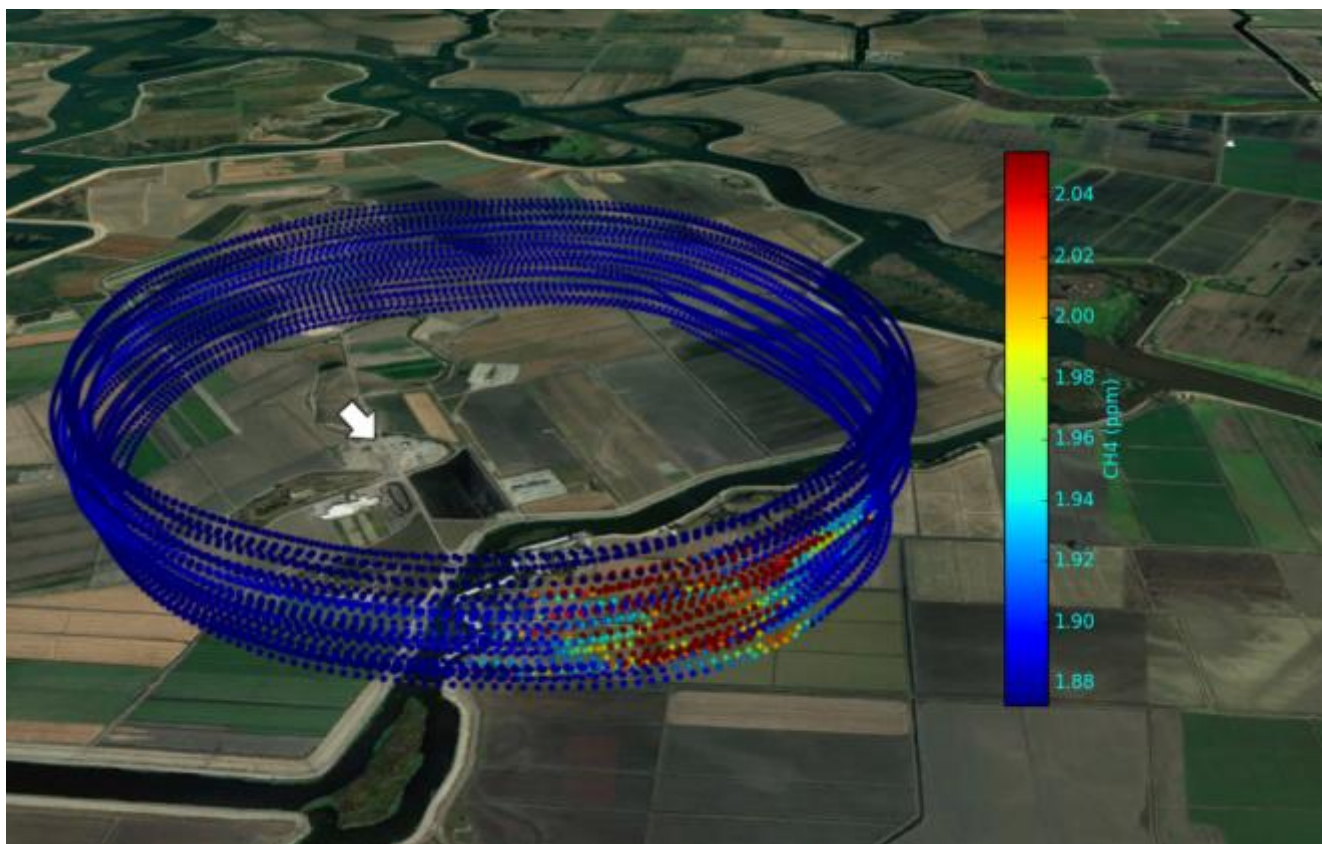


Figure 1 - Map of the airplane flight pattern sampling a methane plume emanating from an underground storage facility. Wind direction is indicated by the white arrow and the methane mixing ratio is given by the color bar to the right. This flight was conducted on June 28, 2016 and took place between 12:46PM and 1:52PM LT at altitudes ranging from 91 m to 560 m with a loop diameter of approximately 3 km. The measured methane emission rate was  $763 \pm 127$  kg hr<sup>-1</sup>.



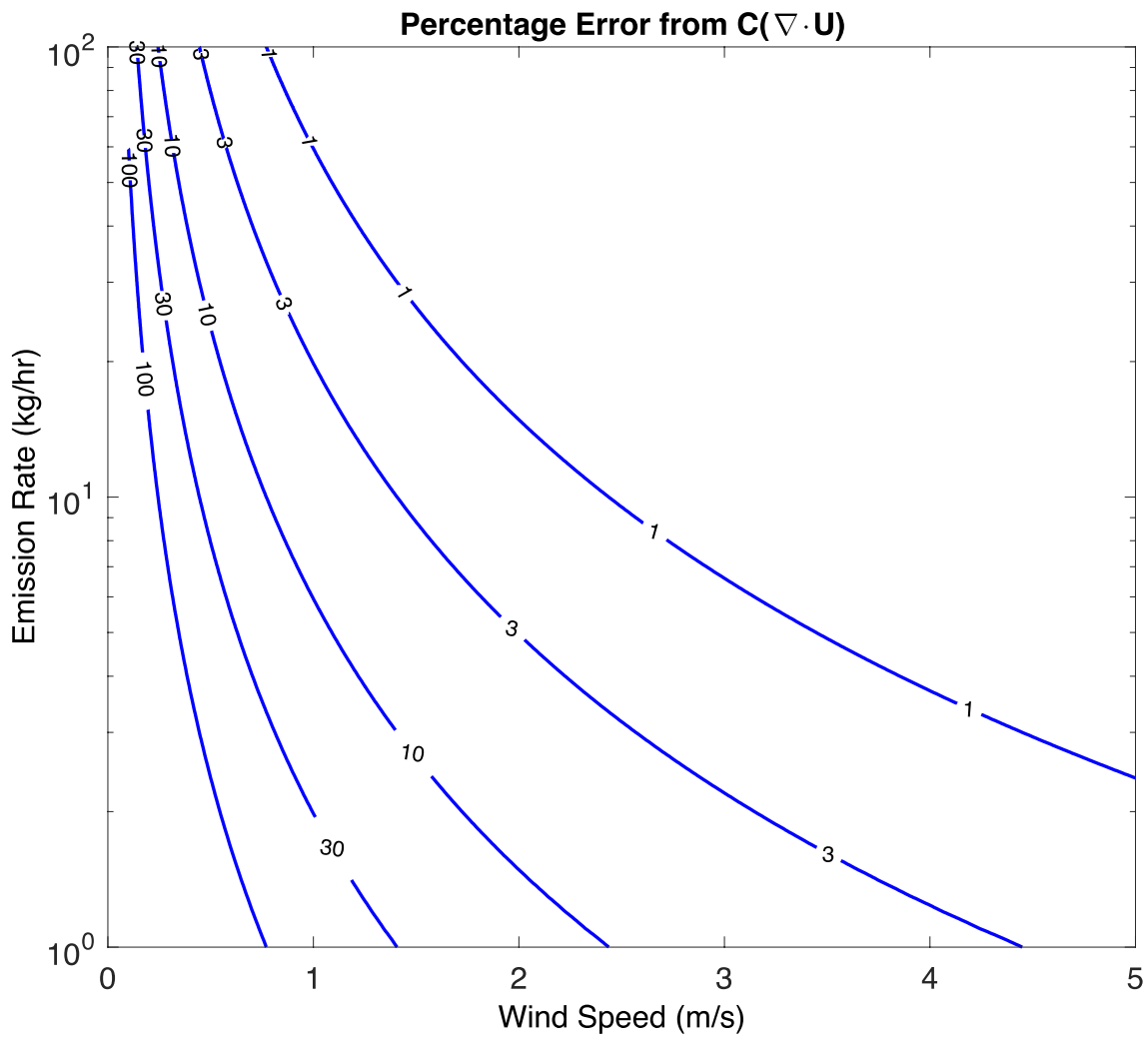


Figure 2 - Graphical representation of the relative magnitude (%) of the contribution of the horizontal wind divergence to the horizontal advective terms in Equation 3, as a function of wind speed and source magnitude for methane, using a typical global background of 1.9 ppm and divergence of 10<sup>-5</sup> s<sup>-1</sup>.

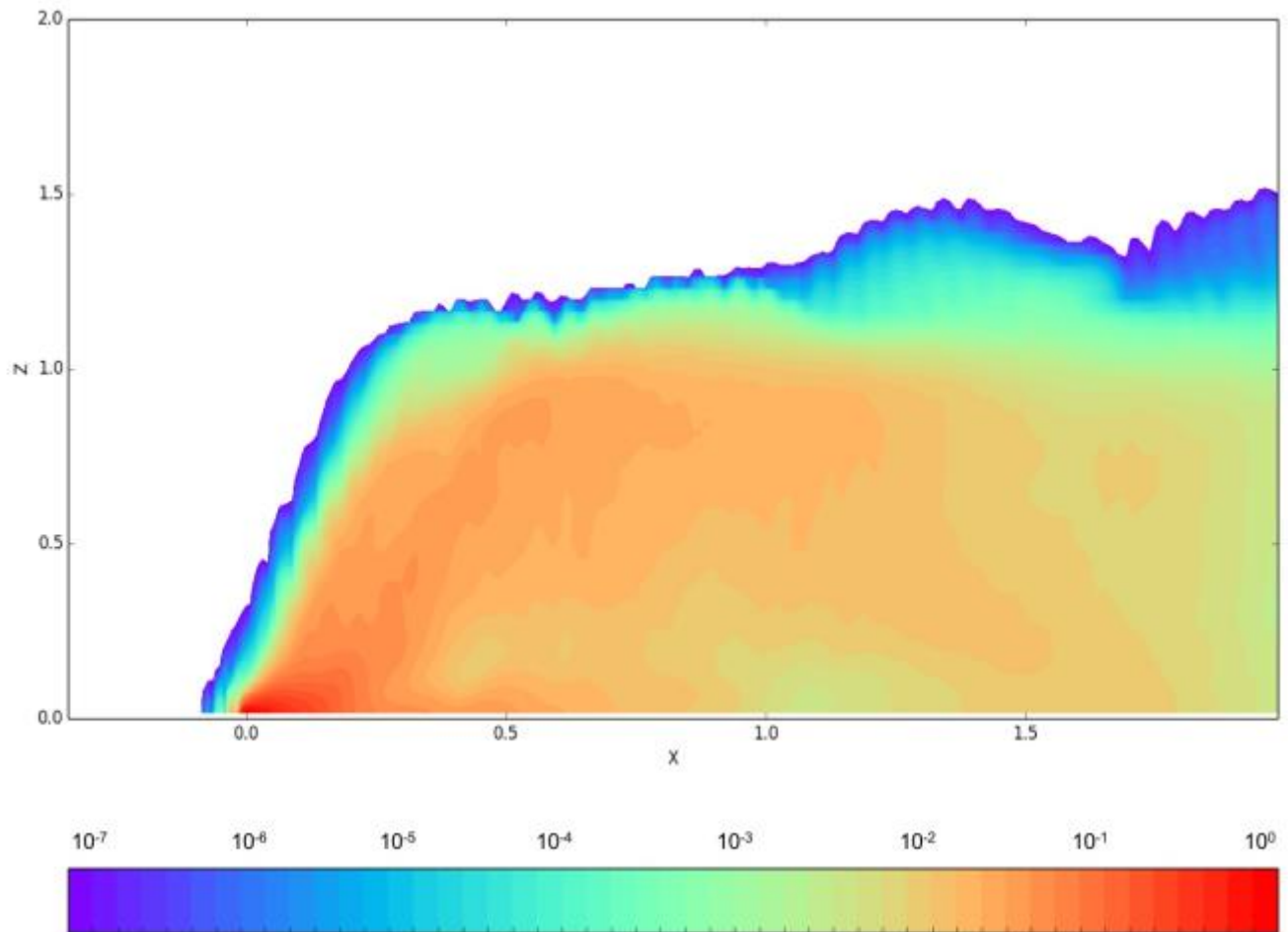


Figure 3 Relative cross wind integrated concentrations of an effluent plume released at the surface in the UCD50B simulation. The data are averaged over 15 minutes of simulation time and normalized by the maximum concentration.

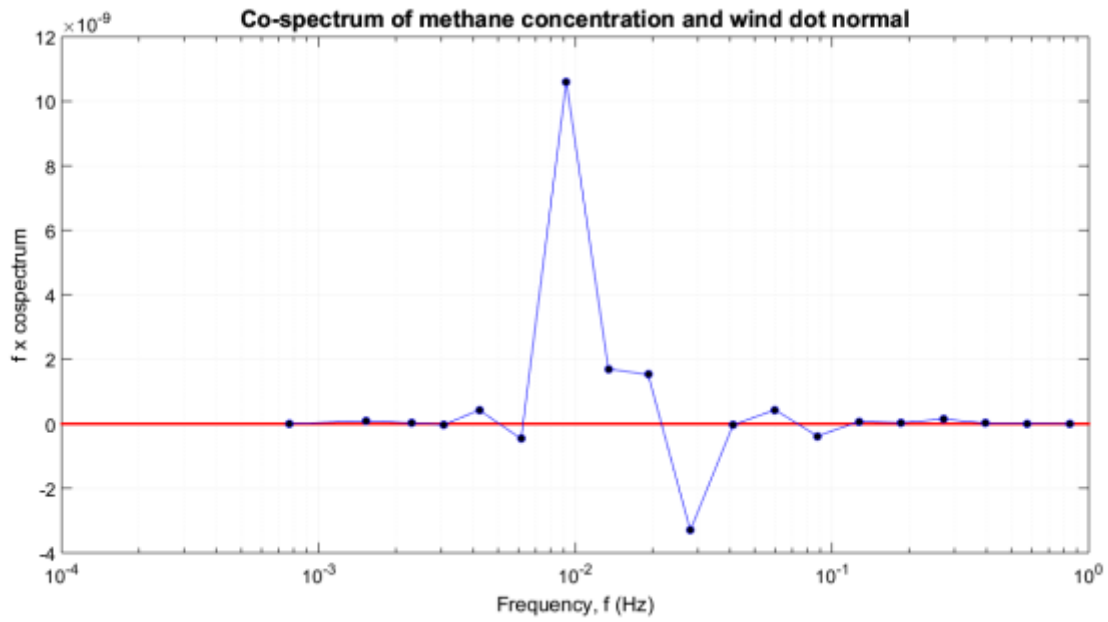


Figure 4 Average cospectrum of the outward directed component of the observed wind and the methane concentration from 70 laps around a point source near San Antonio, Texas. The peak at  $10^{-2}$  Hz corresponds to the period of the circle.

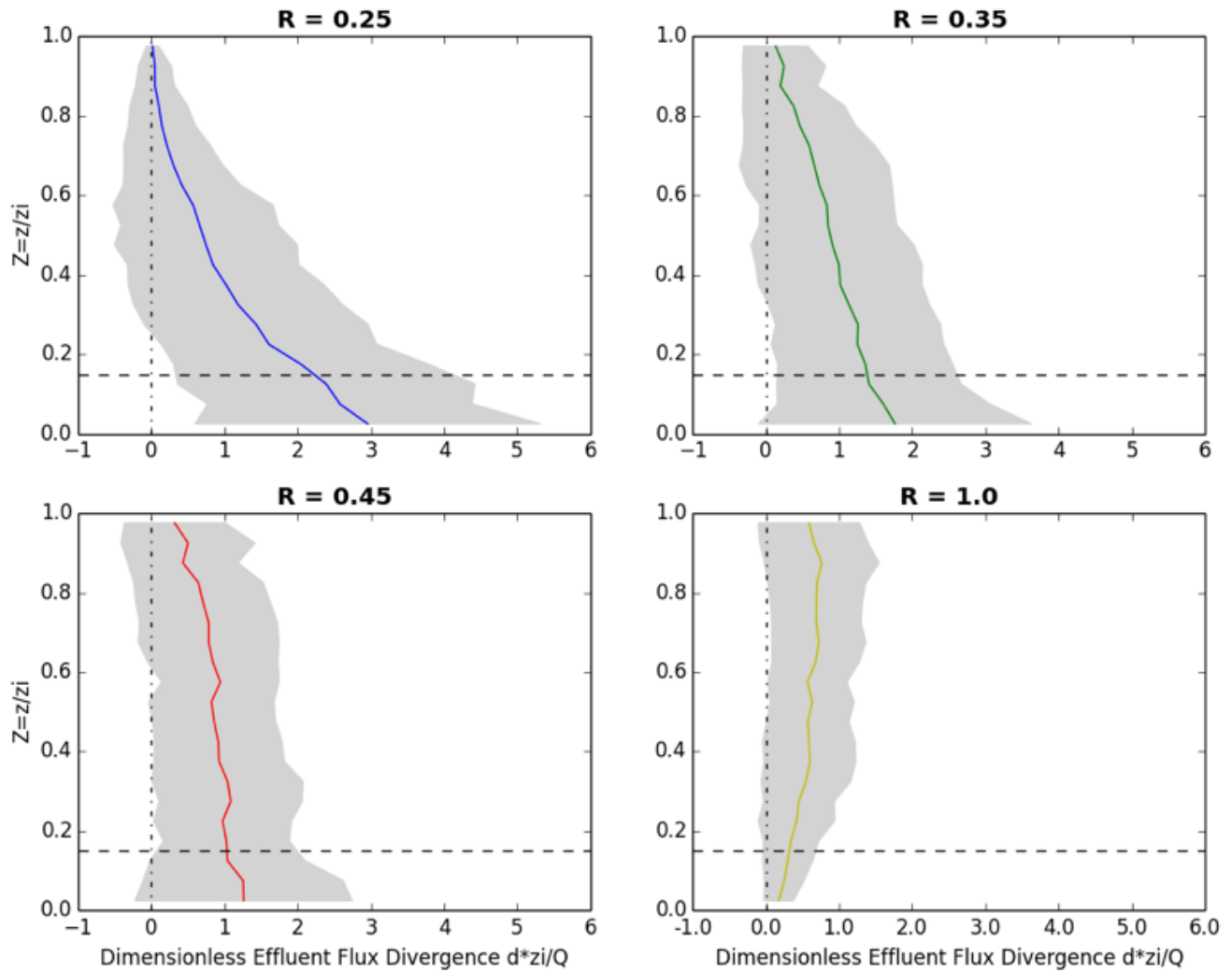


Figure 5 Dimensionless flux divergence profiles generated from averaging over 3 different WRF-LES runs using 30 time steps for each one. The horizontal flux per unit altitude ( $d = F/\Delta z$ ) is normalized by the boundary layer height,  $z_i$ , and source strength,  $Q$ . The colored profiles are averages at various dimensionless distances,  $R=0.1, 0.2, 0.3$ , and  $0.4$  and the gray areas represent one standard deviation about the mean. The horizontal dashed lines are the approximate lowest safe flight altitude.

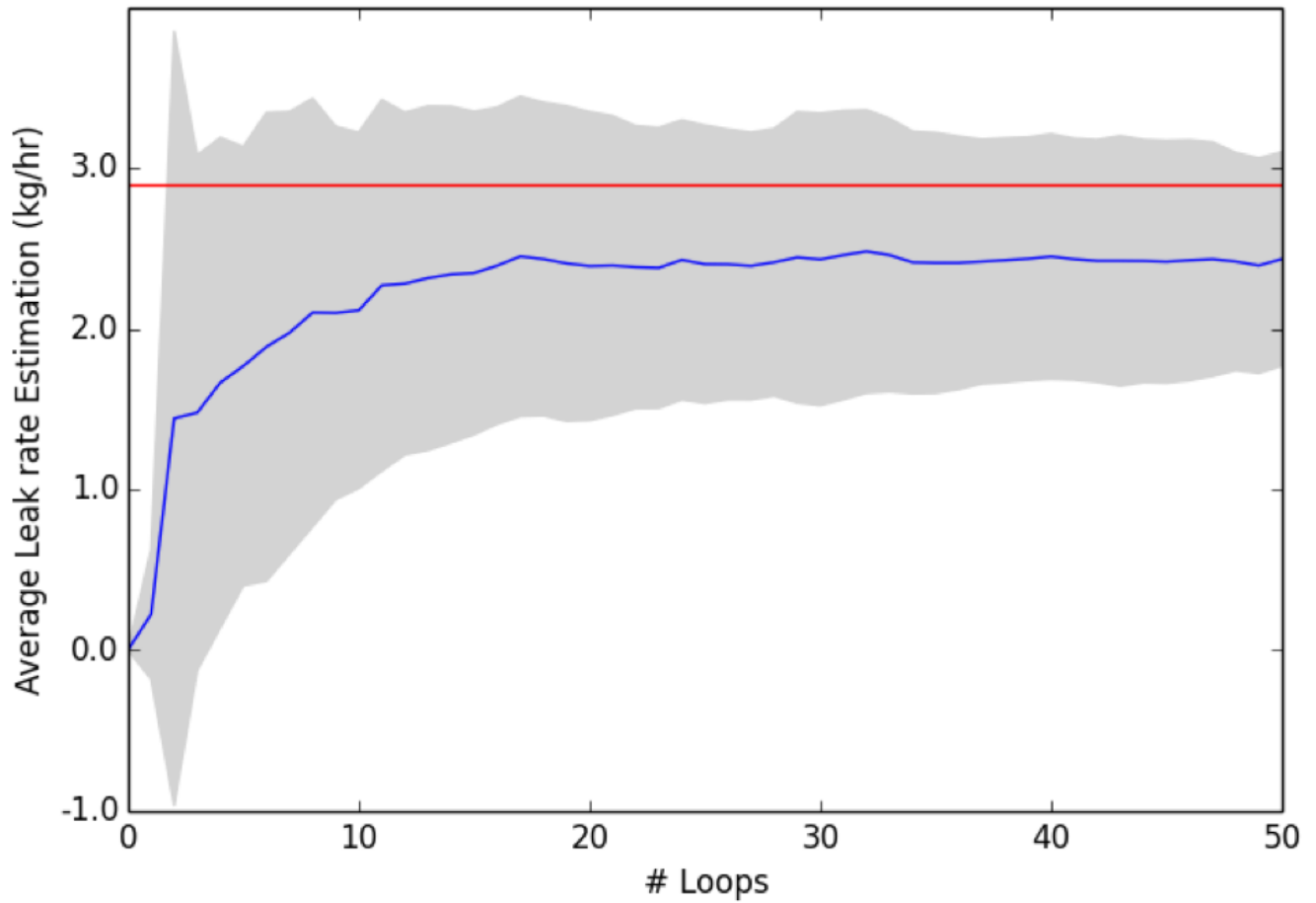


Figure 6 Rate of convergence toward the final leak rate estimation as a function of the number of loops for LES CASE UCD508. By 15 laps, the emissions estimate (blue line) has stabilized to 2.5 kg hr<sup>-1</sup> compared to the actual leak rate (red line) of 2.9 kg hr<sup>-1</sup>. Dimensionless distance R = 0.25, 50 realizations. Grey area represents 1 standard deviation.

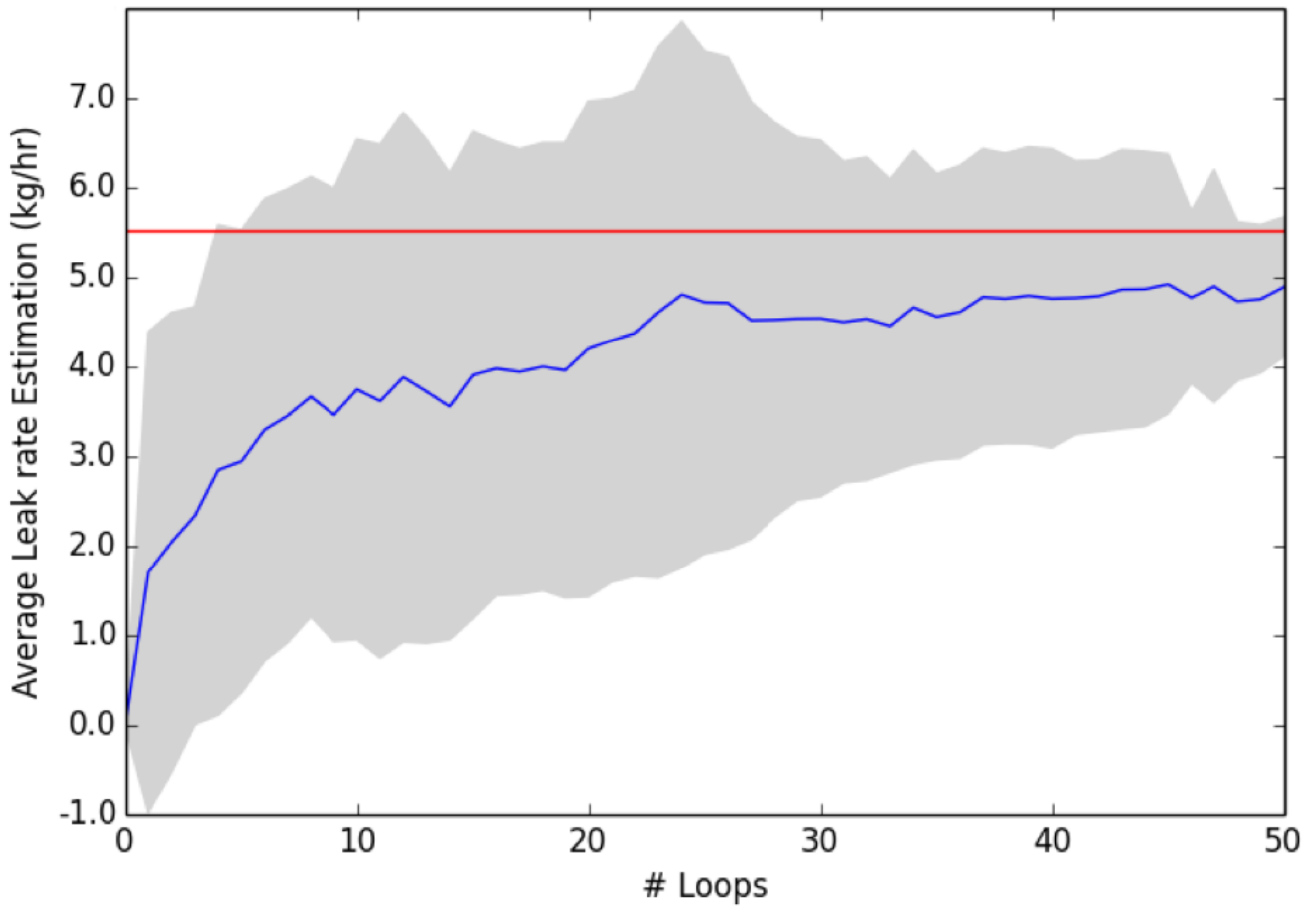


Figure 7 Averaged LES estimates for the Aerodyne case. This leak shows a slightly higher number of laps before convergence (~25 laps). This simulation was performed using the conditions for the Aerodyne controlled release near Denver, Colorado on November 19, 2014.

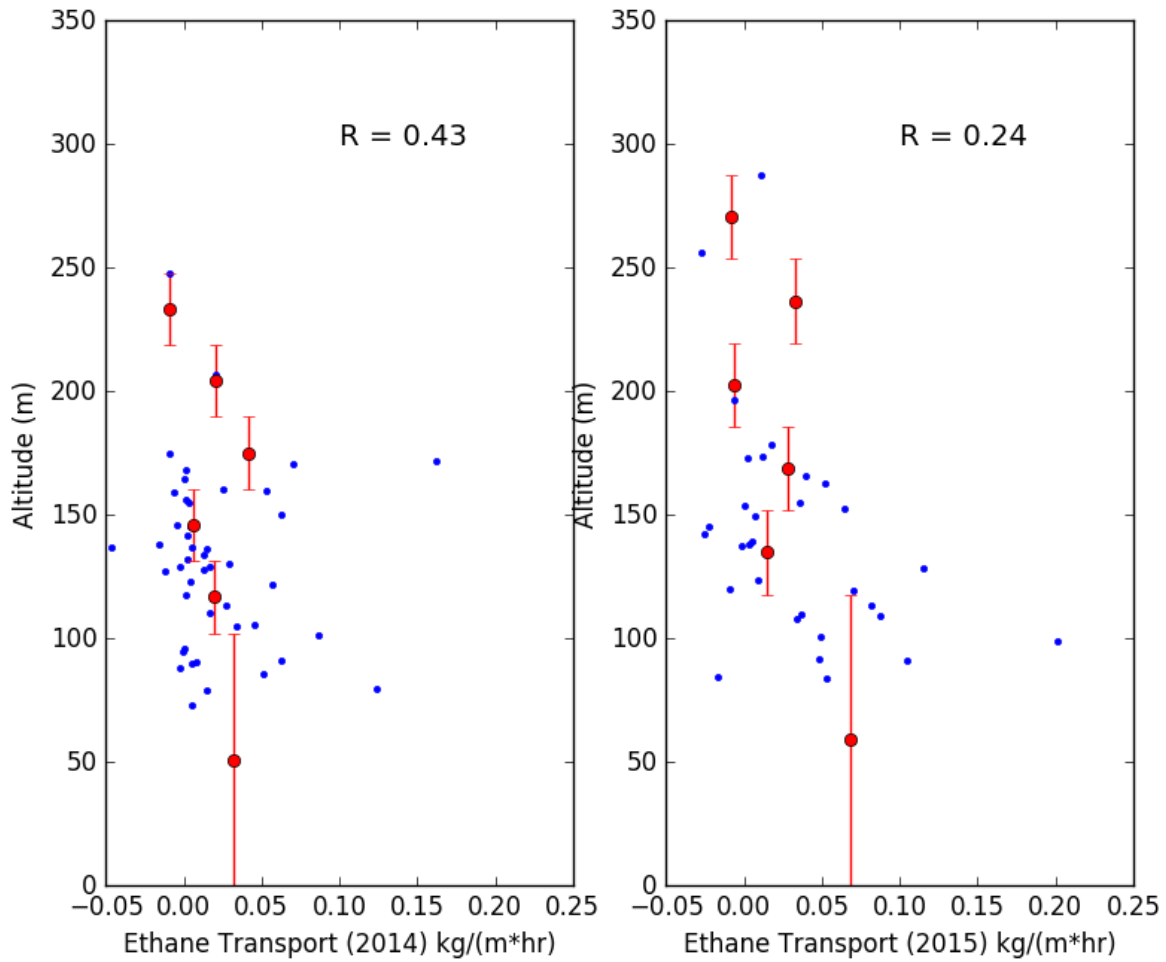


Figure 8 - Ethane horizontal transport profiles for the Aerodyne controlled releases near Denver, Colorado on November 19, 2014 (left) and in Bee Branch, Arkansas on October 3, 2015 (right). Blue dots represent individual flight loop measurements and the red circles represent the bin average values for altitude intervals represented by the red bars.

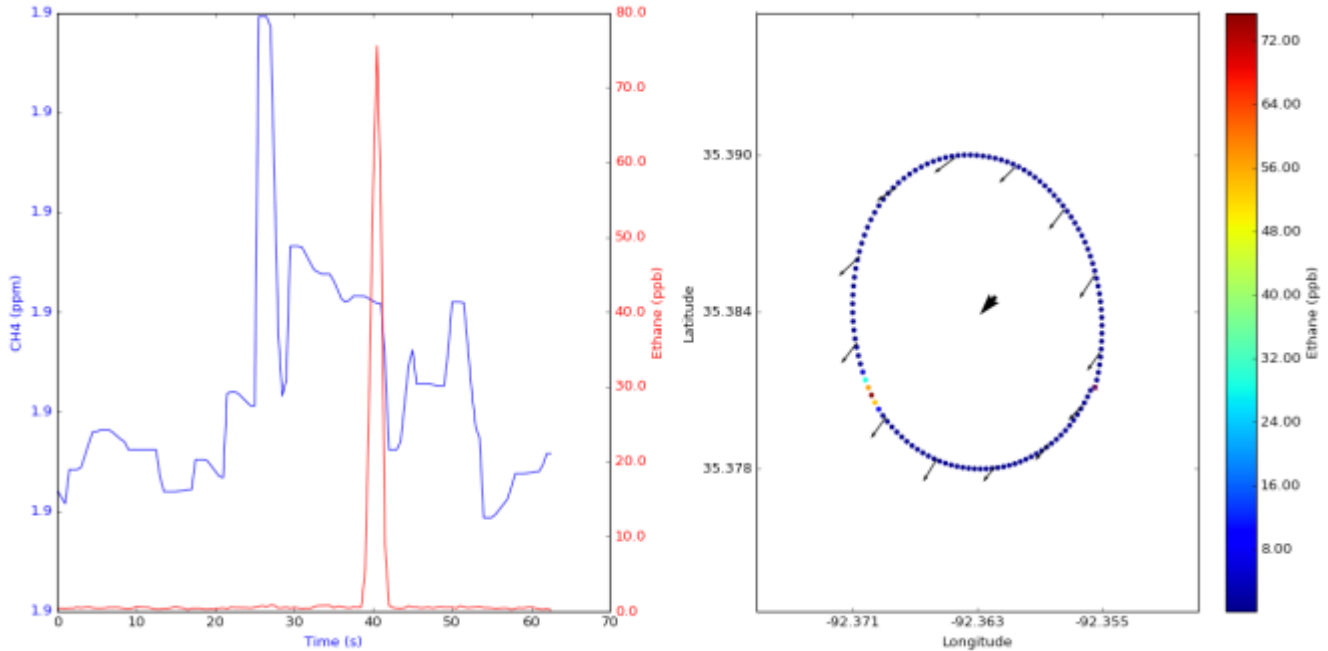


Figure 9 (Left plot) Time series of methane (blue) and ethane (red) along with (right plot) the geographic distribution of ethane (colorbar) and instantaneous winds (arrows) from a single flight loop during the second ethane controlled release.

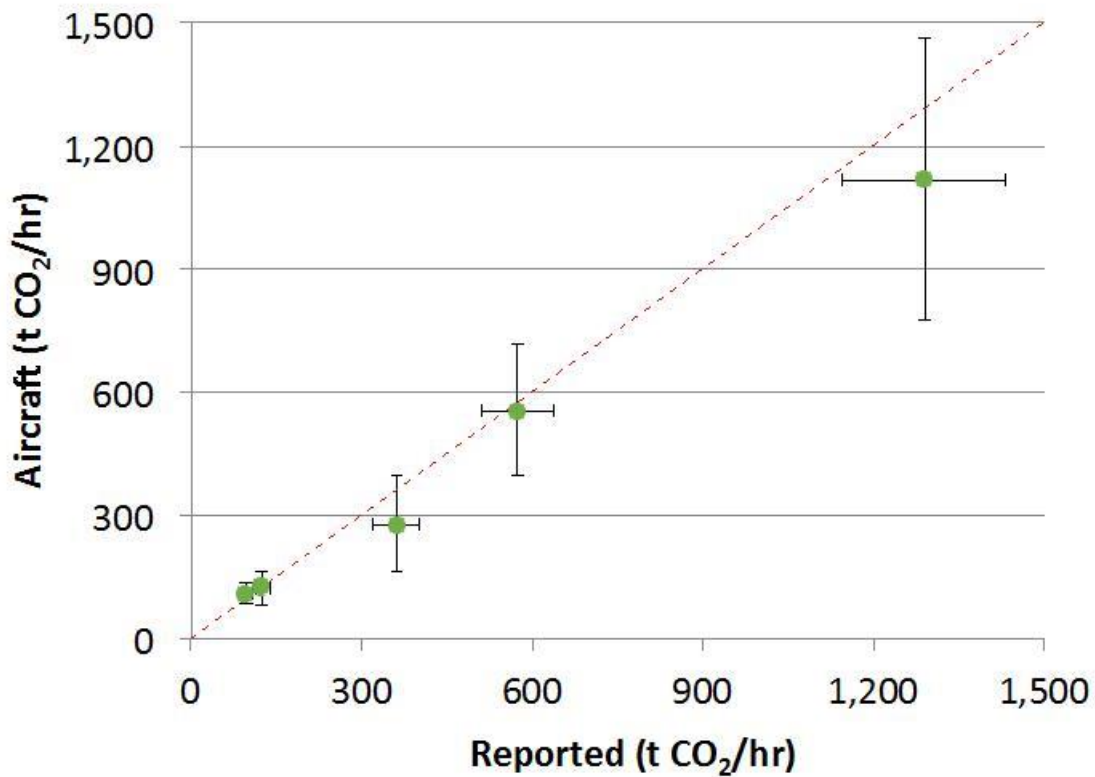


Figure 10 Comparison of aircraft versus reported power plant emissions.





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