1 Supplement of

2	The Importance of Size Ranges in Aerosol Instrument												
3	Intercomparisons: A Case Study for Aerosol Mass Spectrometer in												
4	the ATom Mission												
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16 **1. The cutoff sizes of selected submicron measurements**

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Table S1: The cutoff sizes of AMS, URG PM₁ cyclones, MOUDI 1 µm stage impactor, and SAGA MC at two dry aerosol densities: 1.7 g cm⁻³ of ATom-2 campaign average and 0.9 g cm⁻³ of typical oily particles (Kuwata et al., 2012; Herring et al., 2015). For AMS, the cut sizes in d_{va} are native and the other two sizes are calculated with Eqs. 1-2 in the main text (here only the upper side is listed); for URG, MOUDI, and SAGA MC, the cut sizes in d_{ta} are native; for MOUDI, the cut sizes in d_p are calculated using Eq. 5.28 in Hinds (2012): $d_{50} = \sqrt{\frac{9\eta D_j(Stk_{50})}{\rho_p UC_c}}$. For circular jets such as MOUDI, 50% collection efficiency corresponds to Stokes Number, *Stkso*, of 0.24. η is air viscosity,

24 D_i is the nozzle size (0.78 mm) (Marple et al., 2014), U is air velocity (a nominal volumetric flow

of 30 L m⁻³ gives 26.16 m s⁻¹ with 40 nozzles at the size of 0.78 mm). The equation is also used to estimate the d_{50} for SAGA MC by dividing the formulas between two conditions, and the base

conversion of d_p or d_{va} to d_{ta} is pressure dependent, d_{ta} at sea level, 6 km, and 12 km are calculated

case gives $d_{ta,sea,50}$ of 1 µm (van Donkelaar et al., 2008) (discussed below at Sect. 10). Since the

for AMS, MOUDI, and SAGA MC. The *P* at 6 km and 12 km are based on the U.S. standard

30 atmosphere, 467 mbar and 185 mbar, respectively (NOAA, NASA, U. S. Air Force, 1976).

Dry Aerosol density		1.7 [g cm ⁻³]				0.9 [g cm ⁻³]			
Diameter [nm]		$d_{p,50}$	$d_{ta,sea,50}$	<i>d_{ta,air,50}</i> (6/12km)	$d_{va,50}$	$d_{p,50}$	$d_{ta,sea,50}$	<i>d_{ta,air,50}</i> (6/12km)	$d_{va,50}$
AMS	ATom-1&-2	443	599	624/670	753	836	789	785/775	753
	ATom-3	455	615	639/687	773	859	811	807/797	859
	ATom-4	564	758	782/837	959	1065	1006	1002/991	959
URG	Standard cut ^a	757	1010		1287	1069	1010		962
	Sharp cut ^a	788	1050		1340	1111	1050		1000
	Sea level / 293 K ^b	749	1000		1273	1058	1000		952
MOUDI	6 km / 293 K ^b	663		912	1127	970		912	873
1 μm stage	12 km / 293 K ^b	454		686	772	741		686	667
impactor	6 km / 250 K ^c	612		845	1040	900		845	810
	12 km / 217 K ^c	371		569	631	616		569	551
SAGA	Sea level / 293 K ^a	749	1000		1273	1058	1000		952

МС	6 km / 250 K ^d	576	798	979	849	798	764
	12 km / 217 K ^d	328	507	558	551	507	496

^a At sea level (P = 1013 mbar, T = 293 K).
^b 293 K, a typical cabin temperature.
^c T is based on the U.S. standard atmosphere if the MOUDI impactor operates at ambient conditions.
^d The SAGA MC inlet operates at ambient conditions.

2. Comparison of the observable particle size ranges between instruments or inlets

Table S2: Comparison of the observable particle size ranges (i.e., contributing chemical 36 37 composition information) between instruments or inlets for the conditions in ATom, a summary of the ATom-1 and -2 dataset. For the "submicron" category in the table, e.g., the AMS to URG 38 39 volume ratio is calculated via $V_{phys,TC}$ / $V_{phys,URG}$, the ratio between the fraction seen by each instrument of the AMP volume. For the "AMP full size range" category, e.g., the AMS vs. AMP 40 41 volume ratio is calculated via V_{phys,TC} / V_{phys}. Lastly, for the "Overlap of AMS and PALMS-AMP" 42 category, e.g., the overlap between AMS and PALMS-AMP vs. AMS is calculated via $V_{phys,TC\&PALMS-AMP}$ vs $V_{phys,TC}$, where $V_{phys,TC\&PALMS-AMP}$ represents the volume measured by both 43 44 AMS and PALMS-AMP. Six ratios are listed due to the three PALMS-AMP products discussed 45 here. The PALMS-AMP 3-min and 60-min are calculated at the reported AMP size resolution (20 46 bins/decade), while the combined 4 bins (Fig. S13) are based on Froyd et al. (2019). Both volume 47 and number fractions indicate the aerosol population represented by each instrument or inlet but 48 don't necessarily mean that all of the aerosol population is collected and measured (i.e., depending 49 on the detection technique). These volume fractions are meaningful for comparing the coverage of 50 the size distribution for particle mass products across aerosol instruments.

Cotogomi	Instrument or inlet	,	Volume [%]	Number [%]		
Category	Instrument of Infet	Mean	Median	SD	Mean	Median	SD
	AMS vs. URG	95.1%	95.1%	14.4%	41.2%	40.7%	24.1%
Submicron	AMS vs. MOUDI	85.2%	87.7%	10.2%	43.9%	44.4%	23.7%
	AMS vs. SAGA MC ^a	96.6%	94.7%	13.9%	41.1%	40.6%	24.0%
	AMS vs. AMP	67.8%	74.3%	22.5%	40.9%	40.5%	23.7%
	MOUDI vs. AMP	78.4%	87.0%	23.1%	89.4%	93.2%	10.7%
AMP full	SAGA MC vs. AMP ^a	70.3%	77.1%	23.9%	99.7%	100.0%	0.8%
size range (2.7 nm to	SAGA filter vs. AMP	96.2%	99.9%	12.9%	98.4%	99.1%	1.9%
$4.8 \mu \mathrm{m} d_p)$	PALMS-AMP (3-min) vs. AMP	53.5%	54.5%	23.1%	4.7%	1.2%	8.7%
	PALMS-AMP (60-min) vs. AMP	71.1%	77.0%	22.3%	8.8%	3.8%	12.6%
	PALMS-AMP (4 bins) vs. AMP	76.1%	82.9%	21.4%	11.0%	5.2%	14.6%

	vs. AMS (PALMS-AMP 3-min)	53.6%	56.3%	24.2%	9.6%	5.0%	12.0%
	vs. AMS (PALMS-AMP 60-min)	67.5%	73.4%	22.9%	17.0%	11.7%	16.0%
Overlap of AMS and	vs. AMS (PALMS-AMP 4 bins)	72.8%	79.2%	21.8%	21.3%	15.5%	18.3%
PALMS- AMP	vs. PALMS-AMP (3-min)	65.3%	69.8%	21.6%	96.0%	97.1%	4.0%
	vs. PALMS-AMP (60-min)	64.8%	70.2%	23.9%	97.9%	98.7%	2.7%
	vs. PALMS-AMP (4 bins)	65.8%	71.5%	24.2%	98.4%	99.0%	2.1%

^a Diffusion loss not considered for SAGA MC but expected to be minimal due to the very high airflows and relatively
 large and short sample line.



53 3. AMS total mass, OA/(OA+SO₄), ambient RH, and AMS inlet RH



56 (c) ambient air RH, and (d) AMS inlet RH.

57 **4. AMS inlet configuration and performance**

58 Aerosols were sampled through a window-mounted NCAR High-Performance 59 Instrumented Airborne Platform for Environmental Research (HIAPER) Modular Inlet (HIMIL) 60 inlet (Stith et al., 2009), located 21.5 m behind the DC-8 nose. The inlet (tall HIMIL, 12") (Rogers, 2011) was raised by 4" using a custom mounting plate, so that the sampling axis was 38.8 cm away 61 62 from the plane skin, ensuring no contamination from the boundary layer of the plane at the sampling location, which has been previously characterized (Vay et al., 2003). The HIMIL is a 63 64 sharp-edged diffuser inlet (which could potentially lead to directional losses for larger particles) 65 (Baumgardner and Huebert, 1993), with an estimated slowdown of ambient air from the speed of the plane by a factor of 3-4 inside the diffuser (D. Rogers, NCAR, pers. comm.). The flow was 66 then sampled into a straight, sharp-edged 3.8 mm internal diameter (ID) stainless steel tube 67 68 pointing in the flow direction (called here the "secondary diffuser", although no actual additional 69 slowdown of the flow happens in this part of the inlet since a redesign in 2016). As described in the main text, the flow rate through this tube was prescribed to be 9 sL min⁻¹ ("s" stands for STP: 70 71 T = 273.15 K, P = 1013 mbar), except at high altitude (> 9 km) where a smaller flow rate was chosen (15 vL min⁻¹; "v" stands for volumetric at in-situ P and T) to increase ram pressure and 72 73 hence boost pressure just before the AMS inlet.

74 The inlet plumbing from the tip of the tube inside the HIMIL to the AMS is 1.5 m long (Fig. S2). To minimize residence time, the flow was operated turbulently (linear velocities between 75 5-15 m s⁻¹ and Reynolds numbers between 2000-5000) up to the takeoff of the excess flow. 76 77 However, for the full range of diameters sampled by the AMS, the particle Reynolds number was 78 always <1 and Dean numbers for bends were less <1000, hence calculated particle losses assuming 79 mostly spherical particles are overall modest (Fig. S3). The overall transmission is mostly 80 impacted by the 90° bend inside the HIMIL, the slight oversampling at the point where the main 81 excess flow is taken out ("main takeoff," Fig. S2), and the diffusion losses downstream of the last 82 critical orifice. Overall, the calculated inlet plumbing transmission does not affect the higher end 83 of the instrumental transmission curve used in this work (Fig. S3, bottom). It has a minor impact 84 on the sub-100 nm size range of the transmission curve. However, since that part of the 85 transmission curve a) does not really impact the volume analysis presented in this work (i.e., main text Sect 3.3) and b) was not determined in-situ for ATom (a 20% uncertainty at least; literature 86 87 values are assumed (Zhang et al., 2004a; Knote et al., 2011)), adding this additional correction

does not seem warranted. Also, there are additional uncertainties regarding the turbulent loss
calculations, shown in Fig. S3, that would probably require experimental confirmation.

90 The calculated transmission does not account for transmission prior to the first bend. 91 However, as shown in Fig. S4, using a completely different inlet with the proven supermicron 92 transmission (McNaughton et al., 2007) while maintaining the same conditions in the downstream 93 plumbing, had no appreciable impact on the intercomparisons with other optical aerosol 94 instruments during back-to-back inlet switches. So for the size range of interest, we assume that 95 they are negligible.

96 As described in Bahreini et al. (2008), for reliable airborne AMS performance, especially 97 over the large range of ambient pressure sampled with the NASA DC-8 (down to about 170 mbar), 98 a device is needed that maintains constant pressure in front of the AMS aerodynamic lens and 99 hence ensures a constant sampling flow and, more importantly, consistent aerodynamic focusing 100 of the aerosol onto the AMS vaporizer (Zhang et al., 2004b; Huffman et al., 2005). Bahreini et al. 101 (2008) accomplished this with a pressure controlled inlet (PCI) design consisting of a small volume 102 between two critical orifices (C.O.; hereafter the first one encountered by the airflow is referred to 103 as C.O. #1, and the bottom is referred to as C.O. #2) that is kept at constant pressure and placed in 104 front of the AMS lens. Ideally, the pressure in the volume is lower at all times than ambient 105 pressure, thus ensuring that both orifices remain under critical flow conditions. C.O. #2's size is then chosen to provide a suitable flow into the aerodynamic lens (ideally around $1.5 \text{ scm}^3 \text{ s}^{-1}$, cf. 106 107 Section 2.2) while the top orifice has to be large enough to ensure enough excess flow at all 108 altitudes.

The design in Bahreini et al. (2008) had two main drawbacks: a large residence time (~ 5 s) due to a large internal volume, which could impact the sampling of very volatile aerosol at altitude due to potential evaporation losses, and poor performance at the very low air pressures needed for operation on a plane such as the DC-8. The first one was addressed by reducing both the length and ID and using an improved internal takeoff design, to achieve an internal volume of only 3.5 cm^3 (vs. ~ 30 cm^3 in the original design).

115 The reason for the poor performance at lower pressure is discussed in Chen et al. (2007): 116 for a flat critical orifice, reducing inlet pressure leads to a larger angle for the air expanding behind 117 the orifice and eventually to recirculation and particle loss due to impaction. This depends on the 118 exact parameters of the expansion, so that in general smaller orifice sizes and smaller tubing sizes downstream will increase the likelihood of losses. How this works in practice is illustrated in Fig.
S5: for the regular, ground-based AMS with a PM₁ lens at 1 atm, the large particle losses are
mostly at the back of the C.O. (Williams et al., 2013), due to the small orifice and small expansion
tube. For the larger C.O. #1 facing ambient pressure used in both Bahreini et al. (2008) and this
work, even at low pressures, losses at this orifice are less critical (Fig. S5). This is also the case
for newer AMS lens designs (Williams et al., 2013; Peck et al., 2016) with optimized expansion

126 On the other hand, for C.O. #2, the one at the bottom of the PCI and facing the aerodynamic 127 lens, as the pressure in the PCI decreases, an expansion into a small tube (¼" outer diameter (OD), 128 as used by (Bahreini et al., 2008)) leads to significant losses at low pressures (Fig. S5). To address 129 this, a double diffuser volume was designed and built-in collaboration with X. Wang and P. 130 McMurry (University of Minnesota) (Fig. S6, Volume A1), which allowed for a controlled 131 expansion into a 30 mm diameter volume. This design was successfully flown on the ARCTAS 132 (Jacob et al., 2010) and DC3 (Barth et al., 2015) missions with a PCI pressure of 130 mbar and 133 minimal losses (Cubison et al., 2011; Yang et al., 2015). However, both the ¹/₄" OD tubing present 134 in this design and possibly the overall gradual transition to larger diameters in the initial diffuser 135 can lead to significant evaporation artifacts for the pure dry ammonium nitrate aerosol used to 136 calibrate the AMS, as illustrated in Fig. S7. While there is no evidence that this issue impacted 137 ambient, lower-volatility aerosol, it introduced a significant additional uncertainty on the AMS 138 sensitivity calibration. Hence Volume A was first modified (Volume A2, which had a single fitting 139 as an inlet/C.O. mount) and two other designs (B and C in Fig. S6) were tested in subsequent 140 airborne missions (both on the NASA DC-8 and NCAR/NSF C-130). While Volume A2 still 141 exhibits some evaporation artifacts the later designs did not (Fig. S7), and the transmission of 142 Volume C matched the performance of the Volume A2, as shown in Fig. S8. Nevertheless, it 143 should be noted that overall the performance of both Volume B and C is worse than the originally 144 designed Volume A1, which worked well up to 300 μ m orifice size, for reasons that are still 145 unclear.

Hence for ATom (starting with the Aug 12, 2016, flight on ATom-1, Volume A2 was flown previously to that), Volume C was flown using a 220 μ m C.O. for C.O. #2. As Fig. S8 and this manuscript overall make clear, this resulted in reproducible near-reference PM₁-aerodynamic lens performance (Hu et al., 2017) with no evaporation artifacts (Fig. S7), but with the drawback that 150 constant pressure (in the expansion volume) could only be maintained up to a pressure altitude of 151 about 9 km. While not ideal, this configuration guaranteed that, even at max DC-8 altitude, the 152 pressure in the aerodynamic lens would never go below 1.33 mbar, and hence the aerodynamic 153 focusing into the vaporizer was not substantially impacted (but some additional diffusional losses 154 are shown in Fig. S3).

It should be noted that overall, based on the calculations shown in Fig. S5 and the improved geometry interfacing the PCI with the aerosol lens, both lower PCI pressures and slightly above the reference performance (Hu et al., 2017) should be possible. The observed performance is likely related to the impact of mechanical imperfections on the overall flow profile through the PCI, something that is not unusual for aerodynamic focusing devices (Schreiner et al., 1999) and that is currently being further characterized. The improvement in transmission from ATom 1-3 to ATom-4 is likely related to this.

162 In summary, the current CU-AMS aircraft inlet provides to our knowledge the best 163 transmission of a PM₁ lens-based airborne system with reproducible performance up to 13 km and 164 very low residence times over the full atmospheric column. Recently, Molleker et al. (2020) have 165 described a new airborne AMS inlet system based on PM2.5 lens with a larger size range and 166 comparable residence times to the system described here, although it is currently unclear how well 167 it works for small particles and how well it performs in the field. But it highlights that there are 168 realistic options to expand the airborne size range in the future beyond the limits currently 169 described in this work.



170 171 Fig. S2: Left: Simplified flow diagram for the AMS inlet assembly (not to scale, only the most 172 relevant valves shown). Airflow is turbulent outside the cabin, and laminar inside. Air is pulled constantly through HIMIL at 9 sL min⁻¹ up to ~9 km and 15 vL min⁻¹ above that. The sum of AMS 173 flow and excess flow is 2 vL min⁻¹ controlled by two tandem critical orifices. Also shown is the 174 175 line to the LARGE inlet operated by the AMP team that was used at times instead of the HIMIL 176 to check performance (see the comparison in Fig. S4) (Brock et al., 2019). Right: Total residence 177 time from the tip of the secondary diffuser inside the HIMIL to the AMS, as a function of altitude, 178 color-coded by the different parts of the inlet assembly.



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183 Fig. S3: Top: Calculated aerosol transmission through the AMS inlet plumbing for all ATom 184 altitudes (including aerosol gravitational and diffusional losses and aspirational and inertial sampling efficiencies), assuming the average ATom1-2 mass-weighted density of 1.63 g cm⁻³ and 185 186 using the Pui et al. (1987) parameterization of turbulent losses in bends. Middle: Same as top panel, 187 but using the turbulent loss formulation of McFarland et al. (1997) for describing losses in bends, 188 which predicts higher transmission for intermediate Dean numbers. Bottom: Effect of these losses 189 on the AMS transmission function at sea level and the highest ATom altitude, where the lower 190 flows increase the losses due to diffusional deposition somewhat. Note that these transmission 191 calculations do not include (a) the losses in the HIMIL inlet, which have not been fully 192 characterized, but given the geometry likely has a d_{p50} of around 1-1.5 µm (Porter et al., 1992;

 d_p [nm]

- Baumgardner and Huebert, 1993; Sheridan and Norton, 1998; Hermann et al., 2001), (b) the potential oversampling of large aerosols at altitude in the HIMIL secondary diffuser, and (c) the losses in the PCI. (c) is discussed in detail below (Fig. S5), (b) was not included in the calculations since there is scant experimental data on the validity of the parametrizations normally used for turbulent supersampling at higher Mach numbers (e.g., 0.2-0.4 under ATom conditions) and there is also limited data on what the actual flow speed inside the HIMIL is (slowdown by the primary diffuser is assumed to be about 3-4 times, D. Rogers, NCAR, pers. comm.). However, (a) and (b)
- 200 were indirectly characterized by the comparisons with the LARGE inlet, that do not include either
- 201 of them and showed identical concentrations (Fig. S4).



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204 Fig. S4: Top: Comparison of AMS speciated mass with UHSAS volume and PM1 550 nm 205 scattering (both operated by the NASA Langley Group) for a period during Research Flight 4 of 206 the NSF/NASA DC3 Mission (Barth et al., 2015), onboard the NASA DC-8, with the AMS 207 sampling line being switched between the AMS HIMIL inlet and the LARGE inlet (the same inlet 208 that the AMP Group used during ATom). Bottom: Average UHSAS volume distributions for the 209 five periods shown. While comparisons like these were performed repeatedly during ATom with 210 similar results, this one was chosen for the much higher concentrations and availability of 211 concurrent optical measurements.



213 214 Fig. S5: Top: Calculated aerosol losses for ammonium nitrate aerosol (the standard AMS 215 calibrant) for the CU AMS in the expansion behind the critical orifice (C.O.) facing ambient 216 pressure (C.O. #1, top of the PCI), calculated for 3 different altitudes, based on a sigmoidal fit to 217 the experimental data for the parametrization derived in Chen et al. (2007). Even at sea level, the 218 losses are small compared to both the AMS transmission curve and the inlet plumbing losses. For 219 comparison, the same calculation was performed for a standard AMS operating with a PM1 inlet 220 (smaller orifice, smaller upstream diameter). A comparison with the published transmission curve 221 for that instrument (Hu et al., 2017) suggests that in fact losses at the back of the C.O. are the main 222 reason for the observed shape of the curve on the high end (losses at the front side are 1-2 orders 223 of magnitude less for the sizes shown). Also shown is the performance of the same inlet at 200

224 mbar ambient pressure, suggesting major losses at low pressure/high sampling altitudes. Bottom: 225 Same calculation for the bottom C.O. (C.O. #2, the one facing the AMS aerodynamic lens) of the 226 CU AMS PCI. Three different aerosol densities are shown (Hydrocarbon-like OA, HOA; 227 Ammonium Nitrate, AN and Sulfuric Acid, SA) to explore the impact of density on the overall 228 transmission. Dotted lines show the transmission for a PCI geometry such as in Bahreini et al. 229 (2008), with a ¹/₄" tube behind the critical orifice, which leads to significant losses relative to the 230 regular AMS transmission (the Volume A1 design had a similar restriction right after the 231 expansion). Solid lines show the same calculation for an expansion volume of 15 mm diameter, as 232 used in this work (see Volume C in Fig. S6, the diameter of the expansion fitting throat is used, 233 i.e., the smaller side), showing much improved performance. It should be noted, however, that 234 while these equations capture the trends well, the actual losses observed for these low pressures 235 are significantly larger (see Fig. S8).



236

237 Fig. S6: Schematics of the three expansion volumes that have been flown with the CU AMS PCI 238 over the past decade. Top: Initial design, based on computer fluid dynamics (CFD) calculations of X. Wang and P. McMurry (pers. comm.). This design includes a 7° conical expansion region 239 240 behind the critical orifice, a 30 mm central section, and a 10° conical contraction region. The lens 241 interface is identical to the standard PM1 AMS Inlet (Zhang et al., 2004a; Canagaratna et al., 2007). 242 The critical orifice was initially mounted as shown for the ARCTAS and DC3 missions (Jacob et 243 al., 2010; Barth et al., 2015), which, in some cases, led to evaporation artifacts of ammonium 244 nitrate calibration particles in the small tube immediately downstream of the orifice (Fig. S7). For 245 part of SEAC⁴RS and the first five flights of the ATom mission, it was fitted with a new inlet 246 similar to Volume B (referred henceforth as "Volume A2"), which ameliorated the evaporation 247 artifacts during calibrations, but also for unknown reasons worsened performance at low pressure (Fig. S8). Middle: A redesign first flown during SEAC⁴RS (Toon et al., 2016), with the same inner 248 249 dimensions as Volume A, but a) no obstructions behind the critical orifice and b) a new valve 250 assembly with shorter overall length, more robust mounting and slight prefocusing of the aerosol 251 going into the aerosol lens, which should be beneficial for large particle transmission (Williams et

- al., 2013). While this volume exhibited no obvious evaporation artifacts, its performance at low
- 253 pressure was significantly worse than Volume A (Fig. S8). It was used at higher pressure (430
- mbar) for the WINTER and KORUS-AQ deployments that had a clear lower tropospheric focus.
- 255 Bottom: For ATom, a new expansion volume C was designed for ATom on the assumption that a
- 256 rapid expansion post-orifice at the highest possible angle is preferable. The focusing diffuser angle
- 257 was reduced to 7°, which should minimize losses. The rest of the design is similar to Volume B.



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Fig. S7: Top: Evaporation artifact for ammonium nitrate calibration particles observed for Volume 261 A and its cause: Sampling ammonium nitrate with Volume A shows a clear bimodal distribution, 262 263 with the main peak significantly shifted down from the nominal aerosol size, indicating 264 evaporation/shattering. For Volume B, this artifact is normally not observed, but can be induced 265 by inserting different lengths of 1/4" OD tubing behind the critical orifice mount, mimicking the

- 266 geometry in Volume A. Bottom 3 panels: Comparison of the size distributions observed for all
- three volumes during ATom (e.g., after the inlet of Volume A was replaced), again when sampling
- 268 monodisperse 400 nm (d_m) ammonium nitrate aerosol. Volume A still exhibits an evaporation
- artifact, though considerably smaller. While no bimodal distribution is observed for Volume B, aslight shift of the main distribution is still apparent. Volume C does not show any of these issues,
- 271 except possibly at the lowest pressure.



272

273 Fig. S8: Intercomparison of the performance of the three expansion volumes shown in Fig. S6 as 274 a function of PCI pressures (and corresponding C.O. sizes). Monodisperse 400 nm (d_m) ammonium 275 nitrate aerosol was used and the transmissions were determined by Event trigger/CPC comparison, 276 as done for the derivation of the lens transmission described in the main text. It should be noted 277 that to cover the full altitude range of the DC-8, without losing sampling flow into the AMS, an 278 orifice size of at least 250 µm and a PCI pressure of 135 mbar is needed. All volumes perform well 279 at 430 mbar, roughly comparable to the Bahreini et al. (2008) published data at 467 mbar. For 280 reasons that are still unclear, Volume B performs poorly below that pressure, while for both 281 Volume A and C a more gradual loss in transmission is observed. For ATom operation, Volume 282 C was operated at 250 mbar (220 µm C.O.), as a compromise between good transmission at lower altitudes and some loss in flow at max altitude. 283



Fig. S9: Profiles of the AMS air beam (left) and particle beam (for monodisperse 400 nm ammonium nitrate particles) recorded at the entrance of the AMS ionizer with a beam width probe (Huffman et al., 2005) over the course of ATom-2 (Legend indicates the ICAO code for the airport the profiling was performed). This was a standard calibration taken at the end of most flights during ATom, and served as confirmation that no lens misalignment had occurred.

292 5. AMS composition-dependent CE



 $\begin{array}{ccc}
 & (a) & c_E & (b) & c_E \\
\hline
 & Fig. S10: Frequency distributions of composition-dependent$ *CE* $and altitude for ATom-1 to -4 \\
\hline
 & studies: the left panel colored by the cross frequency of$ *CE* $and altitude and the right panel colored \\
\hline
 & by <math>V_{phys,TC}$.

297 6. AMS in-field calibrations





298 299 Fig. S11: Time series of results from in-field calibrations in ATom-1 and -2 for (a) the ionization 300 efficiency of nitrate (IE_{NO3}), IE_{NO3}/AB (air beam); (b) the relative ionization efficiency (RIE) for 301 ammonium (NH₄), sulfate (SO₄), and chloride (Chl); (c) the ion ratios; (d) the measured artifact 302 signal ratios for CO₂⁺/pNO₃ "Pieber Effect" (Pieber et al., 2016) and Cl⁺/pNO₃ "Hu Effect" (Hu 303 et al., 2017). The most critical calibrations were performed immediately after each flight. The 2σ 304 uncertainties of RIE_{NH4}, RIE_{SO4}, and RIE_{Chl} are 4% (6%), 4% (2%), and 5% (8%), respectively for 305 ATom-1 (ATom-2), all smaller than the reported values from (Bahreini et al., 2009). The 2σ 306 uncertainty of the ionization efficiency (normalized as its ratio to the air beam signal, IE_{NO3}/AB) is

- 307 6% for ATom-1 and 15% for ATom-2 (excluding the two large numbers measured on Jan 29 and
- 308 Feb 1, 2017; if averaging after Feb 1, the uncertainty drops down to only 4%, showing that the
- 309 AMS ionization efficiency performance became very stable for the latter two-thirds of the ATom-
- 310 2 deployment after the unstable start).



313 Fig. S12: Frequency distributions of dry aerosol density (ρ_m) and OA density (ρ_{0A}) for ATom-1 (left) and ATom-2 (right) estimated from the AMS data. ρ_m is calculated via Eq. 5 in the main 314 315 text and doesn't consider BC and sea salt. Including BC and sea salt in the aerosol density calculation only changes the ρ_m averages within 0.01 g cm⁻³. When OA is below DL, ρ_{0A} is 316 317 excluded due to the large uncertainties in the ρ_m estimation. The mass-weighted and the 318 unweighted density campaign averages are plotted. Lower mass concentrations were associated 319 with more aged conditions and more oxidized OA, and thus higher densities. As expected, the 320 unweighted density is higher than the mass-weighted one. High ρ_{0A} , such as 2.0-2.3 g cm⁻³, is 321 predicted for a small percentage of the data points, as the tail of the frequency distribution. The 322 noise in H/C and O/C broadens the distributions of ρ_{0A} and the effect can be reduced by averaging 323 to a longer time scale than the 1 min time resolution shown here.

8. PALMS relative data coverage and chemical information content coverage for PALMS-

325 AMP derived aerosol compositions for ATom conditions



326 327 Fig. S13: PALMS composition coverage across the accumulation and coarse modes (K. Froyd, 328 pers. comm.). For ATom-2, the AMP size distribution (black line, campaign average) is divided 329 into 4 bins. The PALMS fractional composition data is calculated as an unweighted mean within 330 each bin. The red solid line shows the PALMS relative data coverage within each bin, showing variation in relative coverage within each bin. For example, in the smallest bin (Bin 1), the PALMS 331 332 composition means will be weighted to sizes >140 nm. If the aerosol composition is homogeneous 333 within each bin, this uneven composition coverage by PALMS introduces no bias. Actual biases were quantified in Froyd et al. (2019) using atmospheric data. 334



335

336 Fig. S14: (Top) PALMS detected particle numbers per size bin at several size resolutions at a 3-337 min time interval (i.e., the time resolution that the public PALMS-AMP mass products report at) 338 and longer averaging time scales at the AMP reported size resolution (20 bins/decade) for ATom-339 2. Note that each altitude bin is 800 m (16 bins in total), consistently throughout this study. The 340 detected particles per size bin are calculated with AMP number size distributions, PALMS detection efficiency (Fig. 6 in Froyd et al. (2019)), and PALMS flow rate (0.75 L min⁻¹), and then 341 342 scaled to reported PALMS detected spectra to ensure a proper evaluation of the instrument 343 performance. (Bottom) The ATom-2 campaign averaged AMP number size distributions with the 344 fraction that is characterized by PALMS-AMP.



345 346 Fig. S15: PALMS detected particles during a 3-min (top), 60-min (middle), and campaign-wide 347 (bottom) averaging time scale for the conditions in ATom-1 (left) and ATom-2 (right). The 348 detected particles per AMP size bin are calculated as described in Fig. S14. For each altitude, these 349 particle numbers are then scaled to real PALMS detected spectra to ensure a proper evaluation of 350 the instrument performance. Dashed rectangle areas represent the 4 larger bins used for the 3-min 351 PALMS-AMP product (Fig. S13). The solid black lines represent the minimum size ranges to have 352 5 particles (consistent with Froyd et al. (2019)) detected for the range between 100 nm and the left 353 black line, or between the right black line and the upper size limit of AMP.

354 9. SAGA MC inlet design



Fig. S16: Schematic diagram of SAGA MC IC during ATom.

357 **10. SAGA MC inlet transmission**

358 van Donkelaar et al.(2008) estimated the cutoff size to be ~1 μ m ($d_{ta,sea}$) for SAGA MC. 359 Due to the lack of an available SAGA MC inlet transmission profile from literature, we take the 360 MOUDI 1µm stage impactor transmission (discussed in Sect. 3.1 in the main text; Fig. S20) as an approximation to evaluate the change in SAGA MC *d*₅₀ (particle diameter with 50% transmission) 361 362 as DC8 climbs. P, T, and air velocity affect d₅₀. During ATom, the P effect is the same for SAGA 363 MC as compared to MOUDI since the two inlets operate at ambient P. The other two effects make 364 the SAGA MC d_{50} differ from the MOUDI d_{50} . First, the T of MOUDI impactor airflow is assumed 365 to be the cabin T (293 K) (Guo et al., 2016), while the T of SAGA MC air flow is assumed to be the outside ambient T (as discussed in the main text). Second, the air velocity in the SAGA MC 366 manifold is expected to increase with altitude, leading to further shift of cutoff size to smaller sizes. 367 In contrast, MOUDI has a choked airflow of 30 L m⁻³. Here, we take the SAGA MC airflow as an 368 369 approximation to estimate the change in the manifold airflow (only the ratio of air velocity matters 370 in this calculation), since the vacuum of both were provided by venturi pumps. The SAGA MC 371 airflow increased linearly below 10 km and stayed nearly constant above that (Fig. S17b).





Fig. S17: SAGA MC inlet (a) transmission (compared to the MOUDI 1µm stage impactor and AMS), (b) airflow, outside ambient temperature, and true air speed of DC8. Note that the SAGA MC transmission is assumed with the MOUDI transmission profile to investigate the altitude dependency. The SAGA MC and MOUDI transmissions are displayed at sea level (STP), 6 km,

- and 12 km (*P* based on the U.S. standard atmosphere for both; ambient *T* for SAGA MC inlet and
- cabin *T* for MOUDI) (NOAA, NASA, U. S. Air Force, 1976). In contrast, the AMS transmission
- is valid up to ~9 km and a similar performance is expected at the max altitude. The conversions
- between d_p , d_{ta} , and d_{va} are based on the ATom-2 dry aerosol density of 1.70 g cm⁻³. A complexity
- 381 of SAGA MC *d*₅₀ is not considered for this plot, as the SAGA MC inlet size-selects particles at
- 382 ambient conditions (i.e., particles with liquid water if deliquesced) while the MOUDI inlet size-
- 383 selects dry particles (Guo et al., 2016). This effect is larger below 3 km, where ambient RH was
- higher (75% to 40%, from the surface to 3 km), and minimal above 3 km, where ambient RH was
- on average below 40%. So the plotted 6 km and 12 km SAGA MC *d*₅₀ are fairly accurate.

386 **11. SAGA filter inlet transmission**

387 The transmissions for large particles are based on the UNH inlet transmission, Fig. 8C in 388 McNaughton et al. (2007), and adjusted to the ATom conditions, ρ_p of 1.7 g cm⁻³ and true air velocities of DC8. For the coarse mode, the transmissions would shift when the ρ_p of 1.7 g cm⁻³ 389 390 (estimated for submicron) do not represent some pure dust plumes with a density of 2.5 g cm⁻³ or 391 sea salt plumes with a density of 1.45 g cm⁻³ (Froyd et al., 2019). While particles were size-selected 392 at ambient conditions (i.e., with particle liquid water), we apply the transmission to AMP dry 393 particle size distribution without considering the effect of particle water on size. The bias with this 394 simplification is small for less hygroscopic coarse particles (i.e., dust) and large for more 395 hygroscopic particles (i.e., sea salt) (Kumar et al., 2009a, 2009b; Nenes et al., 2014). Furthermore, 396 the bias is larger below 3 km, where ambient RH was higher and decreased from 75% to 40% 397 (surface to 3 km), and smaller above 3 km, where ambient RH was on average below 40%. The 398 diffusion losses (turbulent regime) for small particles are estimated for an approximate 140 cm 399 sample line tubing with an inner diameter of 5 cm (more accurate inlet dimensions discussed 400 below), for which 100 cm are outside of the plane (ambient T) and the rest 40 cm are inside. Here 401 we take the average of ambient T and a typical cabin T of 293 K due to the high air flow of SAGA 402 and the use of Delrin material for most of the in-cabin sample line, which doesn't conduct heat as 403 well as the steel inlet. Note that, using ambient T to estimate the diffusion loss only results in 0.7% 404 lower transmission at the minimum particle size of 2.67 nm d_p , i.e., 79.4% vs. 80.1%). The 405 dimensions of the SAGA filter inlet are as follows. The UNH diffuser-type inlet tip was designed 406 with shrouds (McNaughton et al., 2007). It extends inside of the cabin only 5 cm before it is 407 connected to a large ball valve, ~ 10 cm long, with the same inner diameter as the inlet tube. 408 Downstream of the ball valve, a diffuser, ~ 20 cm long, expands from 5 to 9 cm just before 409 connecting to the filter holder (filter diameter is 9 cm). Both the ball valve and diffuser are made 410 of Delrin. The collection efficiency of the Zefluor filter (1 µm pore size) that SAGA uses is not 411 estimated because Zíková et al. (2015) find it to be significantly underestimated by theoretical 412 calculations and requires in-lab characterization.

Aspiration efficiency is the ratio of the number concentration of particles that enter the sampling probe to that in the ambient air. Here, we estimate the efficiency with Eq. 5 in Weiden et al. (2009) (Belyaev and Levin, 1972, 1974) and the inlet tip internal diameter of 7.77 mm (McNaughton et al., 2007). The range of validity for this equation is shown as the two dashed 417 curves in panel (b) of Fig. S18 due to the Stokes Number. In general, the aspiration effect is less 418 for small particles vs. large particles, and large particles are sub-sampled at low altitudes (< 8 km) 419 and over-sampled at high altitudes (> 8 km). Due to the very limited range of equation validity and 420 the fact that both the nozzle air velocities (120-333 m s⁻¹) and true air velocities (133-234 m s⁻¹) 421 are above the recommended limit of the equation, 30 m s⁻¹ (von der Weiden et al., 2009), the 422 aspiration efficiency is not applied to the SAGA filter inlet transmission.



423 424 **Fig. S18:** SAGA filter inlet transmission (a) and aspiration efficiency (b) plotted with d_p and 425 altitude during AToms.

12. Dust contribution to aerosol volume in the AMS size range



Fig. S19. Frequency distributions of submicron dust volume fraction for ATom-1 (left) and ATom-

429 2 (right). *V*_{dust,TC} is calculated by applying the AMS transmission curve to the PALMS dust volume.



430 **13. URG PM₁ cyclones, MOUDI 1 μm stage impactor transmissions**

431 432 Fig. S20: The transmissions of URG PM₁ standard cut (Model: URG-2000-30EHB) and sharp cut 433 (Model: SCC 2.229) cyclones, and MOUDI 1µm stage impactor vs. transition-regime aerodynamic 434 diameter ($d_{ta,sea}$; as the size range of interest to this study is in the transition regime, requiring a 435 "slip correction"). The MOUDI experimental data is retrieved from Fig. 5 in Marple et al., (1991) using the Data Thief software (version 1.7). The d_{50} (corresponding d_{ta} at 50% transmissions) for 436 this stage at P = 1013 mbar, T = 293 K, and a flow rate of 30 L m⁻³ was reported to be 1.00 µm 437 438 (Marple et al., 1991). Similarly, the URG experimental data are retrieved from an official URG 439 specification sheet for the standard cut version (http://www.urgcorp.com/images/PDF%20Files/Resources/Cutsheets/URG-2000-30EHB.pdf; 440 441 last accessed on July 1^{st,} 2019) and a technical report from BGI Inc for the sharp cut version 442 (https://bgi.mesalabs.com/wp-content/uploads/sites/35/2015/02/scc_btr-2.229.pdf; last accessed 443 on July 1st, 2019). We assume the aerodynamic diameters from the two reports applied the 444 Cunningham slip corrections (C_c) for the transition regime size range since the slip correction factor, 1.166, is not negligible at 1 μ m (and C_c increases non-linearly as the particle size decreases). 445 446 The d_{50} is estimated to be 1010 nm and 1050 nm for the standard cut and the sharp cut cyclones,

447 respectively.



448 14. Predicted decrease in aerosol size when dried from ambient condition

450 Fig. S21: Frequency distributions of the decrease in particle size if the particles lose liquid water 451 content completely due to sample line heating: (a) ATom-1 and (b) ATom-2. The water associated 452 with particulate inorganic species is predicted with AMS plus SAGA-MC aerosol composition 453 (AMS SO₄ and NH₄, SAGA-MC total nitrate (pNO₃+HNO₃)) and E-AIM thermodynamic model 454 (Clegg et al., 1998a, 1998b, 2003; Wexler, 2002; Friese and Ebel, 2010), and the water associated 455 with organics is predicted using Eq. 5 in Guo et al. (2015) (Petters and Kreidenweis, 2007) with 456 the AMS inferred ρ_{0A} from this study and organic hygroscopicity parameter (κ_{0A}) of 0.2 from 457 literature for aged OA (Jimenez et al., 2009; Cerully et al., 2015).







460

Fig. S22: The transmission curves of AMS (valid up to ~ 9 km and expected a similar performance 461 at the max altitude), URG PM₁ cyclones operated at sea level, MOUDI 1 µm stage impactor 462 operated at sea level (i.e., P = 1013 mbar), 6 km, and 12 km (at T = 293 K, typical cabin temperature 463 464 and P based on the U.S. standard atmosphere) (NOAA, NASA, U.S. Air Force, 1976). (a) Results with an aerosol density of 1.70 g cm⁻³, estimated from ATom-2 (ATom-1 is similar); (b) Results 465 calculated with an aerosol density of 0.9 g cm⁻³ (typical of oily particles encountered on some 466 467 laboratory and field studies) (Kuwata et al., 2012; Herring et al., 2015). The contrast in (a) and (b) 468 illustrates the effects of aerosol density on the conversions between geometric diameter (d_p) , 469 vacuum aerodynamic diameter (d_{va}), and aerodynamic diameter (d_{ta} ; for the MOUDI impactor and 470 URG cyclone; note that the MOUDI 6 km and 12 km profiles are slightly off at the plotted $d_{ta,sea}$ 471 axis but precise at d_p and d_{va} axes; the cut sizes for AMS, URG, and MOUDI are summarized in 472 Table S1). It also shows that the AMS cutoff size in d_{ta} depends on aerosol density (also altitude, 473 see Table S1 for examples) and the MOUDI impactor cut size depends on altitude.



16. Volume closure: *V*_{chem} **vs.** *V*_{phys,TC}





479 $V_{phys,TC}$ [µm⁶ scm⁶] $V_{phys,TC}$ [µm⁶ scm⁶] 480 **Fig. S24:** Comparison between V_{chem} and $V_{phys,TC}$ for ATom-1 (left) and ATom-2 (right) at 5-min 481 average level. The top, middle, and bottom panels are colored by sea salt, the fraction of data 482 sampled in clouds (the cloud indicator is from the 2nd generation Cloud Aerosol and Precipitation 483 Spectrometer, CAPS) (Brock et al., 2019; Spanu et al., 2020), and submicron dust volume fraction 484 (Fig. S19), respectively. For the intercomparison, we used 1 min AMS data, for which the raw 485 mass spectra were averaged prior to data reduction and analysis. During the step, the spectra 486 collected in the clouds were not removed. In contrast, the NOAA size distribution was averaged

- 487 from 1 s data with exclusion of cloud impacts (i.e., 13% of the AMS data coverage). Therefore,
- 488 there could be deviation derived from the way the raw data is processed and the impacts are
- 489 investigated via panel (c) and (d).



490 491 **Fig. S25:** Comparison between V_{chem} and $V_{phys,TC}$ for (a) ATom-1 and (b) ATom-2. Data points are 492 colored by altitude and averaged to 5 min resolution. Compared to Fig. 5 (main text), the plotted 493 V_{chem} is recalculated by subtracting the AMS estimated ρ_{0A} with 0.2 g cm⁻³, for a sensitivity test 494 while keeping the other volumes the same.



Fig. S26: Comparison between V_{chem} and $V_{phys,TC}$ for ATom-1 (a) and ATom-2 (b), data points colored by the fraction of V_{phys} removed when applying the AMS transmission. The data points (marker only) are binned by a 20% interval based on the fraction and plotted with the 10th to 90th percentiles in each bin (line and marker) to investigate if any systematic bias exists.



501 **Fig. S27:** Comparisons between V_{chem} and $V_{phys,TC}$ for ATom-1 (blue color) and -2 (red color) data 502 binned by the fraction of V_{phys} removed when applying AMS transmission: (a) 0-20%, (b) 20-40%, 503 (c) 40-60%, (d) 60-80%, (e) 80-100%, respectively. The regressions and correlations are shown 504 for the two ATom studies separately as well as for the combined data set.

505 17. Observable particle size range ranges of the AMP, SAGA filter, MOUDI 1µm stage

506 impactor, AMS, and PALMS (ATom-1)



508 Fig. S28: Campaign-averaged volume (left) and number (right) size distributions observed by 509 AMP in ATom-1 (NMASS measured down to 3 nm and here we only show the subrange starting 510 from 8 nm), together with the approximate particle size ranges contributing chemical composition 511 information (without consideration of the details of the chemical detection) to the AMS, PALMS, and SAGA filter, and size-selected by a MOUDI 1 µm stage impactor. The top panel is one 512 513 dimensional with the campaign average result of each instrument (the transmissions of MOUDI 514 and SAGA filter are altitude dependent and plotted in Fig. 3 and Fig. S18, respectively; PALMS 515 effective detection range depends on counting statistics, and the detected particles given a 516 sampling period are discussed in Fig. S14-15). Note that the top panel shows the fraction of the 517 average, while Fig. 7 shows the average fractions (a summary at Table S2). The right plots 518 represent the size ranges of the number size distribution contributing chemical information to each 519 instrument. The following panels show the vertical profiles of the same quantities for AMP, SAGA filter, MOUDI impactor, AMS, and PALMS-AMP, respectively. The PALMS-AMP product 520

- 521 (Froyd et al., 2019) reports composition above 100 nm, the size range indicated by the dashed
- 522 square in the bottom panels. The plotted altitude bins are 800 m each.

523 18. Observable particle size range ranges for PALMS-AMP: comparing different size and

524 time resolutions



525 526 Fig. S29: Campaign-averaged volume (left) and number (right) size distributions observed by 527 AMP in ATom-1, together with the approximate particle size ranges contributing chemical 528 composition information to PALMS-AMP derived with three methods. Both the 3-min and 60-529 min panels are based on the reported AMP size resolution (as shown in Fig. S15. PALMS detected 530 particle numbers within each AMP size bin are used to infer the fraction represented by PALMS-531 AMP. 100% is assigned to the bin if more than one particle is detected.), and the 4-bins PALMS-AMP product, supposed to have a full coverage of AMP above 100 nm (Froyd et al., 2019). A 532 533 summary of the volume/number fraction vs. AMP (including other instruments) can be found at 534 Table S2 and the vertical profiles are shown in the main text as Fig. 9g&o. The top panel is one 535 dimensional with the campaign average result, and the rest are 2D vertical profiles. PALMS-AMP 536 product (Froyd et al., 2019) reports above 100 nm, the size range indicated by the dashed square 537 other than the top panel. All the plots represent the size ranges contributing chemical information 538 to the PALMS-AMP based on the fractional aerosol population detected by PALMS. In the bottom 539 panels, the volume-weighted (left) or number-weighted (right) diameters for the 4 larger bins 540 (Froyd et al., 2019) are calculated for AMP only and PALMS-AMP, which illustrates the shifted

- 541 weighting on the aerosol population reported by the PALMS-AMP product due to the uneven
- 542 PALMS relative data coverage as shown in Fig. S13.



Fig. S30: Same as Fig. S29 but for ATom-2.

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