

# 1                   **Comparison of Aircraft Measurements during GoAmazon2014/5 and** 2                   **ACRIDICON-CHUVA**

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27 **Abstract.** The indirect effect of atmospheric aerosol particles on the Earth's radiation balance  
28 remains one of the most uncertain components affecting climate change throughout the industrial  
29 period. The large uncertainty is partly due to the incomplete understanding of aerosol-cloud  
30 interactions. One objective of the GoAmazon2014/5 and ACRIDICON-CHUVA projects was to

31 understand the influence of emissions from the tropical megacity of Manaus (Brazil) on the  
32 surrounding atmospheric environment of the rainforest and to investigate its role in the life cycle  
33 of convective clouds. During one of the intensive observation periods (IOPs) in the dry season  
34 from September 1 to October 10, 2014, comprehensive measurements of trace gases and aerosol  
35 properties were carried out at several ground sites. In a coordinated way, the advanced suites of  
36 sophisticated in situ instruments were deployed aboard both the U.S. Department of Energy  
37 Gulfstream-1 (G1) aircraft and the German High Altitude and Long-Range Research Aircraft  
38 (HALO) during three coordinated flights on September 9, 21, and October 1. Here we report on  
39 the comparison of measurements collected by the two aircraft during these three flights. Such  
40 comparisons are challenging but essential for assessing the data quality from the individual  
41 platforms and quantifying their uncertainty sources. Similar instruments mounted on the G1 and  
42 HALO collected vertical profile measurements of aerosol particle number concentrations and size  
43 distribution, cloud condensation nuclei concentrations, ozone and carbon monoxide mixing ratios,  
44 cloud droplet size distributions, and downward solar irradiance. We find that the above  
45 measurements from the two aircraft agreed within the measurement uncertainties. The relative  
46 fraction of the aerosol chemical composition measured by instruments on HALO agreed with the  
47 corresponding G1 data, although the total mass loadings only have a good agreement at high  
48 altitudes. Furthermore, possible causes of the discrepancies between measurements on the G1 and  
49 HALO are examined in this paper. Based on these results, criteria for meaningful aircraft  
50 measurement comparisons are discussed.

51

## 52 1. Introduction

53 Dominated by biogenic sources, the Amazon basin is one of the few remaining continental  
54 regions where atmospheric conditions realistically represent those of the pristine or pre-industrial  
55 era (Andreae et al., 2015). As a natural climatic “chamber”, the area around the urban region of  
56 Manaus in central Amazonia is an ideal location for studying the atmosphere under natural  
57 conditions as well as under conditions influenced by human activities and biomass burning events  
58 (Andreae et al., 2015; Artaxo et al., 2013; Davidson et al., 2012; Keller et al., 2009; Kuhn et al.,  
59 2010; Martin et al., 2016b; Pöhlker et al., 2018; Poschl et al., 2010; Salati and Vose, 1984). The  
60 Observations and Modeling of the Green Ocean Amazon (GoAmazon2014/5) campaign was

61 conducted in 2014 and 2015 (Martin et al., 2017; Martin et al., 2016b). The primary objective of  
62 GoAmazon2014/5 was to improve the quantitative understanding of the effects of anthropogenic  
63 influences on atmospheric chemistry and aerosol-cloud interactions in the tropical rainforest area.  
64 During the dry season in 2014, the ACRIDICON (Aerosol, Cloud, Precipitation, and Radiation  
65 Interactions and Dynamics of Convective Cloud Systems)-CHUVA (Cloud Processes of the Main  
66 Precipitation Systems in Brazil) campaign also took place to study tropical convective clouds and  
67 precipitation over Amazonia (Wendisch et al., 2016).

68 A feature of the GoAmazon 2014/5 field campaign was the design of the ground sites'  
69 location, which uses principles of Lagrangian sampling to align the sites with the Manaus pollution  
70 plume (Figure 1: Source location – Manaus (T1 site), and downwind location – Manacapuru (T3  
71 site)). The ground sites were overflown with the low-altitude U.S. Department of Energy (DOE)  
72 Gulfstream-1 (G1) aircraft and the German High Altitude and Long Range Research Aircraft  
73 (HALO). These two aircraft are among the most advanced in atmospheric research, deploying  
74 suites of sophisticated and well-calibrated instruments (Schmid et al., 2014; Wendisch et al.,  
75 2016). The pollution plume from Manaus was intensively sampled during the G1 and HALO  
76 flights and also by the DOE Atmospheric Radiation Measurement (ARM) program Mobile  
77 Aerosol Observing System and ARM Mobile Facility located at one of the downwind surface sites  
78 (T3 site- 70 km west of Manaus). The routine ground measurements with coordinated and intensive  
79 observations from both aircraft provided an extensive data set of multi-dimensional observations  
80 in the region, which serves i) to improve the scientific understanding of the influence of the  
81 emissions of the tropical megacity of Manaus (Brazil) on the surrounding atmospheric  
82 environment of the rainforest and ii) to understand the life cycle of deep convective clouds and  
83 study open questions related to their influence on the atmospheric energy budget and hydrological  
84 cycle.

85 As more and more data sets are merged to link the ground-based measurements with  
86 aircraft observations, and as more studies focus on the spatial variation and temporal evolution of  
87 the atmospheric properties, it is critical to quantify the uncertainty ranges when combining the data  
88 collected from the different platforms. Due to the challenges of airborne operations, especially  
89 when two aircraft are involved in data collection in the same area, direct comparison studies are  
90 rare. However, this type of research is critical for further combining the datasets between the  
91 ground sites and aircraft. Thus, the main objectives of the study herein are to demonstrate how to

92 achieve meaningful comparisons between two moving platforms, to conduct detailed comparisons  
93 between data collected by two aircraft, to identify the potential measurement issues, to quantify  
94 reasonable uncertainty ranges of the extensive collection of measurements, and to evaluate the  
95 measurement sensitivities to the temporal and spatial variance. The comparisons and the related  
96 uncertainty estimations quantify the current measurement limits, which provide realistic  
97 measurement ranges to climate models as initial conditions to evaluate their output.

98         The combined GoAmazon2014/5 and ACRIDICON-CHUVA field campaigns not only  
99 provide critical measurements of aerosol and cloud properties in an under-sampled geographic  
100 region but also offer a unique opportunity to understand and quantify the quality of these  
101 measurements using carefully orchestrated comparison flights. The comparisons between the  
102 measurements from similar instruments on the two research aircraft can be used to identify  
103 potential measurement issues and quantify the uncertainty range of the field measurements, which  
104 include primary meteorological variables (Section 3.1), trace gases concentrations (Section 3.2),  
105 aerosol particle properties (number concentration, size distribution, chemical composition, and  
106 microphysical properties) (Section 3.3), cloud properties (Section 3.4), and downward solar  
107 irradiance (Section 3.5). We evaluate the consistency between the measurements aboard the two  
108 aircraft for a nearly full set of gas, aerosol particle, and cloud variables. Results from this  
109 comparison study provide the foundation not only for assessing and interpreting the observations  
110 from multiple platforms (from the ground to low altitude, and then to high altitude) but also for  
111 providing high-quality data to improve the understanding of the accuracy of the measurements  
112 related to the effects of human activities in Manaus on local air quality, terrestrial ecosystems in  
113 rainforest, and tropical weather.

## 114         **2. Measurements**

### 115         2.1 Instruments

116         The ARM Aerial Facility deployed several in situ instruments on the G1 to measure  
117 atmospheric state parameters, trace gas concentrations, aerosol particle properties, and cloud  
118 characteristics (Martin et al., 2016b; Schmid et al., 2014). The instruments installed on HALO  
119 covered measurements of meteorological, chemical, microphysical, and radiation parameters.  
120 Details of measurements aboard HALO are discussed in the ACRIDICON-CHUVA campaign  
121 overview paper (Wendisch et al., 2016). The measurements compared between the G1 and HALO

122 are listed in Table 1. Details on maintenance and calibration of the involved instrumentation can  
123 be found in the supplement (Table S1 and Table S2).

#### 124 2.1.1 Atmospheric parameters

125 All G1 and HALO meteorological sensors were routinely calibrated to maintain measurement  
126 accuracy. The G1 primary meteorological data were provided at a one-second time resolution  
127 based on the standard developed by the Inter-Agency Working Group for Airborne Data and  
128 Telemetry Systems (Webster and Freudinger, 2018). For static temperature measurement, the  
129 uncertainty given by the manufacturer (Emerson) is  $\pm 0.1$  K, and the uncertainty of the field data is  
130  $\pm 0.5$  K. The static pressure had a measurement uncertainty of 0.5 hPa. The standard measurement  
131 uncertainties were  $\pm 2$  K for the chilled mirror hygrometer and  $0.5 \text{ ms}^{-1}$  for wind speed.

132 On HALO, primary meteorological data were obtained from the Basic HALO Measurement  
133 and Sensor System (BAHAMAS) at a one-second time resolution. The system acquired data from  
134 airflow and thermodynamic sensors and from the aircraft avionics and a high-precision inertial  
135 reference system to derive the basic meteorological parameters like pressure, temperature, the 3D  
136 wind vector, aircraft position, and attitude. The water vapor mixing ratio and further derived  
137 humidity quantities were measured by the Sophisticated Hygrometer for Atmospheric Research  
138 (SHARC) based on direct absorption measurement by a tunable diode laser (TDL) system. The  
139 absolute accuracy of the primary meteorological data was 0.5 K for air temperature, 0.3 hPa for  
140 air pressure,  $0.4\text{-}0.6 \text{ ms}^{-1}$  for wind, and 5% ( $\pm 1$  ppm) for water vapor mixing ratio. All sensors  
141 were routinely calibrated and traceable to national standards (Giez et al., 2017; Krautstrunk and  
142 Giez, 2012).

#### 143 2.1.2 Gas phase

144 Constrained by data availability, the comparison of trace gas measurements is focused on  
145 carbon monoxide (CO) and ozone ( $\text{O}_3$ ) concentrations. Those measurements were made aboard  
146 the G1 by a CO/ $\text{N}_2\text{O}$ / $\text{H}_2\text{O}$  instrument (Los Gatos Integrated Cavity Output Spectroscopy  
147 instrument model 907-0015-0001), and an Ozone Analyzer (Thermo Scientific, Model 49i),  
148 respectively. The G1 CO analyzer was calibrated for response daily by NIST-traceable commercial  
149 standards before the flight. Due to the difference between laboratory and field conditions, the  
150 uncertainty of the CO measurements is about  $\pm 5\%$  for one-second sampling periods. An ultra-fast

151 carbon monoxide monitor (Aero Laser GmbH, AL5002) was deployed on HALO. The detection  
152 of CO is based on a vacuum-ultraviolet-fluorimetry, employing the excitation of CO at 150 nm,  
153 and the precision is 2 ppb, and the accuracy is about 5%. The ozone analyzer measures ozone  
154 concentration based on the absorbance of ultraviolet light at a wavelength of 254 nm. The ozone  
155 analyzer (Thermo Scientific, Model 49c) in the HALO payload is very similar to the one on the  
156 G1 (Model 49i), with an accuracy greater than 2 ppb or about  $\pm 5\%$  for four-second sampling  
157 periods. The G1 ozone monitor was calibrated at the New York State Department of  
158 Environmental Conservation testing laboratory at Albany.

### 159 2.1.3 Aerosol

160 Aerosol number concentration was measured by different condensation particle counters  
161 (CPCs) on the G1 (TSI, CPC 3010) and HALO (Grimm, CPC model 5.410). Although two CPCs  
162 were from different manufacturers, they were designed using the same principle, which is to detect  
163 particles by condensing butanol vapor on the particles to grow them to a large enough size that  
164 they can be counted optically. Both CPCs were routinely calibrated in the lab and reported the data  
165 at a one-second time resolution. The HALO CPC operated at  $0.6\text{-}1\text{ L min}^{-1}$ , with a nominal cutoff  
166 of 4 nm. Due to inlet losses, the effective cutoff diameter increases to 9.2 nm at 1000 hPa, and  
167 11.2 nm at 500 hPa (Andreae et al., 2018; Petzold et al., 2011). The G1 CPC operated at  $1\text{ L min}^{-1}$   
168 volumetric flow rate and the nominal cut-off diameter  $D_{50}$  measured in the lab was  $\sim 10$  nm.  
169 During a flight, the cut-off diameter may vary due to tubing losses, which contributes less than 10  
170 % uncertainty to the comparison between two CPC concentrations.

171 Two instruments deployed on the G1 measured aerosol particle size distribution. a Fast  
172 Integrated Mobility Spectrometer (FIMS) inside of the G1 cabin measured the aerosol mobility  
173 size from 15 to 400 nm (Kulkarni and Wang, 2006a, b; Olfert et al., 2008; Wang, 2009). The  
174 ambient aerosol particles were charged after entering the FIMS inlet and then separated into  
175 different trajectories in an electric field based on their electrical mobility. The spatially separated  
176 particles grow into super-micrometer droplets in a condenser where supersaturation of the working  
177 fluid is generated by cooling. At the exit of the condenser, a high-speed charge-coupled device  
178 camera captures the image of an illuminated grown droplet at high resolution. In this study, we  
179 used the FIMS 1 Hz data for comparison. The size distribution data from FIMS were smoothed.  
180 Aside from the FIMS, the airborne version of the Ultra High Sensitivity Aerosol Spectrometer

181 (UHSAS) was deployed on G1 and HALO. The G1 and HALO UHSAS were manufactured by  
182 the same company, and both were mounted under the wing on a pylon. UHSAS is an optical-  
183 scattering, laser-based particle spectrometer system. The size resolution is around 5% of the  
184 particle size. The G1 UHSAS typically covered a size range of 60 nm to 1000 nm. HALO UHSAS  
185 covered 90 nm to 500 nm size range for the September 9 flight.

186 Based on operating principles, FIMS measures aerosol electrical mobility size, and UHSAS  
187 measures the aerosol optical equivalent size. Thus, the difference in the averaged size distributions  
188 from those two types of instruments might be linked to differences in their underlying operating  
189 principles, such as the assumption in the optical properties of aerosol particles. The data processing  
190 in the G1 UHSAS assumed that the particle refractive index is similar to ammonium sulfate (1.55),  
191 which is larger than the average refractive index (1.41-0.013i) from a previous Amazon study  
192 (Guyon et al., 2003). The HALO UHSAS was calibrated with polystyrene latex spheres, which  
193 have a refractive index of about 1.572 for the UHSAS wavelength of 1054 nm. The uncertainty  
194 due to the refraction index can lead to up to 10% variation in UHSAS measured size (Kupc et al.,  
195 2018). Also, the assumption of spherical particles affects the accuracy of UHSAS sizing of ambient  
196 aerosols.

197 The chemical composition of submicron non-refractory (NR-PM<sub>1</sub>) organic and inorganic  
198 (sulfate, nitrate, ammonium) aerosol particles was measured using a high-resolution time-of-flight  
199 aerosol mass spectrometer (HR-ToF-AMS) aboard the G1 (DeCarlo et al., 2006; Jayne et al., 2000;  
200 Shilling et al., 2018; Shilling et al., 2013). Based on the standard deviation of observed aerosol  
201 mass loadings during filter measurements, the HR-ToF-AMS detection limits for the average time  
202 of thirteen seconds are approximately 0.13, 0.01, 0.02, 0.01 (3 $\sigma$  values)  $\mu\text{g m}^{-3}$  for organic, sulfate,  
203 nitrate, and ammonium, respectively (DeCarlo et al., 2006). A Compact Time-of-flight Aerosol  
204 Mass Spectrometer (C-ToF-AMS) was operated aboard HALO to investigate the aerosol  
205 composition. Aerosol particles enter both the C-ToF-AMS and HR-ToF-AMS via constant  
206 pressure inlets controlling the volumetric flow into the instrument, although the designs of the  
207 inlets are somewhat different (Bahreini et al., 2008). The details about the C-ToF-AMS operation  
208 and data analysis are reported in Schulz's paper (Schulz et al., 2018). The overall accuracy has  
209 been reported as ~30 % for both AMS instruments (Alfarra et al., 2004; Middlebrook et al., 2012).  
210 Data presented in this section were converted to the same condition as the HALO AMS data, which

211 is 995 hPa and 300 K. Both AMS instruments were calibrated before and after the field deployment  
212 and also once a week during the field campaign.

213 The number concentration of cloud condensation nuclei (CCN) was measured aboard both  
214 aircraft using the same type of CCN counter from Droplet Measurement Technologies (DMT,  
215 model 200). This CCN counter contains two continuous-flow, thermal-gradient diffusion  
216 chambers for measuring aerosols that can be activated at constant supersaturation. The  
217 supersaturation is created by taking advantage of the different diffusion rates between water vapor  
218 and heat. After the supersaturated water vapor condenses on the CCN in the sample air, droplets  
219 are formed, counted, and sized by an Optical Particle Counter (OPC). The sampling frequency is  
220 one second for both deployed CCN counters. Both CCN counters were calibrated using ammonium  
221 sulfate aerosol particles in the diameter range of 20-200 nm. The uncertainty of the effective water  
222 vapor supersaturation was  $\pm 5\%$ . (Rose et al., 2008)

#### 223 2.1.4 Clouds

224 Aircraft-based measurements are an essential method for in situ samplings of cloud properties  
225 (Brenquier et al., 2013; Wendisch and Brenquier, 2013). Over the last 50–60 years, hot-wire probes  
226 have been the most commonly used devices to estimate liquid water content (LWC) in the cloud  
227 from research aircraft. Since the 1970s, the most widely used technique for cloud droplet spectra  
228 measurements has been developed based on the light-scattering effect. This type of instrument  
229 provides the cloud droplet size distribution as the primary measurement. By integrating the cloud  
230 droplet size distribution, additional information, such as LWC can be derived from the high-order  
231 data product.

232 Three cloud probes from the G1 are discussed in this manuscript. The Cloud Droplet Probe  
233 (CDP) is a compact, lightweight forward-scattering cloud particle spectrometer that measures  
234 cloud droplets in the 2 to 50  $\mu\text{m}$  size range (Faber et al., 2018). Using state-of-the-art electro-optics  
235 and electronics, Stratton Park Engineering (SPEC Inc.) developed a Fast Cloud Droplet Probe  
236 (FCDP), which also uses forward-scattering to determine cloud droplet distributions and  
237 concentrations in the same range as CDP with up to 100 Hz sampling rate. The G1 also carried a  
238 two-dimensional stereo probe (2DS, SPEC Inc.), which has two 128-photodiode linear arrays  
239 working independently. The 2DS electronics produce shadowgraph images with 10  $\mu\text{m}$  pixel

240 resolution. Two orthogonal laser beams cross in the middle of the sample volume, with the sample  
241 cross section for each optical path of  $0.8 \text{ cm}^2$ . The manufacturer claims the maximum detection  
242 size is up to  $3000 \text{ }\mu\text{m}$  for the 2DS. However, due to the counting statistic issue, the data used in  
243 this study is from  $10\text{--}1000 \text{ }\mu\text{m}$  only (Lawson et al., 2006). 2DS was upgraded with modified probe  
244 tips, and an arrival time algorithm was applied to the 2DS data processing. Both efforts effectively  
245 reduced the number of small shattered particles (Lawson, 2011). For G1 cloud probes, the  
246 laboratory calibrations of the sample area and droplet sizing were performed before the field  
247 deployment. During the deployment, weekly calibrations with glass beads were performed with  
248 the size variation of less than 5%, which were consistent with the pre-campaign and after-campaign  
249 calibrations. Comparison between the LWC derived from cloud droplet spectra with hot-wire  
250 LWC measurement was made to estimate/eliminate the coincidence errors in cloud droplet  
251 concentration measurements (Lance et al., 2010; Wendisch et al., 1996)

252 On board of HALO, two cloud probes were operated and discussed in this manuscript, each  
253 consisting of a combination of two instruments: Cloud Combination Probe (CCP) and a Cloud  
254 Aerosol Precipitation Spectrometer (CAPS, denoted as NIXE-CAPS; NIXE: Novel Ice  
255 Experiment). The CCP is a combination of a CDP (denoted as CCP-CDP) with a CIPgs (Cloud  
256 Imaging Probe with greyscale, DMT, denoted as CCP-CIPgs). NIXE-CAPS consists of a CAS-  
257 Dpol (Cloud and Aerosol Spectrometer, DMT, denoted as NIXE-CAS) and a CIPgs (denoted as  
258 NIXE-CIPgs). CIPgs is an optical array probe comparable to the 2DS operated on the G1. CIPgs  
259 obtains images of cloud elements using a 64-element photodiode array ( $15\text{ }\mu\text{m}$  resolution) to  
260 generate two-dimensional images with a nominal detection diameter size range from  $15$  to  $960 \text{ }\mu\text{m}$   
261 (Klingebiel et al., 2015; Molleker et al., 2014). The CCP-CDP detects the forward-scattered laser  
262 light by cloud particles in size range of  $2.5$  to  $46 \text{ }\mu\text{m}$ . The sample area of the CCP- CDP was  
263 determined to be  $0.27\pm 0.025 \text{ mm}^2$  with an uncertainty of less than 10% (Klingebiel et al., 2015).  
264 CAS-Dpol (or NIXE-CAS) is a light scattering probe comparable to the CDP but covers the size  
265 range of  $0.6$  to  $50 \text{ }\mu\text{m}$  in diameter, thus including the upper size range of the aerosol particle size  
266 spectrum (Luebke et al., 2016). Furthermore, CAS-Dpol measures the polarization state of the  
267 particles (Costa et al., 2017). Correspondingly to the G1 CDP, the performance of the CCP-CDP  
268 and NIXE-CAS were frequently examined by glass beads calibrations. Prior to or after each HALO  
269 flight, CCP-CIPgs and NIXE-CIPgs calibrations were performed by using a mainly transparent  
270 spinning disc that carries opaque spots of different but known size. The data of the CCP measured

271 particle concentration on board of HALO are corrected to gain ambient conditions using a  
272 thermodynamic approach developed by Weigel et al. (2016). For NIXE-CAPS, the size  
273 distributions were provided where NIXE-CAS was merged with the NIXE-CIPgs at 20  $\mu\text{m}$ .

#### 274 2.1.5 Solar radiation

275 The G1 radiation suite included shortwave (SW, 400 - 2,700 nm) broadband total upward and  
276 downward irradiance measurements using Delta-T Devices model SPN-1 radiometers. The  
277 radiation data were corrected for aircraft tilt from the horizontal reference plane. A methodology  
278 has been developed (Long; et al., 2010) for using measurements of total and diffuse shortwave  
279 irradiance and corresponding aircraft navigation data (latitude, longitude, pitch, roll, heading) to  
280 calculate and apply a correction for platform tilt to the broadband hemispheric downward SW  
281 measurements. Additionally, whatever angular offset there may be between the actual orientation  
282 of each radiometer's detector and what the navigation data say is level has also been determined  
283 for the most accurate tilt correction.

284 The Spectral Modular Airborne Radiation measurement sysTem (SMART-Albedometer) was  
285 installed aboard HALO. Depending on the scientific objective and the configuration, the optical  
286 inlets determining the measured radiative quantities can be chosen. The SMART-Albedometer has  
287 been utilized to measure the spectral upward and downward irradiances; thereby, it is called as an  
288 albedometer, as well as to measure the spectral upward radiance. The SMART-Albedometer is  
289 designed initially to cover measurements in the solar spectral range between 300 and 2,200 nm  
290 (Krisna et al., 2018; Wendisch et al., 2001; Wendisch et al., 2016). However, due to the decreasing  
291 sensitivity of the spectrometer at large wavelengths, the use of the wavelengths was restricted to  
292 300 – 1,800 nm. The spectral resolution is defined by the full width at half maximum (FWHM),  
293 which is between 2 and 10 nm. In this case, the instruments were mounted on an active horizontal  
294 stabilization system for keeping the horizontal position of the optical inlets during aircraft  
295 movements (up to +/- 6 degrees from the horizontal plane).

#### 296 2.2 Flight patterns

297 During the dry season IOP (September 1 – October 10, 2014), two types of coordinated flights  
298 were carried out: one flight in cloud-free condition (September 9) and two flights with clouds  
299 present (September 21 and October 1). In this study, we compare the measurements for both

300 coordinated flight patterns. The discussion is mainly focused on the flights under cloud-free  
301 conditions on September 9 and the flight with clouds present on September 21, as shown in figure  
302 1. The other coordinated flight on October 1 is included in the supplemental document (section  
303 S1, Figure S1, S2, S7, and S8).

304 For the cloud-free coordinated flight on September 9, the G1 took off first and orbited around  
305 an area from the planned rendezvous point until HALO arrived in sight. It then coordinated with  
306 HALO and performed a wing-to-wing maneuver along straight legs around 500 m above sea level,  
307 as shown in Figure 2. The normal G1 average sampling speed is  $100 \text{ m s}^{-1}$ , and the normal HALO  
308 average sampling speed is  $200 \text{ m s}^{-1}$ . During the coordinated flight on September 9, both aircraft  
309 also adjusted their normal sampling speed by about  $50 \text{ m s}^{-1}$  so that they could fly side by side.

310 For the second type of coordinated flights, the G1 and HALO flew the stacked pattern at their  
311 own typical airspeed. On September 21, the G1 also took off from the airport first, followed by  
312 HALO 15 minutes later. Then, both aircraft flew above the T3 ground site and subsequently flew  
313 several flight legs stacked at different altitudes. The two aircraft were vertically separated by about  
314 330 m and sampled below, inside, and above clouds. Due to the different aircraft speeds, the time  
315 difference between two aircraft visiting the same part of the flight paths varied, increasing up to 1  
316 hour at the end of the flight path, as shown in Figure 3. On October 1, the G1 focused on the cloud  
317 microphysical properties and contrasting polluted versus clean clouds. HALO devoted the flight  
318 to the cloud vertical evolution and life cycle and also probed the cloud processing of aerosol  
319 particles and trace gases. The G1 and HALO coordinated two flight legs between 950–1250 m  
320 above the T3 site under cloud-free conditions. Following that, HALO flew to the south of  
321 Amazonia, and the G1 continued sampling plume-influenced clouds above the T3 site, and then  
322 flew above the Rio Negro area.

323 In this study, to perform a meaningful comparison of in situ measurements, all the data from  
324 instruments were time-synchronized with the aircraft (G1 or HALO) navigation system. For AMS  
325 and CPC data, the time shifting due to tubing length and instrument flow had been corrected. For  
326 the coordinated flight on Sep. 9, the data compared was from the same type of measurements with  
327 the same sampling rate. For the measurements with the different sampling rate, the data were  
328 binned to the same time interval for comparison. For the flight with the cloud present (Sep.21 and  
329 Oct. 1), the following criteria are used: 1) the data collected by the two aircraft must be less than  
330 30 mins apart from each other; 2) the comparison data were binned to 200 m altitude intervals; 3)

331 the cloud flag was applied to the aerosol measurements, and the data affected by the cloud  
332 shattering are eliminated from the comparisons of aerosol measurements. Moreover, additional  
333 comparison criteria are specified for individual measurements in the following section. Table 2  
334 shows the total number of points used for the comparison.

### 335 3. Results

#### 336 3.1 Comparison of the G1 and HALO measurements of atmospheric state parameters

337 The atmospheric state parameters comprise the primary variables observed by the research  
338 aircraft. The measurements provide essential meteorological information not only for  
339 understanding the atmospheric conditions but also for providing the sampling conditions for other  
340 measurements, such as those of aerosol particles, trace gases, and cloud microphysical properties.

341 For cloud-free coordinated flights, the comparison focused on the near side-by-side flight leg  
342 at around 500 m, as shown in Figure 2. Table 3 shows the basic statistics of the data for primary  
343 atmospheric state parameters, assuming that two measurements from the G1 and HALO have a  
344 proportional relationship without any offset ( $Y=m_0*X$ ). In general, the atmospheric state  
345 parameters observed from both aircraft were in excellent agreement. The linear regression  
346 achieved a slope that was near 1 for four individual measurements. The regression is evaluated  
347 using the equation below.

$$348 \quad R^2 = 1 - \frac{SS_{regression}}{SS_{Total}} \quad (1)$$

349 Where the sum squared regression error is calculated by  $SS_{regression} = \sum(y_i - y_{regression})^2$ ,  
350 and the sum squared total error is calculated by  $SS_{Total} = \sum(y_i - \bar{y})^2$ ,  $y_i$  is the individual data  
351 point,  $\bar{y}$  is the mean value, and  $y_{regression}$  is the regression value. When the majority of the data  
352 points are in a narrow value range, using the mean is better than the regression line, and the  $R^2$  will  
353 be negative (Neg in Table 3).

354 The difference between the average ambient temperatures on the two aircraft was 0.5 K, and  
355 the difference between the average dew point temperatures was about 1 K. For temperature and  
356 humidity, the G1 data were slightly higher than the HALO data. The main contributions to the  
357 observed differences include the error propagation in the derivation of the ambient temperature  
358 from the measured temperature, instrumental-measurement uncertainty, and the temporal and  
359 spatial variability. The average horizontal wind speed measured by HALO is  $0.4 \text{ m s}^{-1}$  higher than

360 the average horizontal wind speed measured by the G1. The uncertainty source of wind estimation  
361 is mainly due to the error propagation from the indicated aircraft speed measurement and the  
362 aircraft ground speed estimation from GPS. The static pressure distribution measured aboard  
363 HALO showed a smaller standard deviation (0.9 hPa) compared to the value of the G1 (1.5 hPa).  
364 Part of the reason for this difference is a more substantial variation of the G1 altitude during level  
365 flight legs when the G1 flew at around 50 m/s faster than its normal airspeed. Thus, any biases  
366 caused by their near side-by-side airspeeds being different from their typical airspeeds would be  
367 undetected during these coordinated flights.

368 For the coordinated flights under cloudy conditions, we used the criteria from Section 2 to  
369 compare ambient conditions measured by the G1 and HALO aircraft. In addition to the ordinary  
370 linear regression, we also used the orthogonal regression to minimize the perpendicular distances  
371 from the data points to the fitted line. The ordinary linear regression assumes only the response  
372 (Y) variable contains measurement error but not the predictor (X), which remains unknown when  
373 we start the comparison between the measurements from the G1 and HALO. Thus, the additional  
374 orthogonal regression examines the assumption in the least square regression and makes sure the  
375 roles of the variables have little influence on the results. In Table 4, two equations were used for  
376 the orthogonal regression. One assumes that two measurements have a proportional relationship  
377 ( $Y=m_1*X$ ). The other one assumes a linear relationship, which can be described with the slope-  
378 intercept equation  $Y=m*X+b$ . Two regression results in Table 4 don't show a significant  
379 difference. The regression using the slope-intercept equation shows a different level of  
380 improvement in each individual measurement and will be discussed in the corresponding sections.

381 As shown in Figure 4, the linear regression slopes for ambient temperature (Figure 4(a)),  
382 pressure (Figure 4 (b)), and dew point temperature were also close to 1 between the G1 and HALO  
383 measurements during the September 21 coordinated flight. The  $R^2$  value is also close to 1. These  
384 results suggest that the G1 and HALO measurements achieved excellent agreements. Note that the  
385 dew point temperature from the G1 measurement was erroneous and removed from the comparison  
386 the data points between 2200–2700 m and above 3700 m (Figure 4(c)) because the G1 sensor was  
387 skewed by wetting in the cloud. The HALO dew point temperature was calculated from the total  
388 water mixing ratio measured by TDL, and that measurement in the cloud was more accurate than  
389 the measurement made by the chilled mirror hygrometer aboard the G1.

390 The lower value of the  $R^2$  value in horizontal wind speed means the ratio of the regression error  
391 and total error in wind measurement is much higher than the temperature and pressure  
392 measurements. The main contributions to this difference are the error propagation during the  
393 horizontal wind speed estimation and the temporal and spatial variance between two aircraft  
394 sampling locations. We observed differences between the two aircraft data of up to  $2 \text{ m s}^{-1}$ , caused  
395 by the increasing sampling distance as the two aircraft were climbing up. For example, the G1  
396 flew a level leg above T3 around 2500 m between 16:20-16:30, while HALO stayed around 2500  
397 m for a short period and kept climbing to a higher altitude. Due to strong vertical motion,  
398 turbulence, and different saturations (evaporation-condensation processes), the variances in the  
399 horizontal wind speed (Figure 4(d)) were also more significant compared to the variances of  
400 temperature and pressure measurements.

### 401 3.2 Comparison of trace gas measurements

402 For the cloud-free coordinated flight on September 9, ozone is the only trace gas measurement  
403 available on both aircraft. The linear regression slope shows that the HALO ozone concentration  
404 was about 8% higher than the G1 concentration. The difference between the averaged ozone  
405 concentrations was 4.1 ppb. As mentioned in section 2.1.2, each instrument has a 2 ppb accuracy  
406 (or 5%) on the ground based on a direct photometric measurement measuring the ratio between  
407 a sample and an ozone-free cell. The in-flight calibration suggested that the accuracy of each  
408 instrument could raise to 5-7% (or 2-3.5 ppb). Thus, the difference between the averaged ozone  
409 concentrations – 4.1 ppb is within the instrument variation. The primary source of bias is probably  
410 the different ozone loss in the sampling and transfer lines.

411 The comparison made on September 21 flight in Figure 5 shows good agreement for the  
412 vertically averaged ozone measurements. Comparing the statistics data from September 9, the  
413 ozone measurement is not sensitive to the temporal and spatial changes. Although we do not have  
414 the comparison data on September 9, the G1 and HALO CO measurements comparison show a  
415 higher correlation than the ozone data comparison at different altitudes on September 21. Note  
416 that the data points with more substantial variance in CO concentration were excluded because the  
417 G1 and HALO were sampling different air masses between 2000-3000 m, as indicated in Figure  
418 S3-S5. The CO plot in Figure (5b) shows the real atmospheric variability. Around 4000 m, the CO  
419 reading from the G1 and HALO has the minimum variation and is averaged around 85 ppb, which

420 is at the atmospheric background level. At lower altitudes and higher CO concentrations, the local  
421 contribution is not well-mixed, and the inhomogeneity expresses as the more substantial variations  
422 observed in the plot.

### 423 3.3 Comparison of aerosol measurements

424 Aerosol particles exhibited substantial spatial variations, both vertically and horizontally, due  
425 to many aerosol sources and complex atmospheric processes in the Amazon basin, especially with  
426 the local anthropogenic sources in Manaus. Thus, spatially resolved measurements are critical to  
427 characterizing the properties of the Amazonian aerosols. The cloud-free coordinated flights allow  
428 us to compare the G1 and HALO aerosol measurements and thus will facilitate further studies that  
429 utilize the airborne measurements. The vertical profiles obtained using the G1 and HALO in  
430 different aerosol regimes of the Amazon basin have contributed to many studies (Fan et al., 2018;  
431 Martin et al., 2017; Wang et al., 2016).

432 The design and performance of the aircraft inlets can strongly influence measured aerosol  
433 particle number concentration, size distribution, and chemical composition (Wendisch et al.,  
434 2004). Therefore, they need to be taken into consideration when comparing the measurements  
435 aboard two aircraft. The G1 aerosol inlet is a fully automated isokinetic inlet. Manufacturer wind  
436 tunnel tests and earlier studies show that this inlet operates for aerosol particles with a diameter up  
437 to 5  $\mu\text{m}$ , with transmission efficiency around 50 % at 1.5  $\mu\text{m}$  (Dolgos and Martins, 2014; Kleinman  
438 et al., 2007; Zaveri et al., 2010). The HALO sub-micrometer Aerosol Inlet (HASI) was explicitly  
439 designed for HALO. Based on the numerical flow modeling, optical particle counter  
440 measurements, and field study evaluation, HASI has a cut-off size of 3  $\mu\text{m}$ , with transmission  
441 efficiency larger than 90 % at 1  $\mu\text{m}$  (Andreae et al., 2018; Minikin et al., 2017).

#### 442 3.3.1 Aerosol particle number concentration

443 For the cloud-free coordinated flight on September 9, the linear regression of CPC and UHSAS  
444 between the G1 and HALO measurements are also included in Table 3. The total number  
445 concentration measured by HALO CPC was about 20 % lower than that by the G1 CPC, as shown  
446 in Figure 6 (a). The CPC measurement is critically influenced by the isokinetic inlet operation and  
447 performance. During the flights, the aircraft attitude, such as the pitch and roll angles, will cause  
448 the isokinetic sampling under non-axial conditions. The non-axial flow at the probe inlet may result

449 in flow separation, turbulence, and particle deposition. Therefore, quantitative particle  
450 measurements have more substantial uncertainty. As shown in Figure 6 (b), we compared the CPC  
451 data by applying three different data quality criteria. The first criterion is the same criterion  
452 described in the previous section that makes sure all the compared measurements happen in less  
453 than 30 minutes apart, and the linear regression is included in Table 3. The second criterion  
454 constrains the data under the isokinetic and iso-axial condition and the plot in Figure 6(b) shows  
455 the iso-axial criteria reduced the broadness of the scattered data, but no significant change to the  
456 linear regression. We further constrained the data with the averaging. Based on the average wind  
457 speed and distance between two aircraft, we averaged the data into 10 seconds interval and found  
458 that the regression  $R^2$  increase to 0.9392. The typical uncertainty between two CPCs is 5-10% on  
459 the well-controlled environment (Gunthe et al., 2009; Liu and Pui, 1974). Although both CPCs  
460 from the G1 and HALO were characterized in the lab to be within 10% with its respective lab  
461 standard, we observed a 20% variance during the flight. This result suggests the challenging  
462 condition of airborne measurement can significantly increase the systematic uncertainties of CPC  
463 measurements, such as systematic instrument drifts, different aerosol particle losses inside of the  
464 two CPCs, and different inlet transmission efficiencies in the two aircraft.

465 The CPC data in Figure 6 are color-coded with UTC time. The general trend is that the aerosol  
466 number concentration increased with time through the Manaus plume between 15:30 and 15:40.  
467 A similar trend was observed in aerosol particle number concentration (Figure 7) measured by the  
468 Ultra-High Sensitivity Aerosol Spectrometer (UHSAS)-Airborne version (referred to as UHSAS).  
469 The total number concentration data given by UHSAS (Figure 7) is integrated over the overlapping  
470 size range (90 – 500 nm for the September 9 flight) for both the G1 and HALO UHSAS. The linear  
471 regression shows that the total aerosol particle number concentration from HALO UHSAS is about  
472 16.5% higher than that from the G1 UHSAS. The discrepancy between the two UHSAS  
473 measurements is mainly due to the error propagation in the sampling flow, the differential pressure  
474 transducer reading, the instrument stability, and calibration repeatability, consistent with the other  
475 UHSAS study (Kupc et al., 2018). In the airborne version of UHSAS, mechanical vibrations have  
476 a more significant impact on the pressure transducer reading than the case for the bench version of  
477 UHSAS.

478 For the coordinated flight on September 21, the G1 and HALO data are averaged to 200 m  
479 vertical altitude intervals (Figure 8). The data points with an altitude between 2000 – 3000 m

480 altitude were excluded from the comparisons, because the G1 and HALO sampled different  
481 airmasses, as evidenced from trace gas and aerosol chemical composition data (detailed in Section  
482 3.2 and 3.3.3). The UHSAS size range was integrated from 100 to 700 nm on September 21. The  
483 variation of the size range was because the overlap of size distributions from both UHSAS  
484 instruments was changed. Both the CPC and UHSAS measurement comparisons show stronger  
485 variation at the low altitude, especially below 2000 m. Above 3500 m, the variations on the CPC  
486 and UHSAS measured concentration became significantly smaller than the variation at the lower  
487 altitude. This result is consistent with the observation from the trace gas measurement and confirms  
488 that the variability of aerosol properties changes significantly with time and space. It is noticeable  
489 that the discrepancy observed in the UHSAS measurements comparison is larger than that from  
490 the CPC comparison. That is because the aerosol flow control inside the UHSAS can't respond  
491 quickly enough to the rapid change of the altitude and caused significant uncertainty in the data.

### 492 3.3.2 Aerosol particle size distribution

493 For the cloud-free coordinated flight on September 9, the averaged aerosol size distributions  
494 measured by FIMS, G1 UHSAS, and HALO UHSAS during one flight leg are compared in Figure  
495 9. For particle diameter below 90 nm, the G1 UHSAS overestimated the particle concentration,  
496 which is due to the uncertainty in counting efficiency correction. The UHSAS detection efficiency  
497 is close to 100% for particles larger than 100 nm and concentrations below  $3000 \text{ cm}^{-3}$  but decreases  
498 considerably for both smaller particles and higher concentrations (Cai et al., 2008). The aerosol  
499 counting efficiency correction developed under the lab conditions does not necessarily apply under  
500 the conditions during the flight. Between 90 nm and 250 nm, FIMS agreed well with the G1  
501 UHSAS, whereas HALO UHSAS was about 30 % higher than the two instruments. For the size  
502 range of 250–500 nm, FIMS had good agreement with HALO UHSAS and was about 30-50 %  
503 higher than the G1 UHSAS depending on the particle size. Because the UHSAS has a simplified  
504 “passive” inlet, the large size aerosol particle loss in the UHSAS inlet was expected to increase  
505 with the increasing of the aircraft speed. Thus, the lower G1 UHSAS concentrations at a larger  
506 aerosol particle size are likely related to the particle loss correction.

507 For the September 21 flight, the vertical profiles of aerosol size distributions are averaged into  
508 100 m altitude intervals (Figure 10). Overall, all size distribution measurements captured the mode  
509 near 100 nm between 800–1000 m at the top of the convective boundary layer, as indicated by the

510 potential temperature (Figure 10(d)), which starts from a maximum near the ground and then  
511 becomes remarkably uniform across the convective boundary layer. The peak of the aerosol size  
512 distribution shifted from 100 nm to 150 nm with increasing altitude. Note that due to data  
513 availability, the aerosol size distribution data from the HALO UHSAS has a reduced vertical  
514 resolution.

### 515 3.3.3 Aerosol particle chemical composition

516 Figure 11 shows the vertical profiles of the aerosol mass concentrations measured by the two  
517 AMS on September 21. The upper panel shows the medians and interquartile ranges of the different  
518 species (organics, nitrate, sulfate, ammonium) and the total mass concentration for the G1 (circles)  
519 and HALO (triangles). The lower panel shows the difference between the medians of G1 and  
520 HALO. The error bars were calculated using error propagation from the error of the median  
521 (interquartile range divided by  $2 \cdot \sqrt{N}$ ). The data were grouped into 400 m altitude bins. The  
522 total mass concentration is the highest in the lower altitudes between 100 m and 2000 m with a  
523 median value of  $5 \mu\text{g m}^{-3}$  (G1-AMS). At altitudes between 2000 m and 3800 m, the aerosol mass  
524 concentration decreased to a median value of  $1.2 \mu\text{g m}^{-3}$  (G1-AMS).

525 The most significant difference was observed at altitudes below 1800 m. The aerosol mass  
526 concentration measured by HALO-AMS is less than that measured by G1 AMS, likely due to  
527 particle losses in the constant pressure inlet (CPI) used on the HALO AMS. Between 1800 m and  
528 3000 m, the mass concentrations measured by the HALO AMS exceed those measured by the G1-  
529 AMS. This is most likely because the G1 was sampling different air masses than the HALO, as  
530 indicated by the differences in CO mixing ratios and the CPC concentrations for this altitude region  
531 (see Fig. 5 and 8). Above 3000 m altitude, both instruments agree within the uncertainty range.

532 Among individual species, the largest difference above 2000 m is observed for ammonium. The  
533 deployed G1 AMS is a high-resolution mass spectrometer (HR-ToF), whereas the HALO AMS  
534 has a lower resolution (C-ToF). The higher resolution of the G1-AMS allows for a better separation  
535 of interfering ions at  $m/z$  15, 16, and 17 ( $\text{NH}^+$ ,  $\text{NH}^{2+}$ ,  $\text{NH}^{3+}$ ) and thereby a more reliable calculation  
536 of the ammonium mass concentration.

537 Overall the aerosol chemical composition is dominated by organics, as is evident from the  
538 vertical profiles of the relative fractions (Fig. 12). Both AMS show a dominant contribution of

539 organics to the total mass concentration with values around 70 %. This contribution is constant at  
540 altitudes between 100 m and 3500 m and decreases to 50 % at 3800 m altitude. The inorganic  
541 fraction has the highest contribution from sulfate (20 %), followed by ammonium (7 %) and nitrate  
542 (2 – 4 %). For organics, ammonium, and sulfate, both instruments give similar relative fractions,  
543 only for nitrate where a discrepancy is observed between 1000 and 3000 m. Although the absolute  
544 aerosol mass concentration measured by the HALO-AMS was affected by the constant pressure  
545 inlet below 1800 m altitude, the relative fractions of both instruments generally agree well. Similar  
546 results were found for a second comparison flight on October 1, 2014 (see supplemental section 2  
547 and plots S7, S8).

### 548 3.3.4 CCN number concentration

549 These measurements provide key information about the aerosol's ability to form cloud droplets  
550 and thereby modify the microphysical properties of clouds. Numerous laboratory and field studies  
551 have improved the understanding of the connections among aerosol particle size, chemical  
552 composition, mixing states and CCN activation properties (Bhattu and Tripathi, 2015;  
553 Broekhuizen et al., 2006b; Chang et al., 2010; Duplissy et al., 2008; Lambe et al., 2011; Mei et al.,  
554 2013a; Mei et al., 2013b; Pöhlker et al., 2016; Thalman et al., 2017). In addition, based on the  
555 simplified chemical composition and internal mixing state assumption, various CCN closure  
556 studies have achieved success within  $\pm 20\%$  uncertainty for ambient aerosols (Broekhuizen et al.,  
557 2006a; Mei et al., 2013b; Rissler et al., 2004; Wang et al., 2008).

558 According to earlier studies (Gunthe et al., 2009; Pöhlker et al., 2016; Roberts et al., 2001;  
559 Roberts et al., 2002; Thalman et al., 2017), the hygroscopicity ( $\kappa_{\text{CCN}}$ ) of CCN in the Amazon basin  
560 is usually dominated by organic components ( $\kappa_{\text{Org}}$ ). Long-term ground-based measurements at the  
561 Amazon Tall Tower Observatory also suggest low temporal variability and lack of pronounced  
562 diurnal cycles in hygroscopicity only under natural rainforest background conditions (Pöhlker et  
563 al., 2018; Pöhlker et al., 2016).

564 Using FIMS and CCN data from both the G1 and HALO collected during the coordinated flight  
565 leg on September 9, the critical activation dry diameter ( $D_{50}$ ) was determined by integrating FIMS  
566 size distribution to match the CCN total number concentration (section S3). Then, the effective  
567 particle hygroscopicity was derived from  $D_{50}$  and the CCN-operated supersaturation using the  $\kappa$ -

568 Köhler theory. The histogram plots based on the density of the estimated hygroscopicity ( $\kappa_{\text{est}}$ ) from  
569 both aircraft were compared for the flight leg above T3. The  $\kappa_{\text{est}}$  values derived from the G1 and  
570 HALO measurements during the flight leg above the T3 site are  $0.19 \pm 0.07$  and  $0.19 \pm 0.08$ ,  
571 respectively. Those values agree very well with the overall mean value of  $0.17 \pm 0.06$  derived from  
572 long-term measurements at the Amazon Tall Tower Observatory (Pöhlker et al., 2016; Thalman  
573 et al., 2017).

574 A comparison of the vertical profiles of the CCN concentrations at 0.5% supersaturation on  
575 September 21 is shown in Figure 13 as an example. The difference between the CCN  
576 measurements on the two aircraft is about 20% on average. The linear regression slope would  
577 increase to 0.9120 if we focused on the data above 2500 m. The main contributions to the  
578 difference include the difference in aerosol inlet structure, aerosol particle loss correction in the  
579 main aircraft inlet, and the constant pressure inlet, the systematic inlet difference below 2500 m as  
580 shown in AMS data, as well as the error propagation of CCN measurements.

### 581 3.4 Comparison of cloud measurements

582 In situ cloud measurements help to capture the diversity of different cloud forms and their  
583 natural temporal and spatial variability. The G1 CDP and FCDP were deployed under the different  
584 wing pylons, and also on the different side of the aircraft. The G1 2DS was deployed on the same  
585 side of FCDP. The HALO cloud combination probe (CCP-CDP and CCP-CIPgs) and NIXE-CAPS  
586 (NIXE-CAS and NIXE-CIPgs) were deployed under the different wing pylons but on the same  
587 side of the aircraft. On September 21, 2014, based on the aircraft location and elevation  
588 information as shown in Figure 1(b) and Figure 3, both aircraft were sampling above T3 site and  
589 passing through the same cloud field at ~1600 m flight leg and ~1900 m flight leg as shown in  
590 Figure S11 and Figure S12. We used the cloud probes data from ~1900 m flight leg for the cloud  
591 droplet number concentration comparison. Two size ranges were considered: 3-20  $\mu\text{m}$  from light  
592 scattering probes (CDP vs. FCDP on the G1, CCP-CDP vs. NIXE-CAS on HALO) and 2-960  $\mu\text{m}$   
593 from combined cloud probes.

#### 594 3.4.1 Comparison of cloud droplet number concentration between 3-20 $\mu\text{m}$

595 For underwing cloud probes, such as the CDP and the CAS, Lance (Lance, 2012) suggests an  
596 undercounting bias of measured particle number concentration by up to 44% due to coincidence  
597 as soon as the ambient cloud particle density rises to 1000 per  $\text{cm}^3$ . At identical cloud particle  
598 densities, an earlier study (Baumgardner et al., 1985) estimates the coincidence bias for underwing  
599 cloud probes to the range at 20%. Factually, the coincidence correction depends on the  
600 instruments' individual detection volume, the air's volume flow rate through the detector, and the  
601 cloud particles' residence time within the detection volume (Hermann and Wiedensohler, 2001;  
602 Jaenicke, 1972). For this comparison, coincidence bias remained unconsidered for each of the  
603 cloud probe measurements to avoid deviations that are caused by the application of different  
604 corrections.

605 The primary cloud layer was observed by both the G1 and HALO between 1000-2500 m above  
606 ground. Although the two aircraft have sampled along the same flight path, the instruments  
607 probably observed different sets of the cloud due to cloud movement with the prevailing wind or  
608 different cloud evolution stages. Thus, an initial comparison focuses on the redundant instruments  
609 on the same aircraft, that measured truly collocated and synchronous on board of HALO and of  
610 the G1, respectively. In Figure 14 (a), the data of the CCP-CDP and of the NIXE-CAS are  
611 juxtaposed sampled over about 13 minutes for particle detection size ranges which were considered  
612 as most equivalent. The comparison reveals two ranges of particle number concentrations at which  
613 densification of agreeing measurements become visible. At very low number concentrations (about  
614  $10^{-1} - 10$  per  $\text{cm}^3$ ), the presence of inactivated (interstitial) aerosols in the clear air space between  
615 the very few cloud elements should be considered. Over specific ranges, however, the fine  
616 structure of varying cloud droplet number concentration may cause the regression's scattering,  
617 indicated by cloud particle measured by one instrument whilst respective antagonist seems to  
618 measure within almost clear air – and vice versa. At higher number concentrations, i.e., between  
619  $10^2$  and  $10^3$  per  $\text{cm}^3$ , the comparison of the highly resolved data constitutes increasing compactness  
620 with respect to the 1:1 line. The overall data scatter of this comparison, however, may indicate the  
621 highly variable structure within clouds as those investigated over the Amazon basin. The data of  
622 the G1 CDP and the FCDP are juxtaposed as the same as HALO cloud probes. However, the  
623 sampled cloud period was much shorter – about 3 minutes. Similar to the HALO cloud probes  
624 comparison, we observe two ranges of particle number concentrations at which densification of  
625 agreeing measurements become visible, especially for the lower number concentrations (Figure

626 14(b)). At higher number concentrations, only a few cloud elements were observed by the G1  
627 cloud probes. That is because the G1 was about 7-23 minutes later to pass the same location as  
628 HALO, and experienced much fewer cloud elements.

### 629 3.4.2 Comparison of cloud droplet size distribution between 2-960 $\mu\text{m}$ from both aircraft

630 Comparing the cloud probes from the G1 and HALO, the size distributions from the HALO  
631 CCP and NIXE-CAPS probes are in remarkably good agreement between 2-960  $\mu\text{m}$ , and both  
632 peaked around 10  $\mu\text{m}$ , as shown in Figure 15. That is because the potential effects of cloud  
633 elements shattering on the probe measurements were considered similarly for the HALO-deployed  
634 CCP and NIXE instruments. On the G1, the CDP and FCDP had a more significant difference in  
635 the size range of less than 8  $\mu\text{m}$ , although both of them peaked between 10-20  $\mu\text{m}$ . The difference  
636 between the G1 CDP and FCDP is mainly due to the data post-processing. The G1 CDP used an  
637 old data acquisition system from the Science Engineering Associates, which limited its capability  
638 to store the particle-by-particle (PBP) data for further processing. The CDP had placed an 800-  
639  $\mu\text{m}$ -diameter pinhole in front of the sizing detector to minimize the coincidence up to  $1850\text{ cm}^{-3}$ .  
640 On the other side, the FCDP was equipped with new electronics and PBP data was locally stored  
641 in a flash drive onboard the Linux machine. For the G1 flights, a constant probe-dependent  
642 adjustment factor was applied to FCDP to adjust the coincidence further in the final data product.  
643 The G1 CDP and FCDP operated with a redesigned probe tip to minimize the shattering effect. An  
644 additional algorithm was applied to the FCDP data to eliminate particles with short interarrival  
645 times.

646 For cloud droplets larger than 20  $\mu\text{m}$ , the difference between the obtained cloud particle size  
647 distributions from two aircraft becomes substantial (up to two orders of magnitude), which  
648 indicated the observations of two different stages within the progressing development of a  
649 precipitation cloud. The precipitation cloud developing process is evidently expressed in elevated  
650 number concentrations of larger cloud elements observed during the G1 measurement that  
651 happened later. We also observed that the general cloud characteristic is similar at different  
652 altitude levels, as shown in Figure S13. The first two of three averaged periods were chosen during  
653 the flight leg of  $\sim 1600\text{ m}$  and the last average period is for the flight leg  $\sim 1900\text{ m}$  compared in  
654 Figure 15. Due to the averaging, the fine in-cloud structure gets suppressed. The small scale

655 variabilities inside a cloud which are illustrated by the scattering of the highly resolved  
656 measurement data from the instrument comparison (cf. Figure 14) and the temporal evolution of  
657 in-cloud microphysics are not ascertainable and furthermore are beyond the scope of this study.

### 658 3.5 Comparison of radiation measurements

659 In this study, the downward irradiance measured by the SPN-1 unshaded center detector  
660 was compared with the integrated downward irradiance from the SMART-Albedometer between  
661 300–1,800 nm wavelengths in Figure 16. Only measurements from flight legs, where the G1 and  
662 HALO flew near side-by-side and at the same altitude were taken into consideration for analysis.  
663 In Figure 16, the top panel shows the time series of SPN-1 measurements, and the bottom panel  
664 shows the time series of SMART-Albedometer measurements. The black dots represented all data,  
665 and the blue circles identified data when the navigation condition was within +/- 1 degree from the  
666 horizontal level. The large scatter in the data between 15:12-15:28 and 15:35-15:40 is mainly due  
667 to the different sensor trajectories during the maneuvering of the aircraft to get to the coordinated  
668 flight position. Because of the difference of each aircraft position from horizontal, the measured  
669 signal varied from the signal of the direct component of sunlight. Each sensor might look at  
670 different directions of the sky or different parts of the clouds. In addition, both aircraft flew under  
671 scattered clouds, and this uneven sunlight blocking is another contribution to the “drop-off”  
672 behavior in the time series plots of the downward irradiance.

673 Comparing the G1 and HALO measurements between 15:15-1:55 using the restricted  
674 navigation criteria in Figure S14, we observed that the G1 SPN1 irradiance is slightly higher than  
675 the integrated irradiance from the SMART-Albedometer. We used the NCAR tropospheric  
676 ultraviolet and visible (TUV) radiation model estimated the weighted irradiance at 15:42:00 on  
677 Sep 9, 2014, and confirm that the spectral variation in the instruments is the main contribution  
678 to the difference in the comparison.

## 679 **4 Uncertainty assessment**

680 As mentioned in the introduction, a low-flying G1 and a high-flying HALO cover the sampling  
681 area from the atmospheric boundary layer to the free troposphere, and the sampling period from  
682 the dry and wet seasons (Martin et al., 2016a). This spatial coverage provides the user community

683 with abundant atmospheric-related data sets for their further studies, such as for remote sensing  
684 validation and modeling evaluation. However, one critical step to bridge the proper usage of the  
685 observation with further atmospheric science study is to understand the measurement uncertainty  
686 in this data set, especially the variation between the coexisting measurements due to the temporal  
687 and spatial difference.

688 For the majority of the measurements during this field study, three primary sources contribute  
689 to the measurement variation between the two aircraft: the temporal and spatial variations, the  
690 difference in the inlet characterization, and the limitation of the instrument capability. We used  
691 both ordinary least squares (OLS) linear regression and the orthogonal distance regression (ODR)  
692 to correlate the measurements from the G1 and HALO and confirmed that the slope and  $R^2$  are  
693 very similar for the measurements made on September 9. The results from Table 2 confirmed that  
694 the G1 and HALO measurements should be in a linear relationship without an offset if there is no  
695 altitude variation. It also shows the minimum discrepancy between two aerosol instruments (CPC  
696 or UHSAS) could be around 20%, which will include the error caused by the difference in the inlet  
697 characterization and the limitation of the instrument capability. If we assume those two  
698 measurement variation sources are not affected by the altitude, then by comparing the linear  
699 regression data from Table 3 to Table 2, we can estimate the temporal and spatial variation between  
700 two aircraft in a stack flight pattern. Three linear regression approaches were assessed, and the  
701 results are listed in Table 3. If we assume that two measurements from the G1 and HALO should  
702 not have any offsets, the OLS and ODR regressions show similar results. For the meteorological  
703 parameters, this assumption is valid. In addition, good correlations also indicate that there is no  
704 significant temporal or spatial variation during the stack pattern flight. As expected, the wind speed  
705 and the aerosol measurements show that the correlations between the measurements from the G1  
706 and HALO significantly improved with the offset assumption. This result suggests that the  
707 temporal and spatial variation in a half-hour will add an additional 20% variance to the measured  
708 aerosol properties. This will lead to considerable uncertainty when we combine the observation  
709 data between the ground station and the airborne platform. Thus, to evaluate or constrain  
710 atmospheric modeling work, more routine and long-term airborne measurements should be used  
711 to provide statistically sufficient observation.

712

## 713 5 Summary

714 In situ measurements made by well-characterized instruments installed on two research aircraft  
715 (the G1 and HALO) during the GoAmazon 2014/5 and ACRIDICON-CHUVA campaigns were  
716 compared (Table S3). Overall, the analysis shows good agreement between the G1 and HALO  
717 measurements for a relatively broad range of atmospheric-related variables in a challenging lower  
718 troposphere environment. Measured variables included atmospheric state parameters, aerosol  
719 particles, trace gases, clouds, and radiation properties. This study outlines the well-designed  
720 coordinated flights for achieving a meaningful comparison between two moving platforms. The  
721 high data quality was ensured by the most sophisticated instruments aboard two aircraft used the  
722 most advanced techniques, assisted with the best-calibrated/characterized procedures. The  
723 comparisons and the related uncertainty estimations quantify the current measurement limits,  
724 which provide guidance to the modeler to realistically quantify the modeling input value and  
725 evaluate the variation between the measurement and the model output. The comparison also  
726 identified the measurement issues, outlined the associated reasonable measurement ranges, and  
727 evaluated the measurement sensitivities to the temporal and spatial variance.

728 The comparisons presented here were mainly from two coordinated flights. The flight on  
729 September 9 was classified as a cloud-free flight. During this flight, the G1 and HALO flew nearly  
730 side-by-side within a “polluted” leg, which was above the T3 site and across the downwind  
731 pollution plume from Manaus, and a “background” leg, which was outbound from Manaus to the  
732 west and could be influenced by the regional biomass burning events during the dry season. Both  
733 legs were at 500 m altitude and showed linear regression slopes of ambient temperature and  
734 pressure, horizontal wind speed and dew point temperature near to 1 between the G1 and HALO  
735 measurements. These comparisons provide a solid foundation for further evaluation of aerosol,  
736 trace gas, cloud, and radiation properties. The total aerosol concentration from CPC and UHSAS  
737 were compared for the 500 m flight leg above the T3 site. The UHSAS measurements had a better  
738 agreement than the CPC measurements. That is because of the minor difference in the inlet  
739 structure and instrument design between two UHSAS aboard the two aircraft. The average size  
740 distribution from both UHSAS and one FIMS in the G1 suggests that UHSAS had an over-  
741 counting issue at the size range between 60-90 nm, which was probably due to electrical noise and  
742 small signal-to-noise ratio in that size range. Good agreement in the aerosol size distribution  
743 measurement provides a “sanity” check for AMS measurements. A CCN closure study suggested

744 that FIMS provides valuable size coverage for better CCN number concentration estimation. Based  
745 on the  $\kappa$ -Köhler parameterization,  $\kappa_{\text{est}}$  observed at 500 m above the T3 site is  $0.19 \pm 0.08$ , which is  
746 similar to the overall mean kappa from long-term ATTO measurements -  $0.17 \pm 0.06$  (Pöhlker et  
747 al., 2016). This similarity suggests that there is no significant spatial variability along the  
748 downwind transect, although the freshly emitted aerosol particles may have much less  
749 hygroscopicity. The difference in the ozone measurement comparison is about 4.1 ppb, which  
750 suggests that the bias due to the sampling line loss inside of the G1 gas inlet. The irradiance from  
751 the SPN1 unshaded center detector in the G1 was compared with the HALO integrated downward  
752 irradiance between 300–1800 nm and achieved a very encouraging agreement with a variance of  
753 less than 10%.

754 During the second type of coordinated flights on September 21 (with cloudy conditions),  
755 HALO followed the G1 after take-off from Manus airport; then, the two aircraft flew stacked legs  
756 relative to each other at different altitudes above the T3 site. For atmospheric state parameters,  
757 nearly linear correlations between the G1 and HALO were observed for ambient pressure,  
758 temperature, and dew point temperature measurements at an altitude range from ground to around  
759 5000 m. The horizontal wind had more variation than the rest of the meteorological properties,  
760 which is mainly due to the temporal and spatial variability. The aerosol number concentration and  
761 the trace gas measurements both suggest inhomogeneous aerosol distribution between 2000-3000  
762 m altitude. The integrated aerosol number concentration from UHSAS showed consistent  
763 discrepancy at different altitudes. This considerable uncertainty in the UHSAS measurements is  
764 caused by the significant aerosol sample flow variations due to the slow and unstable flow control.  
765 Although the aircraft-based UHSAS is a challenging instrument to operate, a reasonable size  
766 distribution profile comparison was made between both UHSAS and FIMS on the G1. Overall the  
767 chemical composition of the aerosol is dominated by organics. Around 70% of the AMS measured  
768 mass is organic, and this fractional contribution is maintained from the surface to 3500 m, then  
769 decreases to 50% at higher altitudes. The most substantial difference among all the species is  
770 observed for ammonium due to the different mass resolution of the AMS instruments, and more  
771 reliable ammonium mass concentration can be achieved with high resolution mass spectrometer.  
772 Although the absolute aerosol mass concentration measured by the HALO AMS was affected  
773 below 1800 m altitude by the constant pressure inlet, the relative fractions of both instruments  
774 from the G1 and HALO agree well.

775 Cloud probe comparisons were made for the cloud droplet number concentration between 3–  
776 20  $\mu\text{m}$  for the initial comparison between the redundant instruments on the same aircraft. Then the  
777 comparison of cloud droplet size distribution between 2–960  $\mu\text{m}$  for a flight leg around 1900 m  
778 showed a remarkably good agreement. The major cloud appearance was captured by both aircraft,  
779 although the cloud elements observed were affected by the cloud movement with the prevailing  
780 wind and the different cloud evolution stages. Furthermore, the relatively short time delay of 7–23  
781 minutes between the independent measurements may give a hint for the time scales in which the  
782 cloud droplet spectra develop within a convective cloud over the Amazon basin.

783 The above results provide additional information about the reasonableness of measurements  
784 for each atmospheric variable. This study confirms the high-quality spatial and temporal dataset  
785 with clearly identified uncertainty ranges had been collected from two aircraft and builds a good  
786 foundation for further studies on the remote sensing validation and the spatial and temporal  
787 evaluation of modeling representation of the atmospheric processing and evolution.

788 Several efforts made by both airborne measurement teams have significantly contributed to the  
789 overall success of this comparison study, and we recommend them for future field operations.

- 790 1) Characterize instruments following the same established guideline. For example, the  
791 aerosol instruments can follow the guideline from the World Calibration Centre for  
792 Aerosol Physics (WCCAP).
- 793 2) Periodically compare measurements from different instruments for consistency in the  
794 field. For example, we found that comparing the integrated aerosol volume distribution  
795 from the aerosol sizer with the converted total aerosol mass from the AMS  
796 measurement can help check both the instrument performances and the inlet operation  
797 condition. Additionally, measurements from different cloud probes should be  
798 compared in the overlapping size ranges.
- 799 3) Daily calibration would be valuable but likely unrealistic to perform in the field. One  
800 alternative is to daily even hourly monitor the variation of the critical instrument  
801 parameters, such as the aerosol sample flow of the individual aerosol instruments.
- 802 4) For the cases with minor variations in the calibration results, the typical practice is to  
803 use the average calibration results for the variation period. However, we also  
804 recommend documenting the corresponding uncertainty with the data product.

805 5) A side-by-side comparison among the similar instruments deployed at different  
806 platforms, including those at ground sites, is highly recommended and will provide a  
807 comprehensive view of the data reliability.

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1 Table 1. List of compared measurements and corresponding instruments deployed aboard the G1  
 2 and HALO during GoAmazon2014/5. The acronyms are defined in a table at the end of this  
 3 paper.  $D_p$  indicates the particle diameter.  $\Delta D_p$  refers to the size resolution.

Measurement Variables	Instruments deployed on the G1 (Martin et al., 2016; Schmid et al., 2014)	Instruments deployed on HALO (Wendisch et al., 2016)
Static Pressure	Rosemount (1201F1), 0-1400 hPa	Instrumented nose boom tray (DLR development), 0-1400 hPa
Static air temperature	Rosemount E102AL/510BF -50 to +50 °C	Total Air Temperature (TAT) inlet (Goodrich/Rosemount type 102) with an open wire resistance temperature sensor (PT100), -70 to +50 °C
Dewpoint temperature	Chilled mirror hygrometer 1011B -40 to +50 °C	Derived from the water-vapor mixing ratio, which is measured by a tunable diode laser (TDL) system (DLR development), 5-40000 ppmv
3-D wind	Aircraft Integrated Meteorological Measurement System 20 (AIMMS-20)	Instrumented nose boom tray (DLR development) with an air data probe (Goodrich/Rosemount) 858AJ and high-precision Inertial Reference System (IGI IMU-IIe)
Particle number concentration	CPC, cut off size ( $D_p$ ) =10 nm	CPC, cut off size ( $D_p$ ) =10 nm
Size distribution*	UHSAS-A, 60-1000 nm.	UHSAS-A, 60-1000 nm.
	FIMS, 20 nm – 500 nm	
Non-Refractory particle chemical composition	HR-ToF-AMS: Organics, Sulfate, Nitrate, Ammonium, Chloride, 60-1000 nm	C-ToF-AMS: Organics, Sulfate, Nitrate, Ammonium, Chloride, 60-1000 nm
CCN concentration	CCN-200, SS= 0.25, 0.5%	CCN-200, SS= 0.13-0.53%
Gas phase concentration	N2O/CO and Ozone Analyzer, CO, O <sub>3</sub> concentration, precision 2 ppb	N2O/CO and Ozone Analyzer, CO, O <sub>3</sub> concentration, precision 2 ppb
Cloud properties*	CDP, 2-50 $\mu$ m, $\Delta D_p$ =1-2 $\mu$ m	CCP-CDP, 2.5-46 $\mu$ m, $\Delta D_p$ =1-2 $\mu$ m
	FCDP, 2-50 $\mu$ m, $\Delta D_p$ =1-2 $\mu$ m	NIXE-CAS: 0.61 -52.5 $\mu$ m
	2DS, 10-1000 $\mu$ m	NIXE-CIPgs, 15-960 $\mu$ m
		CCP-CIPgs: 15-960 $\mu$ m
Radiation	SPN1 downward irradiance, 400-2700 nm	SMART Albedometer, downward spectral irradiance, 300-2200 nm

4 \*for an individual flight, the size range may vary.

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Table 2. Summary of the total data points compared between the G1 and HALO instruments.

	SEP 9, 2014		SEP 21, 2014	
	G1	HALO	G1	HALO
<b>Atmospheric parameters</b>	2815	2815	7326	12065
<b>Gas phase, CO</b>	N/A	N/A	7326	12065
<b>Gas phase, Ozone</b>	2815	2815	7110	11766
<b>CPC</b>	2043	2043	8466	11646
<b>UHSAS (FIMS)</b>	2031	2031	5841 (9405)	828
<b>AMS</b>	N/A	N/A	587	818
<b>CCNc</b>	663	531	7982	4546
<b>G1: CDP(FCDP)</b>	N/A	N/A	3627(4439)	2051(2260)
<b>HALO: CCP-CDP (NIXE-CAS)</b>				
<b>G1: 2DS</b>	N/A	N/A	2280	2261 (2260)
<b>HALO: CCP-CIPgs (NIXE-CIPgs)</b>				
<b>RAD</b>	1355	1355	N/A	N/A

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Table 3. Summary of basic statistics of data between in situ measurements on Sep 9.

*Comparison of the coordinated flight on Sep. 9*

<i>Variables</i>	G1				HALO					
	min	max	mean	std	min	max	mean	std	slope	R <sup>2</sup>
<i>T, K</i>	297.7	300.2	298.9	0.5	297.2	299.4	298.4	0.4	1.002	Neg.
<i>P, hPa</i>	955	965	960.1	1.5	958	964.9	961.8	0.9	0.998	Neg.
<i>WSpd, m/s</i>	0.3	8.9	3.4	1.2	0.3	7.7	3.8	1.1	0.998	Neg.
<i>T<sub>dew</sub>, k</i>	293	296.5	295.0	0.5	292.9	294.9	294.0	0.3	0.996	Neg.
<i>O<sub>3</sub>, ppb</i>	10.5	58.8	22.2	9.3	18.3	50.8	26.3	6.6	1.082	0.9401
<i>CPC, cm<sup>-3</sup></i>	696.0	3480.6	1591.3	568.7	687.4	2639.4	1313.8	473.5	0.819	0.8508
<i>UHSAS, cm<sup>-3</sup></i>	78.2	1118.	645.5	116.3	504.1	1622.2	756.3	138.6	1.165	0.8193
<i>CCNc (κ)</i>	0.010	0.347	0.1855	0.067	0.012	0.394	0.1890	0.083	0.8937	Neg.

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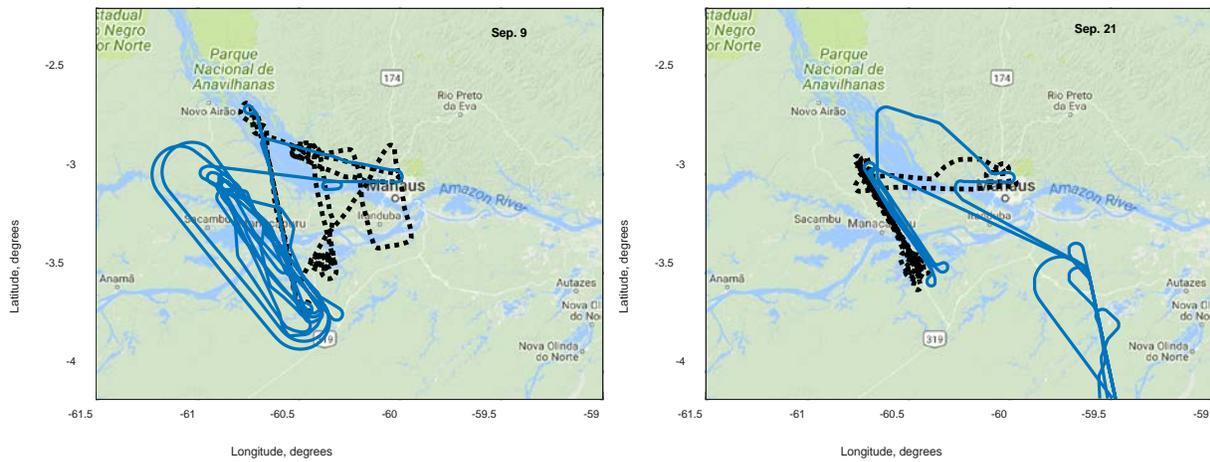
15 Table 4 Summary of three statistics analysis of data between in situ measurements on Sep 21

<i>Comparison of the coordinated flight on Sep. 21</i>							
	m	offset	R <sup>2</sup>	m0	R <sup>2</sup>	m1	R <sup>2</sup>
<i>T, K</i>	0.929	20.0	0.9992	0.999	0.9928	0.999	0.9928
<i>P, hPa</i>	1.001	0.929	0.9998	1.001	0.9998	1.001	0.9998
<i>WSpd, m/s</i>	0.885	1.0	0.7875	1.012	0.5076	1.023	0.5049
<i>T<sub>dew</sub>, k</i>	0.989	3.8	0.9963	1.003	0.9904	1.003	0.9904
<i>O<sub>3</sub>, ppb</i>	1.134	-1.5	0.9598	1.075	0.9369	1.101	0.9208
<i>CO, ppb</i>	0.922	5.4	0.9654	0.966	0.9254	0.967	0.9254
<i>CPC, cm<sup>-3</sup></i>	0.571	199.4	0.9482	0.635	0.8738	0.641	0.8735
<i>UHSAS, cm<sup>-3</sup></i>	1.126	178.0	0.8249	1.293	0.5070	1.384	0.4847
<i>CCNc (κ)</i>	0.766	55.3	0.8330	0.815	0.6544	0.829	0.6521

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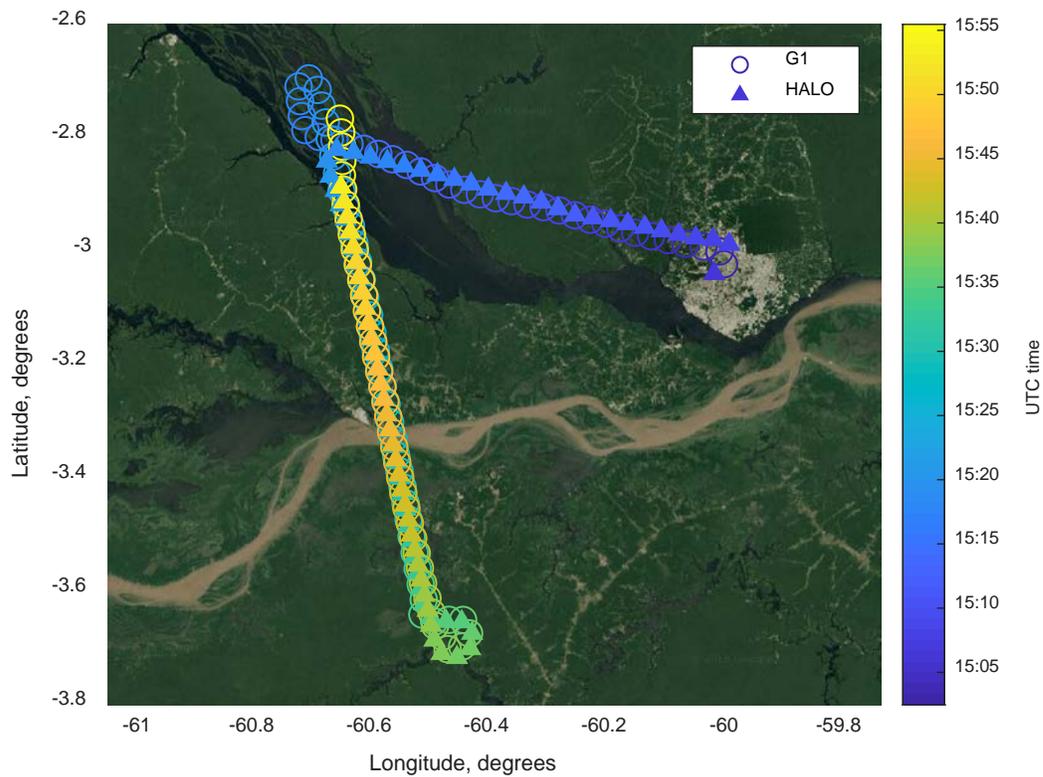


(a)

(b)

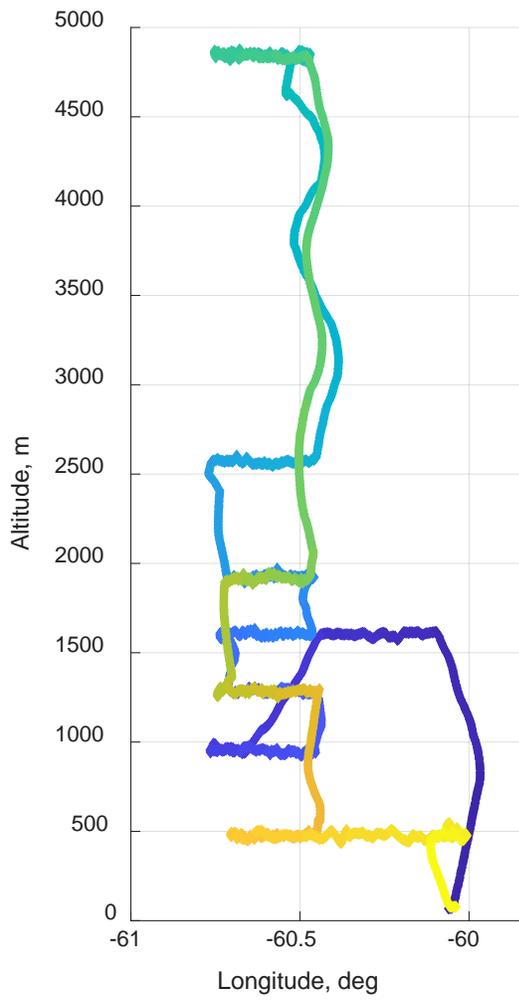
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21 Figure 1. Coordinated flight tracks for September 9 (a) and September 21 (b). The black dotted  
22 line is the flight track of the G1, and the blue line is the flight track of HALO. (This figure was  
23 created using Mapping Toolbox™ © COPYRIGHT 1997–2019 by The MathWorks, Inc.)

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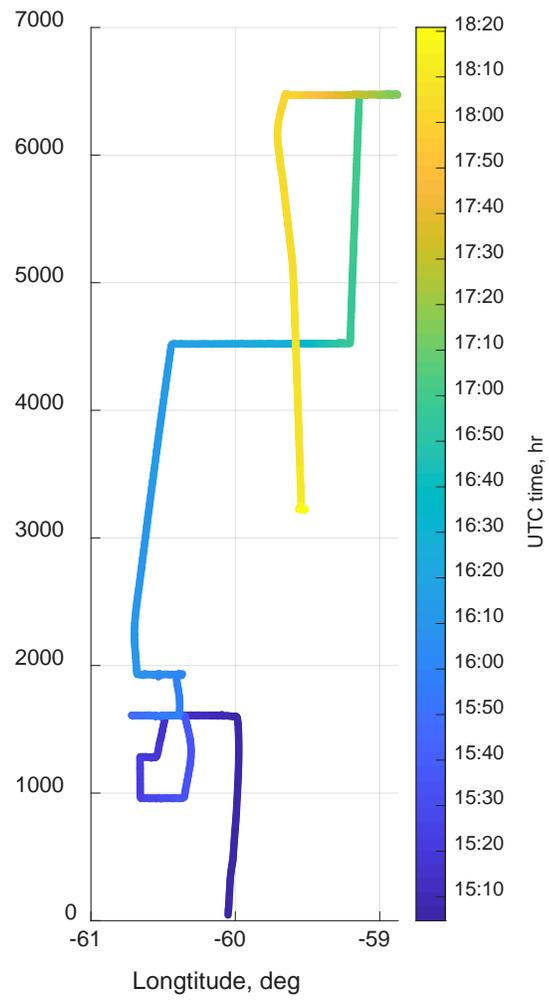


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26 Figure 2. Time-colored flight track of the G1 (circle) and HALO (triangle) on September 9 during  
 27 a cloud-free coordinated flight at 500 m above sea level (50 m apart as the closest distance). (This  
 28 figure was created using Mapping Toolbox™ © COPYRIGHT 1997–2019 by The MathWorks,  
 29 Inc.)



(a)

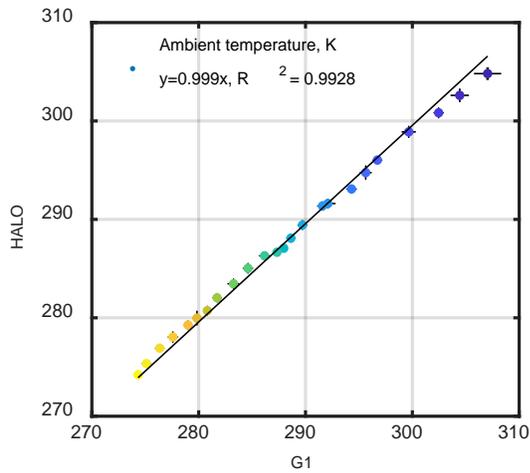


(b)

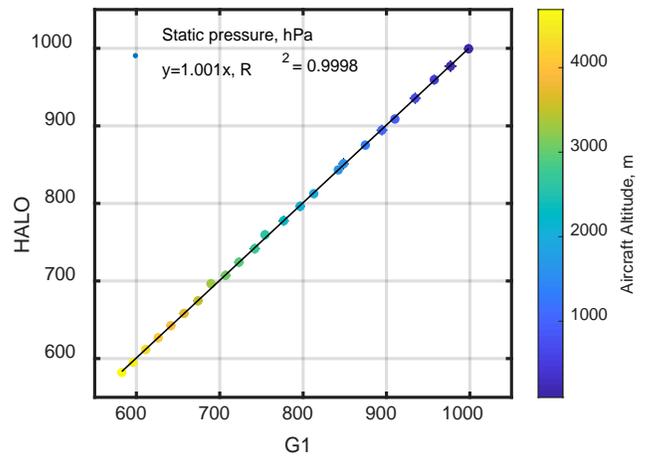
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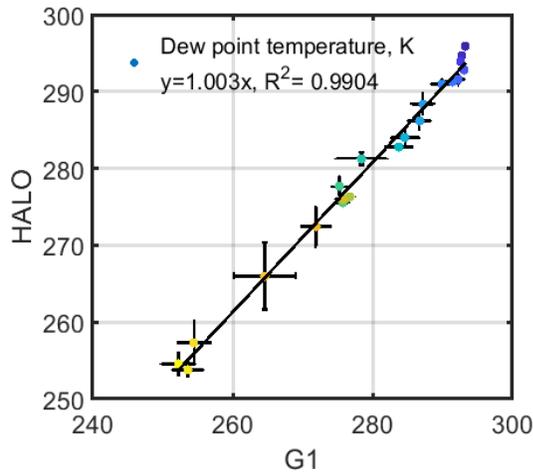
32 Figure 3. Time-colored flight profile of the G1 (a) and HALO (b) on September 21, during a  
 33 coordinated flight.



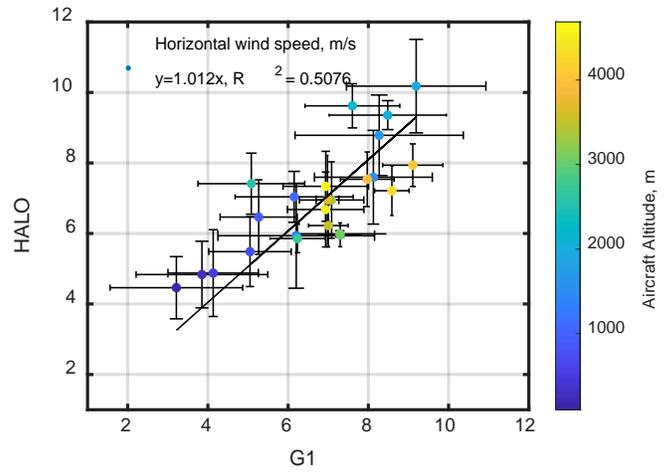
(a)



(b)



(c)



(d)

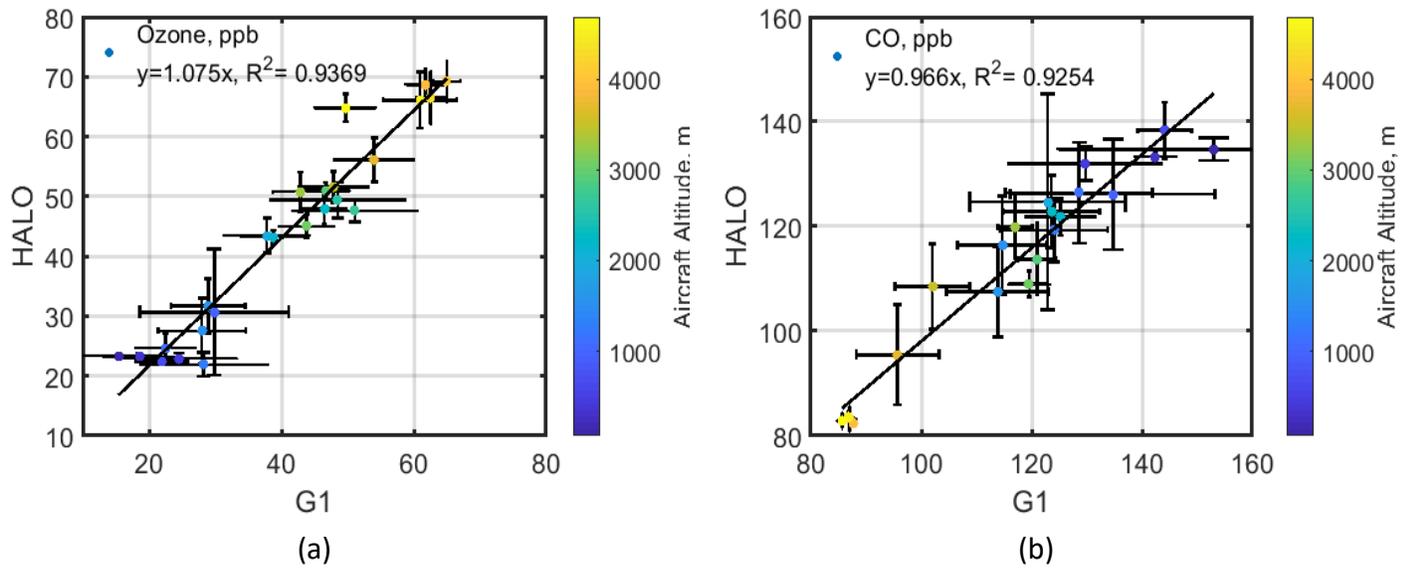
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36 Figure 4. Aircraft altitude-colored plots of (a) ambient temperature, (b) static pressure, (c) dew

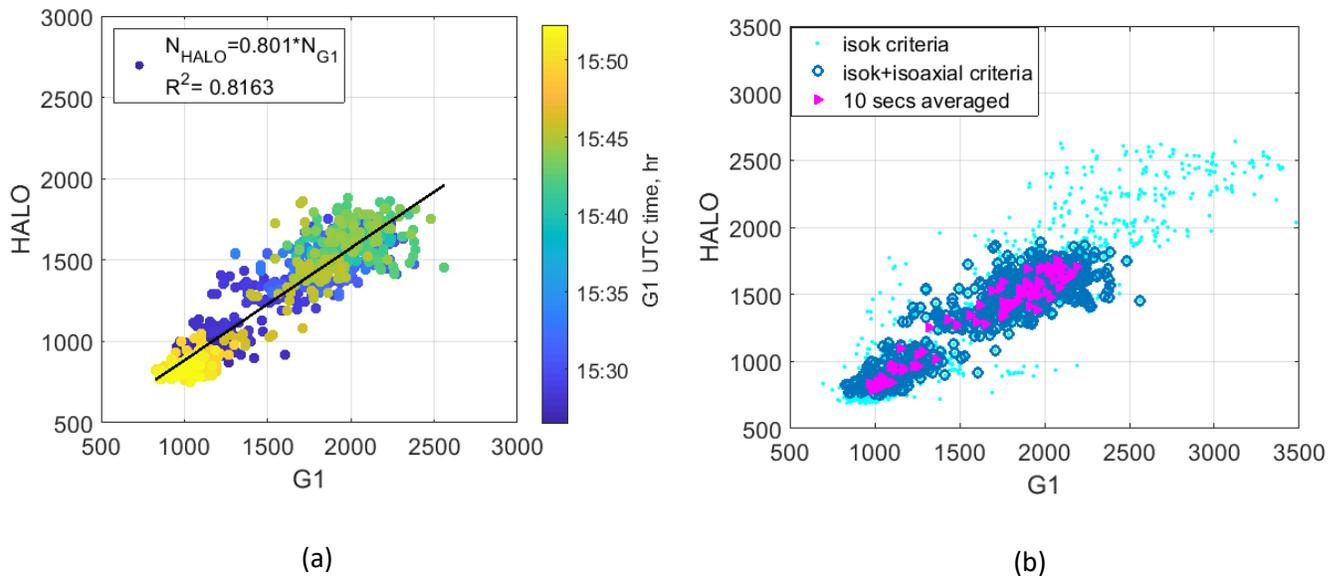
37 point temperature, and (d) horizontal wind speed observed by the G1 and HALO on September

38 21.



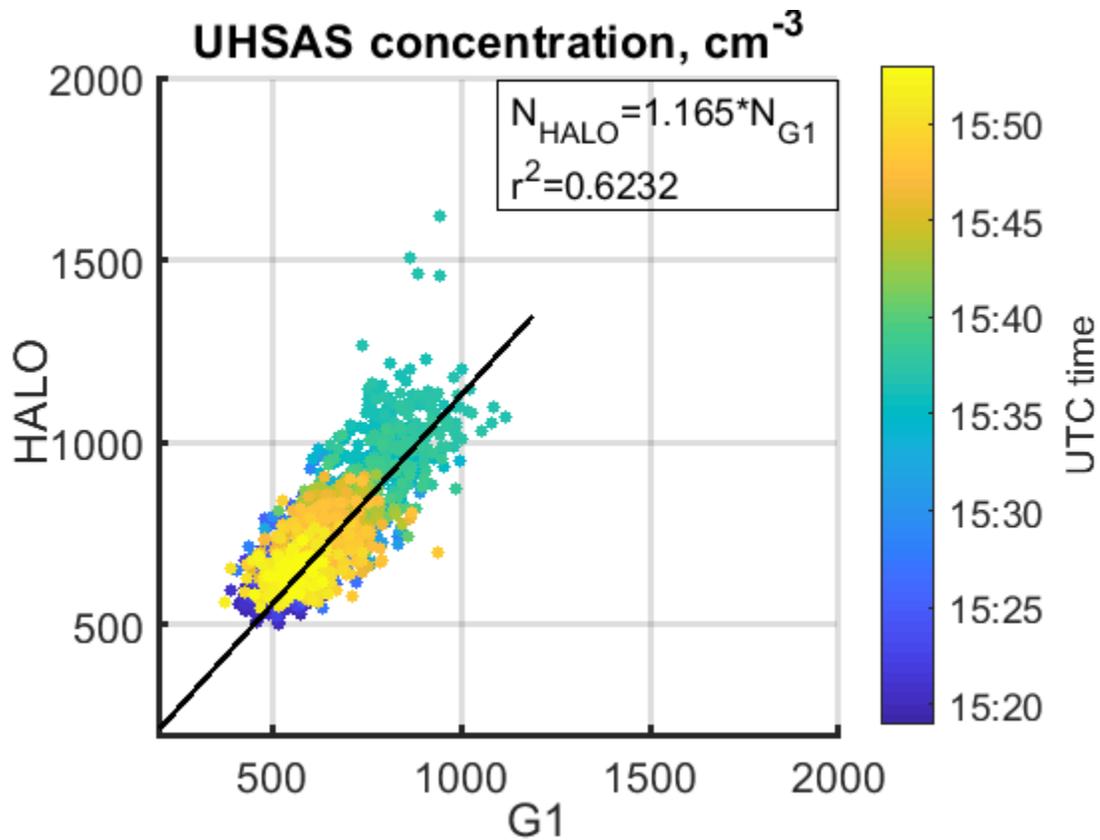
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Figure 5. Aircraft altitude-colored plots of trace gas (a) Ozone, (b) CO, for the coordinated flight on September 21.



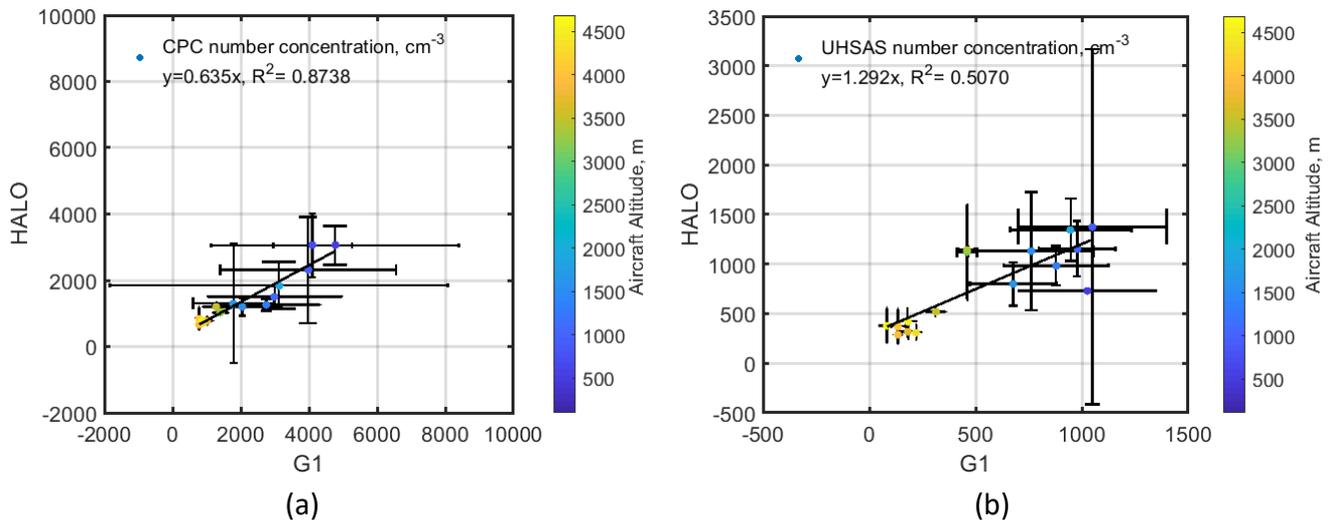
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Figure 6. The G1 and HALO comparison for aerosol number concentration measured by CPC (>10 nm) on September 9: (a) with iso-kinetic inlet constrain; (b) with different criteria.



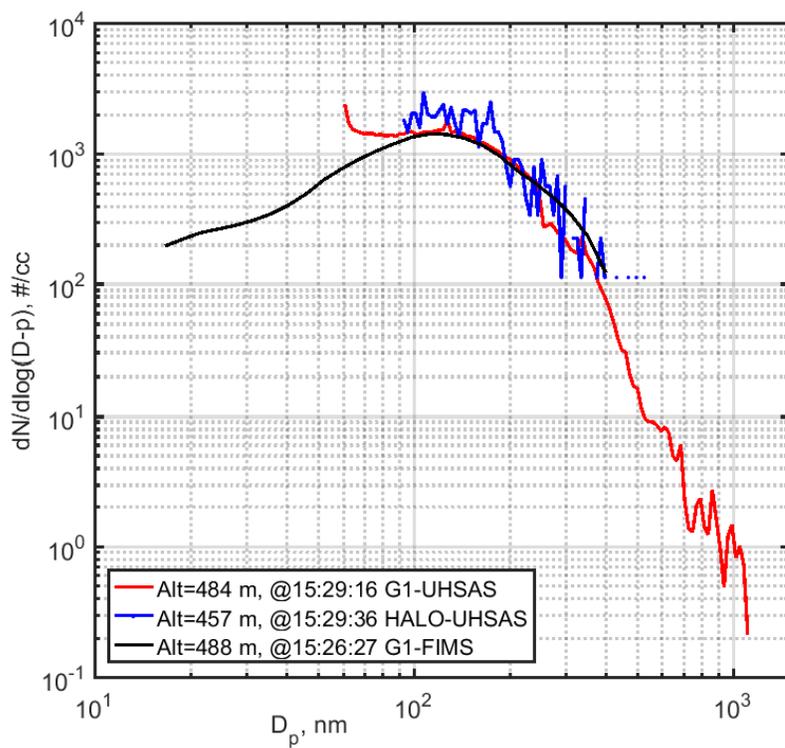
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47 Figure 7. The G1 and HALO comparison for aerosol number concentration measured by  
 48 UHSAS (90-500 nm) on September 9.



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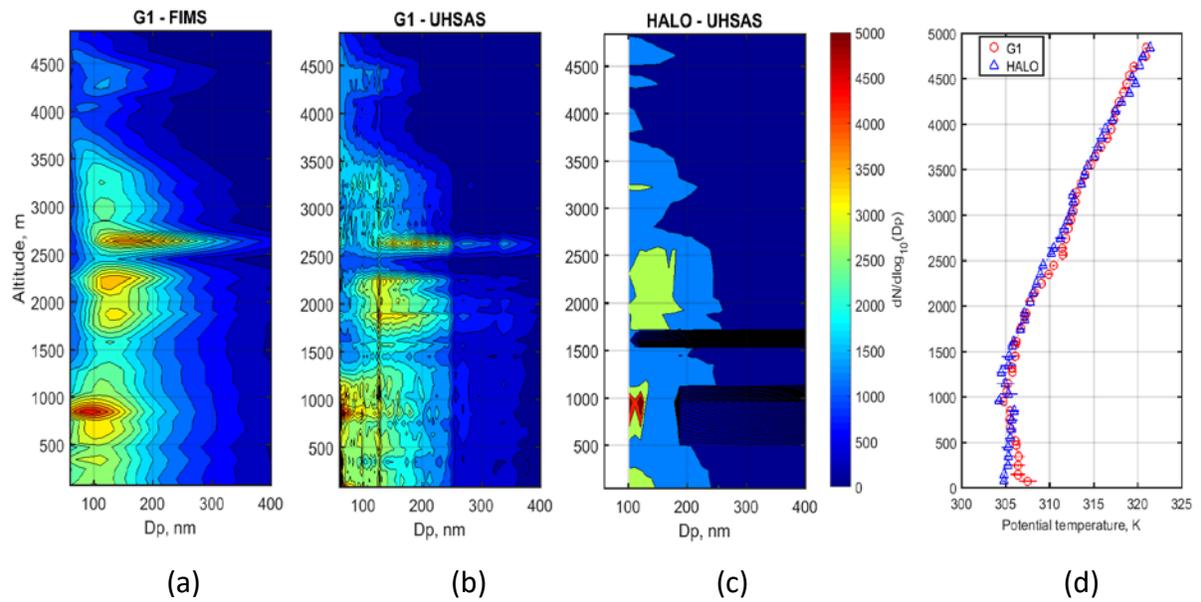
50 Figure 8. The G1 and HALO comparison for aerosol number concentration profiling measured  
 51 by (a) CPC and (b) UHSAS (100-700 nm) on September 21.



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53 Figure 9. The G1 and HALO comparison for aerosol size distribution measured by UHSAS  
 54 (from both aircraft) and FIMS (on the G1) on September 9.

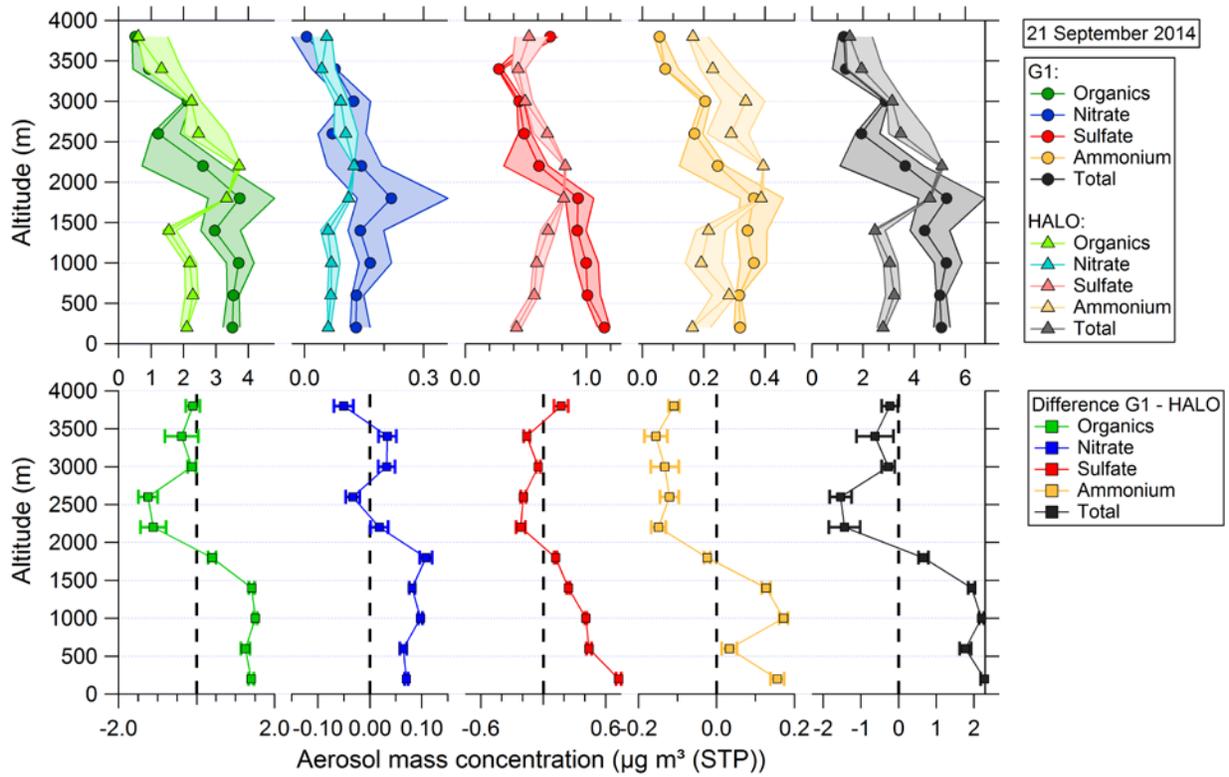
55



57

58 Figure 10. Aerosol size distribution vertical profiles measured by (a) the G1 FIMS, (b) The G1  
 59 UHSAS, (c) the HALO UHSAS, (d) Potential temperature aboard the G1 and HALO on September  
 60 21.

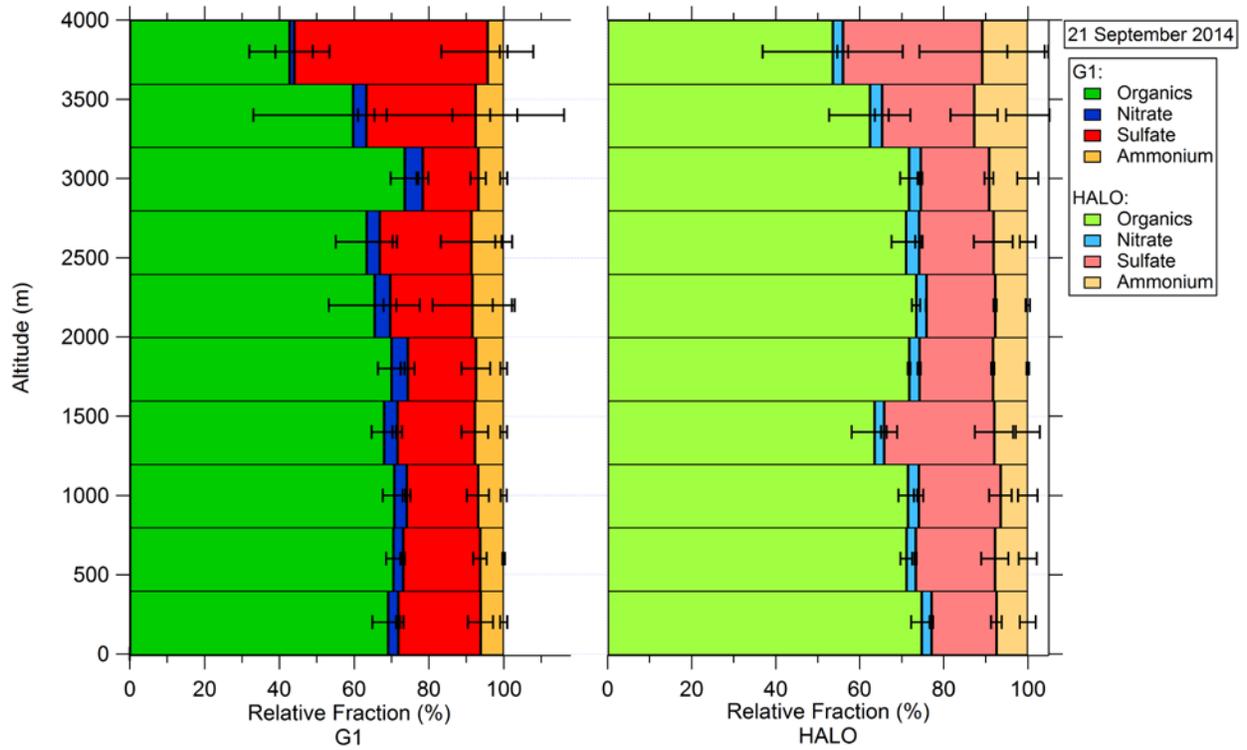
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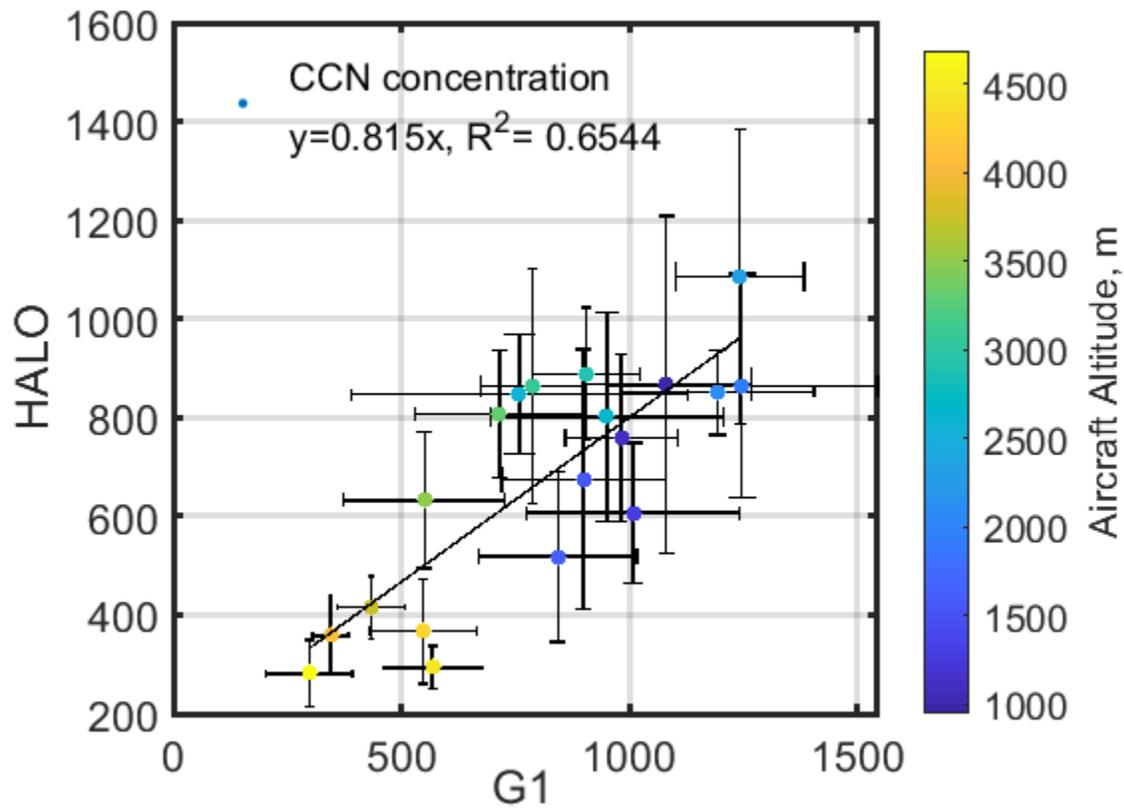
63

64 Figure 11. The vertical profile of aerosol mass concentration measured by the G1 and HALO  
 65 AMS on September 21.



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 67  
 68  
 69  
 70

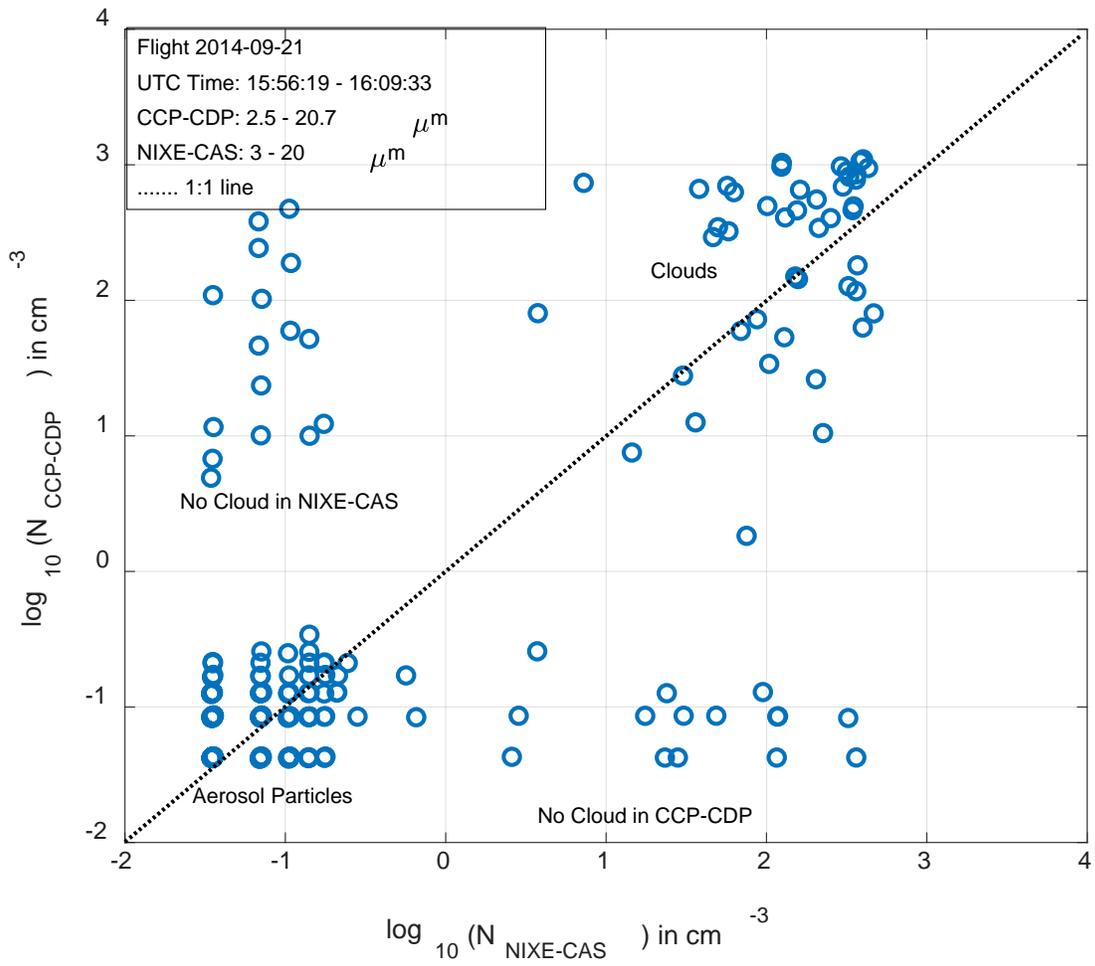
Figure 12. The vertical profile of relative mass fraction of major aerosol chemical species measured by the G1 and HALO AMS, respectively, on September 21



71

72 Figure 13. The G1 and HALO comparison of aerosol CCN concentration ( $S=0.5\%$ ) measured

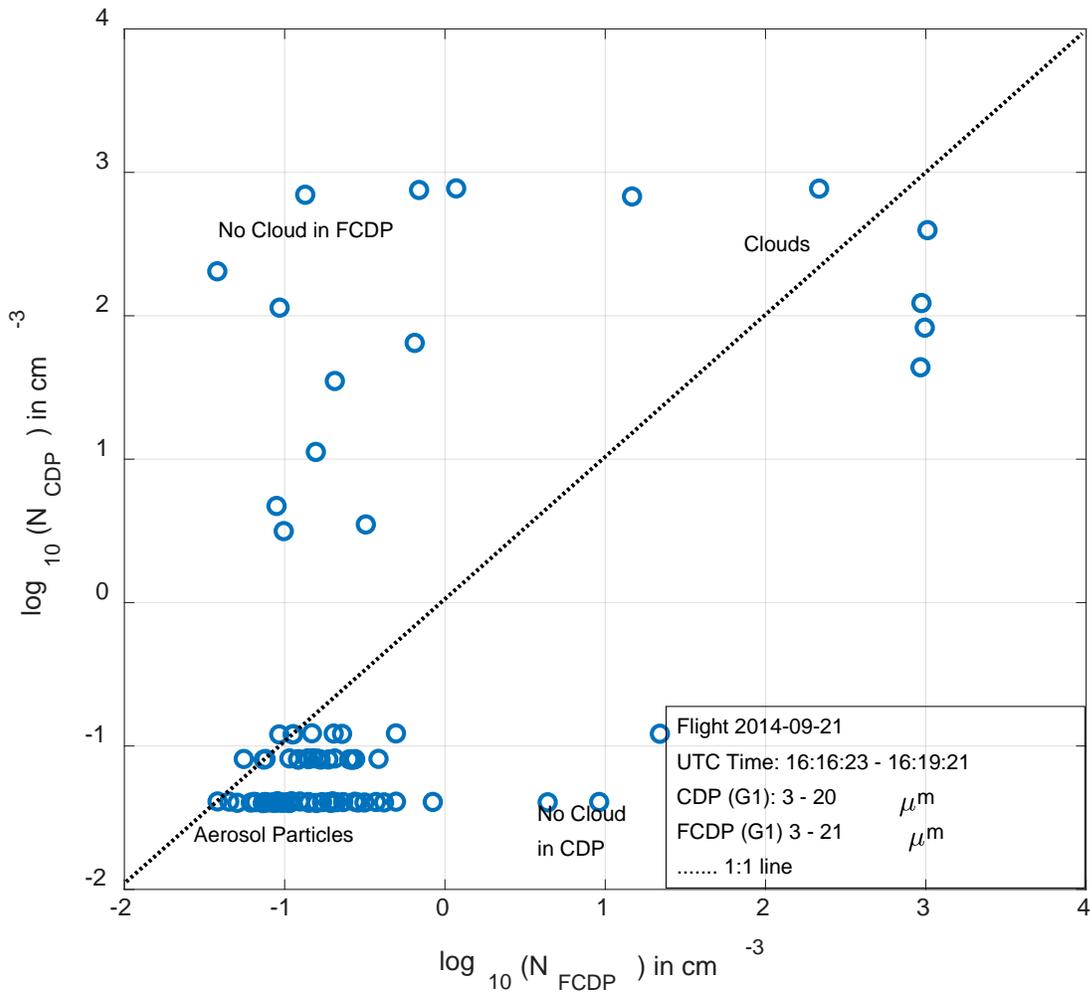
73 on September 21.



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(a)



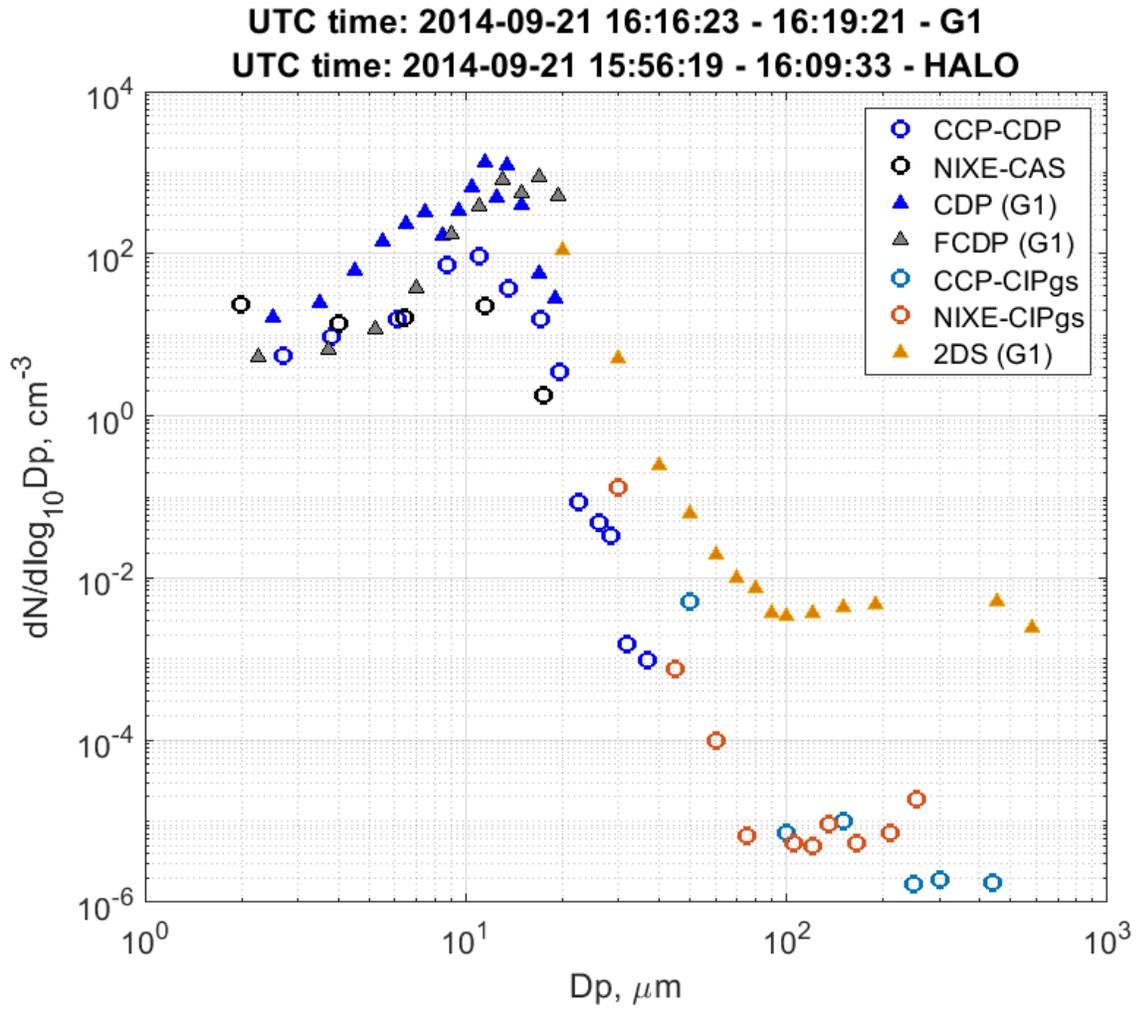
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(b)

78 Figure 14 The comparison of cloud droplet concentrations in the same aircraft (a) between

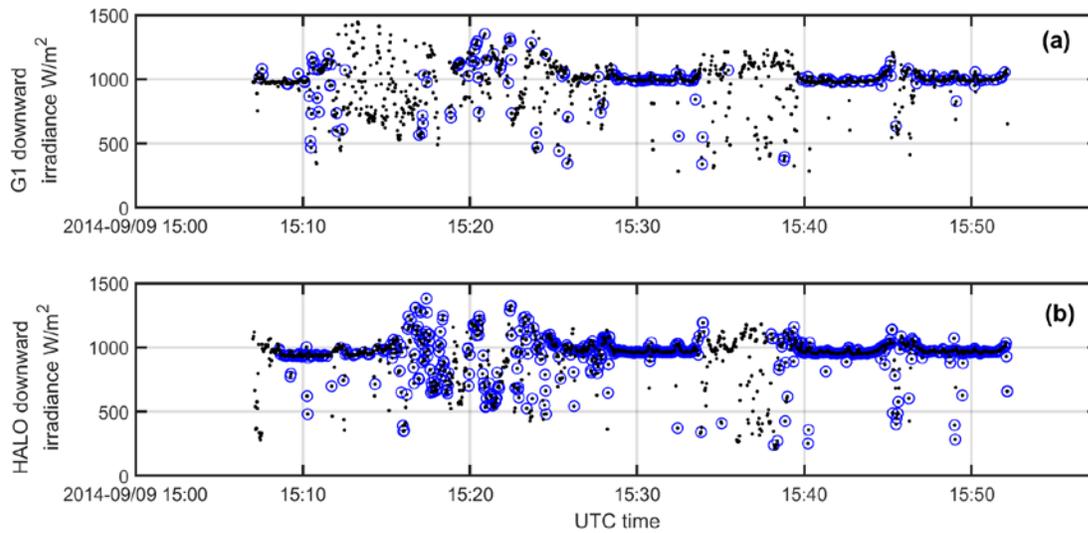
79 NIXE-CAS and CCP-CDP on board HALO; (b) between CDP and FCDP on board the G1.



80

81 Figure 15. The cloud droplet size distribution from the cloud probes on the G1 and HALO.

82



83

84 Figure 16. Time series of the G1 and HALO downward irradiance on September 9. The (a)  
 85 by SPN-1 and (b) by SMART-Albedometer. Black dots represent all data under the general inter-  
 86 comparison criteria. The blue circles represent the restricted navigation criteria.

## Data availability

All ARM datasets used for this study can be downloaded at <https://iop.archive.arm.gov/arm-iop/2014/mao/goamazon/> (DOI: 10.5439/1346559). The full data set from the ACRIDICON-CHUVA campaign is archived and publicly accessible from the HALO database maintained by the German Aerospace Center (DLR) at <https://halo-db.pa.op.dlr.de/mission/5>.

## Competing interests

The authors declare that they have no conflict of interest.